UNCLASSIFIED

	NITI	N AT	CO
AD	NU	IVIE	BEK.

AD802114

LIMITATION CHANGES

TO:

Approved for public release; distribution is unlimited. Document partially illegible.

FROM:

Distribution authorized to U.S. Gov't. agencies and their contractors; Critical Technology; JUL 1966. Other requests shall be referred to Air Force Flight Dynamics Laboratory, FDTR, Wright-Patterson AFB, OH 45433. Document partially illegible. This document contains export-controlled technical data.

AUTHORITY

affdc ltr, 31 may 1973

VERTICAL GUST LOAD ANALYSIS

SUPPLEMENT RESULTS VOLUME II

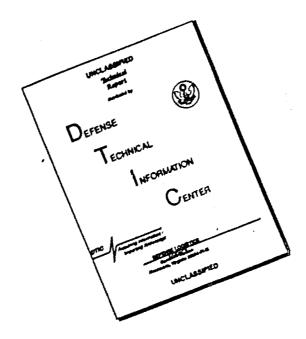
ROBERT N. LATZ
THE BOEING COMPANY

TECHNICAL REPORT AFFDL-TR-66-57, VOLUME II
JULY, 1966

This document is subject to special export controls and each transmittal to foreign governments or foreign nationals may be made only with prior approval of the Air Force Flight Dynamics Laboratory (FDTR), Wright-Patterson AFB, Ohio 45433.

AIR FORCE FLIGHT DYNAMICS LABORATORY
RESEARCH AND TECHNOLOGY DIVISION
AIR FORCE SYSTEMS COMMAND
WRIGHT-PATTERSON AIR FORCE BASE, OHIO

DISCLAIMER NOTICE



THIS DOCUMENT IS BEST QUALITY AVAILABLE. THE COPY FURNISHED TO DTIC CONTAINED A SIGNIFICANT NUMBER OF PAGES WHICH DO NOT REPRODUCE LEGIBLY.

NOTICES

When Government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related Government procurement operation, the United States Government thereby incurs no responsibility nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto.

Copies of this report should not be returned to the Research and Technology Division unless return is required by security considerations, contractual obligations, or notice on a specific document.

AFFDL TR-66-57 VOLUME & Z

VERTICAL GUST LOAD ANALYSIS

SUPPLEMENT RESULTS VOLUME II

ROBERT N. LATZ
THE BOEING COMPANY

This document is subject to special export controls and each transmittal to foreign governments or foreign nationals may be made only with prior approval of the Air Force Flight Dynamics Laboratory (FDTR), Wright-Patterson AFB, Ohio 45433.

FOREWORD

The program described in this report was conducted by the Structural Dynamics Unit, Structures Staff, Commercial Airplane Division, The Boeing Company, Renton, Washington. The program was monitored by Mr. Paul Hasty (FDTR), Air Force Flight Dynamics Laboratory, Research and Technology Division, Air Force Systems Command, Wright-Patterson Air Force Base, Ohio, under contract number AF33(615)-2454, "Investigation to Obtain Specific Design Calculations on Proven Transport Aircraft for the Verification of a Gust Design Procedure Based on Proven Spectral Techniques." The program was accomplished under system number 5(611367 62405334), project number 1367, "Structural Design Criteria", task number 136702, "Aerospace Vehicle Structural Loads Criteria." The time period covered by this final technical report is 1 July 1965 to 1 June 1966. The manuscript was released by the author on 1 April 1966 for publication as an RTD technical report.

Supervising consultant was Dr. John C. Houbolt of Aeronautical Research Associates of Princeton. Robert N. Latz conducted the analysis under the supervision of Arthur J. Kamm, Supervisor of the Structural Dynamics Unit.

This report has been given The Boeing Company document number D6-18252.

This technical report has been reviewed and is approved.

francis (j. /anik/ jr. 🔾

Chief, Theoretical Mechanics Bran

Structures Division

ABSTRACT

This report presents the results of an analysis to obtain the stress response parameters (level of stress per level of turbulence) and zero-crossing rates at two wing stations and two body stations of the KC-135 airplane where the margins of safety for gusts are minimum. Five combinations of gross weight, speed, and altitude were selected. The results of the computer analysis present the effects of changes in scale of turbulence and upper cutoff frequency on the response parameters and zero-crossing rates. Results indicate a large reduction in stress response parameter and small reduction in zero-crossing rate with increasing scale of turbulence. Variations of upper cutoff frequency above the highest modal frequency used in the analysis indicate negligible change in either stress response parameter or zero-crossing rate. The ratios of incremental limit allowable stress to stress response parameter obtained over a wide range of gross weight, speed, and scale of turbulence result in a minimum value of 53. This document (volume I) presents the analyses and specific results described above. Volume II presents response parameters, zero-crossing rates, frequency response functions, and power spectra of bending moment, shear, and torsion.

CONTENTS

Sectio	<u>n</u>		Page
I	INTRODU	JCTION	1
п	ANALYSI	is	2
	a. W b. St c. St d. Ac 3. Atmos	rsis Conditions	2 4 4 6 6 6 7 7
Ш	RESULTS	S AND DISCUSSION	11
IV	CONCLU	sions	29
APPE	NDIX I	WEIGHT DATA	31
APPE	NDIX II	STIFFNESS DATA	39
APPE	NDIX III	AERODYNAMIC DATA	45
APPE	NDIX IV	AIRPLANE FREE-FREE MODE SHAPES	51
APPE	NDIX V	STRESS FREQUENCY RESPONSE FUNCTIONS	67
APPE	NDIX VI	STRESS RESPONSE PARAMETERS AND ZERO-CROSSING RATES	109
APPE	ENDIX VII	INCREMENTAL LIMIT ALLOWAPLE STRESSES	115
APPE	END'X VIII	CORRELATION COEFFICIENTS BETWEEN AXIAL AND SHEAR STRESSES	117
APPE	END IX IX	STRESS INFLUENCE COEFFICIENTS	119
REFE	PENCES		191

ILLUSTRATIONS

Figure	Title							Page
1	Speed-Altitude Conditions							3
2	Gross Weight Conditions Relative to Operational							
	Center-of-Gravity Limits	•		•	•	•	•	5
3	Analytical Representation of Atmospheric							_
	Spectra			•	•	•	•	8
4	Comparison of Steady-State, Static, Elastic Solution							10
_	with Aeroelastic Solution (Analysis Condition 1) .							16
5	Locations at Which Stresses Are Obtained	•	• •	•	•	•	•	12
6	Segment Locations and Typical Distributions of							13
7	Margins of Safety	•	• •	•	•	•	•	10
•	for Shear Stress							14
8	Response Parameters and Zero-Crossing Rates	•	• •	•	•	•	•	1-1
v	for Axial Stress							17
9	Zero-Crossing Rates Versus σ_{-} , η_{-} (Linear Plot)			•				21
10	Zero-Crossing Rates Versus $\sigma_W \eta_D$ (Linear Plot) Zero-Crossing Rates Versus $\sigma_W \eta_D$ (Semilog Plot)							24
11	Stress Interaction Diagram							27
12	Wing Vertical-Bending and Torsion Stiffness					•		39
13	Wing Elastic-Axis Location							40
14	Body Vertical-Bending Section Moment of Inertia .							41
15	Body Elastic-Axis Location							42
16	Wing Lift Distribution							45
17	Local Center-of-Pressure Location							46
18	Body Lift Distribution (Mach 0.85)							47
19	Body Lift Distribution (Mach 0.50)	•		•	•	•	•	48
20	Wing Vertical Displacement in the Normalized							
	Free-Free Airplane Modes; 297, 000-Pound Gross							
	Weight (Weight Condition A)	•	• •	•	٠	•	•	51
21	Wing Torsional Displacement in the Normalized							
	Free-Free Airplane Modes; 297, 000-Pound Gross							
00	Weight (Weight Condition A)	•	•	•	•	•	•	53
22	Body Vertical Displacement in the Normalized Free-Free Airplane Modes; 297,000-Pound Gross							
	Weight (Weight Condition A)							53
23	Wing Vertical Displacement in the Normalized	•	•	•	•	•	•	55
20	Free-Free Airplane Modes; 268, 000-Pound Gross							
	Weight (Weight Condition B)							55
24	Wing Torsional Displacement in the Normalized	•	•	•	•	•	•	00
	Free-Free Airplane Modes; 268, 000-Pound Gross							
	Weight (Weight Condition B)							56
25	Body Vertical Displacement in the Normalized							9.5
	Free-Free Airplane Modes; 268,000-Pound Gross							
	Weight (Weight Condition B)				۵			57
26	Wing Vertical Displacement in the Normalized							
	Free-Free Airplane Modes; 190,590-Pound Gross							
	Weight (Weight Condition C)							59

ILLUSTRATIONS (Continued)

Figure	Title	Page
27	Wing Torsional Displacement in the Normalized Free-Free Airplane Modes; 190,590-Pound Gross Weight (Weight Condition C)	60
28	Body Vertical Displacement in the Normalized Free-Free Airplane Modes; 190,590-Pound Gross	00
	Weight (Weight Condition C)	61
29	Wing Vertical Displacement in the Normalized Free-Free Airplane Modes; 107, 260-Pound Gross Weight (Weight Condition D)	63
30	Wag Torsional Displacement in the Normalized Free-Free Airplane Modes; 107,260-Pound Gross Weight (Weight Condition D)	64
31	Body Vertical Displacement in the Normalized Free-Free Airplane Modes; 107, 260-Pound Gross	
	Weight (Weight Condition D)	65

TABLES

Number	Title	Page
I	Summary of Analysis Conditions	2
п	Summary of Weight Conditions	4
m	Margins of Safety	11
īv	Weight Condition A (Maximum Zero Flap Weight;	
• •	Gross Weight: 297,000 Pounds)	31
V	Weight Condition B (Maximum Transfer Weight;	• • •
•	Gross Weight: 268,000 Pounds)	33
VI	Weight Condition C (Intermediate Gross Weight with	
••	Structural Reserve Fuel; Gross Weight: 190,590 Pounds)	35
VΠ	Weight Condition D (Operating Weight Empty with	00
• • • •	Structural Reserve Fuel; Gross Weight: 107, 260 Pounds)	37
VШ	Naceile Cantilever Mode Shapes and Frequencies	43
IX	Rigid-Airplane Derivatives (1/Radian)	49
X	Rigid-Horizontal-Stabilizer Lift at 24,000-Foot Altitude	49
XI	Nacelle Mode Shapes (Weight Condition A)	54
XΠ	Nacelle Mode Shapes (Weight Condition B)	58
XIII	Nacelle Mode Shapes (Weight Condition C)	62
XIV	Nacelle Mode Shapes (Weight Condition D)	66
XV	Stress Frequency Response Functions (Analysis	30
ΛV	Condition 1)	67
XVI	Stress Frequency Response Functions (Analysis	01
74 V I	Condition 2)	85
XVII	Stress Frequency Response Functions (Analysis	0.0
AVII	Condition 3)	91
хvш	Stress Frequency Response Functions (Analysis	91
VAIII	Condition 4)	97
XIX	Stress Frequency Response Functions (Analysis	91
ALA	Condition 5)	103
XX	Stress Response Parameters and Zero-Crossing Rates	103
ΛΛ		109
XXI	(Analysis Condition 1)	109
VVI	(Analysis Condition 2)	111
XXII	Stress Response Parameters and Zero-Crossing Rates	111
лли		110
XXIII	(Analysis Condition 3)	112
V-VIII	(Analysis Condition 4)	119
XXIV	Stress Response Parameters and Zero-Crossing Rates	113
WVIA		114
xxv	(Analysis Condition 5)	
XXVI	Correlation Coefficients Between Axial and	115
AAVI	Shear Stresses	117
		117

ABBREVIATIONS AND SYMBOLS

A	stress response parameter (rms value of incremental stress for a 1 fps rms random gust) (psi/fps)
N _o	zero-crossing rate (average number of times per second that the incremental stress crosses the 1g mean value with positive slope)
ω	frequency (radians per second)
Η(ω)	absolute value of frequency response function
$\Phi_{\mathbf{i}}(\omega)$	gust spectrum
$^{\omega}\mathbf{c}$	upper cutoff frequency (radians per second)
Ω	reduced frequency (radians per foot)
L	scale of turbulence (feet)
σ	rms level of turbulence intensity (fps)
$^{\sigma}w^{\eta}$ D	measure (fps) of the probability of exceeding limit stress. (It is equal to the ratio of incremental limit allowable stress to stress response parameter.)

SECTION I

INTRODUCTION

A program is being conducted by the U.S. Air Force to establish a simplified procedure to design airplanes for gusts based on power spectral density techniques. To verify the proposed gust procedure, specific design calculations for selected airplanes were obtained from several airplane manufacturers. The Boeing Company was selected to obtain design calculation. for the KC-135 airplane and these design calculations are presented in this report.

The specific design calculations presented are stress response parameters (ratio of rms level of stress to rms level of turbulence), zero-crossing rates, and stress frequency response functions. These data are calculated at two wing stations and two body stations where the margins of safety for gusts are minimum. The free-free mode shapes of the airplane are also included.

In volume II are presented response parameters, zero-erossing rates, frequency response functions, and power spectra of bending moment, shear, and torsion.

SECTION II

ANALYSIS

1. Analysis Conditions:

- a. The selection of flight conditions for analysis is based on two considerations. First, consideration is given to the gust design conditions used in the basic design of the KC-135 airplane (1). These are based on the use of the gust load formula (2). Second, consideration is given to the flight conditions that would result in minimum pitch stability, that contributes to high loads in random turbulence. The critical gust altitude of 24,000 feet was derived from the design gust analysis, and this altitude is selected for the present analysis. Past power-spectral analyses have shown that low pitch stability results in high loads. The conditions for low pitch stability are a high lift coefficient and an aft center of gravity. Since both of these conditions cannot be achieved simultaneously on the KC-135 airplane, five analysis conditions are selected to represent a wide range of gross weight, center of gravity, and speed.
- b. Table I and figure 1 summarize the analysis conditions. Condition 1 is the maximum gross weight, maximum design speed condition and represents the maximum gust force input to the airplane. Using the gust load formula (2), this is the critical gust design condition for the inboard wing. It should be noted that the basic KC-135 wing is designed by maneuver rather than gust conditions. Analysis condition 2 represents a fuel transfer weight of the airplane. Analysis condition 3 represents the airplane with a fuil body and an empty wing, except for structural reserve fuel. Analysis condition 4 represents the operating—weight—empty airplane plus structural reserve fuel (the condition having the most-aft center of gravity). Analysis condition 5 represents the maximum—gross—weight airplane flying at the slowdown speed for severe gust. At this flight condition, the airplane is flying at maximum lift coefficient.

Table 1. Summary of Analysis Conditions

Analysis condition number	Weight condition	Gross weight (1b)	Altitude (ft)	Equivalent airspeed (kn)	Mach number	Body fuei (1b)	Wing fuel (lb)
1	٨	297,000	24,000	350	0.85	83,328	109,512
2	В	268,000	24,000	350	0.85	87,927	75,913
3	С	190 590	24,000	350	0.85	83,323	3,100
4	0	107,260	24,000	350	0.85	0	3,100
5	Α	297,000	24,000	207	0.50	83,328	109,512

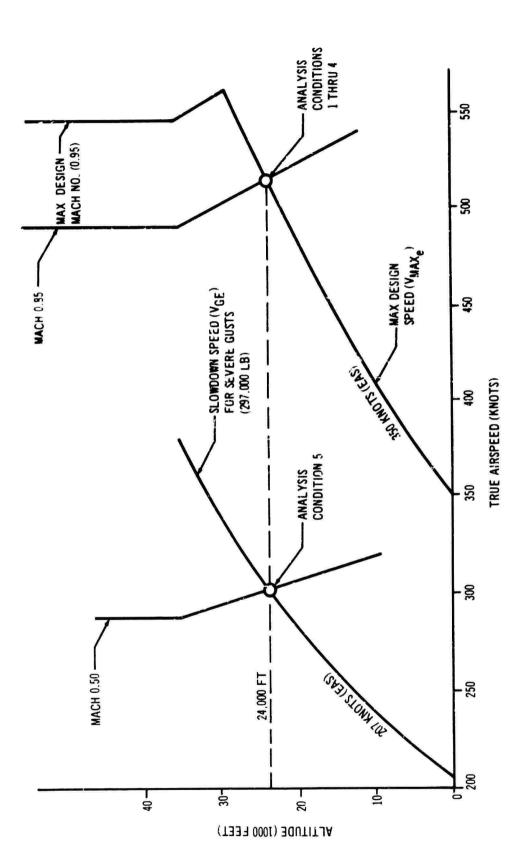


Figure 1. Speed-Altitude Conditions

2. Airplane Representation. The airplane used in this analysis is the KC-135. The airplane has a crew of feur: pilot, copilot, navigator, and boom operator. It normally cruises at altitudes from 25,000 to 45,000 feet, gross weights to 297,000 pounds, and speeds to 525 knots (true air speed). The airplane has a wing span of 131 feet and an overall length of 136 feet. A two-view diagram is shown in figure 5, page 12. All major parts of the airplane except the fin and horizontal stabilizer are considered to be elastic in the analysis. Therefore, a rather comprehensive mass and stiffness description of the airplane is required. Simple beam-bending theory is used to represent the stiffness characteristics of the major components of the structure, such as the wing and forward and aft fuselage. The elastic axes are located approximately along the locus of shear centers of each component, except in the inboard portion of the wing where the elastic axis is determined from static tests.

a. Weights Data:

- (1) The complete detailed description of panel weights (obtained from reference 3) used in this analysis is given in appendix I. The fuselage is divided into 18 weights panels and the panel weight and pitch inertia is determined for each panel. The wing semispan is divided into ten spanwise panels. Each of these panels are divided into five zones: leading edge, front spar, interspar, rear spar, and trailing edge. The weight and center of gravity are calculated for each zone and summed to give the total panel weight and center of gravity. The total panel-weight moments of inertia are computed by rotation and transfer of zone results into axes located parallel and perpendicular to the wing elastic axis.
- (2) The mass properties for each engine, nacelle, and nacelle strut are combined and a lumped center of gravity is determined. Then, the nacelle mass moments of inertia are determined for axes located perpendicular and parallel to the airplane reference axis.
- (3) Table II summarizes the weight conditions shown on the gross-weight-versus-center-of-gravity chart in figure 2.

Table II. Summary of Weight Conditions

Weight Gross weight			Fuel (lb)								
	Gross weight	CG		Wi	ng	Body					
condition	(Ib)	(percent mac) Outboard Inboard mains mains 2 & 3 Center section		Outboard reserves	Forward	Aft	Upper				
A	297,000	21.4	26,806	29,575	47,489	5,642	37,700	41,457	4,171		
В	268,000	23.0	14,212	14,212	47,489		37,700	41,457	8,770		
С	190,590	28.3	1,550	1,550			37,700	41,457	4,173		
D	107,260	35.1	1,550	1,550							
 ,	Capacity of ta	nks	26,806	29,575	47,489	5,642	37,700	41,457	14 131		

Note: Fuel density at 6.5 pounds per gatton.

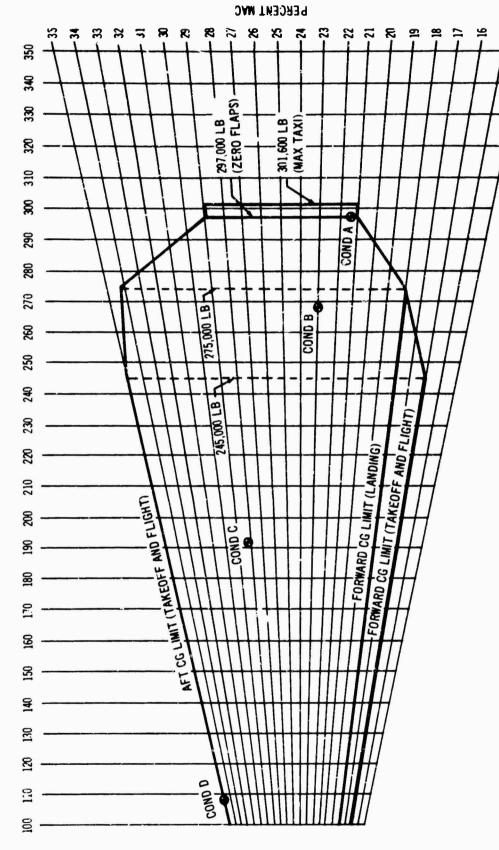


Figure 2. Gross Weight Conditions Relative to Operational Center-of-Gravity Limits

b. Stiffness Data:

- (1) The stiffness of each major component of the airplane (except nacelle struts) is described by a distribution of bending stiffness (EI) and torsional stiffness (GJ) along the elastic axis. The wing-section properties are computed using front and rear spar areas and all in-spar skin for both upper and lower surfaces. Values for modulus and shear modulus of elasticity (E and G) are 10.3×10^6 and 3.8×10^6 psi, respectively. The body-section properties are computed using stiffeners with full-skin effective in tension and a portion of skin effective in compression. The body cutout sections are analyzed individually by special analysis. The body center-section stiffness is estimated on the basis of variation of skin thickness, keel beam stiffness, and stringer size from body stations 620 through 820. The body stiffness is for the 2g diverging maneuver condition (tension in the upper surface).
- (2) The stiffness of the nacelle struts is calculated from the inertia of the nacelle-strut combination and the natural frequency and mode shapes obtained from ground shake tests (4, 5). The detailed stiffness data is given in appendix II.

c. Structural Damping Data:

(1) The structural damping used in this analysis is obtained from the ground vibration test of the 707-320B airplane. These values of damping are considered to be representative of the KC-135 airplane, since the structure of the two airplanes is similar. The values of structural damping coefficient are equal to twice the fraction of critical damping.

Mode 1 2 3 7 8 <u>5</u> <u>6</u> 0.015 0.045 Structural 0.053 0.030 0.025 0.033 0.029 0.028 damping coefficient

d. Aerodynamic Data:

- (1) All of the basic aerodynamic data required for these analyses are obtained from a series of wind tunnel tests (6). Wind tunnel pressure-model test results are used to establish wing- and fuselage-airload distribution. The aerodynamic coefficients are corrected for model flexibility before they are used for full-scale airplane analysis, and are later refined to obtain final agreement between the aeroelastic analysis and actual airplane flight-load survey measurements.
- (2) The unsteady aerodynamics are based on two-dimensional strip theory, based on wind tunnel model-pressure data, and are modified to include aerodynamic induction effects (7). These induction effects account for the aerodynamic pressure carryover between wing panels and between the wing and horizontal tail. This is accomplished by using a downwash matrix based on lifting-line theory. The dynamic downwash matrix includes pressure-carryover and pressure-transmittal functions to provide the proper magnitude and phasing of the carryover pressure. The section (or strip) aerodynamics for zero frequency are made to agree with the comparable aeroelastic solution. Included in the aerodynamics is the effect of gradual penetration into the gust.

(3) Body lift distribution is based on wind tunnel pressure-model data adjusted to make the rigid airplane pitch and lift derivatives match those used in the aeroelastic analysis. Detailed aerodynamic data are given in appendix III. The maximum lift correction due to compressibility occurs at mach 0.85.

3. Atmospheric Turbulence Representation:

a. There are two power spectra that are in current use to represent the atmosphere (8), and the following spectrum was selected by Dr. Houbolt for this analysis:

$$\Phi(\Omega) = \left(\frac{\sigma^2 L}{\pi}\right) \frac{1 + \frac{8}{3} (1.339 L\Omega)^2}{\left[1 + (1.339 L\Omega)^2\right]^{11/6}}$$

This power spectrum is plotted in figure 3 for scales of turbulence of 1,000, 3,000, and 5,000 feet. A value of 1 fps was used throughout the analysis for o.

b. It is assumed that the turbulence is essentially "frozen" in space and is uniform normal to the line of flight of the airplane. The airplane passes over the turbulence much as an automobile would travel over a rough road. This approach assumes that the spanwise variation of turbulence (except for the effect of gradual penetration) is negligible.

4. Equations of Motion:

a. The airplane is represented by ten degrees of freedom: eight symmetrical free-free elastic modes, which are plotted in appendix IV, and rigid-airplane vertical translation and pitch. All flight control surfaces are assumed fixed in the 1g flight position. The response functions and zero-crossing rates are calculated from the following equations:

$$A = \int_{0}^{\omega_{c}} |H(\omega)|^{2} \Phi_{i}(\omega) d\omega$$

$$N_{o} = \frac{1}{2\pi} \left[\frac{\int_{0}^{\omega_{c}} \omega^{2} |H(\omega)|^{2} \Phi_{i}(\omega) d\omega}{\int_{0}^{\omega_{c}} |H(\omega)|^{2} \Phi_{i}(\omega) d\omega} \right]^{1/2}$$

b. To check the equations of motion, the loads are obtained from the equations of motion for a 1g gust condition and compared with those obtained from the aeroelastic solution. This is accomplished by first obtaining the

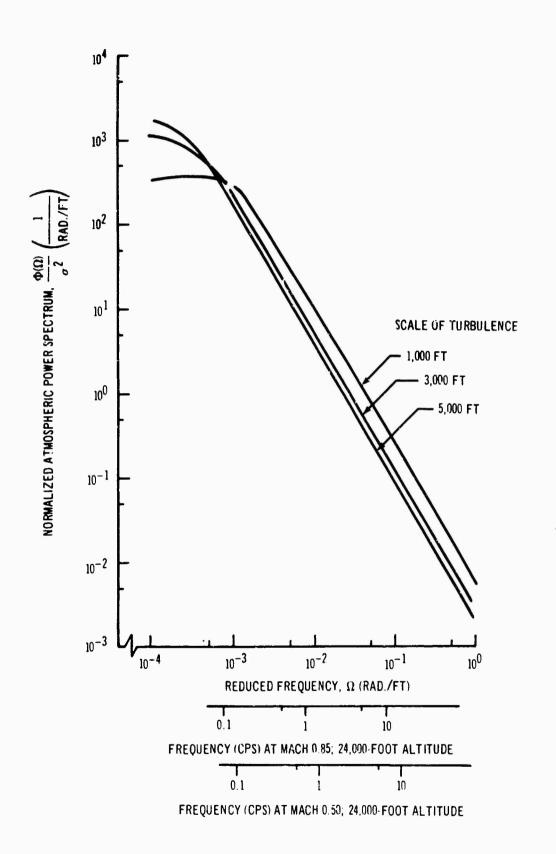
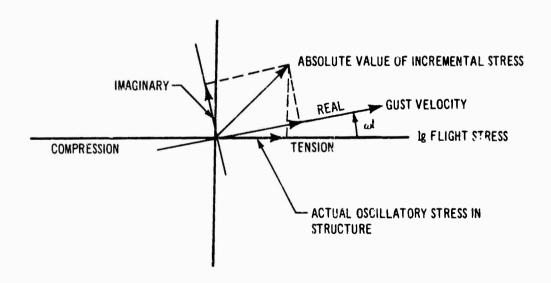
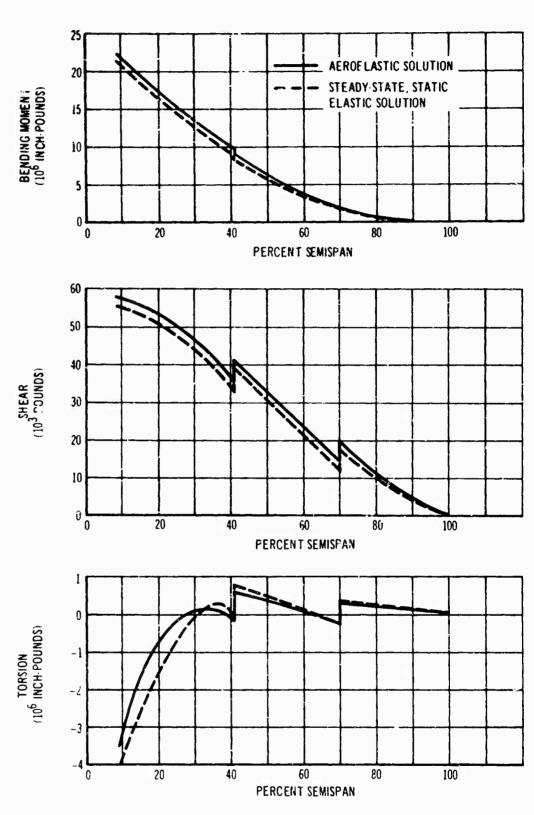


Figure 3. Analytical Representation of Atmospheric Spectra

equations of motion for zero-frequency gust input. Then the pitch- and elastic-mode generalized coordinate accelerations, the pitch and translation displacement, and all of the generalized coordinate velocities are equated to zero. The vertical translation acceleration is equated to 1g. To allow for an airplane moment balance, a tail load is added to the equations as an additional unknown. This tail load represents the change in tail lift required to balance the airplane while the airplane is flying through a gust that gives it a 1g acceleration. The solution of these equations gives the elastic mode deflections, the gust angle required for 1g acceleration, and the tail load required to balance the airplane. A comparison of wing loads is shown in figure 4 for analysis condition 1 (table 1).

c. The stress frequency response functions for the airplane structure are obtained from the complex frequency responses of the generalized coordinates. Shear, moment, and torsion eoefficients are calculated for unit deflections of the generalized coordinates. These coefficients are multipled by the complex frequency responses of the generalized coordinates to obtain load frequency responses. The load frequency responses are multiplied by stress influence coefficients obtained from the airplane stress analysis to give the complex stress frequency response functions. The absolute value of these stress frequency-response functions is then used to obtain A and N . The stress frequency-response functions are given in complex form, and represent the ineremental stress relative to the gust velocity and 1g mean as shown below.





į

Figure 4. Comparison of Steady-State, Static, Elastic Solution with Aeraelastic Solution (Analysis Condition 1)

SECTION III

RESULTS AND DISCUSSION

Loads are obtained at two wing stations and two body stations where the gust margins are minimum. The margins of safety shown in table III are calculated using the gust load formula (2).

The wing and aft body are designed by maneuver conditions. The forward body is designed by braked-roll and pressurization conditions. However, for flight conditions, the margin of safety for gust for the forward body (which includes alleviation due to pitch) is less than for maneuver conditions. The fuselege and wing margins of safety are given in references 9 and 10, respectively.

Table III. Margins of Safety

Location	Segment number	Type of loading	Gust margins of safety	
	WING	,		
27 percent semispan	10	Combined	0.20	
27 percent semispan	14	Primarily tension	0 39	
40.06 percent semispan (inboard of nacelle)	8	Combined	0.24	
40.06 percent semispan (inboard of nacelle)	107	Combined	0.20	
	BOD	ΙΥ		
Body balance station 540	S-17	Clear load only	0.20	
Body balance station 820	S-1	Tension load only	0.14	

Figures 5 and 6 show these locations on the airplane. The curves of margin of safety in figure 6 are included to show the variation alc ig the cross section of the wing. The margins of safety are based on the following equation:

Margin of safety = Allowable ultimate principal stress - 1
Design ultimate principal stress

The results of this investigation are response parameters A , zero-crossing rates $\rm N_{\rm O}$, stress transfer functions, and the ratios of incremental limit allowable stress to stress response parameter. The stress frequency-response functions are tabulated in appendix $\rm V.$

The effects of variations in scale of turbulence on response parameters A and zero-crossing rates \aleph_0 are shown in figures 7 and 8. The stress response parameters and zero-crossing rates are tabulated for each analysis

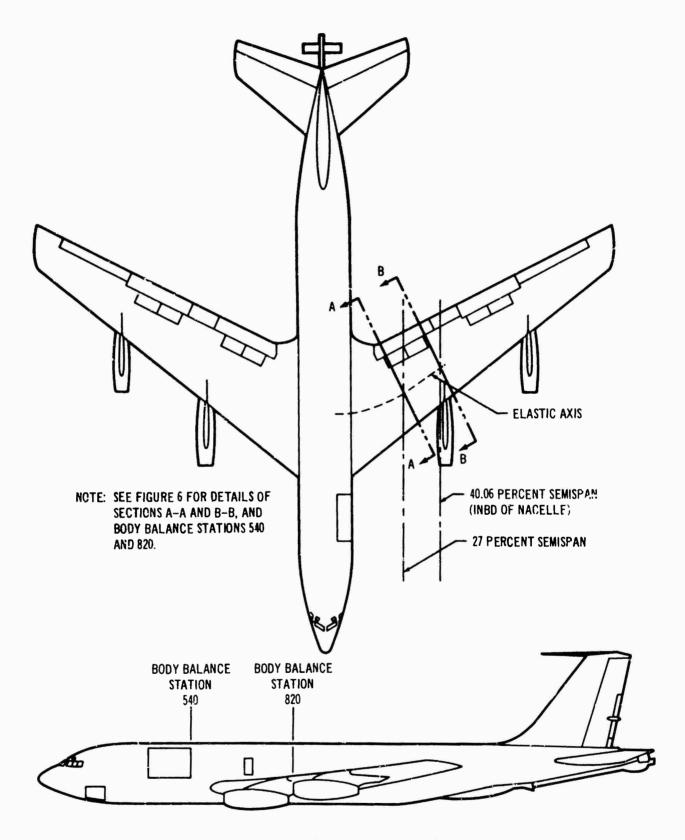


Figure 5. Locations at Which Stresses Are Obtained

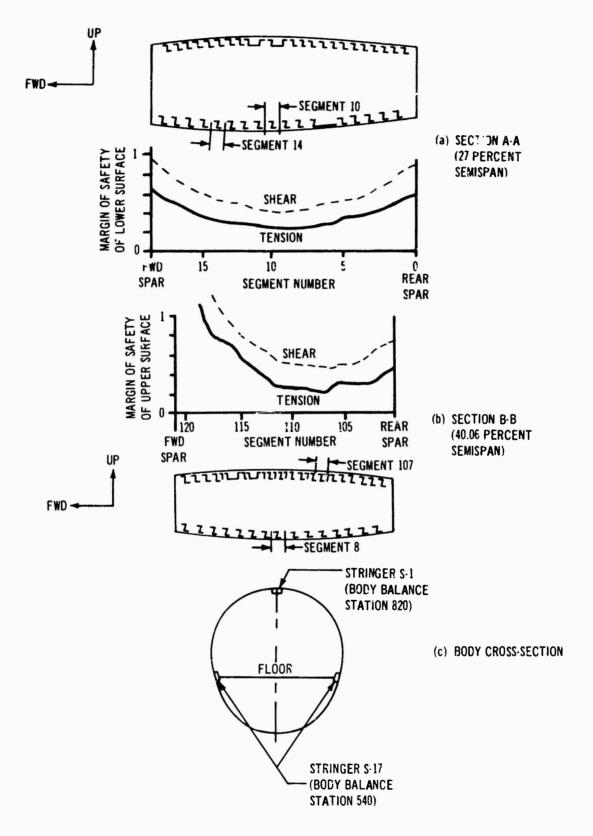


Figure 6. Segment Locations and Typical Distributions of Margins of Safety

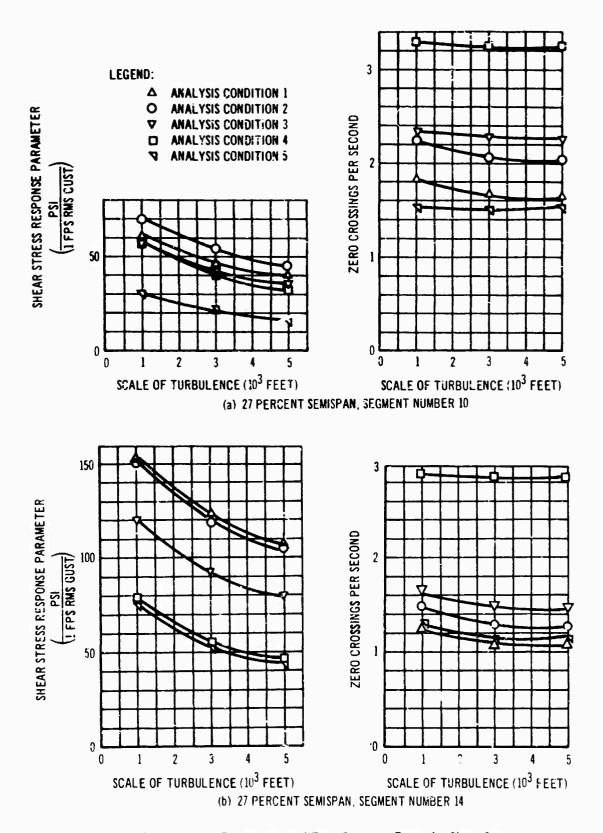


Figure 7. Response Parameters and Zero-Crossing Rates for Shear Stress

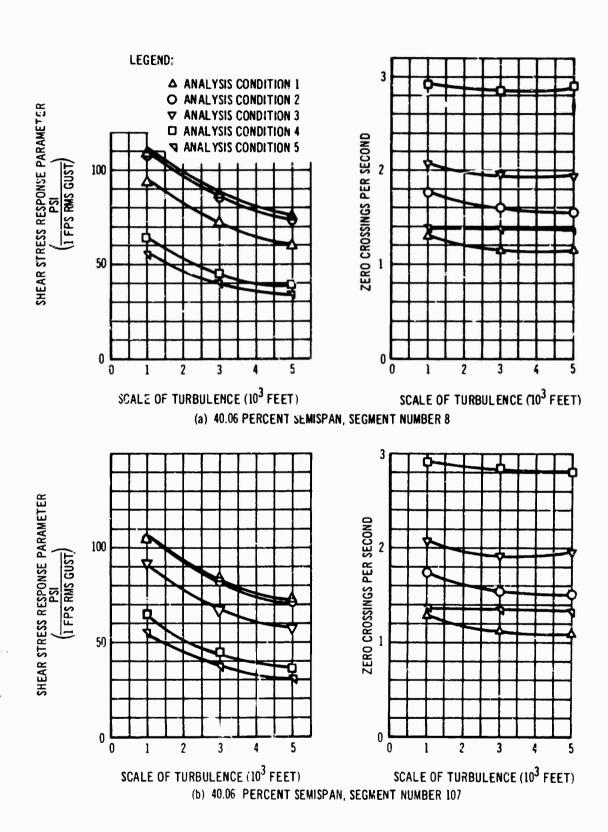
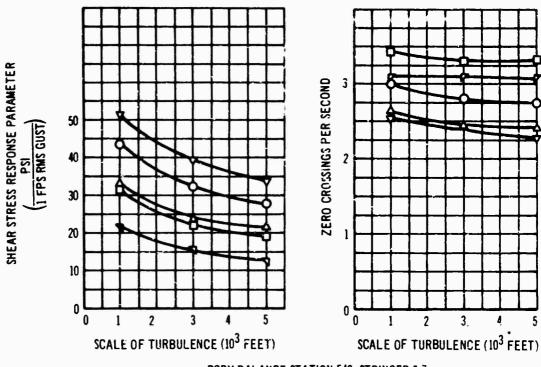


Figure 7 --- Continued



BODY BALANCE STATION 540, STRINGER S-7

LEGEND:

- ANALYSIS CONDITION 1
- O ANALYSIS CONDITION 2
- ▼ ANALYSIS CONDITION 3
- ☐ ANALYSIS CONDITION 4
- ANALYSIS CONDITION 5

Figure 7 --- Concluded

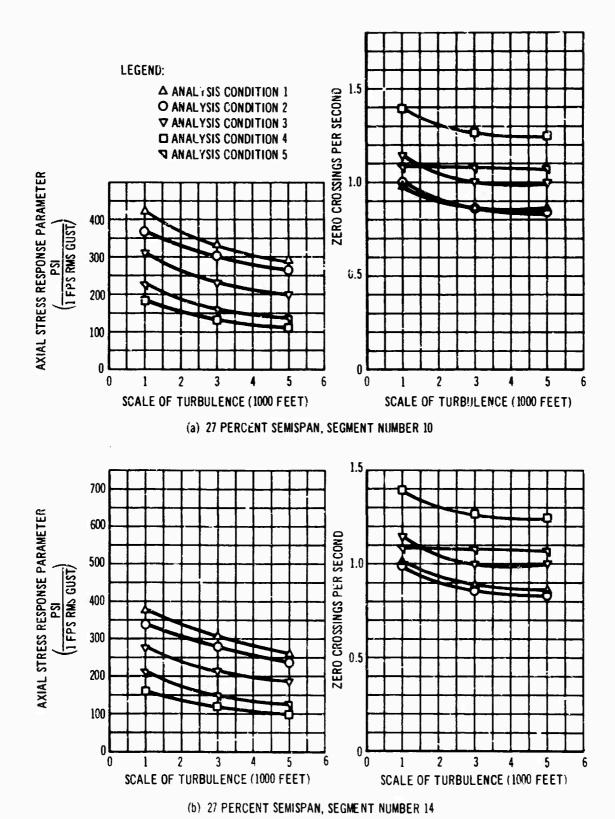
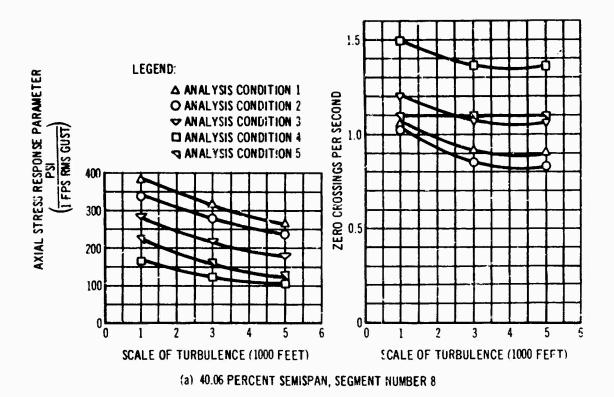


Figure 8. Response Parameters and Zero Crossing Rates far Axial Stress



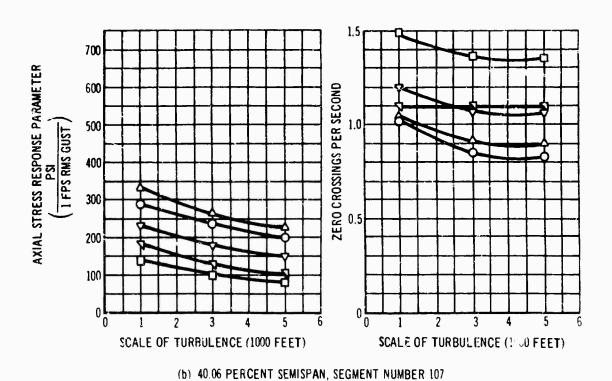
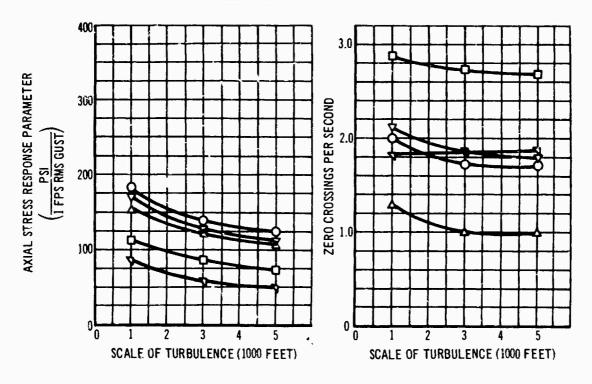


Figure 8 --- Continued

BODY BALANCE STATION 820



LEGEND:

- △ ANALYSIS CONDITION 1
 - O ANALYSIS CONDITION 2
- ▼ ANALYSIS CONDITION 3
- ☐ ANALYSIS CONDITION 4
- NANALYSIS CONDITION 5

Figure 8 --- Concluded

condition in appendix VI. The tabulated data show that changing the upper cutoff frequency from 10 to 15 and 20 cycles per second, as specified by Dr. Houbolt, has a negligible effect on A and $N_{\rm O}$. This is due to the highest elastic modes in the analysis having frequencies less than 10 cps, resulting in little response above that frequency.

Zero-crossing rate versus the ratio of incremental limit allowable stress to stress response parameter is plotted in figures 9 and 10. It should be noted that the critical condition is that of maximum gross weight and high dynamic pressure combined with a scale of turbulence of 1,000 feet. These data are directly comparable to the $\sigma_{\mathbf{W}^{\Pi}\mathbf{D}}$ of reference 11, remembering that in that report the scale of turbulence is 2,500 feet and, whereas the KC-135 airplane is designed to a 2g load factor, the 720 airplane is designed to a 2.5g load factor. The consequence of this is that the KC-135 has lower margins of safety for gust for the maximum gross weight conditions and a resultingly lower ratio of incremental limit allowable stress to stress response parameter.

The incremental limit allowable stresses tabulated in appendix VII are obtained from the stress interaction diagram for each section for which stresses are desired. few definitions follow:

- 1. Allowable stress: the maximum stress at which failure will occur
- 2. Limit allowable stress: the allowable stress divided by 1.50
- 3. Incremental stress: the increment of stress above the 1g flight stress

A typical stress interaction diagram is shown in figure 11. Curve 1 is a plot of the following equation:

$$\left(\frac{\text{Limit allowable}}{\text{shear principal stress}}\right) = \sqrt{\left(\frac{\text{Axial skin tension stress}}{2}\right)^2 + (\text{Skin shear stress})^2}$$

Curve 2 is a plot of:

$$\left(\begin{array}{c} \text{Limit allowable} \\ \text{tensile principal stress} \right) = \left(\frac{\text{Axial skin tension stress}}{2} \right) \\ + \sqrt{\left(\frac{\text{Axial skin}}{\text{tension stress}} \right)^2 + \left(\frac{\text{Skin}}{\text{shear}} \right)^2}$$

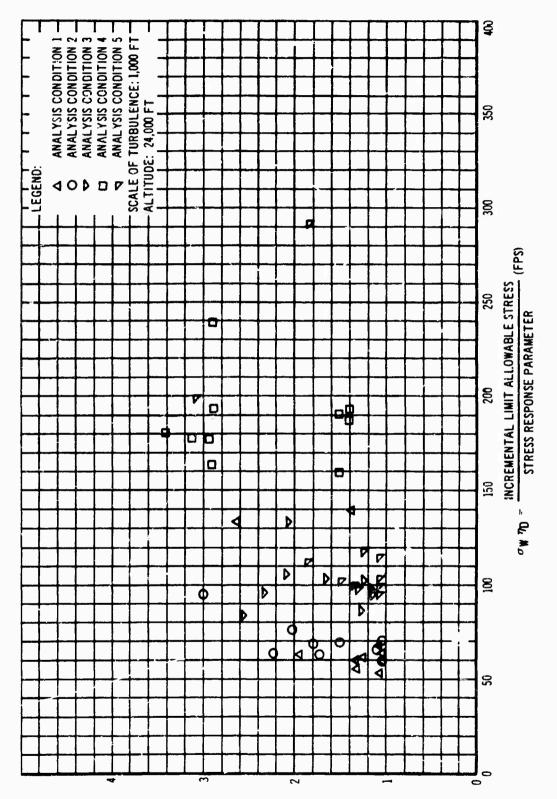
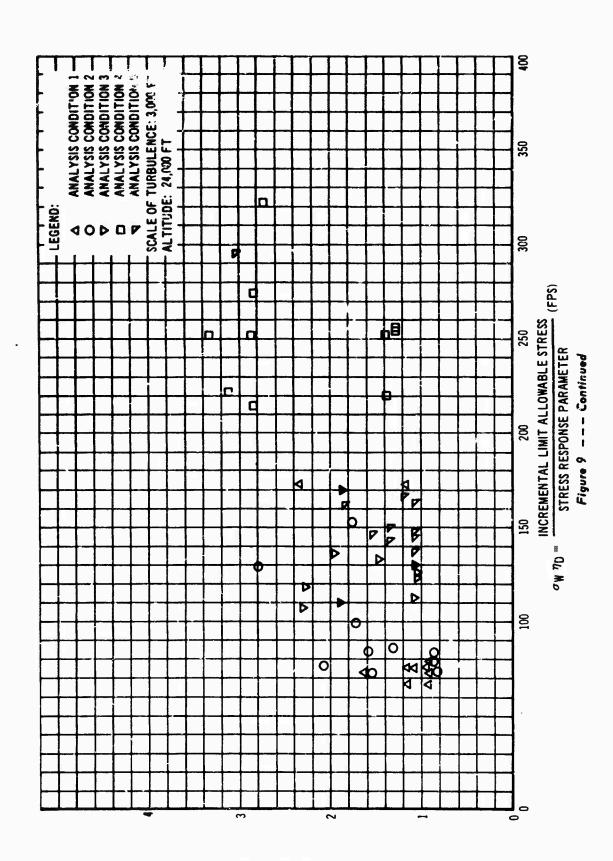
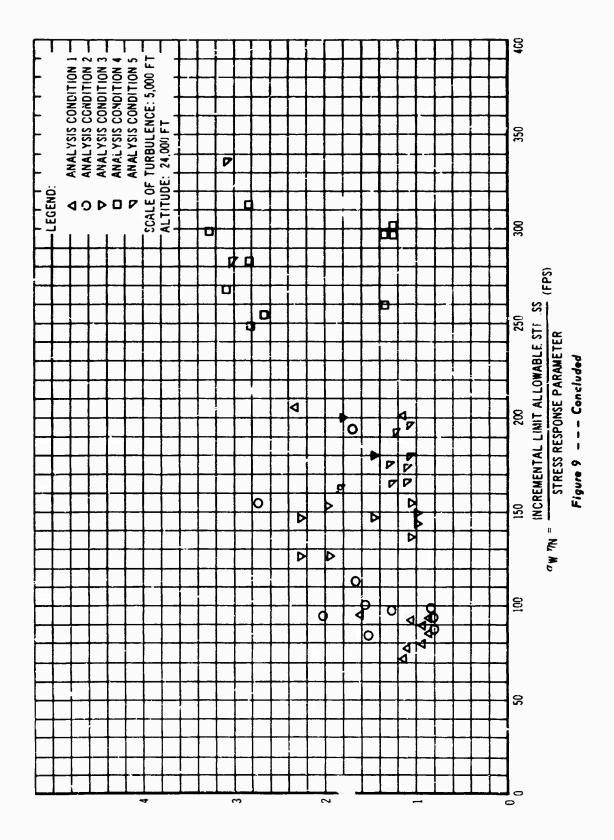


Figure 9. Zero-Crossing Rates Versus aw 110 (Linnar Plot)

ZERO CROSSINGS PER SECOND



SERO CROSSINGS PER SECOND



SERO CROSSINGS PER SECOND

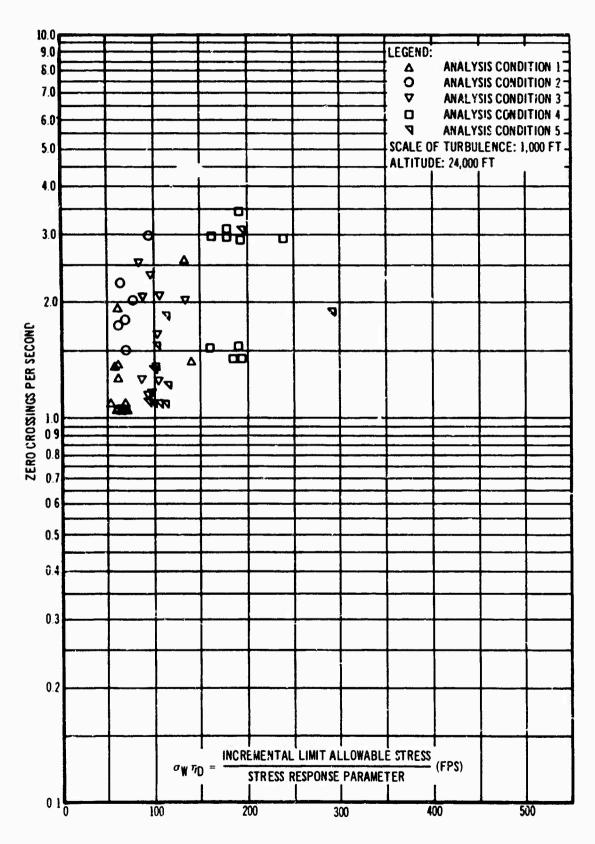


Figure 10. Zero-Crossing Rates Versus σ_{W} η_{D} (Semilog Plot)

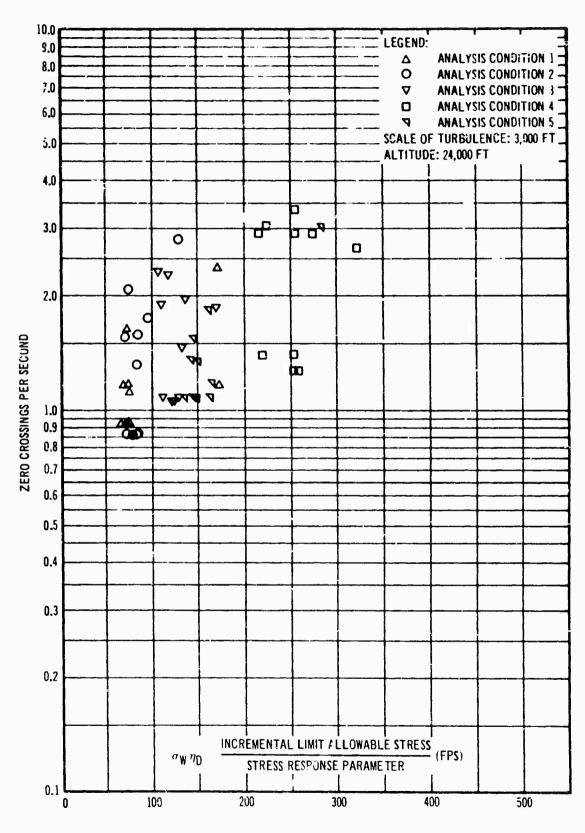


Figure 10 --- Continued

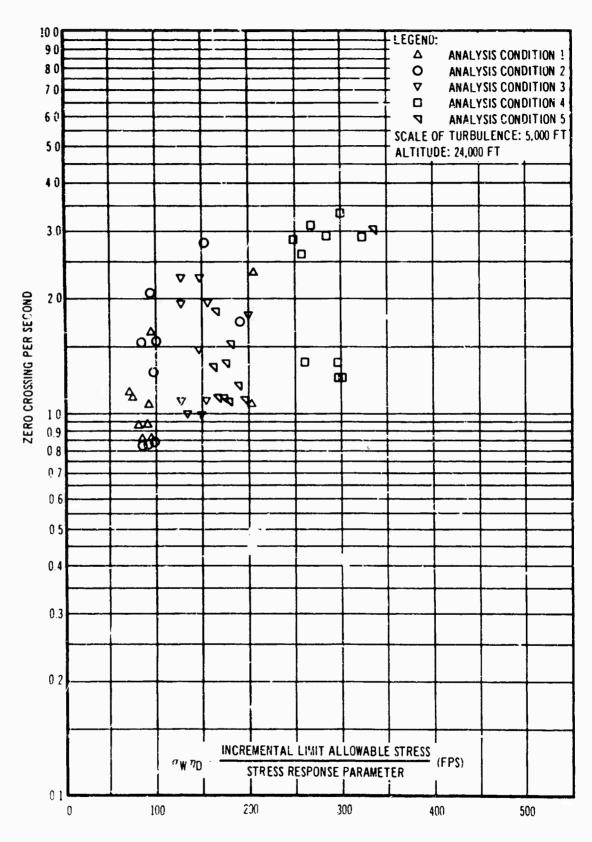
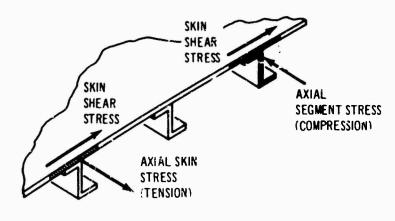


Figure 10 --- Concluded



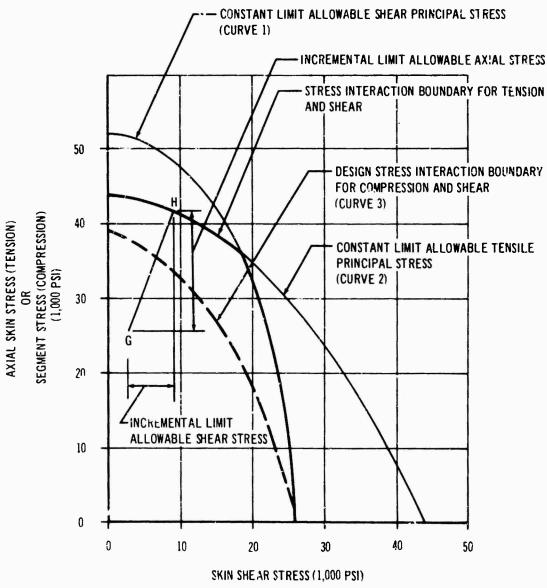


Figure 11. Stress Interaction Diagram

As shown in figure 11, the stress interaction curve used in design is formed by portions of these two curves and represents the combination of axial tension and shear that give the lower of the limit allowable shear or tensile principal stresses. Curve 3, the stress interaction curve for compression and shear, is a plot of:

The segment referred to in the equations includes the stringer area in addition to that of the skin; the area of both is used when computing compressive stresses. The limit allowable compressive stress is not a true principal stress, and its equation is an empirical variation of Moher's circle equation for compression principal stress. Knowing the limit allowable shear, tension, and compression stresses, the stress interaction diagram is drawn. The next step is to determine the incremental limit allowable stresses.

The incremental limit allowable stress is derived in the following manner: The 1g flight axial and shear stresses at a particular point in the wing are plotted as point G in figure 11. Point H in figure 11 is determined by drawing a line from point G having a slope equal to the ratio of axial stress response parameter to the shear stress response parameter. The slope of line G-H is based on the assumption of 100-percent correlation between axial and shear stress. The resulting incremental limit allowable axial and shear stresses are shown in the figure. A method of analysis which includes the effect of correlation between stresses is described in reference 11.

SECTION IV

CONCLUSIONS

The minimum value of $\sigma_W^{\eta}_D$ for all the conditions investigated is 53, which is for the maximum-gross-weight high-speed flight condition and the 1,000-foot scale of turbulence.

The most critical (largest) value of stress response parameter is for the 1,000-foot scale of turbulence and the heavy gross weight, high-speed conditions. Reducing the gross weight, lowering the speed, and increasing the scale of turbulence reduces the value of the stress response parameter.

The largest zero-crossing rate is associated with the lowest-gross-weight airplane and is little affected by the scale of turbulence.

The stress response parameter and zero-crossing rate are little affected by the upper cutoff frequency, where the cutoff frequency is above the highest modal frequency included in the analysis.

APPENDIX I WEIGHT DATA

Table IV. Weight Condition A (Maximum Zero Flap Weight; Gross Weight: 297,000 Pounds)

BODY

Panel number	Body balance	Weight	Center-of-gravity location (in.)			Moment of inertia about cg (fb-in. ² x 10 ⁻⁶)			
	station	(lb)	Body balance station	Body buttock line	Body waterline ^e	Pitch	Roll	Yaw	
1	130 to 259	2,771	220	0	218	4.45	3.95	3.57	
2 ^a	259 to 360	5,290	320	1	207	15.5	14.6	11.3	
3	360 to 420	9,810	393		189	12.1	18.1	12.6	
4	420 to 480	11,715	450		182	11.9	20.5	15.7	
5	480 to 540	11,482	510		181	10.4	18.8	15.3	
6	540 to 620	11,633	571		182	12.3	31	15.9	
7	620 to 755	3,055	689		210	11	11.2	9.33	
8.	755 to 820	1,500	802		208	3.68	5.47	2.88	
gb	820 to 830	9,333	872		185	12.7	27.2	2 6.5	
10	890 to 960	2,037	926		209	5.11	7.43	4.03	
11	960 to 1020	11,823	989		183	10.1	18.7	15.9	
12	1020 to 1080	10,717	1,050		189	8.61	15.9	13.7	
13	1080 to 1140	10,042	1,110		184	8.56	16.3	12.2	
14	1140 to 1200	10,004	1,170		189	10.6	18.3	12.4	
15	1200 to 1280	5,958	1,223		191	5.47	9.94	7.57	
16	1280 to 1360	1,252	1,321		217	2.58	3.79	2.55	
17	1360 to 1440	5,035	1,406	+	220	3.63	6.45	6.85	
18 ^C	1440 to 1676	9,697	1,569	0	288	130	113	102	

WING/SIDE

1 2 ^d 3 4 5 6 7 8 9	0 to 70.5 70.5 to 157.2 157.2 to 235.8 235.8 to 314.4 314.4 to 393 393.0 to 471.6 471.6 to 550.2 550.2 to 628.8 b28.8 to 707.4 707.4 to 786.6		732.2 764.2 805.4 855.1 897.4 950.1 988.7 1,058.2 1,092 1,149.1	36.5 112.7 195.6 271.1 348.6 429.2 496.8 595.2 663.9 735.2	182.7 185.8 192.4 191.3 195.6 205.2 210.7 236.7 243.1 249.7	47.1 92.6 42.2 18.1 12.3 7.56 3.59 1.93 1.37 0.683	7.73 15 6.05 4.04 3.82 2.70 0.818 1.12 0.911 0.314	51.7 103 47.4 21 15.3 9.80 4.23 2.44 2.15 0.970	
--	--	--	--	---	--	---	---	--	--

Fuel density at 6.5 pounds per gallon

Notes: a Includes nose gear in the UP position

- b !ncludes tires, truck assembly, air, wheels, and side strut of main landing gear in the UP position
- c includes the horizontal tail, vertical tail, and refueling boom
- d Includes inner cylinder, collection er cylinder, trunnion, and landing gear support structure in wing
- e Cruise condition

Table IV --- Concluded

Airplane	Weight	Cent	Center-of-gravity location (in.)			Moment of inertia about cg (Ib-in, ² x 10 ⁻⁶)		
section	(1b)	Body balance station	Body buttock line	Body waterline ^e	Pitch	Roll	Yaw	
Body	133,154	858.4	0	196.9	19,200	452	19,200	
Wing	142,164	823.8	212.4	192.9	1,770	3,960	5,650	
Nacelles	10,938	713.7	322	159	16.9	2.64	15.7	
Nacelles	10,744	898.1	552	197	16.2	2.60	15	
Total/avg	297,000	837.9	0	193.6	21,400	15,300	35,900	

Table V. Weight Condition B (Maximum Transfer Weight; Gross Weight: 268,000 Pounds)

BODY

Panel balani	Body halance	Weight	Center of-gravity location (in)			Moment of inertia about cg (lb-in. ² x 10 ⁻⁶)			
	station	(IP)	Pody balance station	Body buttock line	Body waterline ^e	Pitch	Rull	Yaw	
1	130 to 259	2 77 1	220	Û	215	4.45	3 95	3.57	
2 ^a	259 to 360	5,290	3 2 0	Í	207	15.5	14.6	11.3	
3	360 to 420	9,810	393		189	12.1	18.1	12.6	
4	420 to 480	11,715	4 50		182	11.9	20.5	15.7	
5	480 to 540	11.482	510		181	10.4	18.8	15.3	
5	540 to 620	11,633	571		182	12.3	21	15.9	
7	620 to 755	3,055	689		210	11	11.2	9.33	
8 9 ^b	755 to 820	1,500	802		208	3.68	5.47	2 58	
9 ^b	820 to 890	9,333	872		185	12.7	7.2	26.5	
10	890 to 960	2.037	926	l i	209	5.11	7.43	4.03	
11	960 to 1020	11,823	989		183	10.1	18.7	15.9	
12	1020 to 1080	10,717	1,650		189	8.61	15.9	13.7	
13	1080 to 1140	10,042	1,110		184	8.56	16.3	12.2	
14	1140 to 1200	10,004	1,170	l i	189	10.6	18.3	12.4	
15	1200 to 1280	5,958	1,223		191	5.47	9.94	7.57	
16	1280 to 1360	1,252	1,321	}	217	2.58	3.79	2.55	
17	1360 to 1440	9,273	1,408	•	233	6.38	11.6	11.8	
18 ^C	1440 to 1676	10,057	1,564	0	287	136	114	108	

			_	_
w	NG	∕ C I	n	F
77	mu	J	u	_

1							1		
1	ı. İ	0 to 70.5	14,177	732.2	36.5	182.7	47.1	7.73	51.7
2	2d	70.5 to 157.2	18,531	765.4	110.5	181	89	12.8	98.4
3	3	157.2 to 235.8	7,039	819.3	195.5	177.8	27	4.49	30.9
4	1	235.8 to 314.4	6,177	868.5	275.9	185.9	12.4	2.95	14.8
5	5	314.4 to 393	4,521	900	346.6	188.8	8.92	2.45	11.1
6	5	393.0 to 471.6	1,598	965.6	424.8	197	3.38	0.947	4.24
1 7	7	471.6 to 550.2	951	1.016	506.5	205.5	1.60	0.427	1.97
8	3	550.2 to 628.8	647	1,053.6	580.7	220.1	1.09	0.311	1.37
9	9	628.8 to 707.4	378	1,098.5	660	233.4	0.690	0.209	0.885
10	0	707.4 to 786.6	264	1,151.1	744.5	253.9	0.525	0.204	0.721
							·		

Fuel density at 6.5 pounds per gallon

Notes: a Includes nose gear in the UP position

- b Includes tires, truck assembly, air, wheels, and side strut of main landing gear in the UP position
- c Includes the horizontal tail, vertical tail, and refueling boom
- d Includes inner cylinder, outer cylinder, trunnion, and landing gear support structure in wing
- e Cruise condition

Table V --- Concluded

Airplane	Weight	Conter-of-gravity location (in.)			Moment of inertia about cg (lb-in. ² x 10 ⁻⁶)		
section	(ib)	Body balance station	Body buttock iine	Body waterline ^e	Pitch	Roll	Yaw
Body Wing Nacelles Nacelles	137,752 108,566 10,938 10,744	876.9 804.6 713.7 898.1	0 169.4 322 552	198.7 184.3 159 197	20,600 1,110 16.9 16.2	468 2,140 2.64 2.60	20,500 3,220 15.7 15
Tutal/avg	268,000	841.8	0	191.2	22,300	10,200	31,800

Table VI Weight Condition C (Intermediate Gross Weight with Structural Reserve Fuel; Gress eight: 190,590 Pounds)

Panel	Bady	Weight (1b)	Center-of-gravity location (in.)			Moment of inertia about cg (lb-in. ² x i0 ⁻⁶)		
I BUIDDEC!	balance station		Body balance station	Body buttock line	Body waterline ^e	Pitch	Rol!	Yaw
1	130 to 259	2,771	220	0	218	4.45	3.95	3.57
2 ^a	259 ຄົນ 360	5,290	320	1	207	15.5	14.6	11.3
3	360 io 420	9,810	393		189	12. i	18. i	12.ô
4	420 to 480	11,715	450		182	11.9	20.5	15.7
5	480 to 540	11,482	510		181	10.4	18.8	15.3
6	540 to 620	11,633	571		182	12.3	21	15.9
7	620 to 755	3,055	689		210	11	11.2	9.33
8 5 ⁵	755 to 820	1,500	802		208	3.68	5.47	. 2.88
2 ₂	820 to 890	9,333	872		185	12.7	27.2	26.5
10	890 to 960	2,037	926	İ	209	5.11	7.43	4.03
11	960 to 1020	11,023	909		183	10.1	18.7	15.9
12	1020 to 1080	10,717	1,050		189	8.61	15.9	13.7
13	1080 to 1140	10,042	1,110		184	8.56	16.3	12.2
14	1140 to 1200	10,004	1,170		189	10.6	18.3	12.4
15	1200 to 1280	5,958	1,223		191	5.47	9. 94	7.57
16	1280 to 1360	1,252	1,321		217	2.58	3.79	2.55
17	1360 to 1440	5,035	1,406	į į	220	3.63	6.45	6.85
18 ^C	1440 to 1676	9,699	1,569	0	288	130	113	102

WING/SIDE

1 2 ^d 3 4 5 6 7 8	0 to 70.5 70.5 to 157.2 157.2 to 235.8 235.8 to 314.4 314.4 to 393 393.0 to 471.6 471.6 to 550.2 550.2 to 628.8 628.8 to 707.4	1,645 5,673 2,793 2,785 1,569 1,171 951 647 378	1,016 1,053.6 1,098.5	45.6 115.5 196.6 261 344.3 432.6 506.5 580.7 660	173 176.3 176.1 180.8 186.5 196.5 205.5 220.1 233.4	6.13 26.9 13.4 6.24 4.47 2.58 1.60 1.09 0.690	1.10 4.65 2.62 1.51 0.876 0.654 0.427 0.311 0.209	6.73 30.6 15.4 7.50 5.20 3.16 1.97 1.37 0.885
9	628.8 to 707.4	378	1,098.5	660	233.4	0.690	0.209	0.885
10	707.4 to 786.6	264	1,161.1	744.5	253.9	0.525	0.204	<u>0.7</u> 2

Fuel density at 6.5 pounds per gallon

Notes: a Includes nose gear in the UP position

b Includes tires, truck assembly, air, wheels, and side strut of main landing gear in the UP position

c Includes the horizontal tail, vertical tail, and refueling boom

d Includes inner cylinder, outer cylinder, trunnion, and landing gear support structure in wing

e Cruise condition

Teble VI --- Concluded

Airplane	Weight	Cente	r-of-gravity loca (in.)	tion	Moment of inertia about cg ('b-in. ² x 10 ⁻⁶)			
section	(ib)	Body balance station	Body butlock line	Body waterline ^e	Pitch	Roll	Yaw	
Body	133,156	858.4	C	196.9	19,200	452	19,200	
Wing	35,752	876.5	243.7	184.4	422	1,020	1,420	
Nacelles	10,938	713.7	322	159	16.9	2.64	15.7	
Nacelles	10,744	898.1	552	197	16.2	2.60	_15	
Total/avg	190,590	854.6	0	192.4	20,000	8,030	27,400	

Table VII. Weight Condition D (Operating Weight Empty with Structural Reserve Fuel; Gross Weight: 107,260 Pounds)

BODY

Panel Body balance		Weight	Cento	er-of-gravity loc (in.)	ation	Moment of inertia about cg (lb-in. ² x 10 ⁻⁶)		
number station		Body balance station	Body buttock line	Body waterline ^e	Pitc	Roll	Yaw	
1	130 to 259	2,771	220	Q	218	4.45	3.95	3.57
2 ^a 3	259 to 360	5,290	320		207	15.5	14.6	11.3
3	360 to 420	2,258	384		215	6.14	9.33	4.78
	420 to 480	i,495	451		220	3.59	5.45	2.79
5 6 7	480 to 540	1,2€2	513		214	3.03	4.60	2.36
6	540 to 620	1,925	584		212	4.62	7.02	3.60
	620 to 755	3,055	689		210	11	11.2	9.33
8 9b	755 to 820	1,500	802		208	3.68	5.47	2.88
90	820 to 890	9,333	872		185	12.7	27.2	26.5
10	890 to 960	2,037	926		209	5.11	7.43	4.03
11	960 to 1020	1,806	982		185	4.33	6.59	3.37
12	1020 to 1080	1,104	1,051		211	2. 6 5	4.03	2.06
13	1080 to 1140	1,145	1,110		214	2.63	4.07	2.14
14	1140 to 1200	1,834	1,174		221	3.94	6.26	3.43
15	1200 to 1280	1,197	1,238		215	2.74	3.97	2.52
16	1280 to 1360	1,252	1,321		217	2.58	3.79	2.55
17	1360 to 1440	1,273	1,395	l ↓	223	2.07	2.83	2.11
18 ^C	1440 to 1676	9,288	1,574	0	291	121	111	95.3
				WING/SIDE				
1	0 to 70.5	1,645	746.3	45.6	173	6.13	1.10	6.73
2 ^d	70.5 to 157.2		816	115.5	176.3	26.9	4.65	30.6
3	157.2 to 235.8	2,793	842.2	196.6	176.1	13.4	2.62	15.4
4	235.8 to 314.4	2,785	868.7	261	180.8	6.24	1.51	7.50
5	314.4 to 393	1,569	906.3	344.3	186.5	4.47	0.876	5.20
6	393.0 to 471.6		。 974.9	432.6	196.5	2.58	0.654	3.16
7	471.6 to 550.2		1,016	506.5	205.5	1.60	v.427	1.97
8	550.2 to 628.8	647	1,053.6	580.7	220.1	1.09	0.311	1.37
9	628.8 to 707.4	378	1,098.5	660	233.4	0.690	0.209	0.885
10	707.4 to 786.6	264	1,161.1	744.5	253.9	0.525	0.204	0.721

Fuel density at 6.5 pounds per gallon

Notes: a Includes nose gea i. the UP position

b Includes tires, truck assembly, air, wheels, and side strut of main landing gear in the UP position

c Includes the horizontal tail, vertical tail, and refueling boom

d includes inner cylinder, outer cylinder, trunnion, and landing gear support structure in wing

e Cruise condition

Table VII --- Concluded

Airplane	Weight	Cent	er-of-gravity loc (in.)	Moment of inertia about cg (lb-ia. ² x 10 ⁻⁶)			
section	(lb)	Body balance station	Body buttock line	Body waterline ^e	Pitch	Roll	Yaw
Body Wing Nacelles Nacelles	49,826 35,752 10,938 10,744	870.5 713.7	0 243.7 322 552	221.4 184.4 159 197	9,300 422 16.9 16.2	381 1,020 2.64 2.60	9,810 1,420 15.7 15
Total/avg	:07,260	871	0	200.3	10,700	7,910	18,100

APPENDIX II STIFFNESS DATA

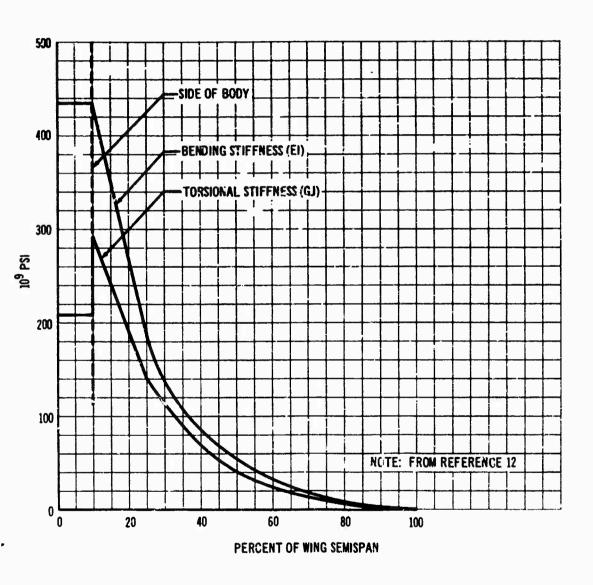


Figure 12. Wing Vertical-Bending and Torsion Stiffness

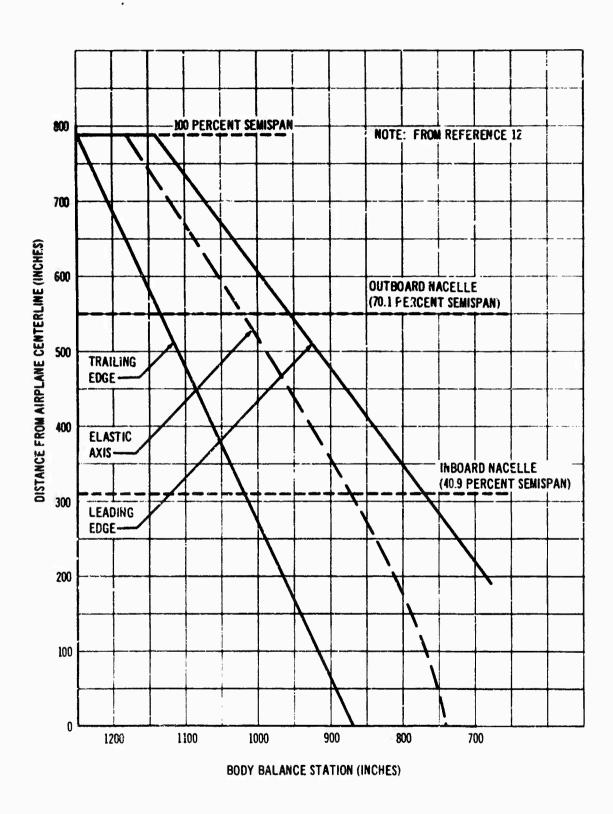


Figure 13. Wing Elastic-Axis Location

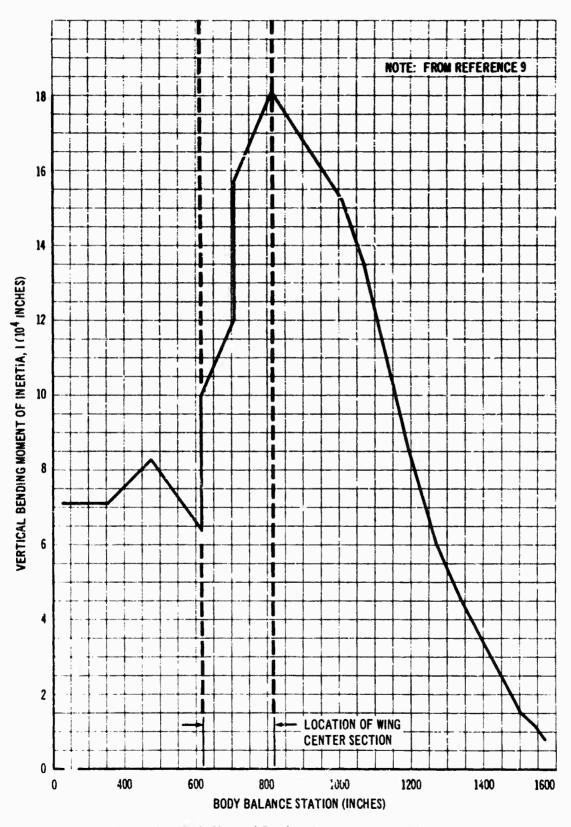
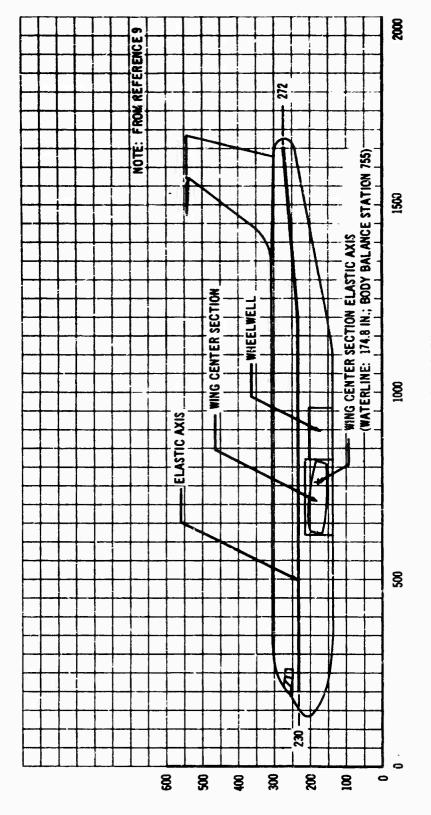


Figure 14. Body Vertical-Bending Section Moment of Inertia



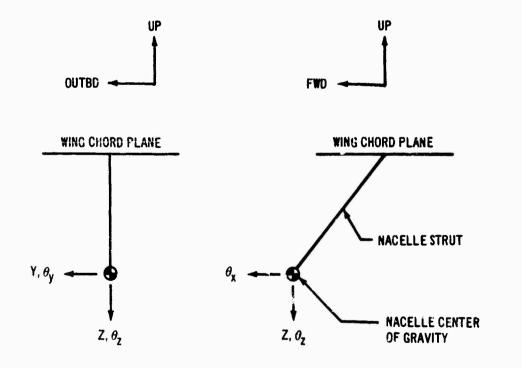
BODY BALANCE STATION (INCHES)

Figure 15. Body Elastic-Axis Location

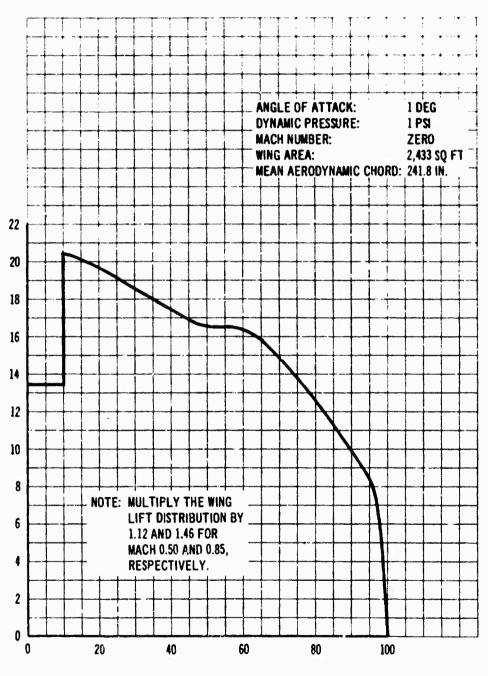
MATERLINE (INCHES)

Table VIII. Nacelle Cantilever Mode Shapes and Frequencies

	Vertical	bending	Side bending							
Location	Z (in.)	θ _y (rad.)	Y (in.)	θ ₂ (tad.)	θ _χ (rad.)					
MODE SHAPES										
inboard nacelle	108	-1	1	0.0136	-0.0057					
Outboard nacelle	98	-1	1	0.0150	-0.0061					
	FREQUENCIES (FR	OM REFERENCE 4)								
Inboard nacelle	4.44	2.31 cps								
Outboard nacelle	4.81	2.50 cps								



APPENDIX III AERODYNAMIC DATA



PERCENT OF WING SEMISPAN

Figure 16. Wing Lift Distribution

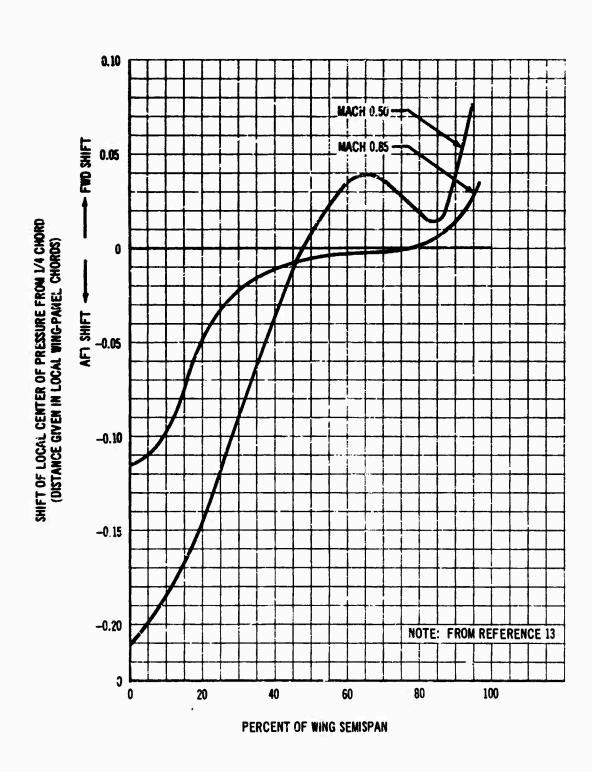


Figure 17. Local Center-of-Pressure Location



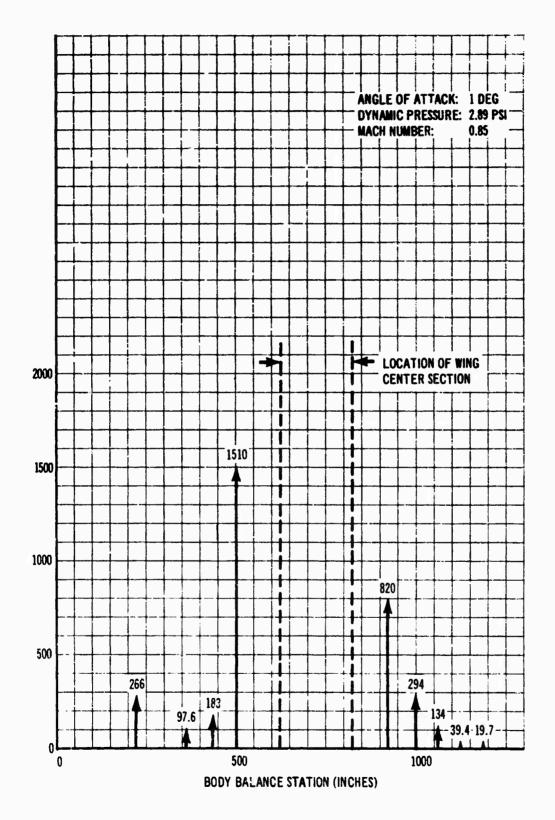


Figure 18. Body Lift Distribution (Mach 0.85)

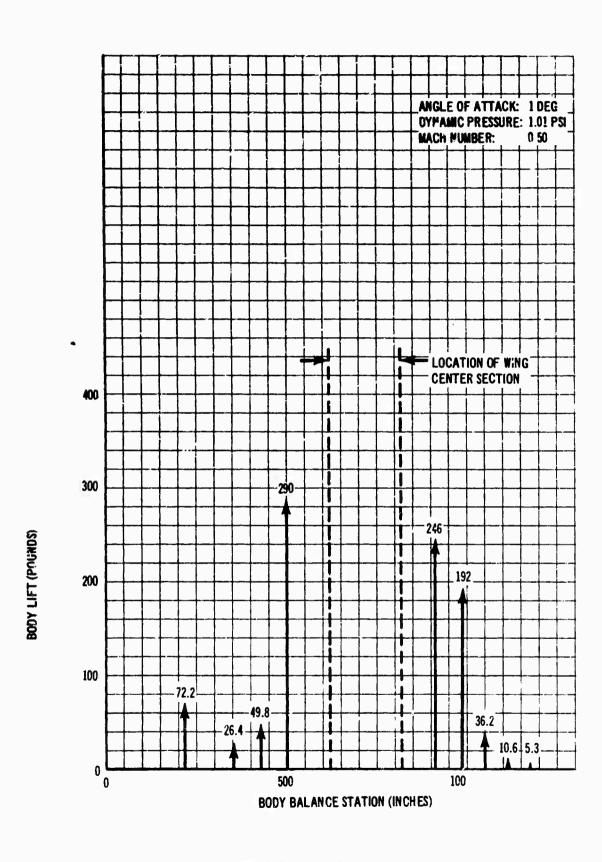


Figure 19. Body Lift Distribution (Mach 0.50)

Table IX. Rigid-Airplane Derivatives $\left(\frac{1}{Radian}\right)$

Mach	Ana	lysis	Wind tunnel (from reference 6)		
number	CLa	C _{Ma}	CLa	CMa	
0.50	5.06	1.51	5.14	1.50	
0.85	6.49	1.68	6.71	1.70	

 $L = \varphi s C_{La} a$

 $M = qs\bar{c}C_{M\alpha}a$

L = lift

M = pitching moment about body station 837.9

q = dynamic pressure

s = wing area = 2,433 square feet

a = angle of attack

c = wing mean aerodynamic chord = 241.8 inches

Table X. Rigid-Horizontal-Stabilizer Lift at 24,000-Foot Altitude

Mach number	Lift (lb/rad.)	For flexible horizontal stabilizer, multiply lift by:
0.50	61,068	0.971
0.80	183,156	0.923

Note: Horizontal stabilizer center of lift is at body balance station 1581.3.



APPENDIX IV AIRPLANE FREE-FREE MODE SHAPES 100 80 MODE 1 (1.34 CPS)-60 MODE 4 (2.58 CPS) 40 MODE 2 (2.30 CPS) 20 MODE 3 (2.44 CPS) 0 MODE 6 (3.52 CPS) -20 MODE 5 (3.26 CPS) -40 MODE 8 (6.08 CPS) MODE 7 (4.96 CPS) --60 0 10 20 30 40 50 60 70 80 90 100 PERCENT OF WING SEMISPAN

Figure 20. Wing Vertical Displacement in the Normalized Free-Free Airplane Modes; 297,000-Pound Gross Weight (Weight Condition A)

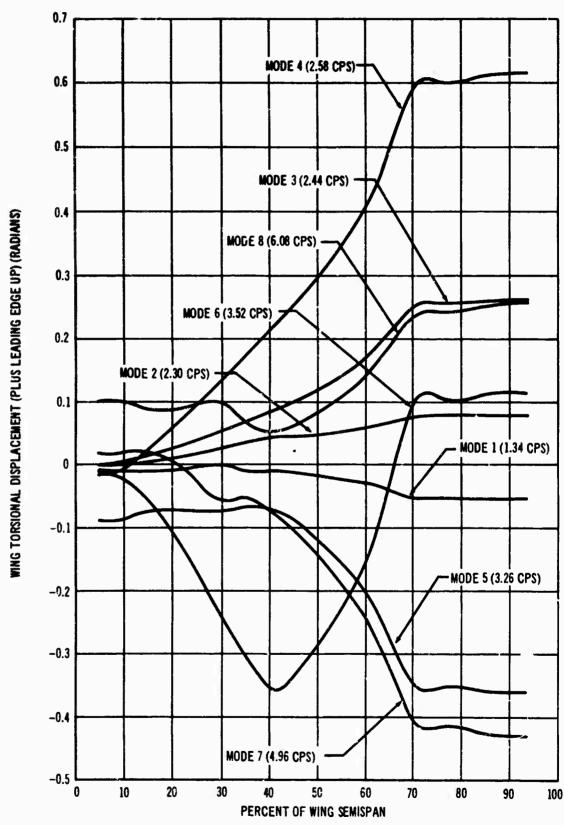


Figure 21. Wing Torsional Displacement in the Normalized Free-Free Airplane Modes; 297,000-Pound Gross Weight (Weight Condition A)

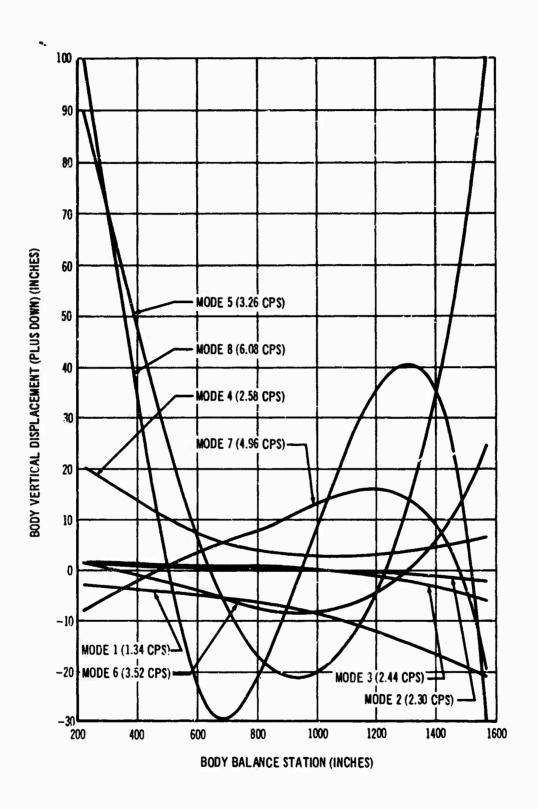


Figure 22. Body Vertical Displacement in the Normalized Free-Free Airplane Modes; 297,000-Pound Gross Weight (Weight Condition A)

Table XI. Nacelle Mode Shapes (Weight Condition A)

Nacelle position	Displace- ment		Mode number								
		1	2	3	4	5	6	7	8		
Inboard	x	-1.78	-1.68	-3.89	-8.30	+2,31	+15.20	+7.51	-5.05		
	ÿ	-5.42	+7 4.5 5	-8.49	-18.88	+0.84	+2.72	-0.16	-0.28		
!	Ī	+0,27	-8.29	-16,25	- 51 .9 5	-31.09	+125.88	-6.64	+18.78		
	$\theta_{\mathtt{X}}$	+0.0923	-0.4413	+0.0280	+0.0091	+0.0170	+0.0884	-0.0754	+0.0824		
	θ_{y}	+0.0375	+9.0565	+0.1293	+0.3345	+0.0751	-1.0573	-0.1073	-0.2043		
	$\theta_{\mathbf{z}}$	-0.0214	+1.0000	-0.1327	-0.3401	+0.0299	+0.1254	-0.0669	+0.0649		
Outhoard	x	3,37	-3.72	-10.29	-28.16	+4.64	-11.00	+8.47	-0.62		
	ÿ	-11.32	+1.70	+70.74	-54.94	+12.57	+4.17	+0.98	+0.58		
	ž	+42.33	+6.97	-21.99	-85.85	+41.71	-55.77	+12.70	-3.20		
	$ heta_{\mathtt{X}}$	+0.2173	-0.0159	-C.5017	+0.2549	+0.1643	+0.0990	+0.3341	-0.1780		
i	θ_{y}	+0.0425	+0.1035	+0.3005	+0.8969	-0.3985	+0.5370	-0.3467	+0.2164		
	$\hat{\theta_{\mathbf{Z}}}$	-0.0407	+0.0206	+1.0000	-0.8936	+0.3983	+0.1712	+0.3107	-0.1431		

Note: Sign convention for nacelle cg positive displacements

x Aft

 $\begin{array}{l} \theta_{\rm X} & {\rm Roll, \ bottom \ inboard} \\ \theta_{\rm Y} & {\rm Pitch, \ nose \ up} \\ \theta_{\rm Z} & {\rm Yaw, \ nose \ outboard} \end{array}$

y Outboard z Down

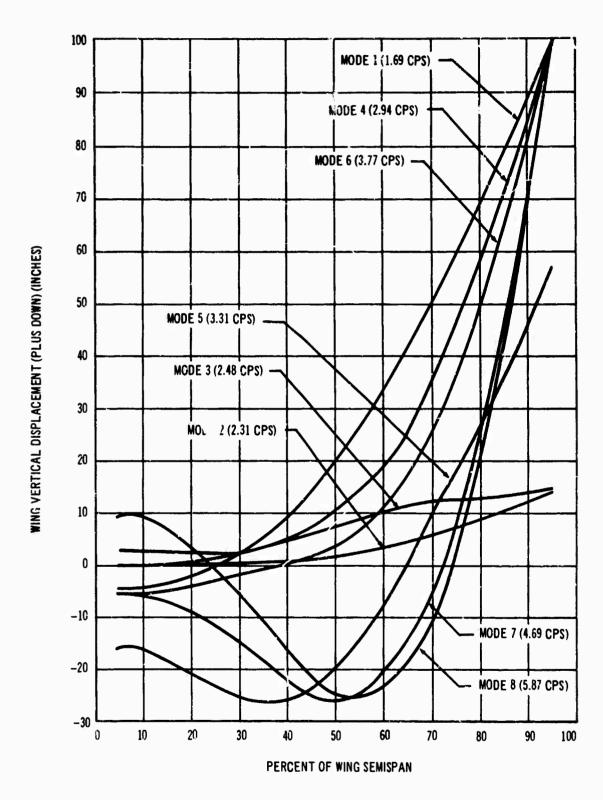


Figure 23. Wing Vertical Displacement in the Narmalized Free-Free Airplane Modes; 268,000-Paund Gross Weight (Weight Canditian B)

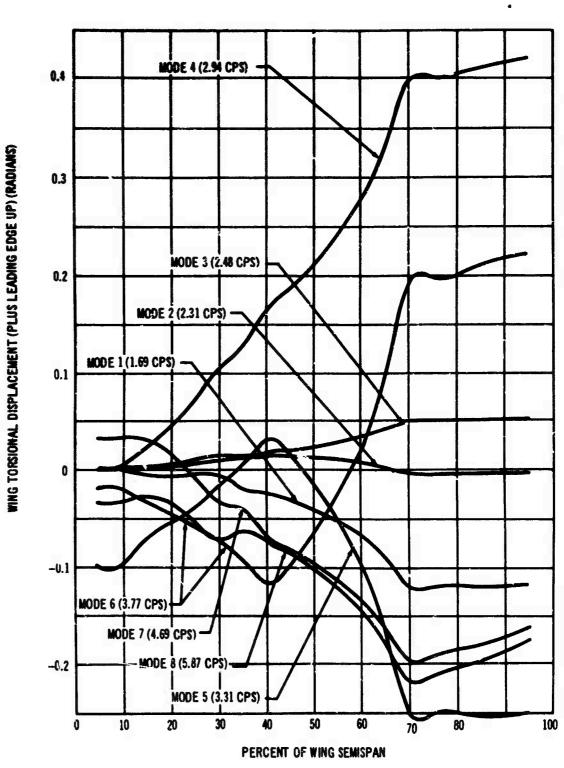


Figure 24. Wing Torsional Displacement in the Normalized Free-Free Airplane Modes; 268,000-Pound Gross Weight (Weight Condition B)

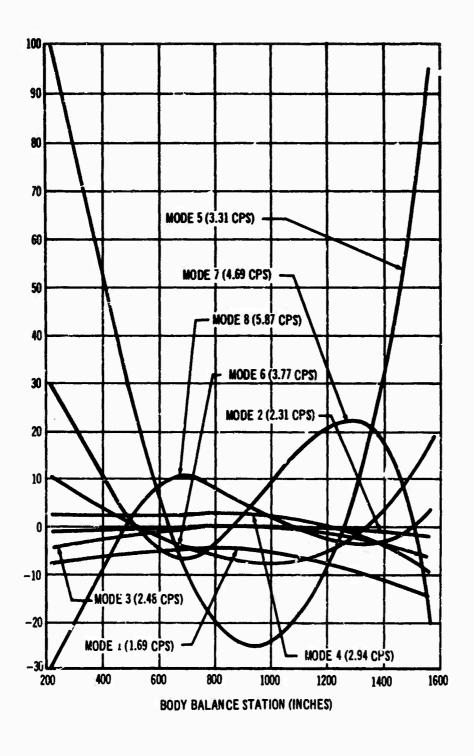


Figure 25. Body Vertical Displacement in the Normalized Free-Free Airplane Modes; 268,000-Pound Gross Weight (Weight Condition B)

Table XII. Nacelle Mode Shapes (Weight Condition B)

Nacerie position	Displace- ment	Mode number								
		1	2	3	4	5	6	7	8	
Inboard	x	-1.42 ·	-0.73	-1.41	-7.79	-3.03	+3,82	+4.94	15.54	
	ÿ	-8.81	+73,67	+3,84	-2.47	+0.38	:1 . 12	+0.04	+0.22	
	Ž	+5.77	~2.51	-0.97	-39.67	-70.28	+51.44	+3.75	-5.20	
	$\theta_{\mathbf{x}}$	+0.1234	-0.4220	-0.0095	-0.0217	±0.0084	+0.0541	-0.0174	-0.0462	
	θ_{y}	+0.0248	+0.0225	+0.0353	+0.3295	+0.3905	-0.4251	-0.1930	-0.0489	
	θ _z	-0.0578	+1.0000	+0.0629	-9.0641	+0.0141	+0.0667	-0.0139	-0.0354	
Outboard	x	-0.19	-0.56	-1.76	-19.06	+0.64	-13.20	+2.21	+0.51	
	ÿ	-15.85	-6.20	+ 67. 73	-3.85	+9.75	+1.29	-0.60	-1.11	
	Ž	+57.66	+5.46	+8.93	-29.74	+45.09	-26.40	+3.84	+3.04	
	$\theta_{\mathtt{X}}$	+0.2614	+0.0571	-0.4315	-0.0129	+0.1461	+0.0487	+0.2417	+0.2426	
	θ_{y}	-0.0576	+0.0029	+0.0251	+0.5765	-0.3250	+0.4707	-0.0855	-0.1253	
	θ_{Z}	-0.0944	-0.0763	+1.0000	-0.0890	+0.3251	+0.0688	+9.1982	+0.1886	

Note: Sign convention for nacelle cg positive displacements

 $\begin{array}{l} \theta_{\rm X} \quad {\rm Roll, \ bottom \ inboard} \\ \theta_{\rm Y} \quad {\rm Pitch, \ nose \ up} \\ \theta_{\rm Z} \quad {\rm Yaw, \ nose \ outboard} \end{array}$

x Aft y Outboard z Down

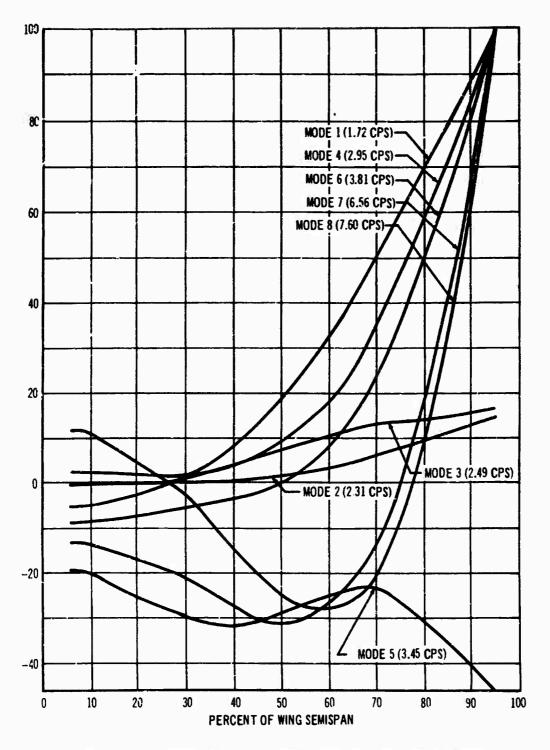


Figure 26. Wing Vertical Displacement in the Normalized Free-Free Airplane Modes; 190,590-Pound Gross Weight (Weight Condition C)

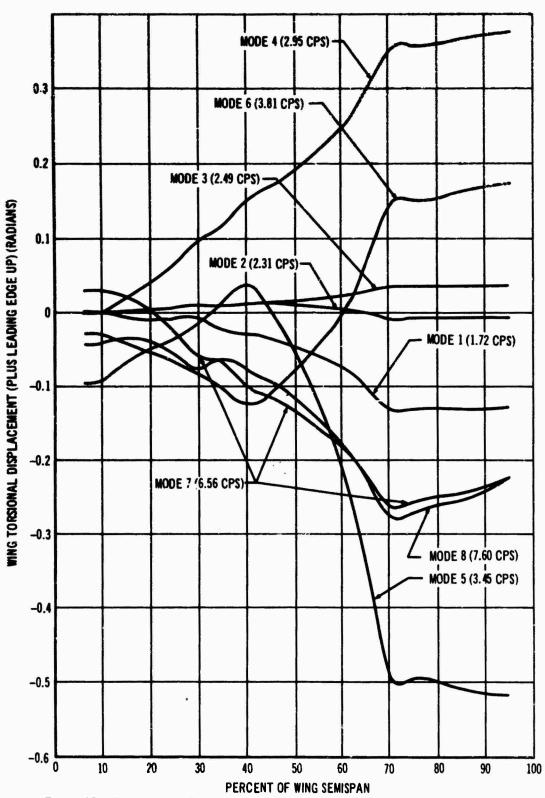


Figure 27. Wing Torsional Displacement in the Normalized Free-Free Airplane Modes; 190,590-Pound Gross Weight (Weight Condition C)

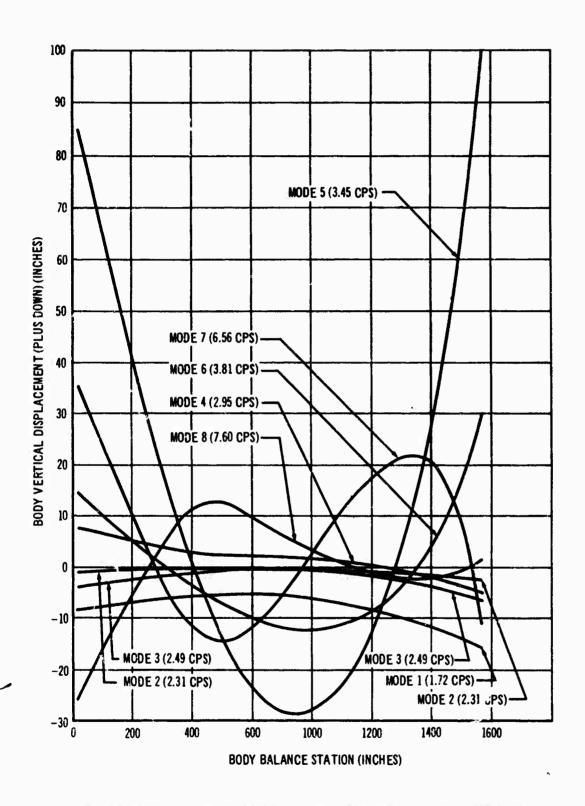


Figure 28. Body Vertical Displacement in the Normalized Free-Free Airplane Modes; 190,590-Pound Gross Weight (Weight Condition C)

Table XIII. Nacelle Mode Shapes (Weight Condition C)

Nacelle position	Displace- ment	Mode number								
		1	2	3	4	5	6	7	8	
Inboard	x	-1.28	-0.72	-1.25	-7.34	-2.69	+4,01	+5.92	+5.88	
	ÿ	-9.2 1	-73.66	+4.31	-2.15	-0.18	+1.10	ZERO	+0.35	
	ž	+5.35	-2.51	-0.31	-38.49	-90.38	+45,24	+1.88	-4.77	
	θ_{X}	+0.1256	-0.4218	-0.0100	-0.0195	-0.0053	+0.0570	ZER0	-0.0530	
	θ_{y}	+0.0219	+0.0222	+0.0298	+0.3146	+0.5241	-0.3998	-0.2097	-0.0338	
	$\theta_{\mathbf{Z}}$	-0.0632	+1.0000	+0.0709	-0.0561	-0.0079	+0.0687	-0.0001	-0.0386	
Outboard	x	+0.04	-0.53	-1.37	-17.78	+14.83	-12.19	+4.54	+3.14	
	ÿ	-16.83	-6.91	+67.23	-1.85	+6.36	+1.68	-1.26	-1.68	
	ī	+57.98	+5.93	+11.59	-23.53	+62,89	-21.60	÷1.54	+0.51	
	θ_{X}	+0.2724	+0.063€	-0.4198	-0.9065	+0.1306	+0.0638	+0.2679	+0.2479	
	θ_{y}	-0.0665	+0.0010	+0.0081	+0.5188	-0.7789	+0.4108	-0.1317	-0.1838	
	θ_{Z}	-0.1049	-0.0849	+i.0000	-0.0430	+0.2427	+0.0944	+0.2075	+0.1816	

Note: Sign convention for nacelle cg positive displacements

x Aft y Outboard z Down

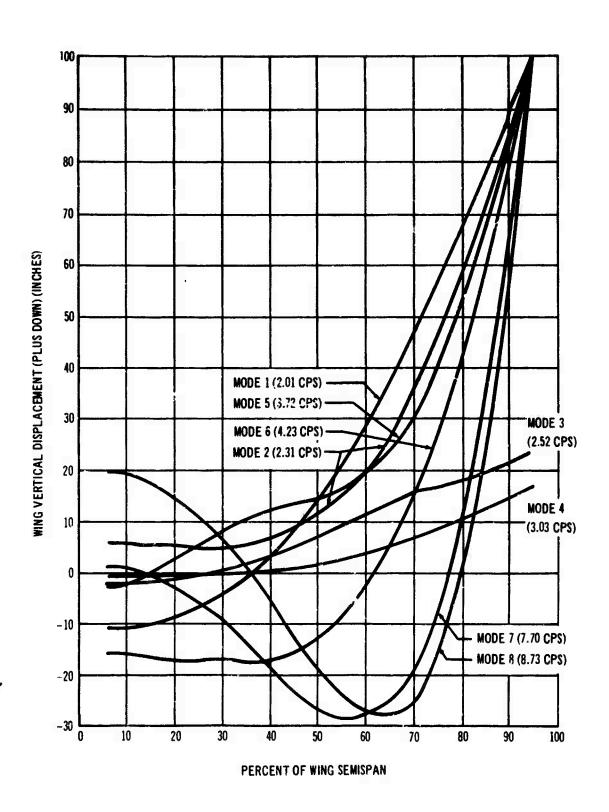


Figure 29. Wing Vertical Displacement in the Normalized Free-Free Airplane Modes; 107,260-Pound Gross Weight (Weight Condition D)

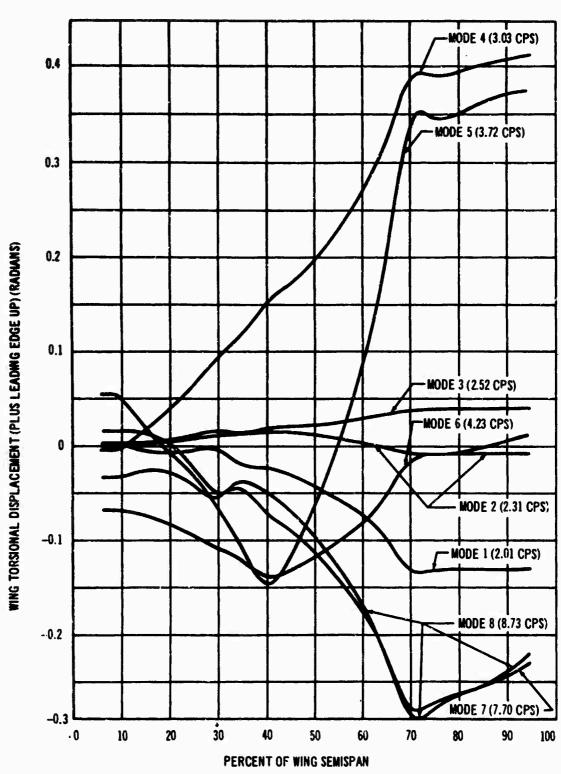


Figure 30. Wing Torsional Displacement in the Normalized Free-Free Airplane Modes; 107,260-Pound Gress Weight (Weight Condition D)

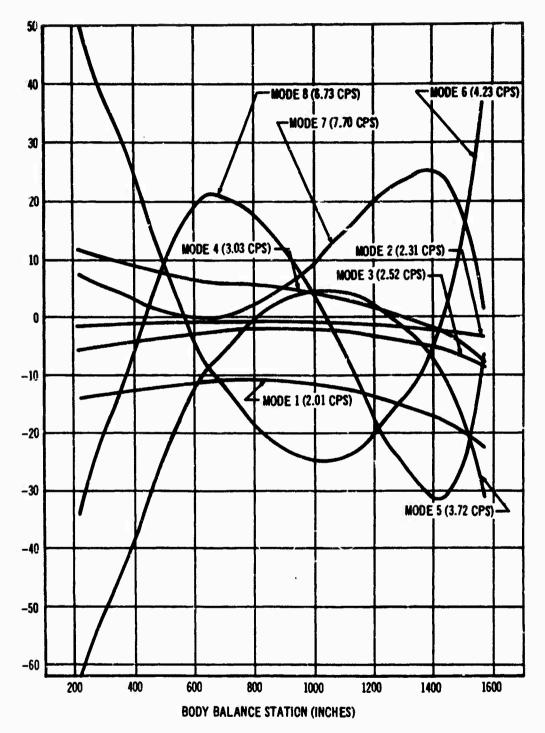


Figure 31. Body Vertical Displacement in the Normalized Free-Free Airplane Modes; 107,260-Pound Gross Weight (Weight Condition D)

Table XIV. Nacelle Mode Shapes (Weight Condition D)

Macelle position	Displare- mer/		Mode number						
		1	2	3	4	5	6	7	8
inboard	Ÿ(-1.53	-0.79	-1,64	-7.12	+5.13	+5.05	÷5.54	+5.79
	ÿ	-10.75	+73.58	+4,77	-2.22	+1.48	+0.66	+0.25	+0.59
	Ž	-1.92	-3.27	-3.52	-33.41	+94.06	+21.32	-2.44	-5.21
	$\theta_{\mathtt{X}}$	+0.1351	-0.4204	-0.0103	-0.0231	+0.0622	+0.0611	-0.0393	-0.0778
	θy	+0.0366	+0.0251	+0.0465	+0.2945	-0.6963	-0.2944	-0.0972	+0.0486
	θ_{Z}	-0.0833	+1.0000	+0.0793	~0.0604	+0.0801	+0.0639	-0.0288	-0.0552
Outboard	x	-0.16	-0.57	-1.84	- 18.93	-15.69	-7.66	+5.19	+5.52
	ÿ	-19.48	-8.26	+66.64	-3.14	-0.43	+2,00	-1.66	-1.75
	Ž	+54.51	+6.54	+13.17	-28.40	-47.22	-5.98	+0.48	-1.07
	$\theta_{\mathtt{X}}$	+0.2969	+0.0759	-0.4062	-0.0146	-0.0133	+0.1489	+0.2558	+0.2235
	θ_{V}	-0.0653	+0.0004	+0.0144	+0.5663	+0.7015	+0.1928	-0.1739	-0.2063
	$\theta_{\mathbf{Z}}$	-0.1371	-0.1016	+1.0000	-0.0763	-0.0202	+0.1700	+0.1889	+0.1589

Note: Sign convention for nacelle cg positive displacements

x Aft y Outboard z Down

 $\begin{array}{l} \theta_{\rm X} \quad \text{Rol1, bottom inboard} \\ \theta_{\rm Y} \quad \text{Pitch, nose up} \\ \theta_{\rm Z} \quad \text{Yaw, nose outboard} \end{array}$

APPENDIX V STRESS FREQUENCY RESPONSE FUNCTIONS

Table XV Stress Frequency Response Functions (Analysis Condition 1)

(PSIFPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

297, QQU LB CUTOF? FREQUENCY: 24,000 FT 0, 85

10 CPS

PERCENT SEMISPAN. 27

SEGMENT NUMBER 10

INCREMENTAL :	SHEAR STRESS	INCREMENTAL	AXIAL STREET			
PFAL	INGINARY	REAL	MAGINARY			This sect
				_	-0.	674 0.10
0.847215-01 0.24347E ; 2	0.35385E 02 0.29947E 02	-0.40681E 00 0.19898E 93	0.30343E 03 0.24210E 03	0. -0.	-G.	6.36
0.44132F 02	0.83492F 03	0.37670F 33	0.74904E 02	-0.	-0. 0.	0.36 0.74
0.5393F 02	-0.51332F 01 -0.16505F C2	0.47339E 03 0.45739E 03	-0.14192E 03	-0. -0.	o. o.	0.50
0.46190F 02	-0.22434F 02	0.400738 03	-0.19746E 03	-0.	0.	3.60
0.412536 02	-0.27016E 02 -0.31773E 02	0.35800E 03 0.32986E 03	-0.24096E 03 -0.28530F 03	-0. -0.	o. o.	0.70 0.40
0.98231F 02 0.36571F 02	-0.37815F 02	0.31144E 33	-0.33932E 03	-0.	0.	0.90
0.35926F 02	-0.59:53F 02	0.29597E 03 0.27761E 03	-0.51729E 03 -0.60276E 03	-0. -0.	e. 0.	1.00
0.36653F 02	-0.571885 02 -0.12897F 03	0.204618 03	-0.10236E 04	0.	0.	1.30 1.34 1.40
0.22848F 02	-0.16771F C3	0.96340E 02 -0.14204E 03	-0.12780E 04 -0.14050E 04	o. o.	-0. -0.	1.49
-0.42896F 01	-0.18946F 03 -0.20625F 03	-0.363518 03	-0.14789E 04	0.	- 0.	1.47
-n.75e85F 03	-0 17111E 03	-0.69482E 03	-0.10976E 04	0.	- C- - O-	1.50 1.55 1.60 1.65 1.80
-0.19911E 03	-0.35902F 02 0.46462F 02	-0.133322 04 -0.16662E 04	-0.10140E 63 0.40630E 03	-0.	0.	1.60
-0.182F7F 03	0.70084E 02	-0.11743E 04	0.40470E 03	-0.	0.	1.65
-0.70840E 02	0.64637F 02 0.62249F 02	-0.36373E 03 -0.20150E 03	0.30779E 03 0.23180E 03	-0. -0.	0. 0.	1.90
-0.3073AF 02	0.63716F 02	-0.12039E 03	0.17513E 03	-0.	0.	2.00
-6.20436F 02	0.69980F 02	-0.77273E 02 -0.54149E 02	0.12793E 03 0.67179E 02	-0. -0.	0. 0.	2.10
-n.9A218F 0:	0.74699E_02 0.70093E_02	-0.520108 92	0.70001E 02	0.	ð.	2.30 2.30
0.172338 02	0.77041F 02	-0.52613E 02	0.46193E 02	ø.	0.	2.39 2.40
0.34477E 02	0.79877F 02 0.79724F 02	-0.63530E 02 -0.856462 02	0.56245E 02 0.35754E 02	0. 0.	0.	2.5
0.390098 02	0.72921F 02	-0.97472E 02	0.56486E 02	0.	0.	2.44 2.47
0.505338 02	0.63392F 02 0.69316F 02	-0.12910E 03 -0.12646E 03	0.10032E 03	o. o.	0.	2.50
0.45535F 02	0. 84011F 02	-0.84240E 02	0.97083E 02	0.	9.	2.3
0.546315 02	0.10597F 03	-0.633122 02 -0.68845E 02	0.66091E 02 0.56672E 02	0.	e. 0.	2.50 2.54 2.56 2.65
0.101025 63	-0.1053 OF 03	-0.962498 02	0.14603E 03	0.	-0.	2.70 2.60
0.29234F 03	-0.14149E 03 -0.11812E 03	-0.14619E 03 0.16352E 02	0.14327E 03 0.11391E 03	-0.	-0. -0.	2,00 3,00
-0.28309F 02	-0.11812F 05	0.24914E 32	0.96723E 02	-0.	-0.	3.30
-0.71372F 02	-0.85784E 02	0.307918 02	0.90310E 02	-0.	- G. - G.	3.30 3.30 3.36
-0.81036E 02	-0.83382E 92 -0.82515E 02	0.33323E 02 0.36221E 02	0.87505E 02	-0. -0.	-0.	
-0.10972F 03	-0.71410F 02	0.43080£ 02	0.62441E 02	-0.	-0.	3.39 3.35
-0.15204F 03	0.69707£ 02 0.16767F 03	0.53623E 02 0.10323E 02	0.43047E 02 0.61216E 02	-3.	0.	3.22
-0.16405F 03	0.21643F 03	0.477208 01	0.77443E 02	-o. -o.	0. 0.	3.32 3.36 3.60
-0.7.035F 02	0.97863F 02 0.12316F 02	0.14542E 02 0.47708E 02	0.451316 02	-0.	0.	3.6
0.47/26F 02	-0.540858 01	D.47620E 02	0.42118E 02 0.32363E 02	0 c	-0. -0.	3.70 3.85 .00
0.103275 02	-0.10415E 02 -0.10380E 02	C.44216E 02	0.25349E 02	0.	-0.	
0.187C7E-00	-0.920a4F 01	0.42629E 02 0.47327E 02	0.18653E 02 0.11968E 82	0. -0.	- 0. - 0.	4.20 4.30
-0.35289F 01	-0.84821F 01	0.401778 02	0.28725E 01	-0.	-0.	4.70
-0.49874F 01 -6.69816F 01	-0.67778E 01 -0.61227F 01	0.73033E C2 0.95124E 02	-0.33318E 02 -0.82493E 02	-0. -0.	0. 0.	4.80
-0.725AIF 01	-0.494398 01	0.842978 02	-0.03031E 02	-0.	ŏ.	4.96 5.60
-0.70801F 01	-0.46116F 01 -3.37278F 01	-0.17010E 02 -0.30270E 02	-0.34016E 02	o. o.	0.	5-15
-0.913796 01	-0.25388F 01	-0.22972E 02	-0.77197E 01	-0.	-0. -0.	5.30 6.67
-0.10893F 0: -0.13055F 02	-0.11493E 01 0.38635E 01	-0.20631E 02 -0.23939E 02	-0.3668/E 81 0.10526E 02	-0.	• • •	5.8
-0-17306F 02	0. 60353F 01	-0.352038 02	0.14942E 02	-0. -0.	· 6.	5.65 6.00
-0.18059F 02	0.89105F 01 0.12317E 02	-0.37039E 02 -0.37163F 02	0.23549E 02 0.33600E 62	-0.	0.	€.93 €.32
-0.17110E 02	0.13721F 02	-0.31713E 02	C.36920E 02	-0. -0.	e. 0.	€.œ
-0.20813F 01	0.49234E 01 0.3636FE 01	0.13415E 02 0.13748E 02	3.936718 01	-0.	0.	6.30
-0.27608F 01	0.38199E 01	0.11146E 32	0.31734E 6 1 3.13424E-06	-0. -0.	6. 6.	6.40 6.40
-0.32540F 01	0.43904F 01 0.51455E 01	0.73743E 41	-0.66774E DB	-0.	٥.	7.60
-0.11105F 01	0.484796 01	0.42403E 01	-0.17344E 01 -0.23299E 61	-e. -o.	e. e.	7.40
0.816F2F 00	0.37366F 01	0.33497E 01	-0.28373E 01	-0.	0-	9.00
0.767896 01	0.	0.22131€ 01	0.	•.	••	0.60

(PSIFFPS SINUSOIDAL GUST)

10 CFS

GROSS WEIGHT: ALTITUDE: MACH NUMBER: 297, 000 LB CUTOFF FREQUENCY: 24,000 FT 0.85

PERCENT SEMISPAN: 27 SEGMENT NUMBER 14

*	- MALE STO-1	TO COMPANY AL	AXIAL STREET			
974	1 PAGE I MARY	RE AL	:RAGIRARY			
						Supplement
0.629636-02	0. 107090 03	-0.73216# 00	0.279590 03	•.	-0.	6.39 6.39 6.35 6.44 7.39 6.66 6.70 6.80
0.721526 02	0. 913296 02	0.19037E 03	0.237056 03	-0.	-6.	139
0.1320% 03 0.1447# 03	0.26474E 67 -0.14484E 62	0.33594£ 03 0.42959£ 03	0.71405£ 02 -0.35429£ 02	-0. -0.	-0. 0.	**
0.14010E 03	-0.491994 67	0.415256 03	-0.12879E 03	-0.		2.99
0.13927E 03 0.12390E 03	-0.07313F 02 -0.01121E 02	0.34345E 03 0.32400E 03	-0.17921E 05 -0.21866E 03	-6. -3.	6. 6.	1.6
0.114146 03	-0.95111E 02	0.299346 03	-0.25091E 03	-0.	6.	i.6
0.10407F 0) G.10451E C3	-0.112356 03 -0.170536 03	0.26203E 03 0.27131E 03	-0.30610E 03 -2.46735E 03	-0. -0.	6. 6.	0.90
0.10109F 03	-0.260075 07	0.251928 03	-6.726496 63	-3:	7.	1,00
6.854056 02 6.524996 02	-0.34669E 05 -0.43977E 03	0.19022E 03 0.27427E 02	-0.73049E 03	••	<u>.</u>	1.89 1.74 1.49 1.49 1.59 1.69 1.69 1.89 2.80
-0.222296 02	-0.409548 05	-0,120996 03	-0.12790E 04	• • • • • • • • • • • • • • • • • • •	-0. -0.	3.40 3.44
-0.95435€ 02	-0.52411E 03	-0.331eff 03	-0.13421E 04	••	-0.	1.47
-0.28712E 03	-0.41390E 03 -0.74547E 02	-0.43053E 03 -0.13914E C4	-0.99007E 03 -0.92016E 02	•• •••	-0. -0.	1.20
-0.583G4E 03	0.1174 WE 03	-0.15120E 04	0.370536 03	-0.	٥.	1.6
-0.42834E 03 -0.14804E 03	0.14756£ 03 0.12261£ 03	-0.10650E 04 -0.33001E 03	0.37009£ 03 0.27932£ 03	-0. -0.	6. 6.	1.65
-0.88042E 02	0.105296 03	-0.162066 03	0.21039E 03	-0.	2	1.00
-8.559797 02	0.95527F 02	-0.10944E 03	0.15093E 03	•••	0.	2.00
-0.3590:E 02 -0.19522F 02	0.9222 0E 02 0.20066 02	-G.70124E 02 -0.49139E 02	0.11611E 03 0.79113E 02	-0. -0.	0. 0. 0. 0. 0.	2.30 2.20
0.498246 01	0. 162 POE 62	-0.47205E 02	0.435246 02	••	6.	2.50
0.02011E 01 0.13072E 02	0.818;2E 02 0.82053E 02	-0.47745E 02 -0.59467E 02	0.41919E 02 0.32091E 02	: :	•	4.35
0.170706 02	0.0404 * 02	-0.777216 02	0.32446E 02	i.	<u>.</u>	2.43
0.19791E 02	6. 80379E G2	-0.00454 E 02	0.51299E 02	•.	•	1.39 1.35 1.40 1.40 2.47
0.224ANE 02 0.23214E 02	0.84092E 02 0.97227E 02	-0.11716E 03 -0.11476E 03	0.91037E 02 0.10414E 03	:	ë.	2.47
0.30142F 02	0.1095#E 03	-0.76446E 02	0.001026 02	J.	•.	1.5
0.40325F 02 0.10242F 03	0.12594€ 03 0.1190€€ 03	-0.57434E 02 -0.62475E 02	0.57976E 02 0.51427E 02	€.	0. 0.	2.96
0.173926 03	-0.165406 03	-0.07343E 02	0.13435E 03	•.	-0.	1.99 1.98 1.98 1.60 1.70 1.88
0.302566 03	-0.172364 03	-0.13263E 83 0.14839E 82	0.13002E 03 0.10337E 03	٥٠,	-0.	2.80
-0.32133E 02 -0.67833E 02	-0.13427E 03 -0.11903E 03	0.24424€ 02	0.377748 03	-0. -0.	-0. -0.	3.00 3.10
-0.84700€ 02	-0.170496 03	0.279436 02	0.01994E 02	-0.	-0.	3.20
-0.10396E 03	-0.12418E 03 -0.12836E 03	0.30421E 02 0.32070£ 02	0.79409E 02 0.73401E 02	-0. -0.	-û. -û.	1.18 1.18
-0.17439E 63	-0.86947E GZ	0.40909E 02	0.56664E 22	-0.	-0.	3.77
-0.74071E 03	0.1619¥ 03 0.24410€ 03	0.40E:3E 02 0.95496E 01	0.40000E 02 0.53552E 02	~0.	0.	3.40
-0.20390(0)	0-29475F 03	0.433056 01	0.702788 02	-0. -0.	0. 0.	3.96
-0.90471F 02	0.1510 M 03	0.131966 02	0.59105E 02	-0.	0.	3.52 3.56 3.70 3.70
0.83432E 07 0.40834E 02	0.45470E GZ 0.19492E GZ	0.43294E 02 0.43214E 02	0.30221E 02 0.29370E 02	•. 0.	- 0. - 0.	3.70 3.84
0.38280f 07	0. 92409E OL	0.401258 02	0.230036 02	0.	-0.	€.66
0.24843F 02 0.19441F 07	0.31220E 01 0.73259E 01	0.38685E 02 0.42948E 02	G.16927E 02 0.10861E 02	-0.	-0. -0.	4.20
0.25751E 07	-0.19877F 01	0.544098 02	0.26867E 91	-0.	-0.	4.50 4.70 4.80 4.96 3.00 5.13 5.30
0.32130F 07	-0.13487F 02	0.662948 02	-0.405676 02	-0.	0.	4.40
0.38 1745 07	-0.50092E 02 -0.31439E 02	0.96322E 02 0.76497E 02	-0.74860E 02 -0.77907E 02	-0. -0.	6. 0.	3.00
-0.212396 02	-0.21733E 02	-0.16162E 02	-0.32604E 02	0.	●.	5-15
-0.28879E 07	-0.83046F 0: -0.27817F 01	-0.27469E 02 -0.20846E 02	-0.73481E 02 -0.70050E 01	••••	- 0.	3.30
-0.73512E N2	0.32690€ 00	-0.187236 02	-C.332936 01	-0.	- 0.	\$.70
-0.7575/E 02 -0.33318F 02	0.10795E 02 0.15306F 02	-0.2174ZE 02	0.95523E 01 0.13393E 02	-•.	-0.	5.70 5.85 6.00
-0.345782 02	0.213764 32	-0.31946E 02 -0.33412E 02	0.23204E 02	- 0. -0.	3. ●.	6.00 6.05
-0.346ROF 02	0.284996 02	-0.33724€ 02	0.32487E 02	-0.	ō.	6.05
-0.37340F 02	0.31501F Q2 0.11071E Q2	-0.30596E 02 0.12174E 02	0.35317E 02 9.05005E 01	-0. -0.	0. 0.	4.0
0.311007 01	0. 79736F 01	0.14291E 02	0.227976 01	-0.	0. 0.	6.40
0.82100F 00 -0.78551E-00	0.430260 01 0.59333E 01	0.10115E 02 0.64097E 01	0.121826-00 -0.787456 00	-0.	ə.	6.60
-0.40016F-01	0. 54699F 01	0.334918 01	-0.13759E 01	-0. -0.	0. 0.	7.00 7.40
0.156F2E 01	0.194996 01	0.364618 01	-0.21143E 01	-0.	0.	1.30
0.24402E 01	0.73047C 01	0.32213E 01 0.2004E 01	-0.25727E 01	-0.	0.	9.00

Table XV - - - Centinues

(PSI/FPS SINUSO/DAL GUST)

10 CPS

GROSS WEIGHT: 297, 000 LB CUTUSF FREQUENCY: ALTITUDE: 24, 000 F? MACH NUMBER: 0. 85

PERCENI SEMISPAN 40.06 SEGMENT NUMBER 8

DETROITAL	. SEEAE SERGO	DOM: N	ATTAL CITIES			
REAL	18461 N/ 17	REAL	1846 1942Y			
-0.134276-00	0.7376# 02	-0.12489E 01	0.270976 03	•.	•	
0.49237F 02 0.90530F 02	0.63188F 62 0.18692E 82	0.10050E 03 0.3300E 05	0.24197E 05 0.75329E 02	:	.	6.30
0.11480F 03	-0. 757618 01	0.43666 05	-0.36997£ 02	i.	-6.	2.55
0 11057E 05 0.96627E 02	-0.73806E 82 -0.46699E 92	0.43006E 05 0.42277E 05 0.37131E 03 0.33236E 03	-0.15336E 93 -0.18676E 93	3.	• I .	i, p
0.844425 02	-0.56777E 02	0.332166 03	-C.229106 03	• • • • • • • • • • • • • • • • • • •	- B.	0.00
0.40623F 02 0.76207E 02	-0.6721 0F 02 -0.603046 02	0.366078 03	-0.27272E 03	•	•	25
0.741776 02	-0.803046 62 -0.123276 03	0.2037E 03	-0.32593£ 05 -0.45926£ 03	i.	-6. -6.	0.00
0.725558 02	-0. 201558 03	0.38236E 03 0.2009PE 03 0.2009PE 03 0.27612E 03 0.27612E 03 0.17001E 03 0.62010E 02 -0.1730PE 03 -0.39572E,03 -0.71632E 03	-0.774036 03	••	-6.	1.30
0.40560+ 02 0.34717E 02	-0.26390E 03 -0.33621E 03 -0.57509F 03	0.17001E 03 0.42019E 02	-0.90754E 03	:	-6. -6.	1.20
-0.244048 02	-0.57509F 03	-0.17300£ 03	-0.322910 % -0.354157 04 -0.146331 04	-0.		12
-0.82484E 02 -0.17080E 03	-0.40102E 03 -0.31401E 03	-0.39372E, 03 -0.71432E 05	→.140339 B4	-0. -0.	-A. -B.	1.49
-0.40744E 01	0.4444E BZ	-0.151376 04	-0.49774E 02	-0.	-6.	1.5
-0.46533F 03 -0.34223F 05	0.10699E 03 0.13051E 03	-0.16147E 84	0.43392E 43 0.40437E 93	-0. -0.	<u>.</u>	1.6
-0.11973£ 05	0.11:04€ 03	-0.34000£ 03	0.204446 02	-6.	2	1.0
-0.71877F 02 -0.4 6094E 02	0.47944E GZ 0.89608E GZ	-0.10770E 05	0.224445 03	-0. -0.	•	1.99
-0.30070f 02	0.053746 02	-0.71632E 09 -0.15137E 06 -0.16147E 06 -0.11241E 06 -0.34608E 03 -0.1127E 03 -0.1127E 03 -0.74624E 02 -0.36642E 02 -0.76510E 02 -0.77510E 02 -0.79500E 02 -0.90500E 02 -0.90500E 02	0.22666E 03 0.16700E 03 0.11651E 05 0.02019E 02	-0.	Ξ.	2.66 2.56
-0.182396 02 -0.118486 07	0.83926E 62	-9.306728 02	0.02019E 02	-0. 0.	•	2.66
-0.40112E 01	0. 96363E 62	-0.76420E 02	0.84967E 02 0.47019E 02	Ĭ.		2.30
0.79364F 01 0.16316F 02	6.97271E 82 0.96724E 62	-0.77310E 02	0.502536 02	•		1,4
0.20197E G2 0.2920% 02	0.91623E 02 0.84823E 02	-0.90331£ 02	0.37577E 02 0.71140E 02	••	- 1	2.45
0.2920% 02	0. 848238 02	-0.903316 02 -0.117536 03 -6.111706 03	0.96361E 02 0.94903E 02	•	i.	2.40
0.30804F 02 0.29494E 02	0.87224E 62 0.95293E 02	-0.11170E 03 -0.64700E 02	6.74999E B2	•. •.		2.90
0.56266E G2 0.65332E 02	0.10797F 03	-0.79607E 02 -0.10917E 03	0.44004E 02 0.45512E 02	•.	Ĩ.	1.9
0.104125 01	0.10000E 03 -0.13261E 02	-0.10717E 03 -0.13902E 03	A 212474 AL	:		2.40
0.17038E 03 6.75319E 01 0.51209E 01	-0.26827E C2	-0.10717E 03 -0.13702E 03 -0.3203E 03 0.22971E 02 0.37666E 02 0.51011E 02	0.21346E 05 0.10451E 05 0.10179E 03			
0.51209f 01	0.11344E 82 0.47821E 82	0.22571E 02 0.37464E 02	0.100516 05	0. -••	-6.	3.00
0.215228 02	0.73985E B2	0.51011E 02	0.192/58 04	•	Ž.	135
0.46046E 02 0.70777E 02	0. 0904 TE 02	0.71164E 02	0.200925 05	•. •.	•	LK
0.13033E 03 0.25790F 03	0.10414E 03 0.52819E 02	0.932738 02 0.167678 03 0.299398 03	0.100936 05 0.10561E 03	ě.	6. 6.	£.
0.25790F 03 0.14849F 83	-0.15260€ 03 -0.21570€ 03	0.255556 03	-0.201145 02	: :	-0. -0.	3.46
0.100416 03	-0.22697£ 03	0.12440€ 02	0.35954F 02 0.07434F 02 0.43257E 02	••	-6.	<u> </u>
0.58461E 02 -0.58454E 02	-0.14463E 05 -0.75612F 62	0.63510£ 02	0.43257E 02 -0.25020E 01	•. •e.	•B.	7.00
-n.57976f 02	-0.499767 02	0.077716 02	-0.11'307E 02 -0.16'354E 02 -0.16'509E 02	-0.	-6.	5.6
-0.509306 02	-0. 93497E G2 -0. 19634E 62	0.667968 02	-0.16934E 02	-0. -0.	-3. -6.	1.00 1.00
-0.460i0E 02 -0.43021F 02	-0.12293E CZ	0.202276 02	-0.21220E 02	-0.	-6.	1,50
-0.46140f 02	-0.71800F 01	0.146216 02	-0.109726 02	-0. -0.	••	2:
-0.49380E 02 -0.54696F 02	0.19053E 02 0.23635E 02	-#,10007E 02	3.25305E 01 0.14755E 02	-0.	2	4.96
-0.50499F 02 -0.17279E 02	0.294846 82	-0.02012£ 01	0.103016 02	-0. -0.	•	5.40
-0.127000 02	0.12101E 62 0.77926E 01	0.552556 02	0.103016 02 -0.309696 01 -0.129406 02 -0.103416 02 -0.170016 02	-0.	<u>.</u>	23
-0.1412W 62	0.75495E 01 0.70174E 01	0.235418 02	-0.16341E 02	•••	•	CC
-0.14000f 02 -0.13001f 02 -0.10344f 02	0.633736 01	0.141946 02	-0.155036 0E	-0. -0.	ī	3.40
-0.10944 02	0.569136 01	0.30760 02 0.124402 02 0.13402 02 0.13402 02 0.14702 02 0.677012 02 0.667962 02 0.467962 02 0.467962 02 0.362702 01 -0.160072 02 0.39222 02	-0.195036 06 20 974641.0- 20 97796 02 -0.190716 08	-0.	6. 6. 6. 6. 6. 6.	1.0
-0.10104E 02	0.43190E 01 0.31225E 01	0.47737E 01	-0.130)7E 02	-0. -0.	-6.	73
-0.10007E 02 -0.10561E 02	0. 3041 96 01	0.91907# 01	-0.14407E 02 -0.16407E 02 -0.16407E 02 -0.15300F 02 -0.17303E 02 -0.07260E 01	-0.	•••	6.00
-0.17259F 02 -0.15964F 02	0.9729 0F 01 0.11746E 02	0.11003£ 02	-0.16762E 02	~0. -0.	Ž.	
-0.116377 02	6.13494F 62	0.75826E 01 0.40050E U1 -0.46244E 06	-0.177032 02	=0,	•	'Lb
-0.11617F 02 -0.96617E 01 -0.61107F 01	6.14439E Q2 0.14214E Q2	-0.60244 E 66 -0.52539 E 61	-0.11334E 02	-0. -0.		7,40 2,44
-0.7342[f-0]	0.1204 OF G2	-0.64070E 01 -0.66579E 01	-0.10070E 01	-0,	<u>.</u>	1.0
0.52004F 01	0.73400E 01	-0.68579E 01 -0.62696E 01	0.196776 01	•		1.00
A.41311E A1	•			••	70	

(PSI#PS SINUSOIDAL GUS%

GROSS WEIGHT: 297,000 LB CUTOFF FREQUENCY: 10 CPS ALTITUDE: 24,000 FT MACH NUMBER: 0.85

PERCENT SEMISPAR: 40.06 SEGMENT WIMSER 107

IN ROCKTA	L SEAR STREET			DC*DEFTAL AXIAL FIRESS		
REAL	15461440*			AFAL	1MAGIMARY	
		_	_			THE PRICE
-0.94903E-01 0.48004E-02	0.717072 62 0.413235 62	0. 0.	•.	\$.10671E 0 -0.15429E 0		0.35
0.60023E 02	0-101416 02	•.	0.	* 9. 20 954£ 0	3 -0.426558 02	0.30 0.36
0.11137# 03 0.10714# 03	-0.97901E 01 -0.32443E 02	•. •.	寸: →:	-0.3725% 0 -0.3612% 0		6.4A C.39
0.935016 02	-0.447266 62	••		-0.517245 0	3 0.159506 09	0.60
0.773206 32	-0.542300 02	•. •.	•	-0.283816 0 -0.261516 0		0.70 9.80
0.734236 02	-0.762925 62 -0.110516 03	•.		-0.24456E 0 -0.23593E 0		G. 90
0.70455# 02	-0.196307 03	•.	- i .	-0.23593R 0 -0.21504E 0		1.00 1.20
0.57704F 02 0.34214F 02	-0.24664F 03 -0.31867E 03	0. 0.	-6. -€.	-0-152106 0		1.16
-0.18687F 07	-0.35531E 03	-9.	⊸.	-0.5299HE @	3 0.1:6622 65	1.40
-0.73%77E 02	-0.36142E 03	-•. -•.	→.	9.33811F 0 0.4120% 0	3 0.119926 04 3 0.063376 03	1.47
-0.379946 63	-0.479426 62	-0.	-0.	0.129516 0	4 0.37402E 0Z	1.99 1.95
-0.43737E 03	0.96901E 02 0.12073E 03	-•. -•.	•:	0.1379TE 0 0.96047E 0		1.60
-0.21340E 05	0.102946 03	⊸.	•.	0.291266 0	5 -0.240156 03	1.65 1.80
-0.400000 02 -0.435200 02	0.90529E 02 0.83204E 02	- 0. - 0.	• • • • • • • • • • • • • • • • • • •	0:14038€ 0 0:94439€ 0		1.90 2.00
-0.24186F 02	0.792406 02	-0.	••	0.63761E 0	2 -0. 993 436 32	2.10
-0.14717# 02 -0.10193# 02	0.792226 02 0.853236 02	-0. 0.	• • • • • • • • • • • • • • • • • • •	0.50131E 0 0.71294E 0	2 -0.70000E 02 2 -0.74324E 02	2.30 2.30
-0.313196 0:	0. 601296 02	•.	7.	0.45302E 0	2 -0.572636 02	2.35
0.75040F 01 0.14571E 02	0.66461E 62 0.67990E 62	0. •.	0. •.	0.66227E 0		2.40 2.43
0.174966 02	0.641726 62	0.	0.	0.84017E 0	2 -0.407916 02	2.44
0.24526E 02 0.25736E 02	0. 00294E 02 0. 04412E 02	•. 0.	0. •.	0.10044€ 0 0.9343乗 0	3 -0. 827 05E 02 2 -0. 81154E 02	2.47
0.766706 02	0, 920775 02	0.	●.	0.726170	2 -0.637068 02	2.94
0.34518F 02 0.45066F 02	0.10295F 63 0.10103E 03	•. •.	0. 0.	0.44010E 0 0.43274E 0	2 -0.39377E 62 2 -0.36667E 62	2.50 2.54 2.58 2.65
0.104696 03	-0.17743E 02	9.	⊸.	0.135476 0	3 -0.162466 03	2.70
0.17355E 03 0.47375E 01	-0.29493E 02 0.44228E 01	0.	-1. -1.	0.1994E 0 -0.1926¥ 0		2.80 3.00
0.47164E 01 0.14904E 02	0.40070E 02 0.43513E 02	-o. o.	•. •.	-0.32164£ 0 -0.43539€ 0		1 10
0.424236 02	0. 777 756 02	•.		-0.60005E D	2 -0.17187E 03	3.20 3.26 5.29
0.44490F 02	0.91010E 02 0.28895F 02	0. 0.	0. 0.	-0.794956 0 -0.1*3265 0		<u> </u>
0.13596E 03 0.23356E 03	-0.18190E 03	0.	⊸.	-0.2183X d	3 0.240216 02	3.39 3.40
0.14575F 03 0.10007F 03	-0.21045F 03 -0.22572E '3	:	-4: -4:	-0.5321e5 0 -0.106295 0	2 -0.30722E 02 2 -0.74879E 02	3.52 3.56 3.60
0.356256 02	-9.14012F 03	0.	⊸.	-0.571636 9	2 -0.36943E 07	3.66
-n.65334f 02 -n.61337E 02	-0.49218E 02	- •. -•.	·1.	-0.99948E 0 -0.74995E 0		3.70 3.85
-0.57110E 02	-0.20084E 02	-0.	⊸•.	-0.53382E 0	2 0.14469€ 02	4.00
-0.451958 02	-0.149775 02 -0.04563E 01	→.	→. -•.	-0.40148E 0 -0.74118E 0		4.20 4.50
-0.4005 9 02	-0.47134E 01	⊸.	⊸.	-0.124936 0	2 0.152116 02	4.00
-0.40855E 02 -0.41978E 02	0.73892E 01 0.12512F 62	-0.	•	-0.46591E 0 0.91314E 0		4.96
-0.395238 02	0.141235 02	4 .	e.	0.70757E 0 -0.2607 € 0	1 -0.15706E 02 2 0.26481E 01	5.00
-0.22051F 02	0.82321F 01 0.73947F 01	3:	0. •.	-0.26414E 0	2 0.110576 02	5.15 5.30
-0.18951E C2	C. 51940E 01	- 0. 1.	••	-0.21823E 0 -0.15898E 0	2 0.13962E 02 2 0.14526E 02	3.70
-0.17951E 02	0.44575f 01 0.41622E 01		•.	-0.12130F 0		5.70 5.85
-0.15248F 02	U. 2473 ME 01	∞ 0. ∀3.	0. -0.	-0.8364% 0 -0.79067E 0		6.00
-0.15049E 02	0. 07355E 01 0. 4-377E 01	-0.	◄.	-0.76864E G	1 0.11140E 0Z	6.03 6.06
-0.1504RE 02	0.94028E 01 0.12237E 02	- •.	4.	-0.78520E 0 -0.10155E 0	1 0.123106 02	6.08
-0.14466 07 -0.14573E 07	0.134455 02	-0.	•	-0.647846 0	1 0.140096 02	6,30 6,40
-0.12476E 02	0.147637 02 0.13110F 02	⇒ :	0. 0.	-7.34227E 0	0.115605 02	6.60
-0.50057E 01	0.14330F G2	→.	••	0.514746 0 0.27432E 0	1 0.576890 01	7.00 7.40
0.97795E 80 8.40945E 01	0.11684E 02 0.48754E 01	+0. •.	•.	0.54743E 0		4.20
3.95326E 01	0.	: :	0.	0.354680		9.00 20.00

(PSI/FPS SINUSOIDAL GUST)

297, 000 LB CUTOFF FREQUENCY: 24, 000 FT 0.85 10 CPS

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

BODY BALANCE STATION: 540 SEGMENT NUMBER 17

INCREMENTA	L SPEAR STREES					
AFAL	IMAGINARY					
						THE
-0.41554F-01	-0-11491F 01	0.	→.	•.	-0.	0.14 0.30 0.35 0.44
-0.17179E 01	-0.77898F 00	0. 0.	→.	٥.	-0.	0.50
-0.199CaF 01	0.91129f-01 0.90408f 00	-0.	-0.	0. -0.	-0. -0.	0. 55
-0.17374F 01	0. 74424F 00	-0.	-0.	-0.	-0.	0.90
-0.1351(F 01 -0.11179F 01	0.80593F 00 0.82421F 00	-0. -0.	-0. -0.	-0. -0.	-e. -o.	0,60
-0.9ACC7E 00	0.84!29 00	-0.	-ŏ.	-0.	-0.	0.50 0.66 0.70 0.50 1.30 1.34 1.49 1.47 1.57 1.55 1.66 1.65 1.80 2.00 2.10 2.30 2.30 2.30
-0.40381E 00 -0.87077E 00	0.87597E 00	-0.	-•. -•.	-3-	- 0.	0.90
-U. 99474F 00	0.10753F 01 0.15150F 01	-0. -0.	-0. -0.	-0. - 0.	-0. -e.	1.49
-0.56510F 00	0.191346 01	0.	-0.	٠.	- 0.	1.34
-0.9716FF 00	0.24331# 01 0.2752# 01	0. -0.	~0. ~0.	0. -e.	-8. -8.	1.40
-0.42273E-00	0.304826 01	-0.	-0.	-0.	-0.	3.47
G.49574E-01	0.29039 61	-0.	-0.	-0.	~ 0.	1.90
0.15715E 01 0.23127E 01	0.13965F G1 0.31194F-00	-n. -0.	-o. -o.	-0. -0.	-0. -0.	1.75
0.140536 01	-0.143336-00	~0.	0.	-0.	0.	1.65
0.459026-00	-0.45090E-01 0.30503E-01	-0. -0.	ø. 0.	-0. -0.	0.	1.80
-0.17078F-00	0.12103F-09	-0.	ö.	-0.	0. 0.	1.90 2.00
-0.341776-00	0.750756-60	-0.	⊸.	-0.	-0.	2.30
-0.539191 00 -0.70718F 00	0.45144F-00 0.54544F 00	0. 0.	-0. -0.	0. 0.	-0. -0-	2.20
-0.71011F 00	0. 70344F @0	ŏ.	-o.	ŏ.	- 0.	2.33
-0.73506F GO	0.79091F 00	0.	-o.	0.	-0.	2.40
-0.46940F 00 -0.41489F 00	0.81300F 00 0.73339E 00	0. 0.	-0. -0.	0. 0.	-0. -0.	2.43
-0.47446F-00	J-42470E-00 0.20201E-00	0.	-0. -0.	0.	-0.	2.43 2.44 2.47
-0.41417E-00 -0.78708F 00	0.20701E-00 0.23893E-00	0. 0.	-0. -0.	0. 0.	-0. -0.	2.50
-0.11547F 01	0.452886-00	0,	-0. -0. -c. -0.	. :	- 6.	₹: 5
-0.17499F 01	0.77007F 00	0.	-ç.	0.	-0.	2.65
-0.23210F 01	0.28963E 01 0.46034E 01	0.	-0. -0.	0. 0.	-0.	2.50 2.54 2.58 2.65 2.70 2.80
-0.67A7RE 00	0. 57247F 01	0.	-ò.	o.	-0.	5.00
-0.439A0F=00 0.778915 00	0.79344F 01 0.10005F 02	∂. -0.	-0. -0.	0.	-0.	3.10
0.27974F 01	0.10009F 02	⊸.	⊸.	-0. -e.	-0. -0.	3.20 5.26
0.49149F DI	0.12C24F 02	-0.	-0.	-0.	-0.	3.49 3.35 3.40 5.52 5.56 3.60
0.1711RE 07 0.20749E 07	0.39513F 01 -0.91543F 01	-0. -0.	-ø. o.	-0. -0.	-0.	3.35
0.214396 01	-0.41543F 01 -0.47554F 01	-0.	ŏ.	-0.	o. o.	5.52
-0.55491F 00 0.95789F 00	-0.47554F 01 -0.18374F-00	0.	-0.	0.	-0.	3.56
0.95789E 00	-0.18467F 01 -0.36728F 01	-0. -0.	o. o.	-0. -0.	0. 0.	3.40
0.314417 01	-C. 34853F 01	-0.	0.	-0.	0.	3.70 3.85
0.17469F 01	-0.25987F 01 -0.24141F 01	-0. -0.	o. o.	-0. -0.	o. c.	4.00 4.20
0.7284AE-01	-0.74141E 01 -0.71506F 01	-0.	0.	-0.	0.	4.30
- 0.13809F- 0C	-0.20910F 01	-to.	0.	-0.	0.	4.50 4.70
- 0.14*57E- 00 - 0.19904E- 00	-0.73517E 01 -0.75313F 01	-0. -0.	0. 0.	-o. -o.	0.	4.80
-0.340726-00	-0.23#03F 01	-o. o.	0.	-0.	0.	5.00
-0.13132E 01	-0.18575F 01	o. o.	0.	0-	0.	5.15
-0.159[6E 0] -0.1865#F 0]	-0.15331F 01 -0.17681F 01	0.	0. 3.	o. o.	0. 0.	5.30 ##
-0.24453F 01	-0.87254F 00	0.	0.	0.	0.	5.70 5.85
-0.33136F 01	0.10737F 01 0.19715F 01	o. G.	-0. -0.	0. 0.	-9. -0.	3. 83 0. 00
-0.542/3F 01	0.31694F 01	0.	-o.	0.	-0.	6.63
-0.54653E 0;	0.45877F 01	0.	-0. -0.	0.	-0.	6.06
-0.50013E 01 0.14277E 01	0.49601F 01 0.91837F 00	0. -0.	-0.	0. -0.	-0. -8.	6.0t 6.20
0.15643F 01	0.1437AF-GO	-0.	-0. -0. -0. -0.	-0.	- 0-	5.40
0.84113F 90	-0.12440F-00 -0.12172F-00	-0. -0.	-0.	-0. -0.	- 8. - 0.	6.60 7.00
0.418406-01	-0.54738F-01	-0. -0.	-0.	-6.	~ 0.	7.00 7.40
-0.21716F-01	-0.377649-01	-0. -0.	-0.	-0.	. D.	8,20
0.17374E-03 0.17713F-01	-0. 6776 F -01	-0. -0.	-o. o.	-0. -0.	-0. 0.	9.60 10.00
	٠.		**	- ••	••	AU, 00

(PSI#PS SINUSOIDAL GUST)

GROSS WEIGHT: 297,000 LB CUTOFF FREQUENCY: 10 CPS ALTITUDE: 24,000 FT MACH NUMBER: 0.85

BODY BALANCE STATION: 820 SEGMENT NUMBER 1

DECREMENTAL ANIAL STREES

		LIKTHUMETAL				
		ne al	IMAGIMARY			PARTMET
٠.	•	0.14020E 01	0.11502E 03	٠.	٠.	6.35
i.	Ŭ.	0.917916 02	3.50364E 02	ŏ.	: :	0.30
٠.	••	0.13394E 03	0.15490E 02	••	0.	0.30 0.36 0.44
0. 0.	- 0. - 0.	0.10530E 03 0.17321E 03	-0.27779E 02 -0.41920E 02	o. o.	- 0. - 0.	0.40
n.	-0.	0.144718 03	-0.77793E 02	0.	-0.	0.50 -0.60
•.	-e. -e.	0.12893E 03 0.11019E 03	-0.90059E 02 -0.102952 03	0. 0.	-0. -0.	0.70 0.80
ö.	-0.	0.111946 03	-0.11949E 03	ö.	-0.	0.40
	- 0.	0.100616 03	-0.17704E 03	· • • • • • • • • • • • • • • • • • • •	-0.	1.00 1.30 1.34 1.40
•. •.	-0, -0,	0.10547E 03 0.88204E 02	-0.27270E 03 -0.34035E 03	0. 0.	-0. -0.	1.40
0.	-0.	0.53972E 02	-C.43528E 03	0.	-0.	1.46
-0. -0.	-0. -0.	-0.21786E 02 -0.94391E 02	-0.48013E 03 -0.50846E 03	~0. ~0.	-0. - 0.	1.40 1.47 1.47 1.55 1.60 1.63 1.80
-0.	-0.	-0.202816 03	-0.390 TOE 03	-0.	-0.	1.47
-0.	-0.	-0.40503E 03	-0.400 10E 02	~O•	-0.	1.55
-0.	9. 0.	-0.53849E 03 -0.378618 03	0.11326E 03 0.11786E 03	-0. -0.	0. 0.	1.60
-0.	0.	-0.1u442E 03	0.02115€ 02	-0.	0.	1.86
-0.	0.	-0.46515€ 02	0.52419E 02	-0.	0.	1.99
-0.	0. 0.	-0.15921E 02 0.16612E 01	0.26898E 02 0.865388 0 0	-o. -o.	0. 0.	2.00 2.10
0.	-9.	0.111130 02	-0,23290E 02	0.	-0.	2.20
o. o.	-0. -0.	0.41003E U1 0.336160 U1	-0.33168E 02 -0.51983E 02	0.	-0.	2.50
-0.	-0.	-0.342498 01	-0.51034E 02	• · · · · · · · · · · · · · · · · · · ·	- 0. -0.	2.35
-0.	-0.	-0.21615E OZ	-0.42007E 02	-0.	~ O.	2.63 2.66
-0. -0.	- 0. - 0.	-0.314408 02 -0.58657E 03	-0.46208E 02 -0.983288 01	-0. -0.	-0. -0.	2.44
-c.	0.	-0.57843E 02	2.23102E 01	-0.	0.	2.47 2.50
-0.	-0.	-0.22121E 02	-0.13984E 02	-0.	. 0.	2.90 2.54 2.54
-0. -0.	-0. -0.	-0.20605E 01 -0.47438E 01	-0.47191E 02 -0.43047E 02	-0. -0.	-6. -0.	2.55 2.65
-0.	-0.	-0.27224E 02	-0.47928E 01	-0.	-0.	2.70
-0. 0.	-0. -0.	-0.80642E 02 0.39611E 02	-0.46099E 02 -0.12611E 03	-0.	-0. -0.	2.70 2. 8 0
0.	-0.	0.601516 02	-0.233128 03	o. o.	-0.	3.00 3.10
0.	-0.	0.19306E 02	-0.32541E 03	0.	-0.	3,30
-n.	-0.	-0.56971E 02	-0.30000E 03	-0.	-0.	3,26
-0. -0.	-0. -0.	-0.14021E 03 -0.41380E 03	-0.41035E 03 -0.95408E C2	-0. -0.	- 0. -0.	3.29 3.35 3.60
-0.	0.	-C. 73783E 03	0.34171E 03	-0-	•.	3.66
0.	0. -0.	0.18019E 02 0.1048ZE 03	0.11410E 03 -0.63286E 02	0.	-0.	3.52 3.56 3.60
٥.	0.	0.13207E 01	0.30239E 02	•.	0.	3.60
-0.	0.	-0.22210E 03	0.15003E 03	-0.	0.	3.70
-0.	o. o.	-0.12256E 03 -0.54874E 02	0.13279E 03 0.13011E 03	-0.	•. •.	3.85 4.00
-0.	ŏ.	-0.10130E 02	0.100 POE 03	-0.	ŏ.	4.20
0.	0.	n.24813E 02 0.42422E 02	0.84378E 02 0.74936E 02	٥.	0.	4.90 4.70
0.	0. 0.	0.322746 02	0.44714E 02	0. 0.	0. 0,	4.80
0.	0.	0.67968E 02	0.31254E 02	0.	0.	4.96
o. o.	o. o.	0.65695E 02 0.311260 02	0.214988 02 0.34112# 02	0. 0.	0. 0.	5.00 5.15
ö.	o.	0.304216 02	0.34783E 02	0.	o.	5.30
0.	0.	0.402728 02	0.29496E 02	₹.	0.	5.60
0. 0.	o. -o.	0.53346E 02 0.68491E 02	0.21124E 02 -0.12977E 02	0. 0.	-0.	5.70
0.	-0.	0.98498E 02	-0.27972E 02	••	-0.	3.45
o. o.	-0. -0.	0.10297E 03	-0.47756E 02	0.	-0.	6.03
n.	-0.	0.10341E 03 0.95463E 02	-0.71724E 02	0.	-0. -0.	6,06 6,08
-0.	- 9,	-0.86315E 01	-0.149922 02	-0.	-0.	5.20
-0. 0.	-0. -0.	-0.10725E 02 0.121458-01	-0.31031E 01 -0.47990E 01	-0,	-0. -0.	6.40
ø.	- 0.	0.631430 01	-0.72223E 61	0.	-0.	6.60 7.00
0.	-0.	0.502310 01	-0.96245E 01	0.	-0.	7.40
0. -0.	-0. -0.	0.43788E-00 -0.50716E 01	-0.43320E 01 -0.43969E 01	-0.	-0. -0.	3.20
-0.	0.	-0.89544E 01	8.	-0.	0.	9.00 10.00

(PSI/FPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

297, 000 LB CUTOFF FREQUENCY: 24, 000 FT 0 55

15 CPS

PERCENT SEMISPANE 27 SEGMENT NUMBER 10

DICTIONS	L SKEAR STRESS	· Incresions	L AXIAL STRESS			
REAL	IRAL (MART	BEAL	IMAG] WARY			
0.847215-61	0.55385F 02	-0.00481E 00		_	_	Common Co
0.205075 02	0.29947F 82	0.198988 85	0.363696 03 0.262106 03	ე. -0.	-3:	6.39
0.44132F 02	0.83492F 01	0.37029€ 03	0.78904E BZ	- 7 :	3 :	
0.553936 02	-0.51992F N1	0.473396 05	-0.39241E 02	-0.	· .	22
0.430946 02	-0.16509F 02	0.457596 85	-0.141928 05	-0.	e.	6.9
0.4619CF 02	-0.22434F 92	0.400736 05	-0.19748E 05	-•.	0.	0.60
0.41253F 02 0.38231F 02	-0.27016F 02 -0.31773F 02	9.35806E 03	-0.24096E 63 -0.28550E 63	-•.	•.	2.79
0.36571E 02	-0.37815F 02	0.51144E 03	-0.33952E 03	-0. -0.	•. 0.	0.40
0.35926F 02	-0.37815F 02 -0.59253F 02	0.29897E 03	-0.51720E 05	- i:	0.	0.9
0.36653F 02	-P. 971 BRE 02	0.2774:E 03	-0.80276E 03	-0.	0.	1.20
0.33495F 02 0.22848F 02	-0.12897E 03	0.20%iF 05	-0.10254E 64	*.	0.	1.34
-0.4289FF 01	-0.16771F 03 -0.18946E 05	0.96340E 02 -0.14204E 03	-0.12780£ 04 -0.14050£ 04	••	-0.	1.40
-0.32143F 02	-0.20623F 03	-0.36551F 03	-0.14789E 04	8 :	-0. -0.	1.45
-0.275685F 02	-0.17111E 03	-9.6948ZE 03	-0.10976E 04	i.	-0.	1.97
-0.19911F 03	-0.35902F 02 0.46462F 02	-9.19332E 04	-0.10, OE 03	-0.	-0.	1.95
-0.23925F G3	0.46462F 02	-0.16662E 04	0.401 50F 03	-0.	•.	1.60
-0.18287F 03	0.70084E 02 0.64637E 02	-C. 11743E 04	0.46 70E 03	- <u>•</u> .		1.45
-0.43308F 02	0.62249F 02	-0.36373E 03 -0.20130E 03	0.347798 05 0.23180E 03	-0. -6.	0.	1.80
-0.30738F 02	0.63716F 02	-0. 12059E 03	0.175130 05	- i.	b. 0.	1.90
-0.204346 02	0.63716F 02 0.69980F 02	-0.77275E 02	0.127955 05	-0.	•.	2.14
-0.9621AF 01	0.746595 02	-0.34149E 02	0.871796 02	-0.	0.	2.20
0.140A2F 02	0.70043E 02	-0.32018E 02	0.70001E 02	•.	0.	2.30
0.17233F 02	0.77041F 02 0.79877F 02	-0.52813E 02 -0.65530E 02	0.46193E 02 0.30245E 02	•	•.	2.35
0.34061F 02	0.79724F 02	-0.85646F 02	0.35754F 02	ð. 0.	•. 0.	2.40
0.390098 02	0.72921E 02	-0.97472F 02	0.544868 02	ě.	•.	2,65 2,66 2,67
0.903536,02	0.43392F 02	-0.12910E 05 -0.12646F 09	0.1005ZE 03 0.11474F 03	0.	ě.	2.47
0.503817 02	0.49316F 02	-0.12646F 03	0.114745 03	0.	0.	2.50
0.43533F 02 0.54431F 02	0.84011E 02 0.10597E 03	-0.84240E 07 -0.43312E 02	0.97005F 02	•		2.5
0.101025 03	0.10496E 03	-0.688458 02	0.46091E 82 0.56672E 82	ð. 0.	6 .	2.55
0.16764F 03	-0.103308 03	-0.962498 02	0.14305E 03	0.	-0.	2.6
0.29234F 03	-0.16149F 03	-0.14613F 03	0.14527E 03	0.	-0.	5.5
-0.28309F C2	-0.11612E 03	0.16332E 02	0.11591E 09	-0.	-0.	i.
-0.36137F 02 -0.71372F 02	-0.93940E 02 -0.85784E 02	0.26914F 02 0.30791E 02	0.96723E 02 0.98310E 02	-0.	-0.	3.30
-0.8103AF 02	-0.83382E 02	0.339238 02	0.903106 02	-0.	· - • .	>-80
-0.88258F 02	-0.025138 02	0.34221F 92	0.80885E 02	-0. -0.	-6. -0.	3.26
		0.45080E 02	0.42441E 02	4 €3	-0.	3467
-0.10972F 03	-0.71410E 02 0.89707F 02	0.53023E 02	0.45047E 02	-0.	· .	2.70
-0.15204E 03	0.18767F 03	0.103256 02	0.41214E 82	-ŏ.	•.	2.93
-n.16405F 03	0.21643F 03	0.47720E 01	6.77443E 02	-0.	0.	3.96
-0.70035F #2	0.97865E 02	D.14 #2E 02	0.45131E 02	-•.	•.	3.60
0.75612F 02	0.12316E 02	0.47708E 02 0.47620E 02	0.421188 Q2 0.323458 Q2	•	-0. -0.	3-72
0.47726F 02	-0.34063F 01 -0.10413E 02	0.442168 02	0.253498 02	ŏ.	-0.	7.00
0.24912F 02	-0.10380E 02	0-424296 02	0.184536 02	ō.	-0.	1.30
0.187C7F-00	-0.920668 01	0.424298 02 0.47527F 02	0.18453€ 02 0.11968€ 02	-0.	-0.	1.30
-n.35269E 01	-0,92066E 01 -0.64821F 01	0.601778 02	0.287258 01	-9.	-8.	1.70
-0.49824F 01	~0.67778F 01	n. 73053E 02	-0.9991 8 € 02	-0. -0.	•. •.	100
-n.69616F 01	-0.61227E 01	0.95124E 02 0.64297E 02	-0.82493E 02 -0.85851E 82	- 7:	•.	
-0.72581E 01	-0.49439F 01 -0.46116E 01	-0.1781CE 02	-0.36016E 02	;		
-0.78159F 01	-0.372788 01	-0.30270E 02	-0.15076E 02	•	-9.	5.30
-0.913291 01	-0.23386E 01	-0.22 9 72E 02	-0.77192E 01	-•.	-0.	1.00
-0.10833F 02	-0.11493F 01	-0.20631E 02 -0.23939E 02	-0.36687E 01	-•.	-1	3.79
-0.130558 02	0.38635F 01 0.60353F 01	-0.234346 02	0.19826E 02 0.16962E 02	-0. -6.	-0. •.	2-57
-0.17386F 02 -0.18059F 02	0.603537 01	-0.370348 02	0.255496 62	-3 :	ě.	-1.01
-0.18057E 02	0.12310F 02	-0.371436 02	0.358006 02	-0.	•.	6.06
-0.17110F 02	0.137218 02	-0.33713F 02	0.309206 02	-4.	0.	8.08
-0.20813F 01	0.49234E 01 0.36368F 01	0.13413E 02	0.93471E 01	∴	¥.	たが、
-0.14308F 01	0.36360F 01	0.197488 02	0.31734E 01 0.13424E-00	-0. -0.	•. •.	6.40
-0.27408F 01	n.>{1 99 f G1 0,439n4f G1	0.11146F 02 0.73719E 01	-C.84774E 60	-4.	į.	7.40
-n.32940E 01 -0.20731F 01	0.514556 01	0.389448 01	-07344E @1	-0.	•.	7.40
-0.111058 nl	0.484798 01	0.424036 01	-0.23299F at	-4.	●.	t.m
0.81682F NO	0.37344E 01	0.35497E 01	-0.285134 01 -0.30441E 01	- 9.	•	9.00
0.2628AF 01	0.19723E 01	0.22131E 01 0.12964E 01	-0.30441E 01 -0.20343E 01	ě.	0. 0.	34.40
0.34104F 01	0.23041F-00 -0.12214F 01	0.12464E 01 0.28049F-00	-0.24430F B1	.	:	.11.00
0.31242F 01	-0.21871F 71	-0.545656 00	-0.189497 61	i.	-0.	11.40
0.21401E N1	-0.25013E 01	-0.54545E 00 -0.1043#E 01	-0.122 99F 31	•.	-0.	16,00
8. 97849F 00	0.	-0.14243F 0 1	••	••	••	16.49 11.40 12.40 13.40 14.40

(PSI#PS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

297, 000 LB 24, 000 FT 0. 85

CUTOFF FREQUENCY:

15 CPS

PERCENT SEMI SPAN: 27 SEGMENT NUMBER 14

THE PROPERTY AL	SEEAR STREET	DESPRESATION	ATTAL STREET			
BFAL	i mag inary	REAL	1846 LMARY			
A					_	PROPERTY CHE
0.62913F-02 0.72352F @>	0.10709F 03 0.91729E 02	-0.73216E 00 0.18057E 03	0.27555£ 03 0.23705€ 03	0. -0.	-0. -0.	0,10 0,50 0,56 0,14 0,50 0,60 0,70 0,80
0.13203F 03	0.24474E 02 -0.14484F 82	0.33594E 03 0.42959E 03	0.71605E 02	-0.	-0.	ĭ.\$
0.1401CE 03	-0.491998 82	0.415258 03	-0.356296 82 -0.128796 93	-e. -0.	*. 0.	0.44
0.13927E 03	-0.67313F 47	7.34345F 01	-3.17921E 03	-0.	0,	0.60
0.11416F 93	-0.8112NE 02 -0.99111E 02	0.924 00 € 03 0.29934€ 0 3	-0.21066E 03 -0.25091E 03	-9. -0.	o. o.	0.70
0.10407F 03	-0.11235E 03 -4.17053F 03	0.282638 07	-0.3001 OE 03	-0.	0.	0.90
0.10109F 03	-0.248377 03	0.27131E 03 0.25192E 03	-0.449396 03 -0.720496 03	-o. -o.	•. 0.	1.00
0.85455E 02 0.52656F 02	-0.346 896 03 -0.439777 03	0.19022F 03 0.87427E 02	-0.939699 03 -0.315978 06	•.	0.	1.75
-0.22229F 02	-0.48954E 03	-0.12890f 03	-0.12750E DA	o. o.	-0. -0.	1,10
-0.45635F 02 -0.20#12# 03	-0.52411E 03 -0.41590E 03	-0.331696 03 -0.630536 03	-0.13421F 04 -0.99607E 03	0.	-0.	1,47
-0.507738 03	-0.74967E 02	-0.139146-04	-0.92014E 02	o. -o.	-0. -0.	1.22
-0.5F304F 03	0.11749E 03 0.14758F 03	-0.15120E 04 -0.10656E 04	0.37053E 83 0.37009E 83	-9.	0.	1.6
-0.14404F 03	0.122617 03	-0.136086 03	0.279326 03	-A. -0.	0. 0.	1.65
-0.9804ZE 02 -0.99979F 02	0.10529E 0? 0.95527F 02	-0.18286E 03 -0.10964E 03	0.23039E 03 0.15099E 03	-0.	6.	1.90
-C.35941E 02	0.92228E 02	-0.70124€ 02	0.11411F 03	-÷. -0.	o. o.	2.60
-0.19577F 02 0.49824E 01	0.88066E 02 C.90290E 02	-0.49139F 02 -0.47203E 02	0.79113F 02 0.63524E 02	-0.	0.	2.20
0.82011F 01	0.41812E 02	-0.477458 01	0.419196 02	0. e.	0. 0.	2.30
0.13072E 32 0.17878E 02	0.82053£ 02 0.81492F 02	- 0.59467E 02 -0.77721E 02	0.328918 02	0.	0.	2.46
9,19791F 02	0.80379F 02	-O. 88454E 02	0.32446E 02 0.31259E 02	0. 0.	o. o.	2.45
0.2246BF 62 0.22214E 02	0.84092F 02 0.97227E 02	-0.11716E 03 -0.11476E 03	0.91037E 02 0.10414F 03	0.	0.	2.47
0.301427 02	0.10458F 03	-0.764447 02	0.88102E 02	9. 0.	o. o.	2.50
0.44375F 02 0.10282E 03	0.12536E 03 9.11968E 03	-0.574348 02	0.599746 02	0-	0.	1.5
0.17392E 03	-0.10440[03	-0.62475F 02 -0.87343E 02	0.51429F 02 0.13435E 03	o. o.	0. -0.	2.65
0.30256E 03	-0.17236F 03 -0.13427E 03	-0.13763E 03 0.14839E 02	0.13002E 03 0.13337E 03	0.	-0.	2.60
-0.67433F 02	-0.119A3E 03	0.244246 02	0.877748 02	-0. -0.	-0. -0.	3.00
-0.447668 02 -0.103968 03	-0.12049F 03 -0.12418E 03	0.27943E 02 0.30421E 02	0.81954E 02 0.79409E 02	-0.	-0.	3.20
-0.12096E 03	-0.12834E 03	0.328706 02	0.73401E 62	-0. -0.	-0. -0.	6.90 1.30 1.30 1.30 1.30 1.50 1.50 1.50 1.50 1.50 1.50 1.50 1.5
-0.17459F 03	-0.86947E 02	0.40969E 02	0.3664E 02	-0.	-0.	L.M
-0.264316 03	0.16193F 03	0.48438 02	0.40880E 02	-0.	0.	3.99 3.40 3.35 3.50 3.70 3.60 4.20
-0.25943F 03	0.244107 03 0.24425E 03	0.43446E 01 0.43305E 01	0.33332t 02 0.7027 8 € 02	-0. -0.	0. 0.	7.31
-0.90AT1E 02	0.15183F 03	0.131948 02	0.34105E 62	-0.	0.	3.66
0.8543ZE 0Z	0.45470F 02 0.19492E 02	0.43294E 07 0.43214E 02	0.38221E 02 0.29370E 02	- 0. 0.	-0. -0.	£22
0.38280E 02	0.92409E 01	0.401298 02	0.23003E 02 0.16927E 02	0.	-0.	4.00
0.24863E 02 0.19441F 02	0.51220F 01 0.25259E 01	0.38685E 02 0.42948E 02	0.100018 02	~0.	-0. -0.	4.20
0.25241E 02	-0.198778 01	0.544098 02	0.24047E-01 -0.48567E-02	-0.	-0.	4.70
0.22150F 02 0.44612E 02	-0.33487E 02 -0.50092E 02	0.66294F 02 0.86322E 02	-0.74860E 02	-0. -0.	0. 0.	4.90 4.70 4.96 5.00 5.13 5.30 5.70
0. 183 145 07	-0.51639F 02 -0.71751E 02	n.76497E 02 -0.16163E 02	-0.77907E 02 -0.32684E 02	-0.	0.	5.00
-0.21204F 02	-0.470466 01	-0.27469E 02	-0.13481E 02	• • •	0. -0.	2.12
-0.24804F 02	-0.27617E 01	-0.20846E 02 -0.18723E 02	-0.70050E 01 -0.33292F 01	-0. -0.	-0. -0.	5.00
-0.23532E 07 -0.25752E 02	0.52490E 00 0.10795E 02	-0.217428 02	0.955238 01	-0.	-0.	
-0.13118F 02	0.1530AF 02	-0.31446F 02 -0.53612E 02	0.15393E 02 0.23204F 02	-0. -0.	0.	4.00
-0.34640E 0Z	0.21326E 02 0.25499E 02	-0.337246 02	0.32487F 02	+0.	0. 0.	6.05
-0.12340F 02	0.31583F 0? 0.11871F 02	-0.30596E 02 0.12174F 02	2.353196 02 0.850056 01	-0. -0.	0. 0.	1.06
0.42538F-00 0.31108F 01	0.747536 01	0.142918 02	0.287978 01	-0.	0.	5.70 5.10 6.03 6.03 6.46 6.10 6.10
0.82100F 00 -0.24551F-00	0.63026F 01 0.59333E 01	0.10173E 02 0.44749E 01	0.121826-00 -0.737456 00	-0. -0.	o. o.	6.60
-0.60416F-01	0.54699F 01	0.534918 01	-0.137590 01	-0.	0.	7.00 7.40
0.15692E 01	0.34444E 01 0.73047F 01	0.38661E 01 0.32213E 01	-0.21143E 01 -0.2592 9E 01	-0. -0.	o. o.	1.2
0.34000F 01	0.62232F 00	0.200048 01	-6.27624E 01	0.	0.	9.00 10.00
0.34533E 01 0.2460#E 01	-0.45267F-00 -0.96154E 00	0.11765E 01 0.2363EE-00	-0.25722E 01 -0.22170E 01	0.	•. •b.	11.00
0.222351 01	-0.1720AE 01	-0.495146-00	-0.17196E 01	0.	-0.	13.00
0.17172F 01 0.13756F 01	-0.12542E 01	-0.94902E 00 -0.12925E 01	-0.11161E 01	0. 0.	-0. 0.	14,4
10.131767 41	٠.	- 44 8 6 7 6 7 6 9 8		٧.	٠.	35.40

(PSI/FPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

297, 000 LB 24, 000 FT 0. 85 CUTOFF FREQUENCY:

15 CPS

PERCENT SEMISPAN: 40.06 SEGMENT NUMBER 8

INCREMENTAL	SEEAR STREET	DICHMINIAL AXI	AL STREET		
#8 AL	IPAG 146PY	REAL	IMAGINARY		nayancı
-0.13427E-00	0.73700F 02	-0.124 89 E 01	0.27097E 03		
0.49232E 02	0.63100F 02	0.180586 03 (.24197E 03	0.	
0.9053RF 02	0.14492F 02 -0.959A1E 01		0.72329E 02	••	
0.1140CE 03 0.11057E 03	-0.33804E 02	0.422778 03 -0	0.133346 03	•	-0. 6.W
0.966278 02	-0.46499E 02	9.371318 03 -6	1.18676E 03	.	-0. 6.60
0.84442F 02 0.80023F 02	-0.56777E 02	0.304078-03 -4	0.22910E 03	:	-0. 0.70
0.762075 02	0-80304E 02	0.200596 03 -0	. 32 593E 03	ě.	-0. 0.99
0.74177E 02 0.72555F 07	-0.12527E 03 -0.20155E 03	n.27612E 03 -0 n.25167E 03 -0	.49924E 03	•. •.	-0. 1.00
0.405408 02	-0.26338E 03	0.17801E 03 -0	. 98754E 03	•	-0.
0.34717E 02 -0.24606E 02	-0.33421E 03 -0.37509E 23			•.	-0. 1166
-0.07404F 02	-0.40182E 3	-0.395728 03 -0	0.13415E 04 0.14635E 04 0.10105E 04	-0. -0.	-0. 1.67
-0.170406 03	-0.314#1E 03	-0.71632E @3 -0	.10105E 04	-0.	-0.
-0.40766E 03	-0.44464E 02 0.10699E 03	-0.15157E 04 -0	0.43774E 02 0.43392E 03	-0, -0,	1.55
-0.34223E 03	0.13051E 03	-0.11741E 04	0.40037E 03	-0.	0. 1.60 0. 1.45
-0.11975E 03 -0.71077E 02	0.11104E 03 0.97544E 02	-0.34000E 93 (30448E 03	•0. •0.	1.00
-0.46094F 02 -0.30070F 02	0.8960BE 02	-0.11207E 03 C	0.14 /8 OE US	-ŏ:	2.40
-0.30070F 07	0.85379E 02	-0.74624E 02 6 -0.58672E 02 6	E0 316917.6	→.	9. 2.30
-0.11848E 02	0.93029E 02	-0.834428 02 (0.82019E 02	-0. 4.	9. 2.30
-0.49112F 01	0.96563E 02	-n.76428E 02	0.47019E 0Z	?.	1.3
0.79364E 01 0.16316F 02	0.97271E 02 0.96724E 02	-0.775102 #2 (-0.90500E 02 (0.57239E 02	•, •,	9. 2.46 9. 2.43
0.20197F 02	0,91623F 02	-0.98331E 02 (0.71140E 02	0.	6.
0.24205E 02 0.30806E 02	0.84825E 02	-0.11755E 03 0	0.94983E 02	6.	2.47
0.294 96F 02 0.36266F 02	0.952938 02	-0.84988E 02 (0.7455 TE 82	0.	2.90
0.16266F 02 0.65852F 02	0.10757E 03	-0.79607E 02	0.4408ef 02	••	2.56
0.106128 03	0.19666E 03 -0.13261E 02	~ 0.15 0028 03 (0.43512E 02 0.21357E 07	•. •.	0.
0.17838F 03	-9.26827E 02	-n.23245E 03 (0.21504E 03	ė.	-0. £.6
0.75319E 01 0.51209E 01	0.11344E 02 0.47821E 02	0.22371E 02 (0.37646E 02 (0.18451E 03 0.18173E 03	-0.	-0. 3,60 3.
0.213227 02	0.71985E 02	0.510118 02 4	0.1924第 醇	•.	i. <u>}</u>
0.46^ 02	0.098678 02	0.71144E 02	0.20092F 03	0.	•. 33
0.15033F 03	0.10414E 03 0.32019E 02	0.93273E 02 0 0.16767E 03 0	0.19993E 03	0.	0. 3.29
0.2579AE 03	-0.19268E 03	0.255536 03 -0	0.28114E 82	0. ●.	0. 3.35 -0. 3.40
0.148698 03	-0.21570F 03	0.300745 02 0	3.35 756E 8 2	€.	-0. 3.52
0.10091F 03 0.34661F 02	-0.72697E 03		0.87434E 02	0. ••	-0. 1.%
-0.58454E 02	-n.75612f 02	0-11498E 03 -C	25820F 01	-0.	-0. 5.79
-0.57976F 02	-0.49976F 02 -0.33437E 02	0.07771E 02 -0	1.12397E 02	-0. -8.	-0.
-0.44010E 02	-0.19654E 02	0.469828 02 -0	SO 398405.C	-0.	-0.
-0.43821E 02 -0.46140F 07	-0.12295E 02 -0.71800E 01	0.78227E 02 -0 0.14621E 02 -0	.21220E 02	~•.	-0.
-0.49380E 02	0.138536 02	0.345298 01 0	. 25395E 01	-0. -0.	6. 4.40
-0.54696F 02	0.236331 02	-0.10487F 02 0	147555 07	-•.	0. 4.95 6. 5.40
-0.50699E 02	0.25984E 02 0.12101f 02	-0.02012E 01 0	3.18301E 02 3.30969E 01 3.12940E 02	-0. -0.	6. 3.80 8.15
-0.12780F 02	0.77928E 01	0.332-36 02 -0	.129406 02	-0.	6. 5.15 6. 5.30
-0.14123E 02	0.75495E 01 0.78174E 01	0.25541E 02 -0 0.18.07E 02 -0	1.14341E 02 1.17001E 02	-0. -0.	£.55
-0.13001f 02	0.45373E 01	0.14196E 02 -0	15583F 02	-0.	5.8
-0.10566E 02	0.569131 01 0.45190E 01	0.57900€ 01 -0 0.92537€ 01 -0	1.14047E 02	-0. -c.	5. 6.66 -0. 6.85
-0.10087E 07	0.312256 01	0.09942E 01 -0	.1 1037E 82	-6:	-0.
-0.10561F 02	0.384.9E 01 0.97290F 01	0.91907E 01 -0 0.11005E 02 -0	1.14407E 02	-0.	-0. 6.0
-0.17239F 02 -0.15966E 02	0.1174AF 02	0 75876F 01 -0	. 14194E 02	-0. -0.	0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0
-0.13637E 02	0.136946 02	0.40058£ 01 -0	139036 02	-0.	6.60
-0.76417E 11	0.14435F 02 0.14714E 02	-0.323396 01 -0	0.11334E 02 0.67283E 01	-0. -0.	0. 7.40 0. 7.40
-0.75921{-0}	0.12040E 02	-0.64070F 01 -0	-14090E 01	-0.	6. 6.30
0.52006E 01 0.91311F 01	0.73404F 01 0.20057E 01		3.19677E 01	-e. e.	9.00
0.10A48F 02	-C.28879E 01	-9.11278E 01 6	. 22374F 01		-0. 11.00
0.94345E 01 0.65287E 01	-0.64974E 01 -0.60264E 01	0.10304F 01 +6 0.30311E 01 -6	. 20573E-00 . 28541E 01	0. •.	-0. 12.00
0.25512F 01	-0.77335E 01	0.24030€ 01 -0	.45301E 01	:	-0.
-0,12972E G1	0.	0.11744E-00 0		-6.	0. 13.00

(PSIFFS S!NUSOIDAL GUSTI

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

297,000 LB CUTOFF FREQUENCY: 24,000 FT 0, 85

15 CPS

PERCENT SEMISPAN: 40.06 SEGMENT NUMBER 107

DECREMENTAL	STAR STREET			INCHMENTAL AXIAL STRESS		
PF6L	1 MAG 1 MBRY			OFAL	1 MAG 1 MARY	
-0.94903E-01	0.71707E 02	0.	•.	0.104716 01	-0.238366 03	7.55 7.55 9.55 9.55 9.55 9.55 9.55 1.50 1.50 1.50 1.50 1.50 1.50 1.50 1.50 1.50 1.50
0.49004F 62	0.61323F 02	0.	••	-0.15429F 03	-0.20474E 03	0.10
0.89023E 02	0.18141E 02 -0.92901E 01	0.	●. -0.	-0.26756€ 03	-0.42455E 02	0.36
0.107145 03	-0.32463E 02	0.	-0.	-0.37259F 03 -0.34123F 03	0.31270E 02 0.11349F 03	0.44
0.945917 02	-0.44728E 82	n.	•••	-0.317246 03	0.13958E 03	0.50
0. #3569F 62	-0.54230E 02 -0.640275 02	o.	- 0. -0.	-0.283216 03	0.17542€ 83	0.70
0.736237 62	-0.742927 02	0.	-0.	-0.24151E 03 -0.24658E 03	0.23302E 03 0.27849E 03	0.80
0.71691E 02	-0.11051F 03	0.	-0.	-0.235936 03	0.424586 03	1.60
0.70455E 02 0.59904F 02	-0.19038E 03	0. 0.	-0. -0.	-0.21304E 03 -0.15210E 03	0.66204E 03 0.84379F 83	1.30
0.36214F 02	-0.31807E 03	0.	-0.	-0.52991E 02	0.10447E 04	1:2
-0.18887E 02 -0.73177E 02	-0.35531E 03 -6.38142E 03	-0.	-0.	0.19079€ 03	0.11462F 04	1.46
-n.155PAF 03	-0.30260F 03	-0. -0.	-0. -0.	0.33811E 03 0.41205E 03	0.11992E 04 0.06337E 03	2.47
-0.37994F 03	-0.479428 02	-0.	-0.	0.127316 04	0.3740ZE 02	1.90
-n.43/37E 03	0.74081E 02 0.12973E 03	-0. -0.	•. •.	0.13 5976 04	-0.370746 02	1.60
-0.117485 03	0.10294F 03	-0.	•:	0.94647E 03 0.29124F 03	-0.34892E 03 -0.24015E 03	3.65
-0.490886 02	0.40524F 02	-0.	ò.	0.16038F 03	-0.19366E 03	1.90
-0.43528E 02	0.8320~F 02 0.74240F 02	-C. -0.	•. •.	0.96433E 02 0.63761E 02	-0.14337E 03 -0.99383E 02	2.30 2.30
-0.28166C 02 -0.16717E 02	0.742228 02	-0.	0.	0.501316 0	-0.700s0€ 02	2.30
-0.101436 02	0.83323E 02	٥.		0.712966 02	-0.74324€ 02	2.30 2.30 2.35
-0.313196 01 0.75040F 01	0.88129F nz 0.88451F 02	n. 0.	7. 0.	0.45302F 02 0.44227F 02	-0.51263E 02 -0.49774E 02	i jo
0.143715 02	0.87 990f 02	ŏ.	o.	0.773266 02	-0.491968 02	2.46
0.1769RE 02 0.24526F 02	0.84172F 02	0.	<u>.</u>	0.840178 02	-0.40791E 02	2.45
0.29734E 02	0.80294F 02 0.84412F 02	0. 0.	♂. 8.	0.10044F 03 0.99439F 02	-0.82285E 02	2.17
0.266705 02	0.920778 02	0.	●.	0.726176 02	-0.63704E 02	2.90 2.96 4.96 2.66
0.3491#E 02	0.10295F 03 0.10103F 01	0.	0.	0.48018k 05	-0.39377E #2	4.56
0.104496 03	-0.17743F 02	o. o.	· · · · ·	0.93274F 02 0.13587F 53	-0.38887F 02 -0.18248F 03	3.5
0.173346 03	-0.294936 02	0.	-0.	0.178415 03	-0.183766 63	2.70 2.80 3.00
0.47373F 01 0.47166F 01	0.64228F 01 0.40020F 02	n. -o.	-o. o.	-0.392836 02	-0.13745E 03	3.00
0.199348 02	0.435338 02	0.	0.	-0.32166F 02 -0.43585F 02	-0.15329E 03 -0.16442E 03	1.10
0.424235 02	0.77773F 02	0.	••	-0.40805F 02	-0.17167E 03	3.30 3.36
0.44690E 02	0.41010F 02	0.	٥.	-0.79495€ 02	-0.17083E 03	5.26
0.13596E 03 0.23354F 03	0.78895E 02 -0.18190E 03	0. 0.	o. -o.	-0.14326F 03 -0.21833F 03	-0.70406E 02 0.24021E 02	2.5
0.145755 03	-0.21045E 03	ŏ.	-0.	-0.332166 02	-0.30722E 02	13
r.10007E 03	-0.22572E 03	0.	-0.	-G.10429E 02	-0.748795 02	3.55
r.15425E 02 -n.65334E 02	-0.14012F 03 -0.64218F 02	0. -0.	- 0. -0.	-0.37183E 02 -0.99948E 02	-0.34943E 02 0.22061E 01	3.60
-0.61337E 02	-0.439948 02	-0.	-6.	-0.74995E GT	" 0.10392€ 02	5.65
-0.52110F 02	-0.28084F 02	-0.	-0.	-0.333626 02	0.144697 02	4,00
-0.45395F 02 -6.48652E 02	-0.14977E 02 -0.84869E 01	-n. -0.	-0. -0.	-0.40148E 02 -0.24118E 02	0.17477F 02 0.10131F 02	4,30 ·
-0.40059E 32	-0.47336E 01	~6.	-0.	-0.12493E 02	0.14211E 02	1,76
-0 40.444 05	0.730827 01	-0.	•.	-0.44391E 01 0.91310E 01	-0.21699F 01 -0.12607E U2	4.80
-0.41078F 02	0.124128 02 0.141258 02	-0. -0.	•. 0.	0.707376 01	-0.157048 02	
-0.22051E 02	0.023218 01	-0.	•.	-0.26079E 02	0.244616 01	5.15
-0.1425AF 02	0.73037E 01 0.41940E 01	-0. -0.	o. •-	-0.26414E 02 -0.21823E 02	0.11057E 02 0.13962E 02	5.30
-0.18951F 02 -0.17951E 02	0.895756 01	-0.	:	-0.138786 02	0.14526E 02	2.00
-n.16427F 02	0.414226 01	-0.	0.	-0.12130€ 02	0.133136 02	\$6.
-0.15788E 07 -0.15069E 07	0.89938E 01 0.87353E 01	-0. -0.	3. -0.	-0.83649E 01 -0.79067E 01	0.127036 02 0.119406 02	6.60
-0.14960E 02	0.84372F 01	-0.	-0.	-0.76866E 01	0.111406 02	73
-0.1904#6 02	0.46028F 01	-•.	-0. 0.	-0.7832#E 01 -0.10155E 02	0.12310E 02 0.14493E 02	6.00
-0.14486F 02	0.12237F 02 0.13445F 02	- 0. -0.	•	-0.64766F 01	0.1400FF 02	6.10
-0.1747RE 07	0.147636 02	-0.	••	-0.34227E 01	0.11880F 62	7.7
-0.85533E 01	0.141106 02	-0. -0.	•:	0.31474E 00 0.27632E 01	0.96840E 01 0.57489E 01	7-42
-n.50057E 01	0.14330F 02 0.11684F 02	-0.	š.	0.347416 01	0.13748E 01	I: E
0.60764F 01	0.66756E 01	0.	••	0.38425€ 01	-0.148126 01	1.00
0.95328F 01	0.11647E 01 -0.11034E 81	:		0.33948F 01 0.74338F 00	-0.28213F 01 -0.19117F 01	30.00
0.911005 01	-0.64079F 01	•.		-0.156196 61	0.179376-00	11.60
#.61152F 01	-0.78380f Ol	0.	-0.	-0.25899£ 01	0.24386E 01	11.00
W. 22178F 01 -0.14018F 01	-0.74A6&E 01 0.	•. -0.	-6.	-0.20332E 01 -0.10034E-00	0.38779E 01	
-40 1 401 105 Of	••	- 40	**		••	12-44

(PSI/FPS SINUSOIDAL GUST)

297,000 LB CUTOFF FREQUENCY-24,000 FT 0.85

15 CPS

GROSS WEIGHT: ALTITUOL: MACH NUMBER:

BODY BALANCE STATION: 540 SEGMENT NUMBER 17

INCREMENTA	L SEEAR STRESS					
REAL	I MAG I MAR Y					
						PRESENCE
-0.41556E-01	-0.11691E 01	0.	-0.	••	-6.	6.30
-0.12179E 01 -0.18598F 01	-0.77898E 00 0.91125E-01	0. 0.	-0. -0.	0. 0.	-0.	8.30 0.36 0.44
-0.19906F 01	0.50408F 00	-0.	-0.	-A.	-0.	0.44
-0.17376F 01 -0.13531F 01	0.746257 00 0.80593E 00	-0, -0.	-0. -0.	-0. -0.	-0. -0.	0.90 0.60
-0.11179F 01	0.424216 00	-0.	-0.	-0.	-0.	0.70
-0.98007E 00 -0.90381C 00	0.84129F 00 0.87597F 00	-n. -o.	-0. -6.	-9. -0.	-0.	0,70 0,80 0,90 1,30 1,34 1,40
-n.87077E 00 -n.90624E 00	0.107536 01	-0.	-•.	-0.	-0.	1.00
-0.56810F 09	0.15150E 01 0.19134F 01	-9. 0.	-0. -0.	-0. 0.	- ?. -6.	1.50
-9.4236PF 00	0.243316 01	0.	-0.	0.	-0.	1:5
-0.69876F 00 -0.42273F-00	0.27529E 01 0.30-82E 01	-0. -0-	- 0 .	-0. -0.	-0. -0.	1.47 1.47 1.59 1.55 1.66 1.65 1.80
0.495746-01	0.290396 01	~0.	-0.	-0.	-0.	1.50
0.15715E 01 0.23197F 01	9.13945E OL 0.31194E-00	-0. -0.	-8. -0.	-0. -0.	-0. -0.	1.95
0.180535 01	-0.14333F-00	-0.	0.	-0.	0.	1.65
0.45002F-00 0.75392F-01	-0.650908-01 0.30563E-01	-0. -0.	0. 0.	-0. -0.	o. o.	1,80
-0.170736-02	0.13103E-00	-0.	n.	-0.	0.	
-0.36177E-00 -0.53919E 00	0.25075F-00 0.45146E-00	-0.	-9. -0.	-0. 0.	-6. -0:	2.18 2.30 2.30 2.35 1.16
-0.70218E 00	0.56544E 00	0.	-0.	9.	-0.	2.30
-0.73011E 00 -0.73506E 00	0.70344E 00 0.79091F 00	0. 0.	-0. -6.	0. 0.	-n. -0.	2.35
-0.66940F 00	0.81300F 00	0.	-0.	0.	-0.	2,45
-0.614RSE 00 -0.12446E-00	0.733396 00	0. 0.	-9. -0.	0. 0.	-0. -0.	2.45 2.46 2.47 2.55 2.56 2.65 2.70 3.80
-0.41417E-00	0.20201E-00	o.	-0.	9.	-0.	2.97
-0.76765F 00 -0.11547E 01	0,73893E-00 0.45268E-00	0. 0.	-0. -9.	0. 0.	-0.	2.54
-0.17489E 01	0.77C02E 00	o.	-0.	e.	-0.	2.66
-0.23210E 01 -0.31240E 01	0.28963E 01 0.46034E 01	0. n.	-0. -0.	0. 0.	-0. -0.	2.70
-0-87678F 00	0.57242# 01	٥.	-0.	· 0.	-0.	3.40
-0.43960F-00 0.77891E 00	0.79344E 01 0.10005E 02	0. -0.	-o. -b.	0. ~0.	-0. -0.	3.00 3.30 3.37 3.37 3.35 3.36 3.70 3.70 4.30 4.70 4.70
0.2 1924E 01	0.11310E 02	-0.	-0.	-0.	-0.	3.25
0.49549E 01	0.120246 02	-0,	-0.	-0.	-9.	3.39
0.1211FF 02 0.20749E 02	0.39513E 01 -0.91543E 01	-n. -o.	-0. 0.	-0. -0.	-0. 0.	7.75
0.21939F 01	-0.42554E 01	-0.	0.	-0.	0.	1,5
-0.55691F 00 0.95739E 00	-0.18374E-00 -0.18467E 01	0. -0.	-0. 0.	0. ~0.	-u. 0.).56 1.60
0.52453F 01	-0.36728F 01	-0,	0.	-0.	6.	3.70
0.31931F 01 0.17466F 01	-0.34853E 01 -0.29987E 01	-0. -0.	0. 0.	-0. -0.	0.	3.85
0.77809E 00	-0.24141E OI	-0.	0.	~0.	0.	4.20
n.72858F-01 -n.13809E-00	-0.21506E 01 -0.20910E 01	-0. -0.	o. o.	-0. -0.	o.	4.90 4.70
-C.14757F-00	-0.295170 01	-0.	٥.	~0.	0.	4.80
-0.19904F-00 -0.34072E-00	-0.25313E 01 -0.23#03E 01	-A. -O.	o. o.	-0. -0.	0. 0.	4.96 5.00
-n.13132F 01	-0.16525E 01	0.	0.	0.	0.	5.00 5.13 5.30
-0.159:6E 01	-0.15331E 01 -0.12681E 01	0. 0.	0. 0.	C. 0.	0.	3.30
-n. 24453F nt	-0.87254E 00	0.	0.	0.	o.	5.00 9.70
-0.33136E 01 -0.51447F 01	0.10737E 01 0.19715E 01	0. 0.	-0. -6.	0. 0.	-0. -0.	5.65
-0.542736 01	G.31494F Ol	ō.	-0.	0.	-0.	6.03
-0.54653E 01 -0.50013F 01	0.45877E 01 0.49601E 01	0. 0.	-0. -0.	0. 0.	-0. -0.	6.06
0.14277F 01	0.91832F 00	-0.	-0.	-0.	-0.	\$.09 6.20
0.15643F 01 0.84113F 00	0.14376E-CD -0.12446E-00	-0. -0.	-0. -0.	-0. -0.	-0. -0.	6.40
0.257447-00	-0.12172E-06	-0.	-0.	-0.	-0.	6.40 7.00
0.61850F-01 -0.212:6E-01	-0.54738E-01 -0.37764E-01	-0. -0.	-0. -0.	-0. -0.	-0. -0.	5.70 5.09 6.09 6.09 6.00 6.00 6.40 7.00 7.00 8.30 9.70
0.123246-03	-0.627696-01	-0.	-0.	-0.	-0.	\$.30 \$.40
0.177137-01	-0.96603E-01 -0.10807E-00	-0. -0.	-0. 0.	-0. -0.	-0. 0.	10.00
-0.392616-01	-0.65250F-01	-0.	0.	-0.	0.	32.00 32.00
-0.74680F-01 -0.3888F-01	-0.48494E-01 -0.17063E-61	-o. -o.	-0. -0.	-0. -0.	-0. -0.	13.40
-0.75923E-01	0.	-0.	0.	~0.	0-	14,09 26,69

(PSIFFPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

297, 000 LB 24, 000 FT 0. 85

CUTOFF FREQUENCY:

15 CP5

BODY BALANCE STATION: 820 SEGMENT NUMBER 1

DICHOGRAL AXIAL STREET

		ALL U. ALL	
	REAL	IMAGIMAN Y	
0.	0.1402 0 F 0 1	0.11502E 03	e.
0.	0.917916 02	0.903446 02	0.
ō.	0.15504E 03	0.154006 02	0.
-A. -A.	0.10550F 03 0.17321F 03	-0.27779E 02 -0.61520E 02	0. 0.
-6.	0.14471E 03	-0.777936 02	0.
-0.	0.120958 03	-C. 90059F 02	0.
-0.	0.11#19F 03	-0.102956 03	0.
••.	0.111945 03	-3.11949E 03	0.
-0. -n.	0.10861F 03 0.10547F 03	-0.17704E 03 -0.27270F 03	
-0.	0. 882046 02	-0.34635E 03	ŏ.
-0.	0.539728 02	-0.43528E 53	. 0.
-0.	-0.21764F 0Z	-0.46013E 03	-0.
-0. -6.	-0.94391E 02 -0.20201E 03	-0.50046E 03 -0.39070E 03	-0. -0.
-0.	-n.48505f 03	-0.40610E 02	-0.
•.	-0.536496 03	0.113246 03	-0.
0.	-0.37861E 03	0.11700F 03	-0.
0.	-0.1044ZE 03	0.02119E 02 0.52419E 02	-0.
0. n.	-0.46515E 02 -0.15921E 02	0.5241 9E 02 0.24898E 82	-0.
o.	0.16612F 01	0.04539€ 00	0.
-0.	n.11133F 0.	-0.25290E 02	0.
-0.	0.41003F 01	-0.3314 0E 0Z	••
-0. -0.	0.33616E 01 -0.54249E 01	-0.51983E 82 -0.61654E 82	-0.
-0.	-0.21619F 02	-0.42007E 02	-0.
-0.	-0.31440F 0Z	-0.46209F 02	-0.
-ń.	-0.30657E 02	-0.963297 01	-0.
0. -0.	-0.57843F CZ -0.22123F OZ	0.25102F 01 -G.13984F 02	-0.
-0.	-0.20405E 01	-0.47191F 02	-0.
-0.	-0.474385 01 -0.272246 02	-0.43847E 02	-0.
-0.	-0.27224E 02	-0.4792 0F 01	-0.
-0.	-0.90642E 02 0.99613E 02	-0.446990 02 -0.12611E 03	-0.
-0. -n.	0.401518 02	-0.233126 03	ě.
-0.	0.19306F 0Z	-0.32541E 83	•.
-0.	0.19306F 02 -n.56971F 02	-0.38909E 03	-0.
-0.	-0.14021E 03	-0.41039E 03	-0.
-0. 0.	-9.41500E 03 -0.73703E 03	-0.45400E 02 0.34171F 03	-0.
ő.	0.100196 02	0.114106 03	-0. 0.
-0.	0.10482F 03	-0.63286E #2	0.
o.	0.13207E 01	0.502396 02	0.
0. 0.	-0.22218F 03 -0.12256F 03	0.15005E 03 0.15279E 03	-0.
ñ.	-0.548748 02	0.130118 03	-0. -0.
0.	-0.10150F 02	0.10090# 03	-0.
o.	0.240336 02	0.84378E 02	0.
o. o.	0.424228 62	0.749368 02	0.
0.	U.52274E 02 0.67968E 02	0.44714F 02 0.31254F 02	o. •.
a.	0.454958 02	0.214906 02	ŏ:
0.	0.31126F 02	0.341127 62	0.
0.	0.304216 02	0.347656 02	0.
6 . 0.	0.40272E 02 0.43146F 02	0.29496E 0Z 0.21124F 0Z	0.
-17.	0.484936 02	-0.12977E 02	•.
-0.	0.984986 02	-0.27972E 02	ŏ.
-0.	0.102976 03	-0.47786E 64	•
-0. -0.	0.10341F 03 0.93663E 02	-0.71000E 02 -0.77724E 02	•
-0.	-0. 06 335E 01	-0.14992E 02	o. -0.
-0.	-0.107298 02	-0.51031E 01	-0.
-0.	0.12145E-01	-0.479406 01	•
-0. -0.	0.43143E 01 0.50231E 01	-0.72223E 01 -0.76245E 01	
-0.	0.457006-00	-0.96249E 01 -0.05320E 01	:
-0.	-0.50716F 01	-0.43969E 01	⊸.
-0.	-0.89644F 01	0.123246-00	-0.
6.	-0.94285E 81 -0.78864E 81	0.26350€ 01	-0.
0. 0.	-0.78866 01 -0.576006 01	0.34722E 01 0.34300E 01	-0.
e.	·0.41276E 01	9.279688 01	-0.
9.	-0.323406 01	0.	-0.

င်ငံခင်ငံခင်ငံခင်နှင့်နှင့်နှင့်သူတို့ ရီခေင်ဝခိုင်နှင့်နှင့်နှင့်နှင့် ငေဝတို့သူတိုင်ငံစစ်ဝတ်ဝတ်ဝင်ငင်ငင်တွင်ဝင်ခေတ်နှင့်ခဲ့နှင့်ချင်ချင်

(PSI/FPS SINUSOIDAL GUST)

297,000 LB 24,000 FT 0.85 GROSS WEIGHT: ALTITUDE: MACH NUMBER: CUI OFF FREQUENCY: 20 CPS

SEGMENT NUMBER 10 PERCENT SEMI SPAN: 27

	SEEAR CORES	-	L AXIAL STRESS			
REAL	3PAG14ARY	REAL	1 MAG 1 MARY			PROGRAMMET
0-0-2716-01	0.35365 6 02	-0.0C081E CO	D. 30545E 05	0.	-0,	0.30
0.24347E 02 0.44132E 02	0.29947E 02 0.83452E 01	0.19898E C3 0.3702CE CS	0.26210E 05 0.78906E 02	-0. -0.	-0. -0.	0.36 0.36 0.44
0.553936 02	-0.51332 f 01 -0.16505 f 02	0.473396 03	-0.59261E 02	-0.	0.	0.44
0.53094E 02 0.46190E 02	-0.16505E 02	0.457548 (3	-0.14192E 05 -0.14748E 03	-0. -0.	o. o.	0.30 3.60
0.41253E 02	-0.27016E 02	0.350COF C3	-0.24396E 05	-0.	0.	0.70
0.30231E U2 0.34521E U2	-0.31273 E 02	0.32986E C3 0.31144E C3	-0.28530E 03 -0.33952E 05	-0. -0.	o. o.	0.80 0.90
0.35924E 02	-0.552531 02	0.248576 63	-0.51720E 05	-0.	0.	1.00
0.36653E C2 0.33495E 02	-0.47188E 02	0.27743E 03 0.20963E 05	-0.274E 03 -0.10254E 04	-o. o.	o. o.	1.30 1.36 1.40
0.2284PF G2	-0.16771 E 03	0.96540F 02	-0.12780E 04	0.	-0.	1.49 1.45
-0.42896E 01 -0.32343E 02	-0.18946E 03 -0.27625E 03	-0.14204E 03 -0.36551E 03	-0-14050E 04 -0-14789E 04	0. 0.	-0. -0.	1.49
-0.75495E 02 -0.39911E 03	-0.17111E 03	-0.69482E 03 -0.15332E 04	-0.10976F 04 -0.30145E 03	0. -0.	-0. -0.	1.99 1.99 1.60
-0.23925E 03	0.464626 02	-0.166626 04	0.40830E 03	-0.	0.	1.6
-0.18287E 03 -0.76840E 02	0.700P4E 02	-0.11743E 04 -0.54373E 03	0.40870E 03 0.30779E 03	-0. -0.	0.	1.69 1.80
-0.4530RE 02	0.627498 02	-D.20140E 03	0.25180E 03	-0.	0.	1.99
-0.3021PE 02 -0.26436E 02	0.63714E 02 0.655#0E CZ	-0.12 059E C3 -0.77273E 02	0.17513E 03 0.12795E 03	-0. -0.	0.	2.00 2.10
-0. S6218E U1	C. /4497E 02	-0.541496 07	0. 671796 02	-0.	0.	2.5
0.34082E 02 0.17233E 02	0.70093E 02 0.27041E 02	-0.52018E 02 -0.52413E CZ	0.70001E 02 0.46193E 62	0. 0.	0.	2.30
0.24472E C2	0.79822E 02	-0.655308 02	0.562458 02	0.	0.	2.40
0.34061F 02 0.39009E 02	0.79724E D2 0.72921E D2	-G-85646E 02 -D-97472E C2	0.35754E 02 0.564#6E 02	0. 0.	٠. ٥.	2,43 2,44
0.505331 02	0.633526 02	- 0-12910E 03	U. 10052E 03	ŏ.	o.	2.47
0.50381F 02 0.45535E 02	0.65314E 02 0.64011E 02	-0-120400 05 -7-842400 02	0.114741 C3 0.970856 02	0. 0.	0. 0.	2.90 2.94 2.98 2.69
0.546311 CZ	0.10592E 03	-3:633126 02	0.660916 02	0.	0.	2.50
0.10102F 03 0.14764F 03	0.10454	-0.48845E 02 -7.96245E C2	0.54672E 02 0.14805E 03	0. 0.	0. -9.	2.65
0.29234F 03	-0.36349E 03	-C.14415F 03	0.143278 03	0.	-n.	2.70 2.80
-0.28309E 02	-0.11012E 03 -0.93940E 02	0.16352E C2 0.26914E 02	0.11391E 03 0.96725E 02	- 0. - 0.	-0. -0.	3.00
-0.233226 02	-0.852846 92	0.307918 02	0-903106 02	-0.	-0.	3.10 3.20 3.26
-0.810361 02 -0.8823#f 02	-0.83362E 02 -0.62515E 02	0.33573E 07 0.36771E 07	0.875056 02	-¢. -0.	-0. -0.	3.26 3.29
-0.10972f U3	-0.71410E 02	0.450808 02	0.424416 02	-0.	-0.	3.35
-0.15204F D1 -C.20169E D3	0.897C7E 02 0.16242E 03	0.53823E 02 0.10523E 02	0.45047E 02 0.41214E 02	-0. -0.	o.	3.40 3.92
-0. 164CSE 01	0.216431 03	0.472206 01	0.774436 02	-0.	0.	3.36
-0.1003°F 02	0.57863E 02 0.12314E 02	0.14542E 02 0.41709E 02	0.45131E 02 0.42110E 02	-9. 0.	0. -0.	3.40
0.4772AE UZ	-0.540#5E 01	0.476206 02	0.32345E 02	0.	-0.	3.50
0.749121 07 0.30327E 02	-0.30415E 02 -0.103201 02	0.44214E 02 0.42425E 02	0.25349E 02 0.39653E 02	o. o.	-0. -0.	4.00
0.347C7f-00	-C.92066f 01	0.473276 02	0.139686 02	-9.	-0.	4,50
-0.392891 C1 -0.49824F 01	-0.64821E 01 -0.62779E 01	0.40177E 02 0.73053E 02	0.78723E 01 -0.53518C 02	-0. -0.	-6. 0.	4.70 4.80
-0.1981/1 01	-0.61727E 01	0.95;246 02	-0.024936 02	-0.	0.	4.96
-0.725011 01 -0.708011 01	-0.45179E 01	0.942977 02 -0.12810E 02	-0.85851E 02 -0.36016E 02	-n. 0.	0. 0.	5.00 5-13
-0.78159E 01	-0. 01	-0.307 266 02	-0.15078E 02	0.	-0.	5. 3 6
-0.413245 01 -0.108935 02	-0. 11 01 -0.11-716 01	-0.27972E 02 -0.20631E 02	-0.77192E 01 -0.36687E 01	-0. -0.	-0. -0.	5.50 5.50 5.10
-0.130551 02	0.366351 01	-0.239596 02	0.10526F J2	~0.	-0.	5.05
-0.173xff 07 -0.180591 67	0.603531 01 0.89105E 01	-0.35203E 02 -0.37039E 02	0.16962E 02 0.25569E 02	-0. -0.	0. 0.	6.00 6.03
-0.18167F 02 -0.17110E 02	0.123101 02	-0.37363f 02 -0.33715F 02	0.35800F 02	-0,	o.	6.06 6.08
-0.20813F 01	0.492346 01	0.13415F 0Z	0.58720F 02 0.93671F 01	-0. -0.	o. o.	6.20
-0.1430#1 01 -0.2760#6 01	0.363681 01 0.381988 01	0.15748E 02 0.11146E 02	0.21754E 01 0.13424E-00	-0. -0.	0.	6.40 6.40
-0. 129401 01	0.439641 01	0.73215F 31	-0.84774# 00	-0.	o. o.	7.00
-0.28731F 01 -0.11105E 01	0.51455E 01 0.484791 01	0.58944E 01 0.42603E 01	-0.17366E 01 -0.25299E 01	-0. -0.	o. o.	7:30
0.8168/1 CO	0.37366 € 01	0.354928 01	-0.20575F 01	-0.	ŏ.	9.00
0.742##F C1 0.3610#F 01	0.157238 01	0.22131F 01 0.179641 01	-0.30441E 01 -0.28345E 01	0. 0.	0. 0.	10.00
0.374816 01	-0.122141 01	0.760496-00	-0.24430E 01	0.	-0.	12.00
0.31242f 01 0.214011 01	-0.218711 01 -0.25813E 01	-0.54565E 00 -0.10458E 01	-5.18949E 01 -0.12299E 01	0. n.	-0. -0.	15.00 14.00
0.478448 00	-0.253941 01	-0.142431 01	-0.46189E 00	0.	-0.	15.00
-0.3364CF-00 -0.10099E 01	-0.20754E 01 -0.13980E 01	-0.35827# 01 -0.14200# 01	-0.19647E-01 0.62906E 00	-0. -0.	-0. -0.	16.00
-0.360591 01	-0.64391 E 00	-0.356148 01	0.128316 01	-0.	٥.	17.00 18.00
-0.182381 01 -0.125916 01	0.77286E-01	-0.12944E 01 -0.78641E 00	0.19489E 01	-0. -0.	o. o.	19.00 80.00

(PSI/FPS SINUSOIDAL GUST)

297, 000 LB CUTOFF FREQUENCY: 24,000 FT 0, 85

20 CPS

GROSS WEIGHT: ALTITUDE: MACH NUMT _R:

PERCENT SEMISPAN: 27 SEGMENT NUMBER 14

TW NOOFTAL OF OL	SEEAR STRESS 1FACTR ARY	ENCREMENTAL DE AL	L AXIAL STREES 1 PAG 1 NARY			
	** = 0 *******					PRESENCE
0.62963E-02	0.107096 03	0.73216E 00	0.27555E 03	0.	-0.	- GP4
0.72152F 02 0.13703E 63	0.51329E 77 0.26474E 02	0.18057E 03 0.33594E 03	0.23785E 03 0.716C5E 32	-0. -0.	-0. -6.	0.30 0.36
0.166709 C3	-0.14496E 02 -0.49199E 02	0.42959E 03 0.41525E 03	-0.35629E 02 -0.12879E 03	-0.	0. 0.	0.44
0.13927E 03	-0.473136 02	0.343456 03	-0.179216 03	-0.	ŏ.	0.50 4. 6 0
0.123946 03	-5.81121E 07	0.32488E 03 0.29934: 03	-0-31866E G3	-0.	0.	0.70
0.11+1/E C3 0.1007E C3	-0.95111E 02 -4.11235E 03	0.24348 03	-0.25891E 03 -0.30610E 03	-0. -0.	G. 0.	0.80 0.90
0.10+515 03	-C.17053E 03	0.271316 03	-0.46955E 33	-0.	٥٠	1.00
0.10109E 03 0.85409E 07	-0.26 907E C3	0.25152€ ö3 0.19922€ 03	-0.72869E 03 -0.93069E 03	-0. 0.	o. o.	1.20 1.7
0.57659E 02	-0.43977E C3	0.874276 02	-0.11597E 04	0.	-0-	1.40
-0.22229E 02 -0.53439E 02	-0.48954E 03 -0.524111 03	-C-12890E 03 -0.33169E 03	-0.12750E 04 -0.13421E 04	0. 0.	-0. -0.	1.45 1.47
-0.20717E 01	-0.41590E 03	-0.63053E 03	-0.99607E 03	0.	-0-	1.50
-0.58304E 03	-0.745671 02 0.11749E 03	-0.13914E 04 -0.15120E 04	-0.92016# 02 0.37053# 03	-0. -0.	-0. 0.	1.25 1.60
-0.42834E 03	0.14758E 03	-0.106568 05	0.37089E 23	-0.	0.	1.65
-0.14804E 03 -0.88C428 C2	0.12261E 03 0.10529E 03	0.J3U08E 03 -0.18286E 03	0.27932E 03 0.21035E 03	-0. -6.	0. 0.	1.80 1.90
-0.559798 02	0.95527E 02	-0.10944# 03	0.15893E 03	-0.	٥.	2.00
-0.35501E 02 -0.19522E 02	0.92228E 02	-0.70124# 02 -0.49139# 02	0.11611E 03 0.79113E 02	-o. -o.	0. 0.	2.10 2.20
0.498246 01	0.AC-30E 02	-0.47205# 02	G-63524E 02	0.	0.	2.30
0.82011E 01 0.13672E C2	C.418;2E 02 0.82053E C2	-0.477451 02 -0.594678 02	0.41919E 02 0.32491E 02	0. 0.	0. 0.	2.35 2.40
0.17E7#E GZ	0.81692E 02	-0.77721E 02	0.324466 02	r.	Ģ.	2.43
0.19751E 02 0.22460E 02	0.80379E 02 0.84092E C2	-0.88454£ 02 -0.11716£ 03	0.51259E 02 0.91037E 02	o. o.	o. o.	2.44 2.47
0.22254E G2	0.572278 02	-0.11476E 03	0.10414E 03	ð.	0.	2.50
0.30142f 02 0.4P325E 02	0.10558E 03 0.125%E 03	-0.76446E 02 -0.57454E 02	0.88102E 02 C.599761 02	0. 0.	0. 6.	2.54
0.10282E #3	0.119427 03	-0.62475E 02	0.51429E 02	٥.	0.	2.58 2.65
0.17352E 03 0.30256F 03	-0.10660E 03	-0.87343E @2 -0.13263E @3	0.13435E 03 0.13002E 03	0. 9.	-0. -0.	2.70
-0.32133E 02	-0.13427E 03	0.14839E 02	0.10337E 03	-0.	-0.	2.80 3.00
-0.62833E 02 -0.84703E 32	-0.11983E 03	0.24474£ CI 0.27943E O2	2.87774E 02 0.81954E 02	-c. -o.	-0. -0.	3.10
-9.103946 03	-0-12418E 03	0.30421E 02	0.79409E 02	-0.	-0.	3. 2 0 3.26
-G.1209(E C3	-0.12836E 03	0.32870E 02	0.73401E 02	-0.	-0.	3.29
-0.17459f 03 -0.26031E 03	-0.86947E 02 0.16193E 03	0.40909E 02 0.48843E 02	0.56664E 02 0.408C0E 02	-e. -o.	-0. 9.	3.35 3.40
-0.25547E 03	0-24410E 0!	0.95496E 01	0.555528 02	-0,	9.	3.52
-0.203901 03 -0.50671E 02	0.29425E 01 0.15183E 01	0.43305E 01 0.13196E 02	0.70278E 02 0.59105E 02	-0. -0.	o. o.	3.56 3.60
0.45432E 02	C. 45470 E 02	0.43294E 02	9.38221E 02	0.	-0.	3.8
0.60834E 02 0.1*280E 02	0.1549ZE 02 0.924C9E 01	0.43214E 02 0.40125E 02	0.29370E 02 0.23003E 02	o. o.	-0. -0.	3.85 6.00
0.248611 02	0.51220E 01	0-38485E 02	0-169276 02	0.	-0.	4.20
0.199417 02	0.25259E 01 -0.15877E 01	0.42948E 02 0.54609E 02	0.10861E 02 0.26067E 01	-0. -0.	-0. -0.	4.50 4.70
0.25251E 02 0.32130E 02	-0.33487E 02	0.66294E 02	-P. 48567E 02	-0.	-0. 0.	4,80
0.44612E 02	-0.50052E 02	0.86322E 02 0.76497E 02	-0.74860E 02 -0.77907E 02	-0.	0.	4.96 5.00
0.38374f 02 -0.21209E 02	-0.51639E 02 -0.21753E 02	-0.161625 02	-0.32684E 32	-o. o.	0. 0.	5-15
-0.28879F 02	-0.83046E 01	-0.27469E 02 -0.20846E 02	-0-13681F 02	0, -0.	-0.	5-15 5-50 5-50
-0.248041 02 -0.23532E 07	-C.27817£ 01 0 52690E 00	-0.127236 02	-0.70050E 01 -0.33293E 01	-0.	-0.	5.70
-0.25752F 02	9.10755E 02 9.151C6E 02	-0.21742E 02 -0.31946E 02	0.95523E 01	-0. -0.	-0.	5.45
-0.333101 02 -0.34570E 02	0.213246 02	-0.336126 02	0.15393E 02 0.23204E 02	-0.	o. o.	6.00 6.03
-0.34680E 07	0.28499E 02	-0.33724E 02 -0.30596E 02	0.32487E 02	•0•	0.	6.06
0.4293#[-60	0.31583£ 02 0.11871£ 92	0.12174E 02	0.35319E 02 0.85005E 01	~e. -0.	0. 0.	6.08 6.20
0.3110#F 01	0.79756E 01 0.63026E 01	0.14791E 02	0.28797E 01	-0.	0.	6.40
0.8210FE 00 -0.2P551E-00	0.593336 01	0.101158 02 0.668958 01	0.12182E-06 -0.78745E 00	-0. -0.	0. 0.	6.60 7.00
-0.604168-01	0.5465VE 01	0.53491E 01	-0.15759E 01 -0.21143E 01	-0.	0.	7.40
0.1%4#2E 01 0.29902E 01	0.39499E 01 0.230471 01	0.38661E 01 0.32213E 01	-0.25929E 01	-9. -0.	0. 0.	9.00
0.3400nE 01	0.42232E 00	0.20084E 01 0.11765E 01	-C.27624E 01	0.	0.	10.00
0.35553E 01 0.298GA1 01	-0.45267E-00 -0.46154E 00	0.23638E-00	-0.257228 01 -0.22170E 01	o. o.	•. -0•	11.00 12.00
0.222356 01	-0.1220eE 01	-0.49\$16E-00	-0.17196E 01	0.	-0.	13.00
0.17177E 01 0.1375#E C1	-0.125421 01 -0.14971E 01	-C.94902E DG -O.12925E GI	-0.21161E 01 -0.60065E 05	o. o.	-0. -0.	14.00 15.00
0.11350F 01	-0.17283E 01	-0.14363E 01	-0.17829E-01	-0.	-0.	14.00
0.86456E 00 0.37448f-00	-0.20385E 01 -0.222271 01	-D.14702E 01 -0.14169E 01	0.57086E 00 0.11626E 01	-0. -0.	-0. 0.	17.00 18.00
-0. 26450E-00	-0.20971E 01	-0-11750E 01	0.17686E 01	-0.	0.	19.90
-0.11484E 01	0.	-0.71+10E 00	0.	-0.	٥.	\$6.00

(PSIFFS SINUSOIDAL GUST)

297, 000 LB CUTOFF FREQUENCY: 24 000 FT U. 85

20 CPS

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

PERCENT SEMISPAN: 40.06 SEGMENT NUMBER 8

ERCHARDITAL REAL	. SARAP STREES IPAGINARY		AFTAL STREET			
46-6	177,14861	REAL	1 MAG 1 MARY			THE PROOF
-0.13427F-00	0.73760E C2	-0.12489E 01	0.27897E 03	0.	•	8,38
0-49232E 02	0.43168 E 02	0.18058F 03	0. 24197E 03	0.	v.	0.50 0.56 0.44
0.50530E G2 0.11480E L3	0.1849ZE 02 -0.95961E 01	0.338 69 E C3 0.434G4E 03	0.73329E 02 -0.34597E 02	o.	-0.	0.50
0.1105/E 53 0.56627: 02	-0.339C6E 02 -3.46659E 02	0.42277E 03 G.37331E 03	-0.33336E 03 -0.38676E 03	n.	-0.	0.50 0.60
0.604478 02	-9.56777E 02	0.332166 63	-0.22938E 03	•. •.	-t. -0.	9.70
0.80023E 02 0.762C7E 02	-0.47218E 02	0.304C7E 03 0.28859E C3	-0. 27272E 03 -0. 32593E 23	o. o.	-0. -0.	0.80 0.90
0 74177E GZ	-0.12527E 03	0.27612E 03	-0.49924E 03	6.	-0.	1.60
0.60560E 02	-0.201:5E C3	0.25167E 03 0.17001E 03	-0.77483E 03 -0.98754E 03	o. o.	-0. -0.	1.30 1.34
0. 34717E C2	-0.33621E 03	0.620:96 02	-0-12251E 64	0.	-0.	1,40
-0.24674E C2 -0.8768eE C2	-0.37509E 03 -0.40102£ 03	-0.17589E 03 -0.39572E 03	-0.33415E G4 -0.34035E G4	-0. -3.	-0. -0.	1.45
-0.1708GF C3	-0.31481E 03	-0.71432E 03	-0.10105E 04	-0.	-0.	1.9
-0.465571 03	0.104598 63	-0.15157E G4 -0.16147E G4	-0.43774E 02 0.43392E 03	-9. -0.	·0.	1.55
-0.34223£ 63 -0.11575£ 03	0.13052 E C3 0.1110 E 03	-0.11241E 04 -0.34000E 03	0.40837E 03 0.30.49E 03	-0. -0.	٥.	1.65
-0.71877E 02	0.97544E 02	-J. 18770E 03	0.22666E 03	-0.	•. •.	1.80 1.90
-0.4605-E CZ -0.30070E 02	0.854CBE 02 0.85379E 32	-0.11207E 03 -0.74624E 02	0.16780E 03 0.1363;E 03	-o. -o.	0. 0.	5.70 5.00
-0.10259E G2	0.85926E 02	-0.58672E 02	0. E2019E 02	-0.	0.	2.20
-0.11848E 07 -0.40112F 01	0.93029E 02 0.56563F 02	-0.83442E C2 -0.76428E O2	0.86987E 02 0.67019E 02	0. 0.	0. 0.	2.39 2.35
0.79364E 01	C.97271E 02	-0.7751 0E 02	0.58253E 02	0.	0.	2,40
0.1631/E C2 0.20157E 02	0.96724f-02 0.51623E 02	-0.70500E 02 -0.98331E 02	0.57577E 02 0.71140E 02	o. o.	0. 0.	2.43 2.44
0.2920°E G2 0.3080¢E G2	0.84825E 02 0.67224E 02	-0.11755E 03 -0.11170E 03	0.96304E 02 G.94983E 02	0.	٥.	2.47
0.294546 02	0.95293E 02	-0-84988E 02	0.74559E 02	o. o.	o.	2.50 2.54
0.36266E 02 0.45857E 02	0.10757E 03 0.10000E 03	-0.79607E 02 -0.10917E 03	0.46086E 02 0.45512E 02	C•	?. 0.	2.9k 2.9k 2.69
0.106178 03	-0.13741E 02	-0.15902E 03	0.21357E 03	0.	-0.	2.70
0.17830F 23 0.75315E 01	-0.26027E 02 0.11346F 02	-0.23245E 03 0.22571E 02	0.21506E 03 0.38451E 03	o. o.	-0. -0.	2.80 3.60
C. 512C9E 01	0.47821 2 02	0.37646E 02	0.10175E 03	-0.	0.	3,10
0.21327E 02 0.45046E 62	0.73985E 02 0.89867E 02	0.51011E 02 0.71164E 02	0.19243E 03 0.20092E 03	o. o.	0. 0.	3.30
0.797777 02	0.10414E 03	0.93173E 02	0.19993E 03	0.	r ,	3.29
0.1503°E 03 0.2575#E 03	0.32819F C2 -0.19768E 03	0.16767E 03 0.2>53F 03	0.10581E 03 -0.28114E 02	o. o.	- 0 .	3.35 3.40
0.148696 03	-0.21570E 03	3.38876E 02	0. 35956E 02	0.	-0.	5.52
0.16us1E 03 0.38651E 02	-0.27697E 03 -0.14463E 03	0.12440E 02 0.43518C 02	0.87634E 02 2.43237E 02	o. o.	-0. -0.	3.52 3.56 3.60
-0.58454E G2	-0.7*612E 02	0.11498E 03	-0.29820E 01	-0.	-0.	3.70 3.89
-0.57976E 02 -0.5095#E 02	-0.45976E C2 -0.33437E 02	0.87771E 02 0.64794E 02	-0.32397E 02 -0.16934E 02	-o. -o.	-0. - 0.	3.09 4.00
-U. 46018E 02	-0.15656E 02 -0.12295E 02	0.46988E 02 0.28227E 02	-0.20489E 02 -0.21220E 02	-0. -0.	-0. -0.	4.20
-0.43521E C7 -0.4614CE 07	-0.7:400E 01	0.14821E 02	-0. 18472E 02	-0.	-0.	4.90 4.70 4.60
-0.49780E 07	0.13953E G2 0.23633E G2	0.54529E 01 -0.10687E 02	0.25395F C: 0.14755E 02	-a. - a.	0.	4.96
-0.506948 02	0.25984E 02	-0.435126 01	0.18301F 02	-0.	o.	9 00
-0.17275E 02 -0.12780E 02	0.12101E 02 0.77925E 01	0.30522E 02 0.33255E 02	-0.30969E 01 -0.12940E 02	-0. -0.	o. o.	3.15 3.30
-0.141236 02	0.25495E 01	0.25541E 02	-0.14341E 02	-0. -0.	0.	3.30
-0.1408#E 02 -0.130G1E 02	0.78174E 01 0.65373E 01	0.18607E 02 0.14156E 02	-0.17001E 02 -0.15503E 02	-0.	o. o.	5-13 5-30 5-50 5-50 5-50 6-60 6-60
-0.10566F 07	0.56913E 01 0.45190E 01	0.97900E 01 0.92537E 01	-0,14867E 02 -0,13975E 02	-r. -o.	0. - 0.	6.00
-0.16087: 02	0.31275E C1	0.89962E 01	-0.13037E 32	-0.	-0.	6.66 6.08
-0.10561: 02 -0.17239F 02	0.38419E 01 0.9729BE 01	0.91907E 01 0.11085E 02	-0.14407E 02 -0.16962E 02	-0. -0.	-0. 0.	6.08 6.30
-0.1596ff D/	0.11746 E 02	0.758246 01	-0.16396E 02	-0.	٥.	1,40
-0.13637E 07 -0.95417t 01	0.13694E 02 0.14435E 02	0.40058E 01 -0.60244F 00	-0.13903E 02 -0.11334E 02	-0. ·0.	o. o.	.60 .00
-0.611076 01	0.142148 02	-0.32339F 01 -0.64070F 01	-0.672E3E ú1	-0.	0.	7.40
-0.75571E-01 0.520CAE 01	0.17040E 02 0.734CBE 01	-0.64379E 01	-0.14090E 01 0.19617E 01	- 0. O.	o. o.	9.00
C. 91311E 01	G. 20057E 01 -0.28879E 01	-0.47096E 01 -0.11278E 01	0.33020E 01 0.22374E 01	°.	o. -0.	10.00 11.00
0.1064#E 02 0.94345E 01	-0.64974E 01	G.18304F 01	-0. 209936-00	0.	-0.	22.00
0.65787E 01 0.25517E 01	-0.80264E 01 -0.77335E 01	0.30311E 01 0.2403 0 E 01	-0.20541E 01 -0.45381E 01	7. 0.	-0. - 0.	15.00 14.00
-0.129728 01	-0.57597 E 01	0.117446-00	-0.44539E 01	-0.	-0.	15.00
-0.47634E 01 -0.59677E 01	-0.30551E 01 -0.51029E-01	~0.29933E 01 -0.55505E 0 1	-0.25030F 01 0.84734E 00	-0. -0.	-0. -0.	16.00 17.00
-0.41324E 01	0.24047E 01	-0.68224F 01	0.45982E 01	-0,	0.	18.00
-0.51799E 01 -0.33513E 01	0.40225E 01	-0.59449E 01 -0.30834E 01	0.758000 01	-0. -0.	8.	19.00 20.00

(PSIFFPS SINUSCIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

297,000 LB CUTOFF FREQUENCY 20 CPS 24,900 FT 0.85

PERCENT SEMISPAN: 40.06 SEGMENT NUMBER 192

DICHESTAL SFAL	SEAR STRESS 1PAG17ABY			15, HA-1577 C1 AL	A, CLIST ALMENA AMBRITANA	
-0. 949C3F-01	0.717676 0		0.	0: 1ce?3E 01	-0.23836E 03	230 CON
0.44004E CZ	0.41323 6 0		••	-0.154245 03	-0.20474E 03	0.30 0.36 0.44
0.88C23E 42 0.11137E 33	0.18141E 0 -0.920C1E 0		0. -0.	-0.2493a8 03 -0.572590 03	-0.62655E 02 0.31270E 02	0.36
0.10714E 03	-0.52463E 0	? 0.	-0-	-0.361735 O3	0.11395E 03	0.50
2-93501E 02	-0.44720E 0		-0.	-5.31726F 53	0.15958E 03	0.60
0.83569E 02 0.27320E 02	-0.54730E 0 -0.64027E 0	? 0. ? 0.	-0. -0.	-0.23581E 03 -04151E GV	0.19582E 03 0.23302E 03	0.70 0.80
0.23425E 02	-0.76292E 0	2 0.	-0.	-0.2+638E 03	0.27849E 03 0.42658E 03	0.90
0.71493E C2	-0.118*1F O	3 0.	-0.	-0.22593E 03	0.42456E 03	1.00
0.70435E F2 0.59906E 02	-0.19038E 0 -0.74886E 0	0. 3	-0. -0.	-0.21504f 03 -0.15210F 03	0.66204E 03 0.84379E 03	1.20 1.34
0.34214E 02	-0.318C?E 0	3 0.	-0.	-0.527416 02	0.10467E 04	1.34 1.40 1.45
-0.189878 02	-0.35531 E O	-0.	-0.	4.15929F 03	0.11467E 04	
-0.73127E 02 -0.15588E 65	-0.38142 E 0	-0. -0.	-0. -0.	3,33911E 03 2,41205E 03	0.11992E 04	1.47 1.50 1.55 1.60 1.63 1.80
-0.324946 03	-0.30200E 0	2 -0.	-0.	0.12951E 04	0.84337E 03 0.37407E 02	1.25
-0.43237E 03 -0.32242E 03	G.94081E 0	2 -0. 3 -0.	0. 6.	3:13797E 04 3:94047E 03	-0.37076E 03 -0.34892E 03	1.60
-0.11349E 03	0.10294F 0	3 -9.	0.	0.291245 03	-0.26015E 03	1.60
-0.48CBRE 02	0.90529E 0 0.43204E 0	2 .0.	0.	9.14038E 03	*0.17364E 03	1.90
-0.4352PE 02	0.43204 F 0	? -0. ? -0.	o. o.	0.96435E 02 0.63761E 02	-0.14337E 03	2.00
-0.28166E G2 -0.16717E 02	0.29240E 0 0.29222E 3	2 -0.	0.	0.50131E 02	-0.99383F 02 -0.20080E 02	2.10 2.30 2.30 2.30
-0. 10193F 02	0.853236 '	2 0.	0.	9.712965 02	-0.24324E 02	2.30
-0.31319E 01	0.88129k	0.	0.	C.63302F 02	-0.57263E 02	2.35
0.75040E 01 0.14571E 02	0.88461 E	0.	0.	0.66227E 02 c.77324E 02	-0.497246 02 -0.4919:# 02	2.40 2.43 2.44 2.47
C-1249PE 02	0.82990E 0 0.84172E 0	2 0.	o.	3.840176 02	-0.4919:# 02 -0.40791E 02	2.44
0.245Z4E 02	0.80294E 0	2 0. 2 0.	0.	0.10044E 03 J.#5439E 02	-0.82285E 02	2.47
0.25736E 02 0.26670E 02	0.920726 0	2 0.	G.	3.72612E 02	-0.82285E 02 -0.81154E 02 -0.63704E 02	2.50 2.54 2.58 2.65 2.70 2.80
0.3451 02	0.10295 E 0	3 0.	0.	0.44018F 02	-C.39327E 02	2.56
0.45C40F 07 0.10469F 03	0.101016 0	3 0. 2 0.	0. -0.	9.93276E 02 9.33587E 63	-0.38887E 02 -0.18248E 03	2.65
0.17355E 03	-0.12743E 0 -0.29493E 0	2 0.	-0.	0.17461E 03	-0.18374(03	2.40
0.473256 01	0.66228E 0 0.40070E 0	1 0.	-0.	-a.19285E 02	-G.15765E 63	3.00
0.47144F C1 0.19904F 07	0.40070 € 0	? -0. ? 0.	0. 0.	+0.32166E 02 -0.53585E 02	-0.155298 03 -0.164428 03	3-10
0.424/3E 02	0.227736 0	2 0.	0.	-0.0 .805E 02	-0.17167E 03	3.20 3.26
0.646906 02	0.91010F 0	2 0.	0.	-0.29695F 02	-0.12063E 03	3.29 3.26 3.26 3.26 3.25 4.20 4.20 4.20 4.30 5.00
0.135966 03	0.28895 0	2 0.	0. -0.	-0.14326E 03 -0.21833E 03	-0.90406E 02 0.24021E 02	3.35
0.21356F 03 0.14525F 03	-0.18150E 0 -0.21045E 0) 0.	-0.	-0.33214E 02	-0.30722E 02	7.0
0.10007f 03	-0.22522E 0	3 0.	-0.	-U.10629E 02	-0.74879F 02	5.54
0.35625E 02 -0.65334E 02	-0.14012E 0	9 0. 2 -0.	-0. -0.	-0.37183E 02 -0.99945E 02	-0.36943E 02 0.22061E 01	3.60
-0.41337f 02	-0.43994 E 0	2 -0.	-0.	-0.749956 02	0.105928 02	3.6
-0.52110f 02	-0.28084 F O	2 -0.	-0.	-0.553626 02	0.10592E 02 0.14469E 32	4.00
-0.453951 02 -0.406521 02	-0.14977E 0	? -0. 1 -0.	-0. -0.	-0.40148E 02 -0.24118E 02	0.17672F 02 0.18131F 02	4.20
-0.40C5 F 02	-0.42336 E	1 -0.	-0,	-0.12493E 02	0.16211E 07	4.70
-0.4UR55E 02	-0.42336 E 0 0.73882 E 0	1 -6.	0.	-0.465916 01	-0.216994 01	4.80
-0.41978E 02 -0.39523E 02	0.12512E 0 0.14125È 0	2 -0. 2 -0.	0. 0.	0.91316E 01 0.70257E 01	-0.12607E 02 -0.15706F 02	1.96
-0.22951E 02	0.023216 0	1 -0.	0.	-0.26079E 02	0.264616 01	5.15
-0.19252E 07	0.13037E 0 0.81940F 0	1 -0.	0.	-0.28414E 02	0.11057E 02	2.20
-0.14951E 02	0.89525 0	1 -0. 1 -0.	0. 3.	-0.71873E 02 -0.15898 02	0.13962E 02 0.14526E 02	5.15 5.20 5.70 5.85
-0.17951E 02 -0.16827E 02	0.91622E 0	1 -0.	0.	-0.12130E 02	0.133156 02	5.85
-0.1528FE 02	0.89938E 0	1 -0. 1 -0.	0. -0.	-0.83649E 01 -0.290628 C1	0.12703E 02 0.11940F 02	6.00 6.03 6.06 6.08
-0.14960F 02	0.841726 0	1 -0.	-0.	-0.74866 01	0.11440E 02	4.04
-0.1504Pf 02	0.960786 0	1 -0.	-0.	-0.285286 01	0.12310E 02	6.08
-0.16494E 02 -0.145731 02	0.12232E 0	2 -0. 2 -0.	0. 0.	-0.10155F 02 -0.44784E 01	0.14493E 02 2.14009E 02	6.40
-0.12478F 02	0.14763E 0	2 -0.	0.	-0.342274 01	0.118805 02	6.60
-0. #5533F 01	0.14763E 0 0.15110E 0	2 -0.	0.	0.514741 00	0.46840E 01	7.00
-0.50057f 01 0.57795F 00	0.14330E 0	? -0. ? -0.	0. 0.	0.27433E 01 0.54743F 01	0.57489E 01 0.13748E 01	7.40
0.609695 01	0.66756 E	1 0.	0.	0.58473E 0i	-0.16812E 01	9.00
0.953286 01	0.13647E 0	1 0.	0.	0.359698-01	-0.28213E 01	10.00
0.106451 0? 0.41100E 01	-0.33034E 0	1 0.	-0.	0.963562 00 -0.15639E 01	-0.19112F 01 0.17937E-70	11.00 12.00
0.41100E (1	-0.78380E 0	ù 0.	-0,	-0.25877E 01	0.24586E 01	13.60
0.72120F 01	-0.73844 F G	1 0.	-0.	-0.20532F 01	0.387756 01	14.00
-0.1405PE 01 -0.40910E 01	-0.54432E 0	1 -0.	-0.	-0,10034E-00 0.25576E 01	0.38055E 01 0.21584E 01	15.00 16.00
-0.560701 01	- 0. 174 01 E-0	0 -0.	-0.	0.42476E 01 0.42427F 01	-0.74110E 00	16.00 17.00
-0.570216 61	Q.20822E 0	1 -0.	0.	0.582936 01	-0.39289E 01	18.00
-0.48292F 01	0.35429E 0	1 -0.	0. 0.	0.50812F 01	-0.64766E 01	19.00
-0.32850F 01	0.1	-0.	٠.	0.26362E 01	0.	20,00

(PSI/FPS SINUSOIDAL GUST)

BODY BALANCE STATION: 540 SEGMENT NUMBER 17

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

297, 900 LB CUTOFF FREQUENCY: 20 CPS 24,000 FT

0.85

INCREMENTAL RE61	L SEPAR STRESS 1 PAG 1 4 ARY					
-0.41556E-01	-0.114916 01	0.	-0.		-0.	670 CAF
-0.12179E 01 -0.18598E 01	-C.77898E 00 0.91125E-01	0. 0.	-0. -0-	0. 0.	-0. -0.	6.10 6.73 6.43 6.45 6.49 6.49 6.40 6.40 6.40 6.40 6.40 6.40 6.40 6.40
-0.199CEE 01	0.504CBE 00	-0.	-0.	-0.	-0.	i i
-0.1737eE 01 -0.13531E 01	0.74625E 00 0.80553E 00	-0. -0.	-0. -0.	-0. -6.	~0. -0.	0.55
-0.111746 01	C.82421 E 00	-0.	-0.	-0.	-0.	2.79
-0.98007E 00 -0.90381E 00	0.84129E 00 0.87597E 00	-0. -0-	-0. -0.	-0. -0.	-0. -0.	0.80
-C. 97077E OC	G.16753E 01	-0.	-0.	-0.	-0.	1.00
-0.90624E 00 -0.5681C5 00	0.15150E 01 0.19134E 01	-0. v.	-a. -o.	-0. 0.	-0. -0.	1.34
-0.9236°E 00	0.24331E 01	0.	-0.	0.	-0.	1.75 1.49 1.49
-0.69876E CO -0.42273E-GO	G.37529E 01 0.30482E 01	-0. -0.	-0. -0.	-0. -0.	-0. -0.	1.49
0.49574E-01	C.29039E 01	-0.	-0.	-0.	-0.	1.47 1.59
0.15715E 01 0.23187E 01	0.;3955E 01 0.31194E-00	-c. -o.	-0. -0.	-0. -0.	-0. -0.	1.90 1.55
0.186535 01	-0.14333 E-00	-ə.	ŏ.	-0.	0.	1.60
0.45002E-00 0.75392E-01	-9.650596-01	-0. -0.	o. o.	-0. -0.	0.	1.80
-0.17G7#E-00	0.30583E-01 0.13103E-00	-0.	ö.	-0.	0.	1.90 2.60
-0.36177E-0C	C-25075 E-00	-0.	-c.	-0.	-0.	2.10
-0.53919E CO -0.70218E 00	0.42146E-00 0.56544E 00	0. 0-	-0. -0.	0. 0.	-0. -0.	2.30
-0.73011F 00	C_70344E 00	0.	-3.	0.	-0.	2.30
-0.735C6E 00 -0.66940E 00	0.79091 6 00	0. 0.	-0. -0.	0. 0.	-0. -0.	20
-0.61489E 00	0.73339 € 00	0.	-0.	0.	-0.	2.43 2.44
-0.42446E-00 -0.4;417E-00	0.424706-00	0. 0.	-0. -0.	0. 0.	-0. -0.	2.47
-0.7870PC 00	0.234936-00	ŏ.	-0.	ŏ.	-0.	2.90 2.94 2.98 2.69
-0.11547f 01	0.45288E-00	0.	-0. -0.	0.	-0.	2.56
-0.174696 C1 -0.232106 C1	0.770CZE 00 0.28963E 01	0. 0.	-0.	0. 0.	-0. -0.	2.65
-0.31240t 01	G.46034E 03	9.	-0.	0-	-0.	2.70 2.80
-0.87678E 00 -0.4396CE-00	0.57242 E 01 0.79344 E 01	0. 0.	-0. -0.	0.	-0. -0.	3.00 3.16
0.778918 00	0.100058 02	-0.	-0.	-0.	-0.	5.20
0.27924E 01 0.49545E 0:	0.11310E 02 0.12024E 02	-0. -0.	-0. -0.	-0.	-0.	3,36
0.1211#6 07	0.39513E 01	-0.	-0.	-0. -0.	-0. -0.	3.2) 3.7)
0.20749E 02	-0.91543 E 01	-0.	0.	-0.	0.	3.35 3.40
0.21839E 01 -0.55691E 00	-0.42554E 01 -0.12374E-00	-0. 0.	0. -0.	-0. 0.	0. -0.	2.72 2.76 2.70 2.70 2.70
0.95789E OG	-0.18467E 01	-0.	0.	-0.	0.	3.6
0.52653E 01 0.31931E 01	-0.36728E 01 -0.34853E C1	+0. -0.	0. 0.	-0. -0.	0.	7.70
0.17466F 01	-0.259878 31	-0.	0.	-0.	0.	4.00
0.778U9E 00 0.7285PE-01	-0.24141E 01 -0.215G6E 01	-0. -0.	o. o.	-0. -0.	0. 9.	4.30
-0.13809E-00	-0.20910 E O!	-0.	0.	-0.	0.	6.70 6.70 6.60
-0.16757F-00 -0.19904E-00	-0.23517E 01 -0.25313E 01	-0. -0.	0. 0.	-0. -0.	0. 0.	\$. 6
-0.340721-00	-:.238C3E 01	-0.	ö.	+0.	0.	3.00
~0.131326 01 0.139166 01	-0.18525E 01 -0.15331E 01	0. 0.	0. 0.	o. o.	o. o.	3.23
-0.18658E 01	-0.12681E 01	ŏ.	0.	0.	0.	5.30 8.50
-0.74453E, C1 -0.33136E 01	-0.87254E 00 0.10737E 01	0. 0.	0. -0.	0.	0. -0.	3.50 3.70
-0.514476 01	0.197156 01	ŏ.	-0.	ŏ.	-0.	5.85 6.00
-0.542736 01	0.31694 € 01	0.	-0.	0.	-0.	6.03
-0.54653E 01 -0.50013E 01	0.45#77E 01 0.496ClE 01	o. o.	-0. -0.	0. 0.	-0. -0.	6.05 6.08
0.142776 01	0.91832 8 00	-0.	-0.	-0.	-c.	6,20
0.15643E U1 0.84113E 00	0.14376E-00 -0.12415-00	-0. -0.	-0. -0.	-0. -0.	-0. -0.	1.40 4.60
0.257448-00	-0-12172 E-CO	-0.	-0.	-0.	-0.	7.00
0.618506-01	-0.54738E-01 -0.37764E-01	-0. -0.	-0. -0.	-0. -0.	-0. -0.	7.40
0.123246-03	-0.627698-01	-0.	-0.	-0.	-6.	9.00
0. ;7713f-01 0. 29369[-02	-0.96605E-01 -0.108C7E-00	-0. -0.	-0.	-0. -0.	-0. 0.	30.00
-7.392616-01	-0.85250F-01	-0.	0.	-0.	0.	11.00 12.00
-0.74680F-01 -0.88868E-01	-0.48454E-01 -0.17083E-61	-0. -0.	-0. -0.	-0. -e,	-0. -0.	13.00
-0.759236-01	-0.72081 E-02	-0.	-0.	-0.	-0.	14.00 15.00
-0.50305E-01 -0.34450E-03	-0.19938E-01 -0.43249E-01	-0. -0.). 0.	+0. +0.	•. 0.	16.00
-0.386231-01	-0.54482 F-G1	-0.	0.	-0.	0.	17.00
-0.47129E-01 -0.10530E-00	-0.41030f-01	-0. -5.	0. 3.	-o. -o.	0.	19.00
-0.10330f-00	••	- % •	4 •	-0.	0.	30.00

Table XV --- Concluded

(PSI#PS SINUSOIDAL GUST)

297, 900 LB 24, 000 FT 0, 85 CUTOFF FREQUENCY:

20 CPS

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

SEGMENT NUMBER 1 BODY BALANCE STATION: 820

		DUMBETAL AXIAL STREES REAL 3 RAGINARY					
•.	o.	0.1407#E CI 0.11507e 03	o. o.	3.			
0. •.	0. 0.	0.91791E 02	o. o.	?. 0.			
9.	-0.	0.18558E 03 -0.27779E 02 0.17321E 03 -0.61520E 02 0.14671E 03 -0.77793E 02 0.12895E 03 -0.77793E 02 0.12895E 03 -0.77793E 02 0.11819E 03 -0.10295E 03 0.11146E 03 -0.13949E 03 0.10547E 03 -0.27277E 03 0.082206E 07 -0.34535E 07 0.59772E 02 -0.4952CE 03 -0.21784E 02 -0.49013E 03 -0.21784E 02 -0.49013E 03 -0.20281E 03 -0.39070E 03 -0.49591E 02 -0.50044E 03 -0.37841E 03 -0.11784E 03 -0.37841E 03 -0.11784E 03 -0.37841E 03 -0.11784E 03 -0.10447E 03 -0.2218E 02 -0.15921E 02 -0.2498E 02 -0.15921E 02 -0.2498E 02 -0.15921E 02 -0.2498E 02 -0.15921E 02 -0.2598E 02 -0.31401E 01 -0.33168E 02 -0.50472E 01 -0.33168E 02 -0.50472E 01 -0.33168E 02 -0.50472E 01 -0.3408E 02 -0.50472E 02 -0.4928E 01 -0.57845E 02 -0.4920E 02 -0.31440E 02 -0.46208E 02 -0.31440E 02 -0.46208E 02 -0.31440E 02 -0.46208E 02 -0.21212E 02 -0.4928E 01 -0.27224E 02 -0.47929E 01 -0.27224E 02 -0.47929E 01 -0.8047E 02 -0.33512E 03 -0.41580E 03 -0.33512E 03 -0.50971E 02 -0.33512E 03 -0.50971E 02 -0.33502E 03 -0.41580E 03 -0.49309E 02 -0.41580E 03 -0.49309E 02	ŏ.	-0.			
٥.	-0.	0.17321E 03 -0.61520E 02	0.	-o.			
o. o.	-0. -0.	0.14471E 03 -0.77793E 02 0.12895E 03 -0.90099E 02	0.	-0. -0.			
o.	-0.	0.110196 03 -0.102956 03	o.	-c.			
0-	-0.	0.111946 03 -0.139496 03	0.	-0.			
o. o.	-0. -0.	0.108e1E 03 -0.177@4E 03 0.10547E 03 -0.27279E 03	0.	-0. -0.			
ö.	-0.	0.882046 07 -0.348356 09	0.	-0.			
0.	-o.	0.53972E 02 -0.4352EE 03	0.	-0.			
-0. -c.	-0. -0.	-0.21786E 02 -0.45013E 03 -0.94391E 02 -0.50846E 03	-0.	-0. -0.			
-0.	-0.	-0.20201E C3 -0.39070E 03	-0.	-0.			
-0.	-0.	C.40505E 03 -0.40810E 02	-0.	-0.			
- 0.	o. o.	-0.53849E 03 0.11326E 03 -0.37861E 03 0.11786E 03	-0.	0. 0.			
-0.		-0.104426 03 0.021156 02	-0.	0.			
-0.	0. 0. 0. -0.	-0.46515E 02 0.52419E 02	-0.	€.			
-0. 0.	٥.	-0.15721E 02	-0.	0. 0.			
ě.	-0.	0.11133E 02 -0.25290E 07	ů. 0.	-0.			
o.	-0.	0.41083E 01 -0.33168E 02	0.	-0.			
¢.	-0.	0.33616E 01 -0.51983E 02	0.	-0.			
-0. -0.	-0. -0.	-0.542492 01	-0.	-0. -0.			
-0.	-0-	-0.31440E 02 -0.46208E 02	-0.	-c.			
-0.	⊸ç.	-0.58657E 02 -0.98328E 01	-0.	-0.			
-0. -0.	0. -0.	-0.57843E 02	-0.	0. -0.			
-0.	-0.	-0.280CSE 01 -0.47191E 02	-0.	-0.			
-0.	-0.	-0.47438E 01 -0.63847E 07	~0.	-0.			
-0. -0.	-0. -0.	-0.27224E 02 -0.47929E 01 -0.80642E 02 -0.46099E 02	-0.	-0. -0.			
ŏ.	-0.	0.59613E 02 -0.12611E 09	0.	-0.			
0.	-0.	0.40151E 02 -0.23512E 03	0.	-0.			
0. -0.	-0. -0.	0.19306E 02 -0.32561E 03 -0.56971E 02 -0.38009E 03	-0.	-0. -0.			
-0.	-0.	-0.14021E 03 -0.43035E 03	-0.	-0.			
-0.	-0.	-0.41580E 03 -0.9540BE 02 -0.73783E 03 0.34171E 03	-0.	-0.			
-0.	0.	-0.73783E 03 0.34171E 03	-0.	o. o.			
o. o.	0. -0.	0.18019E 02	0.	-0.			
o.	o.	0.13207E 01 0.50239E 02	0.	0.			
-9.	0.	-0.22210E 03	-0.	o. o.			
-0. -0	o. o.	-0.54874E 02 0.13011E 03	-0.	0.			
-ō.	0.	-0.10158E 02 0.10090E 03	-0.	0.			
٥.	0.	0.24833E 02	٥.	0. 0.			
o. o.	0. 0.	0.57274E 02 0.44714E 02	o.	0.			
o.	0.	0.67968E 02 0.31246E 02	0.	0.			
0.	0.	0.65655E 02	٥.	o. o.			
0. 0.	0. 0.	0.30471E 02 0.34785E 02	0.	0.			
ō.	0.	0.40272E 02 0.29496E 02	0.	0.			
٥.	0.	-0.12238E 03	0. -0. -0. -0. 0. 0. -0. -0. -0. 0. 0. 0. 0. 0.	0. -0.			
o. o.	-0. -0.	0.98498E 02 -0.27972E 02	ö.	0.			
ō.	-0.	0.10297E 03 -0.477862 02	ð.	-0.			
0.	-0. -0.	0.10341E 03 -0.71080E 02 0.95663E 02 -0.77724E 02	o.	-0. -0.			
o. -o.	-0.	-0.86355E 01 -0.14992E 02	-0.	-0.			
-0.	-0.	-0.86335E 01 -0.14992E 02 -0.10725E 02 -0.51051E 01	-0.	-0.			
0.	-0.	0.12145F-01 -0.47990E 01 0.43149E 01 -0.72223E 01	o. c.	-0. -0.			
0.	-0. -0.	0.50231F 01 -0.96245E 01	o.	-0.			
ŏ.	-0.	0.457888-00 -0.853788 01	0.	-0.			
-n.	-0.	-0.50716E 01 -0.43969E 01	-0.	-0.			
-0. -0.	-0. 0.	-0.89844E 01 -0.12324E-00 -0.94285E 01 0.26358E 01	-3. -0.	-0. 0.			
-0.	0.	-0.74864E 01 0.36723E 01	-0.	0.			
-0.	0.	-0.5768VE 01 0.36388E 01	-0.	0.			
-0. -0.	o. o.	-0.41276E 01	-0. -0.	0.			
-0. -0.	0.	-0.28365C 01 0.17949E 01	-0.	ŏ.			
-0.	0.	-0.25374E 01 0.13026E 03	-0.	0,			
-9.	0. -0.	-0.71031F 01 0.35563E-00 -0.17228E 01 -0.10769E 01	-0. -0.	0. -0.			
-0. -0.	-0. 0.	-0.100335 01 0.107445 01	-0. -0.	0.			
••				-			

Table XVI Stress Frequency Response Functions (Analysis Condition 2)

(PSIFFPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

268, 000 LB 24,000 FT 0.85

CUTOFF FREQUENCY:

10 CPS

PERCENT SEMISPANE 27

SEGMENT NUMBER 10

DICTIONSTA	L SEEAR 577988	DICHERTAL	AXIAL STREET			
*EM	1 MAG 1 MAR 7	atal	1RAG1MARY			Paramet
1.304477-00	0.333276 02	0.610316 00	0.290298 03	•.	-0.	G78 6.39 6.39 6.36 6.46 6.39 6.60
0.33941E 02	0.21879E 02	0.28946E 03	0.190226 03	-•.	-0.	4.30
0.49502E 02 0.53714F 02	0.23367E 01 -0.76171E 01	0.42891E 03 0.46704E 03	0.25025F 02 -0.45014F 02	-0. -0.	-0.	6.5
0.506256 02	-0.16911E 02	0.436796 03	-0.14570E 03	-0.	6. 6. 6.	12
0.45020E 02	-0.22383E 02	0.36534E 03	-0.19233E 03	-0.	0.	0.60
0.41137F 02	-0.269676 02	0.34486E 03	-0.22029E 03	- •.	9.	0.70 0.8p
0.38690F 02 0.37171E 02	-0.31709E 02 -0.37361E 02	0.31485E 03 0.29137E 03	-9.26202E 03 -0.29845E 03	-0. -0.	e. e.	
0.34289F 02	-0.53894E 02	0.271326 03	-0.39104E 03	-0.	0.	0.99 1.00
0.355538 02	-0.74831E 02	0.234336 03	-0.49344E 03	0.	-0.	1.20
0.34211F 02	-0.00353E 02	0.196406 03	-0.55562E 0B	••	-6.	1.35 1.40
0.32314E 02 0.29130E 02	-0.10304E 03 -0.11197E 03	0.17419E 03 0.14477E 03	-0.42029E 03 -0.43043E 03	• •	- 4. - 6.	1.40
0.266028 02	-0.12219E 03	0.12495E 03	-0.70130E 03	•.	-0.	1.49
0.23094E 02	-0.14740E 03	0.100126 03	-0.00339E 03	●.	- O.	1.9
0.11364F 02	-0.100196 03	0.274346 02	-0.12044E 03	••	-0.	1.55 1.66 1.69
-0.11964F 02 -0.59543E 02	-0.21054E 03 -0.33323E 02	-0.99:13E 02 -0.33015E 03	-0.10307E 04 -0.11037E 03	-0. -3.	-0.	1.60
-0.339276 03	0.130736 03	-0.137196 04	0.450376 03	-0.	6.	1.60
-0.20047E 03	0.133846 03	-0.69391E @3	0.373786 03	-0.	0.	1.90
-0.10276F 03	0.11989E C3	-0.304306 03	0.26271E 05	-•.	4.	2.40
-0.57829£ 02 -0.32122F 02	0.10923E 03 0.99165E 02	-0.15237E 03 -0.84126E 02	0.1742 6E	-0. -0.	B.	2.30
-0.10603F 02	0.90912E 02	-0.64087E 02	0.011736 02	-0.	0. 0. 6. 0. 0.	2.39
-0.66524F 01	0. F801 6E 02	-0.300296 02	0.500426 02	•.	٩.	2.35
-0.45882F OT	0-86493E 02	-0.61991E 02 -0.74849E d2	C.33374E 02	••	•	2.40
-0.34658E 01	0.03913F 02 0.054E3E 02	-0.03481E 02	0.28717E 02 0.29683E 02	•. 0.	0. 0. 0. 0.	*3
-0.42973F 01	0.00673E 02	-0.11255E 0#	0.59433E 02	•.	0.	2.47
-0.56243E 01	0.93866E 02	-0.13199E 03	0.97700E 02	0.	0.	2.30
-0.14194E 01 0.59393E 01	0.99622E 02 0.10232E 03	-0.10600E 03 -0.70676E 02	0.95698E 02 0.47747E 02	o. o.	٥.	4.2
0.179106 02	0.10232E 33	-0.437916 02	0.488976 02	0.	6. 6.	2.36 2.37 2.45 2.45 2.46 2.99 2.39 2.56 2.60 2.10
0.26114F 1/2	0-10472E 03	-0.30264E 02	0.20273E 02	ŏ.	0.	2.70
0.4360 PF 02	0.12633E 03	-0.40236E 02	-0.18527E 02	0.	6. 6.	2.80
0.102b2E 03 0.16432E 03	0.13734E 03 0.11642E 03	-0.71193E 02 -0.10347E 03	-0.29113E 02 -0.194378 #2	6. 0.	8.	3.00 5.30 3.80 3.85
6.20027F 03	0.222096 02	-0.16171E 03	0.27554E 02	:		35
0.40981E 03	-0.10634E 03	-0.21267F 33	e.00333E 02	••	-0.	3.36
0.439256 03	-0.36059£ 03	-0.22818E 03	0.19966E 03	••	-0.	3.80
9.30453E-03 0.99036F 02	-0.30336E 03 -0.31219E 03	-0.13726E 63 -0.19672E 02	0.15690E 03 0.27800E 02	•. 0.	-4. -0.	7.5
0.375916 02	-0.24040E 03	-0.29131E 02	0.154026 02	o.	-0.	Ç.
-0.436596 02	-0.24686E 03	-0.45690E 0Z	0.70146E 01	-0.	-0.	3.5
-0.89493E 02 -0.12734E 03	-0.15219E 03 -0.69139E 02	-0.41206E 02 -0.10414E 03	-0.18427E 01 0.22626E 02	-0. -0.	-0. -8.	3.60
-0.13169E 03	0. 7029 OE 00	-0.103416 03	0.14940E 03	-0.	-0.	£2
-0.12068E 03	0.53387E 02	-0.27300E 63	0.444826 @3	-0.	-0.	€.
-0.31757E 02	- 0. 8266 BE 01	-0.49196E 02	0.157926 03	-0.	-0.	4.20
-n.52292F 01 -n.15472F 02	-0.17621E 02 -0.11747E 02	0.0034E 03 9.17910E 03	0.63275E 02 0.36703E 02	-0. -0.	-0. -0.	1.00 1.20 1.25 1.00 1.20 1.20 1.30 1.30 1.30 1.30 1.30 1.30 1.30 1.3
-0.18705F 02	-0.90481E 01		0.721926 01	-0.	-0,	
-0-22215F 02	- 0. #210 9E 01	0.143426 03	0.11231E 01	-0.	-0.	4.96
-0.27900F 02 -0.24849F 02	-0.50000E 01 -0.11934E 01	0.13877E 03	-0.17231E 02	-0.	-0.	5.00
-0.26447F 02	0.31097E 01	0.12434E 03 0.11177E 03	-0.3362 02 -0.34695 02	-0. -0.	6.	5-75
-0.283508 02	0.16764E 02	0.97817E 02	-0.04253E J2	-0.	0. 6.	
-0.296726 03	0.34278E 02	0.014376 02	-0.11731E 03	-0.	0.	5.70 3.80 6.80
-0.22481E 02	0.31057E 02 0.24143E 02	0.50620E 02 -0.23361E 02	-0.99944E 02 -0.00007E 02	-0.	0. 0.	3.40
0.176106 02	0.174326 02	-0.255012 02	-0.70143E 02	•: •:	ö.	6.49
0.140558 07	0.122225 02	-0.26155E 02	-0.71159E 02	0.	0.	6.45 6.46
0.169036 02	0.84014E 00	-0.21663E 02	-0.660466 02	••	0.	5.44
0.69738F 01 -0.31961F 01	-0.31186E-00 0.33698E 01	-0.14597E 02 -0.47786E 02	-0.79876E 03 -0.221482 61	0 €	0.	6.30
-0.62815F 01	0.306336 01	-0.103776 03	0.321766 02	- C.		5.06 6.20 6.40 6.60
-0.54870E 01	0. 63166E 01	-0.323476 01	0.118666 02	-0.	- 2.	7.4
-P.48608E 07	0.777096 01	0.06016 01	-0.10499E 01	-0.	0.	7.40 7.40
-0.20678E 01 0.98984C 00	0.74488E 01 8.36939E 01	0.02708E 01 0.40337E 01	-0.39169E 01	-0. -0.	:	1.20
0.307716 01	0.	0.16409E 01	0.	•	0,	9.60

Table X ✓ I --- Continued

(PSI#PS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

268, 000 18 24, 000 FT 0. 55

CUTOFF FREQUENCY:

10 CPS

PERCENT STAILSPANE 27

SEGMENT NUMBER 14

ERCHINE ZAL	SEEME STREET	THE REPORT AL	AXIAL STRESS			
DEAL	184639487	0.354826 00 0.262478 03 0.394726 03 0.394726 03 0.39476 03 0.39476 03 0.312956 03 0.262626 03 0.24626 03 0.24626 03 0.150076 03 0.150076 03 0.150076 03 0.150076 03 0.13106 03 0.150076 03 0.13106 03 0.249956 02 0.249956 02 0.249956 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.13106 03 0.141376 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.141476 03 0.1414776 03 0.141476 03	1 MAG : MARY			
0.57676E 00	0.10581E 03	0.354826 00	0.27049E 03	0. - 8 -	-0. -0.	
0.154075 03	0. #3650€ 01	0.30722E 03	0.23436E 62	-0.	-0.	
0.144036 03	-C.23503E 02	0.423826 03	-0.59000E 02	-0.	0. 4.	
0.137792 03	-0.477346 02	0.349696 03	-0.174546 6)	-0.	ö.	
0.12407F 03	-0.804216 02	0.312956 03	-0.20717E 03	-0.	6.	
0.10732E 03	-0.106438 03	0.264626 03	-0.270648 03	-0.	6.	
0.101#2E 03	-0.143" OF 03	0.24621E 03	-0.35484E 03	-5.	0.	
G. 84241E 02	-0.21501E 03	0.2:245E 03	-0.50421E 03	: :	-0.	
0.777946 02	-J. 24434E 03	0.150076 03	-0.54290E 03	••	-0.	
G.68136E 02	-0.26173E 03 -6.28189F 03	0.131306 03	-0.59733E 63	0.	-0.	
0.52734E 02	-0.330466 03	0.90568 02	-0.72905E 03	ŏ.	-0.	
0.25431F 02	-0.392206 03	0.24095E 02	-0.04234E 03	0.	-0.	
-0.12304E 03	-0.73232E 02	*0.29940E C3	-0.10017E 03	-0.	-0.	
-0.44330E 03	0.221472 03	-0-12450E 04	0.40076E 03	-0.	٥.	
.0.34701E 04	0.210.86 03	-0.42971E 03 -0.27794E 03	0.337178 03	-0. -0.	9. 0.	
-0.92510E C7	0.14231F 03	-0.138446 03	0.150148 03	-0.	0.	
-0.508+5E 02	0.11571E 03	-0.743426 02	0.947186 02	-0.	0.	
0.214446 02	0. 727116 02	-0.52384E G2	0.454126 02	ŏ.	č.	
-n.198836 02	0. 84801F Q2	-0.54256E 0Z	0.304686 02	0.	٥.	
-0.25793E C7	0.855206 02	-0.400196 32	0.249398 02	0.	ŏ.	
-n.35505E 02	0.98948E 02	-0.10214E 03	0.337366 02	0.	٥.	
-0.432836 07 -0.311386 02	0.11985E 03	-0.11978E 03	0.88948E 02	0. 0.	0. A.	
-0.11847E 02	0. 1231 EF 03	-0.44137E 0Z	0-614796 02	o.	0.	
0.77748E 01	0-11727E 03	-0.397416 02	0.443736 02	0.	o.	
0.410736 02	0.12324E 03	-0.365136 02	-0.16013E 02	ŏ.	ŏ.	
00184E 03	0.12443E 03	-0.44524 6 02	-0.24419E 02	3.	٥.	
0.294476 03	-0.127296 02	-0.14674E 03	0.250056 02	i:	6.	
0.422246 03	-0.15037E 03	-0-192996 03	0.061628 02	ō.	-0.	
0.471802 03	-0.44684E 03	-0.20707E 03	0.10119E 03	:	-6. -8.	
0.326066 02	-0.31484E 03	-0.18034E 02	0.252338 02	i.	- 0.	
-0.54752E 07	-0.28491E G3	-0.74436E 02	0.139776 02	0.	-0.	
-0.1548GE 03	-0-15054E 03	-0.55416E 32	-0.16722E 01	-0.	-0.	
-0.195506 03	-0.54974E 02	-0.94508E GZ	0.20533E 02	-0.	-0.	
-0.21A99E 03	9-23035E 03	-0.24047E 0)	0.403446 03	-0.	-0.	
-0.57753E 02	0.576548 02	-0.4444E 02	0.143316 03	-0.	-0.	
0.553426 02	0.17857E 02	0.142536 03	0.37421E 02	~0. ~0.	-0.	
0.47773E 07	-0.44 044E 01	0.140016 03	0.633136 01	-0.	- 0.	
0.184498 02	- 0. 44014E 01	0.130156 03	0.102106 01	-9.	-0.	
0.31333E 02	-0.1784RE 02	0.112046 03	-0.30317E 02	-0.	o.	
0.270166 02	-0.25741E 02	0.101436 03	-0.49633E 02	-0.	0.	
0.16 TORE 02	-0.519746 02	0.739028 02	-0.10443F 03	-0. -0.	0. 0.	
0.27470F 01	-0.42A72E 02	0.45934E 02	-0.90697E 02	-0	ŏ.	
-0.271116 0; -0.278316 02	- 0.30147E 07	-0.21200E 02 -0.24149F 02	-0.79864E 02	ð. 0-	0.	
-0.27106F 02	-3.32350E 02	-0.237356 02	-0.443736 02	o.	ŏ.	
-0.25730E 02	-0.34407E 02	-0.21474E 0Z	-0.39933 € 02	٥.	0.	
-7.497196 02	0. 1 2 04 7 E 02	-0.433446 02	-0.20098E 01	0.	-0.	
-0.92591E 02	0.411226 02	-0.94171E 02	0.291998 02	0.	-0.	
0.61434 00	0.139236 02	-0.27534E 01 0.87899E 21	-0.970000 02	-0. -0.	-0.	
0.48715E 01	0.74401E 01	0.750456 01	-0.35345E 01	-0.	ō.	
0.790275 01	0. 27042E 0t	0.43845E 01	-0.34302E 01	-0. 8.	0. 0.	
					-	

(PSIFPS SINUSOIDAL GUST)

268,000 LB CUTOFF FREQUENCY: 24,000 FT 0. 8:-GROSS WEIGHT: ALTITUDE: MACH NUMBER: 10 CPS

PERCENT SEMITIPAN: 40.06 SEGMENT NUMBER 8

THE RESERVE	AL SHEAR STREES	INCREMENTAL	LAKIAL STREET			
REAL	IMAG I MAGY	REAL	IMAGINARY			/memor
0.25514E-00	0.6667 06 02	-0.944345-01		_	_	77
0.67833E 02	0.455326 02	-0.24416E-01 0.26127E 03	0.27155E 03 0.10101E 63	• • • • • • • • • • • • • • • • • • •	0. 0.	4.30 4.30
0.79953F 07 0.10077F 03	0.58171E 01 -0.15047E 02	0.38703E 03 0.42488E 03	4.23330E 02 -0.60332E 02	9. 0.	0.	6.30 6.36 6.44
0.10256E 03	-0.33733E 02	0.399716 03	-0.13492E 03	•.	-0. -0.	0.55
0.90973F 02 0-92594E 02	-0.44660E 02 -0.53954E 02	0.35073E 03 0.31326E 03	-0.17844E 03 -0.21214E 03	0. •.	-0. -0.	6.35 6.66 6.70
0.761,93F 07	-0.63673E 02	0.26500E 03	-0.24366E 03	••	- 0.	0.00
0.72918F 02 0.69965E 02	-0.73401F 02 -0.10300F 03	0.26243E 03 0.24250E 03	-0.27747E 03 -0.36223E 03	•	-0. -0.	0.50 0.50 1.60 1.30 1.34
0.65407F 0;	-0.13663E 03	0-20463E 03	-0.455038 03		-0.	1.30
0.59659F 02 0.54661F 02	-0.16104E C3 -0.18510F 03	0.16740E 03 0.14283E 03	-0.31033E 03 -0.56791E 03	•• •• ••	-0. -0.	3.25
0.47754E 02	-0.19955t 03	0.113566 03	-0.60174E 03	0.	-0.	1.49
0.425067 02	-0.21598E 03 -0.25600E 03	0.94103E 02 0.69973E 02	-0.63960E 03 -0.72923E 03	:	-0. -0.	1.47 1.90
0.13627F 02	-0.30693F 03	0.397096-00	-0.83754E #3	••	-0.	1.95
-0.27767F 02	-0.36420E 03 -0.41617E 02	-0.31847E 03 -0.33154E 03	-0.94701E 03 -0.37372E 02	-0. - 0.	- 0. - 0.	1.60 1.63
-0.53774E 03 -0.24973E 03	0.207576 03	-0.12366E 94	0.422ZDE 03	-8.	5.	3.80
-0.14539E 03	0.19305E 03 0.16062E 03	-0.60791E 03 -0.26340E 03	0.32870E 83 0.23199E 83	-0. -0.	0. 0.	1.90 2.00
-0.79111E 07	0-13232E 03	-0.13106E 03	0.14432E 85	-0.	6. 6. 6.	2.10
-0.39047F 02	0.11204E 03 0.11552E 03	-0.76822E 02 -0.86203E 02	0.87647E 82 0.927632 82	-1. -1.	0. 0.	2.30 2.30
-0.30166F 02 -0.16179E 02	0.11201E 03 0.10695E 03	-0.75352E 02 -0.43434E 02	0.67486E @2 8.33440E 02	-9.	0. 0.	2.35
-n.154316 02	0.1050 9F 03	-0.67709E 02	0.509748 02	-0. -0.	0.	2,40 2,43
-0.15197E 02	0.1074@E 03 0.10569E 03	-0.71029E 02 -0.67142E 02	0.47 999 E 82 0.62213E 82	••	•.	2.44
-0.16466E 07	0.11236E 03	-0.96842E 02	0.79548E 02	0. 0.	6. 6. 1.	2.47
-0.12326F 02 -0.26409F 01	0.11298E 03 0.10892E 03	-0.80160E 02 -0.59834E 02	0.73594E 02 0.51141E 02	o. o.	1. 0.	2.56
0.52304F 01	0.10409E 03	-0.46733E 02	0.36934E 02	0.	●.	2.95 2.94 2.98 2.65
0.25449F 02	0.10258E 03 0.10247E 03	-0.46202E 02 -0.33964E 02	0.16323E 02 -0.49960E 01	0. ●•.	0. 0.	2.70
0,45097F 02	0,10616E 03	-0.92988E 02	-0.226247 01	••	0.	2.70 2.80 3.00 3.86 3.86 3.70 3.70 3.70 3.70 3.70 3.70 3.70
0.54454F 02 0.6994;E 02	0.11452E 03 0.12727E 03	-0.13006E 03 -0.19510E 03	0.32477# 8 2 0.313 0 1# 8 3	•. •.	0. 0.	ž
D. 78267E 02	0.14287E 03	-0.24764E G3	0.205778 03	0.	0.	3.8
0.85546E 02 0.13359E 03	0.16599E 03 0.16429E 03	-0.23412E 03 -0.90160E 02	0.36471E 03 0.30493E #3	: :	0. 0.	3.29
0.23602F 03	-0.88743E 02	0.12737E 03	-0.59766E 04	0.	-0.	5.6
0.26845F 03 0.19899F 03	-0.10154F 03	0.13236E 03 0.34535E 02	-0.63403E 02 -0.63552E 02	0. 0.	-0. -0.	3.52
0.16050F 03	-0. 53533E 02	0.708148 00	-0.39541E 02	0.	-0.	3.60
0.14089F 03 0.17677E 03		-0.78477E 62 -0.15973E 03	-0.06023E 01 0.11720F 03	o. P.	-0. -0.	7.70°
0.22645F 03	-0.36710E 03	-0.227146 03	0.34043c 03	0.	-0.	\$.00
0.21434F 07 -0.16377F 03	-0.12274E 03 -0.51119E 02	-0.16088E 02 0.14932E 03	0.87302E 02 0.22172E 02	0. -0.	-8. -0.	4.30
-0.13721E 03	-G. 31646E 02	0.107225 03 0.882976 02	0-72409E 01	-0.	-8.	6.70 4.80
-0.17490F 03		0.621976 02	-0.52309E 81 -0.68999E 01	-0. -0.	-0. -0.	4.50 4.56
-0.10511F 03	0.65744E 01	0.56401E 02 0.37199E 02	-0.893775 01 -0.34680E 01	-0. -0.	0.	5.00
-0.940ADF 02	0.29945F 02	0.195368 02	0.86046E 01	-0.	0. 8.	5.15 5.30
-0.74715F 02 -0.64591F 02	0.4628BF 02 0.64364F 02	-0.74234€ 00 -0.13630E 02	0.43012E 02 0.10220E 03	-0. -0.	0. 0.	5.78
-0.48679F 07	0.58020E 02	0.41215E 01	0.063176 02	-0.	•.	1.65
-0.93065F 01 -0.70227F 01	0.57013F 02 0.4686E 02	0.11946E 03 0.12642E 03	0.66528E 02 0.66786E 02	-0.	e. 0.	§.00
-0.67294E 01	0.431096 02	0.125986 03	0.31176E 02	-0.	0.	6.05 6.06
-0.75866F 01 -0.11821F 02	0.39121F 02 0.44974E 02	0.12002E 03 0.09145E 02	0.637338 01 0.211698 02	-o. -o.	ę.	6.06 6.20
0.60610f 00	0.125566 02	0.910036 02	-0.64859E 02	0.	t	6.40
0.22794F 02 -0.99091F 01	O. 66152E 01	0.14063E 03 0.23288E 02	-0.85430E 02 -0.58602E 02	o. -o.	0. 8.	i.
-0.10278F 07	0.1901 OE 02	-0.66789E 01	-0.20210E 02	-0.	٥.	7.00 7.40
-0.19616F 01 0.60764F 01	0.16923E 02 0.10529E 82	-0.23609E 02 -0.24913E 02	-0.79133E 01 0.63937E 01	-1. 0.	0. 5.	8.20 9.60,
0.12184E C2	0.	-0.17164E 02	0.	ě,	6.	30,66

(PSIFPS SINISOIDAL GUST)

240,000 LB CUTOFF FREQUENCY: 24,000 FT 0.85

10 CPS

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

PERCENT SEMI SPAN: 40.06 SEGMENT NUMBER 107

THE RESIDENCE PARTY.	METAL STREET			DECREGATAL AXIAL STREETS	
BFAL	TRACINARY			REAL IMAGINERY	
				THE CON.	
0.790116-00	0.675036 02	•.	•.	0.20062E-01 -0.232@2E-05 0.30	
0.44454E 02 0.10047E 02	0.44423E 62 6.57401E 81	•. •.	•.	-0.22324F 03 -0.13464E 03 0.30 -0.53244F 03 -0.20105E 02 0.30 -0.3640F 03 -0.20105E 02 0.40	
0.10659F 03	-0.145896 82	i.	-i:	-0.36/0)E 03 3,51294E 02 0.44	,
0.10040F 03	-0.32725F 02	9.	-0.	-0.5415 X 03 0.11520£ 03 0.50	1
0.80764E 02	-0.43449E 02 -0.52164E 62	0. •-	- ?. - 4.	-0.29966E 03 0.15246E 03 0.60 -0.29764E 03 0.10126E 03 0.70	
0.751866 07	-0.408606 02	•.	⊸.	-0.24352E 03 0.20019E 05 0.80	1
0.00453E 02	-0.799321 02 -0.99010E 02	ø. D.	-3:	-0.22421E 03 0.23707E 03 0.90 -0.20727E 03 0.30952E 03 1.00	
0.642118 02	-0.133086 03	ő.	-0.	-0.17484F 03 0.38879F 03 1.20	
0.491198 07	-0.15451E 03	0.	⊸.	-0.14303E 05 0.4360EE 05 1.3% -0.12204E 05 0.40524E 03 1.40	
0.544 <i>82</i> £ 02 0.47851E 02	-0.17753E 03 -0.19137E 03	0. 6.	-3:	-0.12204E 05	
0.427346 02	-0.20710E 03	0.	-5.	-0.404046 A7 0.544496 A3 1.47	1
0.36411E 02 0.15733E 02	-0.24544E 03 -0.29437E 03	6. 0.	-0. -0.	-0.39707E 02 0.62307E 05 1.39 -0.39707E-00 0.71564E 03 1.59 0.1012H 05 0.80919E 03 1.60	
-0.23448E 02	- 0. 34957E 03	-ŏ.	-0.	0.101236 05 0.809196 03 1.60	
-0.10010E 03	-0.44512E 02	-0.	-0.	0.203206 03 0.490206 02 1.65	
-0.51105E 03	0.1899/E 03 0.18164E 03	-0. -0.	o.	0.10564E 04 -0.36074E 03 1.80 0.51942E 03 -0.20939E 03 1.90	
-0.130016 03	0.15000F 03	⊸.	0.	0.225126 03 -0.198226 09 2.00	
-G.74397E 07	n.17375E 03	-0. -0.	0. 0.	0.11148E 03 -0.12519E 05 2.10 0.65639E 02 -0.74889E 02 2.30	
-0.34745F 02	7. 109206 07	-0.	ö.	0.73654E 02 -0.79259E 02 2.30	
-0.28450E 02	0. 1074 85 03	-o.	٥.	0.64307E 02 -0.59371E 02 2.35	
-0.17993# 02 -0.14190# 02	0.97350E 02 0.95520F 02	-0. -0.	0. 0.	0.54202E 02 -0.47369E 02 2.40 0.57653E 02 -0.43554E 02 2.45	
-7.16441E 07	0. 935006 02	0.	0.	0.613736 02 -0.410116 02 2.44	
-0.19613E 07	0.9632;E 02 0.1072 % 03	0. 0.	٥.	0.74457E 02 -0.53157E 02 2.57 0.82744E 02 -0.67948E 02 2.50	
-0.147348 07	0.100756 03	0.	o. o.	0.66491E 02 -0.62661E 02 2.54	
-0.47879E 01	0.103756 03	0.	0.	0.51141E C2 -0.45497E G2 2.58	
0.8484#F 01	0.10056E 03 0.96393E 02	0.	o. o.	0.39947E 02 -0.51550E 01 2.65 0.39476E 02 -0.13947E 02 2.70	
0.247668 02	0. 94570E 02	0.	0.	0.461006 02 0.426946 81 2.80	
0.433875 07 0.53849F C2	0. 16897F 82 0.10325F 03	0. 0.	6.	0.79452E 02 0.19330E 01 3.00 0.11113E 03 -0.27749E 02 3.30	
0.030134 02	0.1144 96 03	ŏ.	ŏ.	0.16670E 03 -0.96550E 02 3.30	
0.709268 02	0.12972E 83	0.	0.	0.16670E 03 -0.76550E 02 3.20 0.21170E 03 -3.17502E 05 3.26 0.21713E 05 -0.31500E 03 3.25 0.777030E 02 -0.26050E 03 3.25 -0.1000ME 03 0.51065E 02 3.40	
n.7n192E 02 0.11940E 03	0.17171E 03 0.17109E 03	0. 0.	0. 0.	0.21713E 05 -0.31589E 03 3-29 0.77034E 02 -0.24054E 03 3-25	
n.21455F 03	-0.74991E 02	6.	-0.	-0.1000 X 03 0.51065E 02 3.40	
0.74597€ 03	-0.76632F 02 -0.74771E 02	0. 0.	-0. -0.	-0.11309E 03 0.72971E 02 3.52 -0.46614E 02 0.71309E 02 3.55	
0.18470F 03 0.15119E 03	-0.47739F 02	ŏ.	⊸.	-0.404045 00 0.404745 02 %-60	
0.135446 01	-0.57072E 02	0.	⊸.	0.67053E 02 0.74186E 01 3-7-3	
0.17041F P3 0.21598F 03	-0.15274F 03 -0.34741F 03	o. o.	⊸. ⊸.	0.13648E 0? -0.10021E 03 3.85 0.19407E 03 -0.29105E 03 4.00	
0.14003F 02	-0.1174 NF 03	0.	⊸.	0.137446 02 -0.745936 02 4.30	
-0.15412E 03	-0.45029E 02 -0.2004E 02	-0. -0.	-0. - 4 .	-0.17750E 03 -0.18945E 02 4.50 -0.91615E 02 -0.61937E 01 4.70	
-0.11472E 03	-0.09511E 01	⊸.	-6:	-0.754436 02 0.446946 01 4.80	
-0.90513F 07	-0.51949F 01	-o.	-0 .	-0.53514E 02 0.50955E 01 4.96	
-0.94905F 02	0.55711E 01 0.14164F 02	-0. -0.	6. •.	-0.40191E 02 0.76366E 01 3.00 -0.31704E 02 0.46721E 01 5.15	
-0.747A3F 07	0.23403E 02	-0.	•.	-0.144925 02 -0.715205 01 3-30	
-0.84094F 02 -0.54495E 02	0.33505£ 02 0.43256£ 02	-0. -0.	0. ●.	0.63423E 00 -0.38663E 02 -0.67992E 62	i
-0.34478 07 -0.43541F 07	0.40046F 02	.0.	0.	-0.3521ME 01 -0.75461E 02 3.89	,
-0.21740F 07	0. 371426 02	-9.	٥.	-0.10207E 03 -0.56844E 02 \$.00	•
-0.20370F 02	0.34720F 02 0.32994E 02	-e. -0.	0. •.	-0.10001£ 03 -0.39976£ 02 \$.05 -0.10764£ 03 -0.26437£ 02 6.06	í
-0.20121E 02	0. 31441F 02	-0.		-0.1037 % 03 -0.54457E 01 6.00	1
-0.20703F 02	0.746166 02 0.20805E 02	-0. 0.	o. o.	-6.76160€ 02 -0.10105€ 02 6.20 -0.70500€ 02 0.30329€ 02 6.40	
0.64920E 00	0. 16407F 02	o.	0.	~0.120146 05 0.729946 02 6.60	•
-0.17101F 07	0.204736 -32	- e . -≎.	0.	-0.1989#F 02 0.50671F 42 7.40	•
-n.875335 01 0.51427F 00	0.1707: 62	-ō.	•:	0.57067E 03 0.24111E 02 7,40 0.20172E 02 0.67631E 01 8.80	,
9.F: 671F OI	0.977596 01	3.	•.	0.212000 02 -0.534050 01 9.00)
0.133275 02	0.	••	•.	0.144496 82 0. <u>10.90</u>	ı

(PSI/FPS SINUSOIDAL GUST)

GROSS WEIGHT: 268, COD LB CUTOFF FREQUENCY: 10 CPS ALTITUDE: 24, COD FT MACH NUMBER: 0.85

BODY BALANCE STATION: 540 SEGMENT NUMBER 17

THERMALITAT	SEEAR STRESS (MAS) 46RY					
-0.49923E-01	-0.137vac 01	•.	-0.	••	~ 0.	2000年,11、11、11、11、11、11、11、11、11、11、11、11、11、
-0.17275E 01 -0.22837F 01	-0.46 32 76 00	•.	-• .	₩••	-5. -1	•.5
-0.2251RE 01	0.14195E-00 6.55171E 00	ŏ.		• • • • • • • • • • • • • • • • • • •	-6.	0.44
-0.20058F 01	0.812406 08	0.	4. 4. 4. 4. 4. 4. 4. 4. 4. 4. 4. 4. 4. 4	••	-0.	0,50
-0.14459E 01	0.921396 00 0.990046 00	0.	•••	0. 0.	-0. -6	A.65
-0.13060F 01	0.10542F 01	•. 0.	₹:	: :	-6.	8,60
-0.12197F 01	0.1132if 01	•.	-0.	ě.	-0.	0.90
-0.11463F 01 -0.11312F 01	0.13785E 01 0.17035E 01	0. 0.	⊸.	0. 0.	- 0. -0.	1,40
-0.112996 01	0. 191746 01	ŏ.	-3 :	ě.	-0.	1.3
-0-11193F 01	0.21477E 01	0.	-o.	ę.	-0.	1.40
-0.10925E 01 -0.10475E 01	0.72072F 01 0.74469F 01	0. 0.	-0. -0.	•. •.	-0. -0.	1,49
-0.10301F 01	0.284126 01	0.	⊸.	0.	-0,	1.50
-0.89339F 00	0.33576F 01	0,	-0.	•	-0.	1.55
-0.59813E 00 0.37873E-01	0.39790E 01 0.16984F 01	-0. -0.	-9.	-0. -0.	- 0.	1.66
0.41681F 01	-0.446336 00	-0.	-ô. ô. ô.	-0.	0.	1.80
0.230396 01	-0.665736 00	-0.	Q.	-0.	•	1.90
0.48277E 00 0.705667~00	-0.33028E-00 0.48503E-01	-0. -0.	3. 0.	-0. -0.	9. 9.	2.30
-0.16677E-00	0.472006-00	-0.	-0.	-0.	- 0.	2.30
-0.2780AF-00	0.404526 00	-0.	-0.	-0.	-0-	2.30
-0.30556F-00 -0.29295F-00	0.85997F 00 0.10167F 01	-0. -0.	-0. -0. -0. -0.	-0. -0.	-0.	2.20
-0.17504F-00	0.10454F 01	~. -0.	-0.	-0.	-0.	2.43
-0.91#34F-01	0.105778 01	-0.	-o.	- <u>o</u> .	-0.	2.44
0.29746E-00 0.42741E-00	0.74317E 00 0.30614F-00	-0. -0.	-0. -0. -0.	-0. -0.	-0. -0.	2.47 2.49
0.136A0F-30	0. 31 66 0F - 00	-ŏ.	~ŏ.	-0.	-0.	2.5
-0.793761-00	0.87677E 00	-0.	-o.	-0.	-6.	2.56
-0.70151F 00 -0.85476E 00	0.975456 00	0. 0.	-0. -0.	0. 0.	-0. -0.	2.65
-0.10220F 01	0.331996 01	ŏ.	-4	ŏ.	-0.	ž. 6
-0.10703F 01	0.4746 of 01	٥.	-0. -0. -0. -0.	o.	-0.	3.00
-0.889599 OG -0.38419F-00	0.71972E 01 0.10033F 02	0. -0.		•. •0.	-0. -0.	3.30
0.477386-00	0.124146 02	~o.	- i :	-0.	-6.	3.8
0.18362E 01	0.174336 02	→.		-0.	-0.	3.29
0.91356E 01 0.21350E 02	0.15730E 02 -0.20200E 02	-1. -1.	- .	-0. -0.	-0. 0.	3.33
0.217506 02	-0.217946 02	-6.	٥.	-0.	0.	3.9
0.110ARE 07 0.43ER6F 01 -0.27374F 01 -0.64150F 01	-0.19559F CZ	-o.	e. e.	-0.	0. 0.	3.56
0.430366 01	-0.13334F 02 -0.76597F 01	-0. 0.	0. 0.	-0. 0.	0. 6.	3.60
-0.64150F 01	-0.19742F 01	ŏ.	ŏ.	.	ŏ.	3.85
-0.85201F 01	0.59906F 01	0.	-0.	•	-6.	1,00
-0.74609F 01 0.74986F 01	0.16795E-01 -0.13046E-01	0. -0.	-0. 0.	0. -0.	-n. 0.	1.39
0.14379F 01	-0.16489F 01	-0.	i.	-0.	ö.	1.7
0.94302F 00	-0.10127E 01	-0.	0.	-0.	6-	1.6
0.25993E-00 0.95385E-01	-0.10212E 01 -0.17674E 01	-e. -e.	e. 9.	-0. -0.	o. •-	1.76
0.467477-00	-0.157:0F 01	-	••	-0.	ŏ.	3.13
-0.10507F 01	-0.97463F 00	0.	•.	••	0.	فلاءؤ
-0.19087F 01 -0.78670F 01	0.80594E 00 0.42903E 01	0. •.	∹: →:	0. 0.	-0. -0.	(2
-0.70976F 01	0.384556 01	ŏ.	-0.	ő.	-0.	5.65
0.57327F 01	0.27594E 01	-9.	-	-0.	-0.	1.0
0.64181F 01 0.65675F 01	0.69949E 00 -0.66135E 06	-0. -0.	7.	- 0. -0.	-0. 0.	7:2
6.43150F 01	-0.40920E 01	⊸.	٥.	-0.	0.	6.06
0. 161675 01	-0.53717E G1	-0.	•.	-0.	0.	6.20
-0.13054F 01 -0.52076F 01	-0.49104F-00 0.13681F 01	•. P.	•• ••	0. 0.	e. - 0.	i.Z
-0.19495F-00	0. 46 39 TF - 00	-4.	-6.	-0.	-0.	7.00
0.246976-00	0.11715F-01	-0.	·••.	-0.	-0.	7.40
0.17297F-00 0.56220F-01	-0.85347E-01 -0.11990F-00	- 4 .	-0. -0.	-0. - 0.	-0. - 0.	1.0
0.97472F-01	0.	-0. -0. -0. -0.	ě.	-0.	ě.	10.00

Table XVI --- Concluded

(PSIFPS SINUSOIDAL GUST)

GROSS WEIGHT: 268,000 LB CUTOFF FREQUENCY: 10 CPS ALTITUDE: 24,000 FT MACH NUMBER: 0.85

BODY BALANCE STATION. 820 SEGMENT NUMBER 1

		THE REPORT AL	AXIAL STRESS			
		REAL	1 MAG I MARY			materi
	_			_	_	- 0.10
•. •.	o. o.	0.19861E 01 0.15570E 35	0.12202E 03 0.74433E 02	o. o.	o. o.	0.30
•.	0.	0.191196 05	0.12575E-00	ö.	0.	0. 50 0. 54 0. 44
ě.	- 0.	0.199996 03	-0.36505E 02	o.	-0.	0.44
0 -	-0.	0.184266 05	-0.66417E 02	0.	- 0.	0.50
	-0.	0.158746 03	-0.02477E 02	0.	-0.	0.60
•.	- O. - O.	C-14090E 05	-0.94470E 02 -0.10579E 03	0.	- 0. - 0.	0.70 b, 80
•:	-0-	0.12854E 03 0.11962E 03	-0.11033E 03	ö.	-0.	0.90
ě.	-0.	0.11:716 03	-0.15150E 03	0.	-0.	1.00
٥.	- 0.	0.101766 03	-0.15150E 03 -9.18952E 03	o.	-0.	1.20 1.34 1.40 1.45
•.	-0.	C. 91757E 02	-0.21247E 03	o.	-0.	1.54
•.	-0. -0.	9.84650E G2	-0.23452F 03	0.	-0.	1.40
•. •.	-0.	/ \32E 02	-0.25071: 03 -0.26666E 03	0. 0.	-0.	1.47
•.	-0.	0.614058 04	-0.30471E 03	ö.	-0.	1.90
ñ.	-0.	0. 37466 02	-0.35162E 03	0.	-0.	1.90 1.95 1.60
-0.	-0.	-0.53421E 01	-0.40167E 03	-0.	-0.	1.60
-0.	-0.	-0.05172E 02	-0.00783E 02	-0.	-0.	1.65
-0.	0. 9.	-0.46169E 63 -0.22474E 05	0.12920E 03 0.90614E 02	-0. -0.	0.	1.80
-0.	0.	-0.84471E 02	0.444746 02	-0.	0.	1.90 2.00
-0.	0.	-0.27704E 02	0.254946 31	-0.	0.	2.10
-0.	~O.	-0.25E55E G1	~2.34904E 02	-0.	-0.	2.20
-8.	-0.	-0.48540E 01	-0.43254E 02	-0.	-0.	2,30
-0.	-0.	-0.58407E 01 -0.68058E 01	-0.63865E 02 -0.75735E 02	-0.	-0.	2.30 2.35 2.40
-n. -o.	-0. -0.	-0.17599E 02	-0.79236E 02	-0. -0.	-0. -0.	2.40
-0.	- 0.	-0.245946 02	-0.77000E 02	-0.	-0.	2.45
-0.	-0.	-0.482286 02	-0.54000E 02	-0.	-0.	2.67
~0.	-0.	-0.64105E 02	-0.24223E 07 -0.27360E 02	-6.	-0.	2.50
-0.	-0.	-0.45754E 02	-0.27360E 02	٠0.	-0.	2.94
-0. 0.	- 0. - 0.	-0.15611E 02 0.52787E 01	-0.54645E 02 -0.75316E 02	~0. 0.	-0. -0.	2.50 2.54 2.58 2.65
•:	-0.	0.95254E 01	-0.11449E 03	ŏ.	-0.	2.00
ō.	-5.	0.594998 01	-0.209228 03	ō.	-0-	2.70
-0.	-0.	-0.354646 02	-0.209228 03 -2.27514E 03	-0.	-5.	3.00
-0.	-0.	-0. 87805E 02	-0.35 846 E 83	-0.	-9.	3.10
-0.	-0. -0.	-0. 19445E 95	-0.40564E 03 -0.41705E 03	- 6. -0.	-t. -t.	3.20 3.26
-A.	- O.	-6.51064E 03 -0.39597E 03	-0.43848E 03	-3.	-0.	3.29
-0.	-0.	-0.373646 05	-0.41330E 03	-J.	-0.	3.35
-0.	0.	-0.910306 03	0.965836 03	-0.	•	3.35 3.40
-0.	0.	-0.000106 03	0.10054E 04	-0.	0.	3.52 3.56 3.60
-0.	0.	-0.45510E 03	0.80094E 03	-0.	0.	3.56
-0.	0. 0.	-0.15554E 03	0.38044E 03 0.33450E 03	-0.	0.	3.90 3.70
0.	ŏ.	0.12050E 03 0.22216E 03	0.142168 03	0. 0.	0. 0.	3.6
ŏ.	-0.	0.244928 03	-0.24609E 02	ŏ.	-0.	4.00
0.	0.	0.800816 02	0.83463E 02 0.84314E 02	0.	0.	4.20
0.	9.	0.235528 02	0.843146 02	0.	0.	4,50
٥.	0.	0.54239E 02 0.65869E 02	0.77761F 02	0.	0.	4.70 4.80
0. 0.	0.	0.2000E 02	0.63041E 02 0.59751E 02	0. 0.	o.	1.2
•.	ŏ.	0.845536 02	0.439710 02	ě.	ŏ.	4,96 5.00
0.	0.	0.255798 02	0.243608 02	0.	ō.	3.15
0.	-0.	0.10493E ul	-0.10964E 02	0.	-0.	5.30
٥.	-5.	0.177756 03	-0.82769E 02	•.	-0.4	**
0.	-0. -0.	0.15.19E 05 0.42745E 02	-0.19580E 03 -0.17331E 03	o. o.	-0. -0.	5.70 5.85
-0.	- 0.	-0.1496+E 03	-0.12745E 03	-0.	-0.	6.00
-0.	-0.	-0.167e4E 03	-0.84Z40E 02	-0.	-0.	6.05
-0.	- o.	~0,17007E 03	-0.40207E 02	-0.	-0.	6.06
~0.	0.	-0.161926 03	G.27433E 02	-0.	٥.	6.00
-n. -0.	0. 0.	-0.73456E 02 -n.25055E 02	0.33772E 02 0.15525E 02	-0. -0.	0. 0.	6.20
-0.	ů.	-0.790056 01	0.14231F 02	-0.	ö.	6.40 6.60
0,	0.	3.22013E 0i	0.798788 01	0.	ŏ.	7.00
0.	-0.	0.996128 01	-0.38849E 01	٥.	-0.	7.40
٥.	-0.	0.887146 01	-0.88779E 01	0,	٠0.	1.20
•. -0.	-o. o.	0.25679E @1 -0.46770E 01	-0.04422E 01	•.	- 8. 0.	9.00 17.00
-5.	٧.	-0.441.105 01	▼•	- ••	₹.	\$74.00

Table XVII Stress Frequency Response Functions (Analysis Condition 3)

(PSI/FPS SINUSOIDAL GUST)

10 CPS

GROSS WEIGHT. 190, 590 LB CUTOFF FREQUENCY:
ALTITUDE: 24,000 FT
MACH NUMBER: 0.85

PERCENT SEMI SPAN: 27 SEGMENT NUMBER 10

THORPMENTAL OPEAR STRESS		INCR INCIPI A	INCRIMIZATAL AXIAL STRESS			
REDL	V RAW I JAN I	REAL	IMAGIMARY			
						PARTIES 1
0.144RRF 01 0.23312F 02	0.16311E 07 0.10499E 07	0-15029E 02 0-26795E 03	0.19340F 03	-0.	-0. -6.	0.30
0.294785 02	0.18434E 01	0.33465E 03	0.126025 03 0.24623F 02	-0. -0.	-6.	12
0.323105 02	-0.356515 01	0.359736 03	-0.38487F 07	-0.	j.	0,44
0.319956 07	-0.96994F N1	0.341798 63	-0.10708E 03	-0.	Ö.	0.50
1.304 27E 02	-0.13961F 07	0.33501E 03	-0.15287E 03	-0.	e.	0.6
0.2908AF 02 0.28241F 02	-0.17861F 02 -0.21951F 02	0.31032E 03 0.20052E 03	-0.16951E 03 -0.22319E 03	-0. -0.	0.	0.70 0.80
1.27451F 02	-0.268136 02	0.269676 03	-0.25*236 03	-0.	0. 0.	0.90
0.27835F 02	-0.41892F 07	0.252946 03	-0.34250E 03	-0.	0.	. 60
0.28468F 02	-0.58514F 0Z	0.220656 03	-0.43151E 03	-0.	0.	1.20
0.28343F 02	-0.497526 02	0.192226 03	-0.46371E 03	-0.	-0.	1.56
0.27376F 02 0.25354E 02	-0.81903E 02	0.17022E 03	-0.53764E 03	0. 0.	-0. -0.	1.40
9. 236906 02	-0.976176 02	0.131616 03	-0.60452E 03	o.	-0.	1.47
0.213478 07	-0.11825F 03	0.11270E 03	-C.68844F 03	ŏ.	-2.	1.90
0.11389F 07	-0.14534E 03	0.58705€ 02	-0.79270F 03	0.	-0.	1.95
-0.25248F 01	-0.17904E 03	-0.32451E 02 -0.19561E 03	-0.41156E 03	-0.	-0.	1.60
-0.30471F 03	0.1:509E 03		0.373966 03	-0. -0.	-0. 70.	1.65 1.97 1.90 1.95 1.60 1.63
-0.212991 03	9.13190€ 03		0.34606E 03	-0.	0.	1.90
-0.107996 63	0.11983F 03	-0.32130E J3	0.24510F 03	-0.	0.	2.00
-6.59444F N2	0.1C924E 03		0 - 160 00E 03	-0.	0.	3.10
-0.32384E 02	0.98963E 02 0.90871E 02		0.92564E 02 0.70562E 02	-0.	0.	2.20
-0.66259F 31	0. 404716 02	-0.58481E 02	0.407688 02	-0. 0.	0. 0.	3.20
-0.40966F 01	0.85642F 02	-0.60136 02	0.24295# 02	0.	0.	2.50 2.55 2.46
-0.30234F 01	O.84923F 02	-0.71197E 02	0.189226 02	o.	0.	2.43
-0.74447F 01	0.84074E 02	-n.76696E 02	0.156558 02	0.	0.	2.44
-0.47778F 01	0.87021F 02	-0.10559E 03 -0.1297\E 03	0.39935E n2 0.82225E n2	0.	0.	2.47
-0.31523F 01	0.999371 02	-0.112416 93	0.457406 02	0. 0.	0. 0.	2,50
0.527146 01	0.10202F 03	-0.768938 02	0.606236 02	o.	o.	
0.184046 02	0.103116 03	-0.479/36 02	0.42636E 0.	0.	0.	2.69
0.269685 02	0.10649F 03	-0.41786F 02	0-15434E + 3	0.	٥.	2.95 2.94 2.95 2.65 2.70 2.80
0.44515F 02 0.10112F 03	0.11789E 03 0.17020F 03	-0.43386E 02 -0.74153E 02	-0.20305E /3 -0.26794E 02	0.	0. 0.	3.00 3.00
1.14853F 03	0.870715 02	-0.1046ZE 03	-0.14418E 02	0. 0.	0.	1.10
0.27243E 03	-0.73706E 07	-0.16623E 03	0.41240E 02	ō.	- 0.	5.5
0.37741E 03	-0.1+321E 03	-0.21576E 03	010876E 03	•.	-0.	3-86
0.40213F 03	-0.38437E 03	-0.22334E 03 -0.11240E 03	0.20511E 03 0.15194E 03	٥٠	- e. -0.	}- <u>17</u>
0.14671F 03	-0. 42738F 02		0.351398 02	0. -0.	-0.	- 22 -
-0.58021F 02	-0.176496 02		0.165926 02	-0.	0.	1.32
-0.7484F B?	0.35874E-00	-0.22697E 02	0.7035#E 01	0.	0.	3.96
0.18882F 07	-0.7799AE 02	-0.36277E 02	-0.11750E 02	0.	-0.	3.60
0.13456F 03	-0.19200E 03	-0.735845 02 -0.152216 03	0.23942E-00 0.87604E 02	0. -0.	-0. -8.	2.72
-n.11979F 03	0.243336 02	-0.231176 03	0.353276 03	-0.	-0.	
-0.60218F 02	-0-12288E 02	-n.14905E 03	0.18742E 03	-0.	-0.	1.10 1.20 1.20 1.20 1.20 1.20 1.00 1.00
-0.78982E 01	-0.167316 02	0.16127E 03	0.774616 02	-0.	-0.	4.50
-0.175956 02	-0.15116E 02	0.16537E 03 0.1506DE 03	0.47575E 02 0.16191E 02	-9.	-0. -0.	4.70 4.70 5.00 5.13 5.30
-0.20640F 02 -0.23440E 02	-0.10142F 02	0.170656 03	0.101366 02	-0. -0.	•0.	12
-D.23860E 02	-0.62945F 01	0.1234UE G3	-0.67784E 01	-0.	-0.	3.00
-0.24455F 02	-0.24302E 01	0.10700E 03	-0.19493E 02	-0.	0.	9.19
-0.24501E 02	0.25695E 01	0.921366 02	-0.31394E 02 -0.40357E 02	-0.	0.	5.30
-0.23706F 02 -0.22012F 02	0.102296 02	0.75442E 02 3.40075E 02	-0.40357E 02 -0.45416E-02	-0. -0.	0. 6.	1.5
-0.20307F 02	0.13951F 07	0.497446 02	-0.49940E 02	-0.	i.	
-0.17861E 02	0.146305 02	0.30004E 02	-0.5C655E 02	-0.	3.	8.00
-0.17343F 02	0.15325F 02	0.3400AE 02	-0.51353E 02	-0.	0.	6.03 6.06 6.08 6.38
-n.16786E 07	0.16036F 0Z	0.34947E 02 0.37976E 02	-0.52037E 02 -0.54905E 02	-0. -0.	0. •.	1.05
-0.16185E 07 -0.12773E 07	0.199801 02	0.234676 02	-0.57259E 02	-0.	0.	4.34
-0.947045 00	0. 6901 96 01	0.25455E 01	-0.34020E 02	0.	ŏ.	i.u
0.240266 02	-0.360ABE 01	-0.23228E 02	-0.36045E 02	0.	-0.	1.40
-0.94513F 01	0.54895E 01	-0.68377E 01	-0.27879E 02	-0.	0.	7.00
-0.73624E 01	0.424385 01 0.490255 01	-0.52461F 02 0.23464E 02	0.19606E 02 -0.45361E 01	0. -0.	-0. •.	7:20
-0.19473E 01	6.747136 01	0.126946 02	-0.434996 01	-6.	:	1.30
0.28407F 61	0.	0.337636 01	0.	ě.	0.	16.00

(PSI/FPS SINUSOIDAL GUST)

190, 590 LB 24,000 FT 0.85

CUTOFF FREQUENCY:

10 CFS

,而是我们的自己的自己的自己的自己的自己的对象,但是不是不是不是不是不是不是不是不是不是不是不是不是不是不是不是不是不是,我们们的自己的心理,也可以是不是一个人的人,也可以是一个人的人,也可以是一个人的人,也可以是一个人的人,也可以是一个人的人的人,也可以是一个人的人的人,也可以是一个人的人,也可以是一

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

PERCENT SEMISPANE 27

SEGMENT NUMBER 14

INCREMENTAL	SEEMAR STREETS	l	INCHMODITAL	ANTAL STREET				
RFAL	18451946		RFAL	18461844	l v			
								71
0.54070# 01	0, 64 91 25	02	0.134396 02	0.17550E	03	-0.	-0.	_
0.90784E 02 6.11906E 03	0.419896	02	M. 24135E D3 M. 18713E D1	0.114367	73 0>	-0.	-0. -0.	
0.12470F 01	-0-110676	02	0.33461E 03	-0.34926F	02	- 0.	••	
0.12237F 03 0.11386F 03	-0.357376	02	0.32031E 03	-0.97169F	93	-0.	•. •.	
0.10583F 01		02	0.201616 03	-0.1E197E	•1	. 0.	ō.	
0.99335E 02 0.94168F 02	-0. 74 85 TE	02	6.26163E 63	-0.292548	•3	• • •	0. 0.	
0.899136 07	-0.119698	03	0.217576 03	-0.310616	03	-6.	ŏ.	
0.0283 W 07	-0.14642F	63	3.29924E 83	-0.391598	03	-0.	0.	
0.76033F 02 0.70883F 02	- 0, 1700 FE	03	0.172426 03	-0.401895	•3	-0.	-0. -0.	
0.440900 02	-0.7167EF	03	0.133336 03	-0.51655E	03	•.	- 0.	
0.59269F 02 0.53033F 02	-0,23280E	03	0.11943E #3	-0.348348	8 3	0.	- 0. - 0.	
0.139600 02	-0.321558	73	4.33274E 02	->.71935E	03	ō.	- 0.	
-0.83919F 00	-0.3816BE	03 -	0.294486 82	-0.027228	03	10.	-0. -0.	
-0,56721F 03	0.143026	03 ~	0. 11314E 04	0.33930€	03	-0.	0.	
-0.371095 03	0.197426	03 -	0.470216 83	0.315058	03	-0.	0.	
-0.17714F 03 -0.97932E 02	0.116096	01 -	0.142348 03	0.145196	03	-0.	o. o.	
-0.40202f 02	0.111046	03 -	0.77533E 02	0.04000E	02	-0.	o.	
-0.25263E 12	C. 99114E	02 -	0.52198E 02	0.640366	02	-0.	•. •.	
-0.17451F 02	0. #34ADE	07 -	0.54572E 02	0.220476	02	0.	ŏ.	
-0.19592F n2	9. 4164 7F	02 -	0.446335 02	0.171716	02	0.	0. 0.	
-0.21621E 02	0. 910945	02 -	3.94094E 02	0.342408	02	0.	0.	
-0.39992F 02	0.11313F	03 -	5.11771E 03	0.766175	02	0.	0.	
-0.31835F 02 -0.12741F 02	0.170166	03 -	0.10231E U3 0.4927AF 02	0.778638	02	0.	0. 0.	
0.101406 02	0.114298	01 -	0.434716 07	0.304716	02	o.	ő.	
0.21745F 02 0.42117F 02	0.117126	03 -	0.379226 02	0.140045	02	0.	0. 8.	
0.102975 05	0-116476	03 -	0.472426 02	-0.261306	•2		i.	
0.16343F 03	0. 737275	02 -	a. 20434E 05	-0.13004€	92	•.	6.	
0.79219F 03	-0.33710E	01 -	0.170432 03	0.987146	92	•	-0. -8.	
0.43726F 01	-0.677336	01 -	0.202446 81	0.106136	03	Ö.	-0.	
0.145F2F 03 -0.76967F 07	-0. 177136	03 -	0.10237E G3	0.137886	03	0. -0.	-6. -0.	
-n.11959F 03	-0.347592	02 -	0.853596 01	0.160726	03	-0.	o.	
-0.88949F D7	-0.307245	02 -	0.705978 07	0.638495	01	0.	0.	
0,556775 07	-0.184746	01 -	0.667756 02	0.217276-	-00	φ,	- 0. - 0.	
-n.13206F 03	-0.350956	٠,	n. 13013F 03	0.79680E	02	-0.	-0.	
-0.21133F D3	0.14493E	03 -	0.209738 03	0.32059E	03 01	-0.	-0. -0.	
0.44173F 02	0. 905 958	Ŏ1	0.26450E 03	0.70294E	02	-0.	- 0.	
0.302P4F 02 0.22917E 02	7. 84373F	20	0.15007E 33	0.431736	02	-0.	-0. -0.	
0.137476 07	-0.465718	01	0.116736 03	0.42000E	01	-0.	-0.	
0.11835F 02	-D. P4646F	01	0.112366 03	-0.413126	01	-0.	-9.	
0.63032C 01 0.20073F 01	-0.947716	01	0.83411E 02	-0.204898	02	-0.	9. 0.	
-0.21144F 01	-0.891526	01	0.684638 32	-6.346236	02	-0.	ō.	
-0.53395E 01 -0.72440F 01	-0.27711E	01	0.51517E 02 0.45105F 02	-0.41714E	07 07	-0.	e. 0.	
-0.40107F 01	-0.7222AF	Öİ	0.352146 02	-0.439682	02	-0.	ŏ.	
-0.47943E 01	-0.704088	01	0.314736 02	-0.46602E	02	-0.	e. 0.	
-0.484515 01	-0.384196	01	0.249236 02	-0.49025E	02	~O.	0.	
-0.1099RF 02	-0.37944F	01	0.211148 07	-0.5i961E	02	-0.	ō.	
-0.12237F 07	-0.17017F	01 -	0.731336 01	~0.30877E	02	0.	0. -ú.	
-0.14701F 07	- 0. 95591 F	00 -	0.420445 01	-0.25300E	02	-0.	0.	
-0.37345F 07	0.7507AF	07 -	0.47607E 02 0.21703F 02	-0.411445	01	-0.	-0. 0.	
0.86884F 81	0. 541 997	01	0.113206 02	-0.560046	01	-0.	ě.	
0.78205F 01	n,		0.30185E 01	•.		•.	••	

(PSI#PS SINUSOIDAL GUST)

GROSS WEIGHT: 190, 590 LB CUTOFF FREQUENCY:
ALTITUDE: 24, 000 FT
MACH NUMBER: 0. 85 10 CPS

PERCENT SEMISPAN: 40.06 SEGMENT NUMBER 8

::::::::::::::::::::::::::::::::::::::	SEEAR STREET	INCHESTAL	AXIAL STRESS			
REAL	1 MAG 1 MARY	REGL	IMAGINARY			Tinana.
						1.30
0.374f6F 01 0.570A]F 02	0.41220F 02 0.26936E 02	0.12531E 02 0.23143E 03	0.16923F 03	•. •.	0.	1.30
0.7766AF 02	0.548186 01	0.231436 03	G. 11044E 03 0.21133E 02	6.	0.	4.5
0.79345F 02	-0.79120F 01	0.32334E 03	-0.349498 02	•.	-0.	0.44
0.78304E 02	-0.22687E 02 -0.32936E 02	n.31793E ü3 n.29533E 03	-0.96617E 02 -0.13846E 03	•. ¢,	-0. -0.	0.40
0.49474E 07	-0.41 708E 02	n, 27373F 03	-0.17243E 03	e.	-0.	0.70
0.66131F 02 0.63713F 02	-0.50397E 02	n.25457£ 83 n.237832 83	-0.20400E 03 -0.23703E 03	0. 0.	-0. -0.	6.30 6.40 6.30 1.40 1.45 1.45 1.45 1.45 1.45 1.46 2.20 2.20 2.20 2.20 2.20 2.20 2.20 2.2
0.419106 02	-0.86653E 02	0.22245E 03	-n. 31656f 03	0.	-0.	1.00
n.491n0F 02	-0-11779F 03 -0-13704E 03	0.19150E 03 0.16102E 03	-0.46817E 03 -0.44894E 03	0. 0.	- 0. - 0.	1.30
0.51734F 02	-0.15751F 03	0.14114E 03	-0.49914E 03	0.	-d.	1.6
0.46554E 02 0.42722E 02	-0.16973E 03 -0.;0337E 03	0.11700F 03 0.107616 33	-0.32844E G3 -0.56112E 03	7. 0.	- 0. - 0.	1.5
D. 37659F 02	-0.21721E 03	0.83875E 02	-0.43042F 03	0.	- 0.	1.59
0.2178#F 02 -0.77569F 01	-0.26049E C3 -0.31270F O3	6.30413E 02 -0.57713E 02	-0.73357É 03 -0.04009E 03	0.	-0.	1.55
-7.44920F 02	-0.134966 03	-0.2141?E 03	-0.20700E 03	-0. -0.	-0. -0.	1.45
-0.49710F 03	0.17736F 03	-7.11798E 04	0.37720€ 03	-0.	•.	1.00
-0.32534F 03 -0.15567F 03	0.15859E 03	-9.60193E 03 -0.28179E 03	0.33776E 03 0.23344E 03	-0. -0.	•. 0.	1.50
-0-417518 02	0.13000E 03	-7.14217E 03	0.14630F 03	-0.	0.	2.30
-0.45563F 0? -0.39274E 02	0.16861F C3 0.11451E G3	-0.01303E 02 -0.00321E 02	0.04409E 02 0.00161E 02	-0. -0.	ə. 0.	2.5
-0.309665 02	0.10732F 03	-0.776406 02	0.43333E 02	-0.	•.	2.35
-0.10769E 02	0.10170E 03 0.995A3E 02	-0.63751E D2 -0.70343E 02	0.47863E 02 0.42624E 02	-0. -0.	0. 0.	2.49 2.43
-0.14307F 37	0.96271E 02	-n.74767E 02	0.37169E 02	-0.	•.	2.44
-n.19011E 02 -n.22772E 02	0.99411E 02 0.10#89E 03	-n.92214E 02 -n.10737E 03	0.31041E 02 0.75742E 02	o. o.	0. 6.	2.47
-0.17559F 02	0.11045E 03	-0.93069E 02	0.73373E 02	ŏ.	0.	2.5
-0.63644F 01	0.105966 03	-n.69478E 02 -n.32393E 02	0.49230E 02 0.32033E 02	0.	J.	2.98
n.13713F 62	0.10200E 03 0.95916E 02	-0.50741E 02	0.797196 01	ა. ₀.	•. 0.	2.45 2.70
0.23086F 02	C. 07453E 02	-0.5935AE 02	-0.23605E 02	e.	٥.	2.00
0.36302F 02	0.85624E 02 0.87210E 02	-n.10746€ 03 -n.13537€ 03	-0.27187E 02 0.62513F 01	0. 0.	0. 0.	3.00 1.16
0.29907F 02	1.10948E 03	-n.24669E 03	0.10336E 03	0.	•.	3.20
0.13974F 02 0.89375F 01	0.17974E 03 0.20339E 03	-0.32144E 03 -0.32893E 03	0.27672E 03 0.37572E 03	0. •.	0. 0.	a a a a a a a a a a a a a a a a a a a
0.591125 02	0.200756 (/3	-0.13436E 03	0.209316 03	0.		3.35
0.13193F 03 0.22222F 03	0.15240E 03 0.13719E 03	0.33097E 02 0.54230E 32	0.10442E 03 0.70756E 02	0. 0.	0. 0.	3.46
0.25433F 03	0.11434E 03	0.42232E 02	0.396736 02	ŏ.	ŏ.	3.5
0.29349F 03	-0.45428E 02 -0.19171E 03	n.32092E 02 n.85973E 01	0.72726E 01 -0.16408E 02	0. 0.	-0. -0.	3.60
0.17073F 03	-0.105566 03	-0.12123E 03	0.930946 02	ŏ.	-0.	
0.16147F 03	-0.31380F 03 -0.15529F 03	-0.21A24E 03	0.34612E 03 0.17406E 03	0.	-0.	£
0. F2691F 02 -0.15528E 03	-0.177271 03	-0.11639E 03 0.20291E 03	0.437426 02	Ů. -0.	- 0. - 0.	1.2
-0.13908F 03	-0.441 73E 02	0.173346 03	0.378486 02	-0.	-0.	4.5
-0.12704F 03 -0.16976F 03	-0.10903E 02 -0.13980E 02	0.15509E 03 0.13063E 03	0.10150E 02 0.51289E 01	-0. -0.	-0. -3.	1.00
-0.10576E C3	-0.097926-01	0.12459E 03	-0.793656 01	-0.	٥.	3.00
-0.93147F 02 -0.81938F 02	0.10527F 02 0.20709F 02	0.10371E C3 0.49334E 02	-0.16264E 02 -0.21930E 02	-0. -0.	•. 0.	รห
-0.69575F 02	0.20723E 02	0.72232E 0 2	-0.23711E 02	-0.	0.	2:18 2:20
-7.58474F 07 -0.50994F 07	0.33544E 22 0.38203E 02	0.38433E 02 0.30478E 02	-0.23016E 02 -0.20609E 02	-0. -0.	•. 0.	5.70
-0.43196F 02	0.30903F 02	0.445276 02	-0.23014E 02	-0.	0.	2.0
-0.41A17E 02	0.39761F 02 0.40538E 02	0.43719E 02 0.42996E 02	-0.19362E 02 -0.10652E 02	-0. -0.	0.	2. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1.
-0.40417F 02 -0.38890F 02	0.440336 02	0.42370# 02	-0.14707E 02	-0.	0.	6,06
-6.32011F 02	0.48814F 02	0.41053E 02	-0.67301F 01	-0.	0.	6.3
-0.15548F 02 0.93037F 01	0.32265E 02 0.27155E 02	0.49416E 02 0.79697E 02	-0.32034E 02 -0.40902E 02	-0. 0.	0.	6.46
-0.93228E 01	0.29140F 02	n.13339F 02	-0.32193E 02	-0.		7.66
0.14961F 02 -0.12339F 02	0.78401F 01 0.17690F 02	0.17867E 02 -0.20234E 02	-0.40533E 02 -0.13303E 02	0. -0.		7.40
-0.55730F 00	0.13539F 02	-0.243266 02	-0.500096 00	-0.	é.	9.00
0.96464F 01	0.	-0.10032E 02	••	2 .	•.	24.25

Teble XVII - - - Continued

(PSIFFPS SINUSOIDAL GUST)

GROSS WEIGHT: 190,590 LB CUTOFF FREQUENCY: 10 CPS ALTITUDE: 24,000 FY MACH NUMBER: 0, 85

PERCENT SEMISPAN: 40.06 SEGMENT NUMBER 107

INCOMPANY SEEAR STREET				TECHNOTAL	AXIAL STREET	
9544	1 MAS MAS Y			• F61	18451 MARY	
*****	1144914004			•601	114911464	PROCURING
		_	_			0.15
8.37974F 01 9.57361F 07	0.41183F 02 0.76906F 02	o. c.	o. o.	-0.10707E 02 -0.19776E 02	-6.14460E 03 -0.94366E 02	0,15
0. F7544F 07	0.454796 01	e.	o.	*-0.25297E 03	-0.18054E 07	0. %
O. TYPARE OZ	-n. 772766 01	0.	-0.	-0.27429E 03	0.29342E M2	6.44
0.77919F 07	-0.22271E 02 -0.32276F 02	n.	-o.	-0.27167E 03	0.82553E 02	ð. 5 0
0.731424 42	-0.407178 62	n. C.	-0. -0.	-0.25277E 03 -0.233 40E 03	0.1:031E 03 0.14733E 03	0.50
0.654766 02	-3,4903 DE 02	o.	⊸o.	-0.217516 03	0.17430E LD	0.80
0.62935 02	-0. 58355E 02	n.	-0.	-0.203276 03	0.20253E 03	0.90
0.41070F 02 0.58245F 02	-0.83437E 02 -0.11291E 03	°.	-0. -0.	-0.19007E 03 -0.16369E 03	0.770-8E 03 0.341 92E 03	1.40
0.564916 02	-0. 13114E A3	0.	-ŏ.	-0.1375@ 03	0.363596 03	1.35
A. 5' 145F 02	-0.15053E n3	n.	-o.	-0.120616 03	G. 42 64 86 03	1.50
0.46571F 02 8.42949F 02	-0.14211E 03 -0.17523F 03	n. n.	-0. -0.	-0.10072E 03 -0.87676E 02	0.45152E 03 0.47944E 03	1.45
0.392236 02	-0.207148 03	0.	-û.	-0.718A4E 02	0.545486 03	2. 67
0.2337 0 E 02	-0.24875E 03	0.	-O.	-0.2632 9E 02	0.42478E 03	1.50 1.55 1.60
-0.43341F 01 -0.58102F 02	-0.29795E 03 -0.13254E 03	-0. -1.	-0. -0.	0.49309E 02 0.10299E 03	0.71780F 03 0.24329F 03	1.60
-0.4670)E 03	0.163758 03	-n.	-0.	0.10001E 04	-C. 37730F 03	1.65
-n.30745f 03	0.17542E 03	-r.	٥.	0.361846 03	-0. 20057E 03	1.99
-0.14713F 03	0.1475% 03	-0. -0.	0.	0.74931E 03	-0. 99486 03	2.00
-8.77093F 62	0.12037E 03 0.44791E 02	-n.	o. o.	0.17147E 63 0.4943Œ 62	-0.12500£ 03 -0.72190£ 02	2.20
-n.34593E 02	0.10489E 03	-o.	•.	0.75464E 02	-0.75327E 02	2.50
-n.24454F 07	0.97696E 02	-ņ.	0.	0.4433 0€ 07	-0.54114E 02	2.55
-0.1806RF 02 -0.14391F 02	n, 92149E 92 n, 90074E 02	-n. -n.	o. o.	0.56188E 07 0.60193E 02	-0.40895E 07 -0.34419E 07	2.45 2.45
-0.1647RE 02	0. 97: 405 02	-9.	o.	0.63893€ 02	-0.31775E 07	2.44
-0.264335 02	0. 91513F 02	0.	Ģ.	0.707916 02	-0.434116 07	7.47
-n.25071E 02 -n.19545E n2	0.13272E 05	e.	0. 0.	0.91740E 07 0.80204E 02	-0.64714E 87 -0.62693E 07	2.50
-0.75212F 01	0.100376 03	0.	o.	0.593448 02	-C. 42 063E 02	2.94 2.98
0.468951 01	0. 963436 07	0.	0.	0.44511E 02	-0.280726 02	2.45
0.13315F 02 0.22492F 02	n. 401 47F 02 0. 81 946E 02	o. o.	o. o.	0.43372E 02 0.50717E 02	-0.64114E 01 0.70237E 07	2.70
0.35875F 02	0.792065 92	0.	o.	0.913716 07	0.732296 07	3.00
0.3630AF 02	0. 814076 02	ç.	0.	0.133619 03	-9.534136 01	3.10
0.30520F 02	0.99761E 07 0.12787E 03	€. 0.	0. 0.	0.21070E 03 0.27469E 03	-0.00313E 07 -0.10517E 03	3.20
0.910796 01	C.14879E 03	0.	١٥.	0.2610¥ 03	-0.37103€ 03	3. 36 3.29
0.35745F N2	0.1 R764E 03	Ĉ.	0.	0.110726 0)	-0.24719E 03	7.77
0.12-0AF 03	0.147176 03	ç.	••	-0.2844E 02	-6.84550E 05	3.35 3.40
0.2106#F 01 0.24111F 03	0.12724E @3 0.10531F 03	o. o.	o.	-0.4435年 02 -0.3405年 02	-0.677 926 07 -0.509886 07	3,52 3,96 3,60
0.277966 03	-0.43840F 07	0.	-0.	-0.27412F 02	-0.67139F 01	3.60
n.36322E 03	-0.18190F 03	0.	-0. -0-	-0.73458E 01	0.1.01 - 02	3.70 3.65
0.170ALF 03	-0.16079E 03 -0.30552E 03	0. 2.	-r. -0.	0.1035aE 03 0.1847aE 03	-0.807746 07 -0.317876 03	}. 55
0.77004F 02	-0.1490AE 03	o.	-o.	0.94184 02	-0.14877E 03	4,00 4,20
-0.191235 61	-0.64483F DZ -0.41404F DZ	-n. -n.	-0.	-0.17337F 01	-0.56172E 02	.50 .70
-0.13609F 03	-0.41404F 07	-0.	-0. -c.	-0.149856 03 -0.144056 03	-0.3233#E 02 -0.86721E 01	•.70
-0.10472F 03	- 0. 12854E 02	-r.	-o.	-0.11162 03	-0.438236 01	4.95
-0.197745 03	0.191876-00	-ņ.	0.	-0.104496 03	0.678128 01	5.00
-n.40233F 02 -n.79206F 03	0.10013E 02 0.19234F 02	-e. -e.	o. o.	-0.90321E 02 -0.76347E 02	0.150%E 02 0.18737E 02	5.15
-0.A7176F 02	0.262736 02	-ñ.	ö.	-0.617346 02	0.202396 02	9.30 6.79
-0.56527F 02	J. 30371 6 02	-0.	0.	-0.499276 02	0.1966FE 02	5.76
-0.49594F 02	0.34214E 02 0.34845E 02	-0. -0.	o.	-0.43301f 02 -0.3804% 02	0.17609E 02 0.17100E 02	5.15 5.76 5.76 5.86 6.00
-0.41297F 07	0.354706 02	-n.	0.	-0.37344F 02	0.143436 07	6.0t
-0.40057F 02	0.36091 02	-0.	0.	-0.367375 32	0.15936E 02	6.05 6.06
-0.3480AF 02	0.388418 02 0.425496 02	-e. -o.	0.	-0.26202E 02 -0.33077E 02	0.173446 07	5.08
-0.19700F 07	0.311716 02	-0. -0.	o. o.	-0.330772 02	0.37744E 01 0.27371E 02	6.30
-9.83171F 00	0.276628 02	0.	0.	-0.480946 02	0.41783E 02	6.40
-0.11336F 02	0.287196 02	-₽.	٥.	-0.13081F 02	0.27304E 02	7.40
0.74852F 01 -0.78935F 01	0.13463E 02 0.18373E 02	?. •0.	o. o.	-0.15766E 02 0.1778E 02	0.34631F 02 0.13146F 02	7.46
0.2853ME 01	0.12817F 02	-r.	0.	0.203296 02	0.434131-00	1.26 3.40
0.114379 02	0.	e.	0.	0.140716 02	0.	14.40

(PSI/FPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER: 190, 590 LB 24,000 FT 0, 85 CUTOFF FREQUENCY: 10 CPS

BODY BALANCE STATION: 540 SEGMENT NUMBER 17

THE PERSON	L SEEAP STREES					
RFAL	1 PAG1 WARY					THEORET
						974 0.14
-0.33429£ 03 -0.44446 02	-0.27740E 02 -0.14055F 02	n. B.	J. 0.	0. 0.	9. 9.	6.30
-0.53940€ 02	-C.53280F 00	٨.	-0.	0.	- 0,	6.30 0.36 0.44 0.50 0.60
-0.5591CF 02 -0.53027F 02	0.81572F 01	0. 0.	-0. -0.	0. 0.	- 0. - 0.	0,44
-0.472716 02	0-210076 02	n.	-0.	0.	-0-	9.60
-0.47415E 07 -0.38631E 02	0.24020E 02 0.24390F 02		-0. -0.	0. 0.	-9. -0.	0.70 0. 8 6
-0.35713F 02	0.28630E 0Z	0.	-0.	0.	-0.	0.90
-0.334196 02 -0.30272F 02	0.33825E NZ 0.39486E 02	n. n.	-0. -0.	0.	- 0. - 0.	1.00
-0.28593# 02	n.42912F 02	0.	-o.	ō.	-0.	1.20 1.34
-0.27780F 02	0.46530E 02 0.48684E 02		-0. -0.	0. 0.	- 0. - 0.	1.40
-0.262A9E 02	0.511256 02	0.	-0.	0.	-0.	1.65 1.67
-0.23595E 02 -0.23189F 02	0.57078E 02		-o. -o.	0. 0.	-0. -0.	1.90 1.95 1.60
-0.185BAF 02	0.74557F DZ	٠.	-0.	0.	- 0.	1.75
-0.10357F 02	0.55150E 02 0.29499E 01		-0. 0.	0. -u.	- 0.	1.65 1.80
0.416476 02	-0.91740£ 00		2.	-0.	0.	1.40 1.90
0.151356 02	0.38670E 01 0.93878F 01	-0.	-0.	-0.	~0.	2.00
0.30424F 01 -0.30149F 01	0.15383F 02	-0. -0.	-0. -0.	-0. -0.	-0. -0.	2.10 2.10
-0.47536F 01	0.16946F 02	-0. -0.	-n. -n.	-0.	-0.	2.50
-0.50770F 01 -0.51403F 01	0.20127F 02	-n. -n.	-n. -0.	-0. -0.	- 0. - 0.	2.55 2.40
-0.37822F 01	0.22987E 02	-n.	-0.	•0.	-0.	2.65
-0.27654F 01 0.12123F 01	0.23472F 02 0.19804c 02	-n- -0-	-0. -0.	-0. -0.	-0. -0.	2.44 2.47
0.51168F 01	0.12524E 02	-0.	-0.	-0.	-0.	2.90
0.28327E 01 -0.31574E 01	0.11013F 02 0.14212F 02	-n. -0.	-0. -0.	-0. -0.	-0. -0.	2.56
-0.94759F 01	0.170706 02	0.	⊸.	0.	-0.	2.56 2.56 2.65
-0 12062F 02 -0.1566F 02	0.22418E 02 0.34825F 02		-0. -0.	0. 0.	- 0. - 0.	2.70 2.80
-0.25105F 02	0.44944E 02	n.	⊸.	0.	-0.	2. 00 3.00
-0.35952F 02 -0.41154F 02	0.70125F 02 0.11789F 03	e. e.	-•. -•.	0. 0.	-0. -0.	3.10
-0.84501F 02	0.1710AE 03	e.	-0.	0.	-0.	3.20 3.26
-0.83635F 02	0.29455F 03 0.23261E 03	ი. ⊸o.	-o. -o.	0. -0.	-0. -3.	\$2.26 \$1.29 \$1.29 \$1.60
0.104516 03	0.17010F 03	-0.	٠0.	-0.	-0.	3.6
0,1976AF 03 0,23654F 03	0.15501F 03 0.13112F 03	-n. -n.	-0. -0.	-0. -0.	-0. -0.	3.52 3.96 3.60
n.24479F 03	- 0. 98427E 02	-0.	0.	-0.	0.	3.60
0.42353F 03 0.20886F 02	-0.32151F 03	-0. -0.	0. 0.	-0. -0.	0.	3.70 3.85
-0.115855 03	0.278515 02	0,	-0.	0.	0. -0.	3.67 4.60
-0.73841F 42 0.21205F 02	-0.33063F 01	0. -0.	0.	0. -0.	0.	4.20
0.121215 02	-0.202595 02	-0.	0.	-0.	ρ. 0.	4.50 4.70
0.74143E 01	-0.194956 02	-n.	0.	-0.	0.	4.80
0.15380F 01 _0.31168F-00	-0.19716F 02	-ņ. -n.	0. 0.	-0. -0.	o. o.	4.96 5.30
0.31168F-00 -0.32136F-01	-0.14060F C2	-n.	0.	-0.	o.	5.15
-0.58646F 0) -0.8)363F 01	-0.97526F 01 -0.45393F 01	0. 0.	0. 0.	0. 0.	0. 0.	5.50 5.40
-0.92912F 01	-0. 47969F-01	0.	0.	v.	0.	5.70
-0.92211F 01 0.78214F 01	0.479705 01 0.496466 01	n. 0.	-0. -0.	0. 0.	-0. -0.	3.85 6.00
-0.73685F 01	0.818976 01	0.	⊸.	0.	-0.	6.09
-0.68797E 01 -0.61916E 01	0.947696 01 0.161506 02	n. 0.	-0. -0.	0. 0.	- 0. - 0.	6.06 6.08
-0.15724F 01	0.290936 02	-0.	-0.	-0.	-0.	6.20
0.20643F 0. 0.78494F 02	-0.98080E 01	-0. -0.	0. 0.	-o. -o.	o. o.	6,40 6,60
-0.180477 02	-0.22341E 02	0.	0.	0.	0.	7.00
-0.49646F 07	0.13420E 02 0.11442E 01	0. -0.	-0. -0.	0. -0.	-0. -0.	7.40 8.20
0.34034F 01	-0.13286E 01	-0.	⊸.	-0.	-0.	9.00
0.17133F 01	n.	-0.	0.	-0.	ō.	10.00

Table XVII - - - Concluded

(PSIFFPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

190, 590 t 8 24,000 FT 0.85

CUTOFF FREQUENCY:

10 CPS

BODY BALANCE STATION: 820 SEGMENT NUMBER 1

INCOMPTAL ANIAL STITUTE								
		• FAL	1 MASI MARY			IRECTORY		
·-	<u>^</u>	0.10020E 02	0.103496 03	0.	0.	6.30		
n.	o. e.	n.15919F 03 n.19704F 03	0.64263F 07 0.65361E 01	ሳ. 0.	0. 0.	0.30 0.36		
D.	-0.	0.26835E 03	-0.27207E 02	0.	-0.	0.44		
o. n.	-0. -0.	0.20098E 03 0.18263E 03	-0.61645F 02 -0.82745E 02	0. 0.	-0. -0.	0.50		
n.	-0.	0.14635€ 03	-0.984049 82	0.	-0.	9.70		
٥.	-0.	0.15325F 03	-0.11204F 03	0.	-0.	0.80		
0. n.	- 0. - 0.	0.14269E 03 0.13383E 0;	-0.12577E U3 -0.15838E 03	0. 0.	-0. -0.	0.90 1.00		
•.	- 0.	0. 11930F 03	-0.191786 03	ŏ.	- 0.	1.30		
n. o.	-0. -0.	0.107246 03	-0.21139E 03	0.	-0.	1.34		
n.	- D, - O.	0.10022F 03 n.92133F 02	-0.23156E 03 -0.24334E 03	o. o.	-0. -0.	1.40 1.45		
n.	- 0.	0.868476 02	-0.25648E 03	ŏ.	-0.	1.47		
0. 4.	- 0. - 0.	0.403526 02	-0.28768F 03	0.	-0. -0.	1.50		
: :	-0.	9.41926F 02 0.30443F 02	-0.32649F 03 -0.37122F 03	o. o.	-0.	1.5÷ 3. £ 3		
-0.	- 0.	-0.261236 02	-0.18998E 03	-•.	-0.	1 65		
-0. -0.	o. o.	-0.40201E 03 -0.23132E 03	0.67341E 02 0.57189E 02	-3. -0.	o. o.	1.80 1.90		
-0.	n.	-0.83442 02	7.16329F 02	-0.	0.	2.00		
-0.	•0.	-0.25459E 02	-0.21511E 02	-0.	-0.	2.10		
-0. -0.	-9. - 0.	-D.12175E 01 -0.20741E 01	-0.54631E 02 -0.61124E 02	-•. -0.	-0. -0.	2.30 2.30		
-0,	- 0.	-0.21705E 01	-0.78169E 02	-0.	- 0.	2.30 2.35 2.40		
-0.	-0.	-0.30220F 01	-0.80525F 02	-0.	-0.	2.40		
-0.	-0. 0.	-n.11997E 02 -n.17435E 02	-0.91937F 02 -0.93544F 02	-0. -0.	-0. -0.	2.43		
-9.	-0.	-0.35945E 02	-C.76598E 02	-0.	-0.	2.47		
-0. -0.	- n. - o.	-0.54243E 02 -0.43224E 02	-0.44795E 02 -0.44453E 02	-0. -0.	-0. -0.	2.90		
-0.	- 0.	-0.19055€ 02	-0.63860F 02	-0.	-0.	2.9		
٥.	-0.	n.#2100F 00	-0.78791F 02	0.	-e.	2.50 2.54 2.58 2.65		
n. n.	- 0. - 0.	0.51077E G1 0.44563E 31	-0.10512E 03 -0.15824E 03	¢. 0.	-0. -0.	2.70 2.80		
-0.	-0.	-n.13n93E 02	-0.189906 02	-0.	-0.	3.00		
-n.	- 0.	-0.27894F D2	-0.23365E 03 -0.27969E 03	-0.	- 0.	3.10		
-n.	-0. -1.	-0.43217E 02 -0.58646E 02	-0.320 7 03	-0. -0.	-0. -0.	3.26		
-0.	-0.	-0.76737E 02	-0.36766F 03	-0.	-0.	5.20 5.27 5.75 5.76 5.76 5.70 5.70 5.70 6.20		
-0.	-0.	-0.16294F 03 -0.26033E 03	-0.39898F 03 -0.40773E 03	-0. -0.	-0. - 0.	3-25		
-n. -n.	-0. -0.	-0. 100356 03	-0.39481F 03	-0.	-0.	3.92		
-0.	-0.	-0.62885E 03	-0.34048F 03	-0.	- 6.	3.36		
-n. -n.	0. n.	-n.79791E 03 -n.12327E 04	0.31623F 03 0.97089F 03	-0. -0.	0. 0.	7.00		
-0.	n.	-0.704096 02	0.42332F 03	-0.	õ.	5.65		
n.	0.	n. 29261E 03	0.56792F 02	0.	0.	4.00		
n. 0.	r. r.	0.17317E 03 0.26537E 02	0.43433E 02 0.41176E 02	o. o.	0. 0.	1.30		
0.	0.	0.51053E N2	0.81304E 02	0.	0.	4.90 4.70 4.8e 4.96 5.13 5-30		
n.	0.	0.59477E 02 0.67733E 02	0.637 99 F 02 0.59135F 02	0. 6.	0. 0.	1,80		
n. n.	n. n.	0.690296 02	0.43090E 02	0.	0.	3.00		
n.	0.	0.71725E 02	0.27133F 02	0.	0.	3.15		
n.	0. -0.	n.72029E 02 0.69562E 02	0.73240F 01 -0.12331F 02	o. o.	o. - o.	5.30		
n. n.	-0.	0.63984F 02	-0.27050F 02	o.	- 0.	3.70		
n.	- n.	n.57747F 32	-0.44053E 02	0.	-0.	3.85		
n.	- ñ, - 0,	0.47970E 02 0.49792E 02	-0.47223F 02 -0.50481F 02	e. 0.	· 0.	5.70 5.70 5.80 6.00 6.03 6.08		
e.	-n.	C.43417E 02	-0.31139F 02	0.	-0.	6.06		
0.	- 0.	0.40814E 02 0.25167E 02	-0.76352E G2 -0.98879E G2	o. o.	-0. -0.	6,08		
0. -0.	-n. -n.	-0.32324E 02	-0.11754F 02	-0.	-0.	6.30 6,40 6,40 7,00 7,40		
-0.	e.	-0.15911E 03	0.57432F 02	-0.	0.	6.60		
••	0.	0.24374F 02 0.23203F 02	0.77446F 01 0.30191F 01	0. 0.	o. o.	7,00		
e. o.	0. -0.	0.71905E 02	-0.11644F 02	0.	-0.	1,20		
0.	-0,	0.111666 82	-0.12938E @2	•.	-•.	9,80		
-0.	e.	-0.15005E 01	0.	-0.	•.	10.00		

Toble XVIII Stress Frequency Response Functions (Analysis Condition 4)

(PSI/FPS SINUSOIDAL GUST)

10 CPS

CROSS WEIGHT: 107, 260 LB CUTOFF FREQUENCY:
ALTITUDE: 24,000 FT
MACH RUMBER: 9, 85

PERCENT SEMI SPAN: 27 SEGMENT NUMBER 19

	STRAK STRAKS					
			AXIAL STRESS			
eral	YRAKIDARI	REAL	IMAG I MARY			DECEMENT
				_	_	CPS
-0.15909F 0:	-0.63291F 01 -C.5:535E 01	0.20823E 02 0.13478E 03	0.92277E 02 0.76787E 02	-0. -0.	- 0. -0.	0.30
-0.1:74FE 02	-0.30347E 01	0.164845 03	0.462356 62	-0.	-0.	0.36
-0.128598 02	-0.12354E 01	0.194146 03	0-E8100F 02	*-0.	-a.	0.44
-n.11314E 02	0.126746 01	0.206145 03	-0.23857E 02	-0.	٥.	0.30
-0.12970F 02 -0.11845F 02	0.29736F 01 0.39679F 01	0.21A10E 03 0.21280F 03	-0.43495E 02 -0.95232F 02	-0. -0.	o. o.	0.70
-0.103736 02	0.476566 01	0.204118 03	-0.12245E 03	- ö .	0.	0.20
-0.272365 01	0.39136E 01	0.19329E 03	-0.14809E 03	- e.	0.	0.90
-7.70*68E 01 -0.37917F 01	0.98525E CC -0.38857E 01	0.18149E 23 0.15867E 03	-0.19880E 03 -0.24329E 03	-0. -0.	0.	1.00 1.30
-0.142725 01	-0.715648 01	0.14C71F 03	-0.26699E 03	-0.	ŏ.	1.34
-0.93969F 00	-0.1C716E 02	0.13148E 03	-0.29022F 03	-0.	0.	1.40
-0.4019£ 00 -0.4179£500	-0.12462E 02 -0.15335E 02	0.12222E 03 0.11677E 03	-0.30337F 03 -0.31774F 03	~0. -0.	0.	1.43
-0-41/4#1-00 	-0.212918 02	0.11056E 03	-0.35108E 03	-0.	•. •-	1.47 1.30
-0,103256 01	-0.292016 02	0.94919E 02	-0.392E3F #3	-0.	٥.	1.55
-0.778528 01	-0.39750E 02	0.72441E 02	-0.44331E 83	-0.	0.	1.50
-0.47236E 01 -0.73520E 02	-0.86952E 02 -0.12126E 02	0.37728E 0/ -0.31112E 03	-0.42765E 03 -0.22483E 03	-n. -0.	-0.	1.65 1.80
-0.17410F 03	0.9995AF 02	-0.758028 03	0.2100ef 03	-0.	0.	1.90
-0.121366 03	0.108598 03	-0.429918 03	0.18239F 03	-9.	٥.	2.00
-0.e1C41F 02 -0.31533E 02	0.10337E 03	-0.16373E 03 -0.48224E 02	0.163296 03 0.489876 02	-0. -0.	0. 0.	2.10
-0.104736 02	0.9174AE 02	-0.604436 02	0.367856 02	-∛:	o.	2.20 2.30
-0.616938 01	9.90334F 02	-0.539048 02	0.6424BE 01	0.	0.	2.95
-0.278968 01	0.49603E 02	-9,51216E 02	-0.90399E 01	0.	٥.	2.40
-0.81701E On -0.15629E-00	0.89266F 02 0.88577F 02	-n.57793E 02 -0.62866E 02	-0.64855E 02 -0.24242E 02	0.	0. 0.	2,43 2,44
0.73417F 00	0.893576 02	-0.036215 02	-0.14570F 02	ō.	0.	2.47
0.27300F-00	C. 94704E 02	-0.11070E 03	0.27374E 02	0.	٥.	ā.50
0.11002F 01 0.45924F 01	0.10006E 03	-0.11641E 03 -0.85736E 02	0.44526E 02 0.31185F 02	0.	0.	2.54
0.177486 02	0.107598 03	-0.509656 32	0.1548SF 02	š.	ŏ.	2.58 2.65
0.254885 02	0.11429E 03	-0.42208E 02	-0.94437E 01	0.	٥.	2.70
0.41496F 02 0.412275 02	0.13628E 03 0.15240E 03	-0.40590F 02 -0.63471E 02	-0.45468F 02 -0.39627F 02	0. 0.	0. 0.	2.80
0.117766 03	0.152400 03	-0.87237E 02	-0.700486 02	0.	å.	3.00 3.10
0.223956 03	0.16555E 03	-0.127448 03	-0.693348 02	0.	0.	3.20
0.71P66F 03 0.39747E 03	0.14692E 03 0.19024E 02	-0.16762E 03 -0.19707E 03	-0.41455E 02 -0.10730E 02	0. 0.	0. 0.	3.26
n. 487195 03	-0.35587E 03	-0.147072 03	0.12374F 03	0.	-0.	3-29 3-25
0.68269F 03	-0.49585E 03	-0.294048 83	0.14049 03	0.	-0.	3.40
-0.111726 03	-0.359816 03	-0.25656E 02	0.894186 02	-0.	-0.	3.52
-0.170356 03 -0.181636 03	-0.26391E 03 -0.12547E 03	-0.19929E 02 -0.30093E 02	0.55206F 02 0.14090E 02	-0. -0.	-0. -0.	3.56 3.60
-0.16011F 03	-0.305R7E 02	-9.76811E 02	0.34549E 02	-0.	-0.	3.70
-0.103718 03	-n.32449F 02	-9-17132E 63	0.21034E 03	-0.	-0.	3.85
-0.72707F 02	-0.38240E 02	-0.170425 03 -0.26585E 02	0.18421F 03 0.24454E 03	0. -0.	-0. 0.	4.00
-0.14764F 02	-0.47359E 01	0.110978 08	0.971736 02	-0.	-0.	4.30 4.50
0.101685 01	-0.11272F 02	0.17516F 03	0.534346 02	-0.	-0.	4.70
-0.56336E 01 -0.13357E 02	-n.:2461E 02 -n.11997F 02	0.15928E 03 0.13202E 03	6.14974E 0? 0.83307E 01	-0. -0.	-0. -0.	4.80
-0.14704E 07	-0.44890E 01	9.12577F 01	· 0. 10252E 0:	-0.	-0.	4.96 5.00
	-0.639666 01	0.104436 03	-6.210387 02	-0.	-0.	3.15
-0.19387E 02 -0.19757F 02	-0.25278E 01 0.10244E 01	0.90391E 02 0.73162E 02	-0.31439E 02 -0.59047E 92	∹ 0. −Ç.	o. o.	3.30
-0.1904#8 02	0.32590# 01	0.57948F 02	-0.428976 02	-0.	٥.	5.70
-0.18111E 02	0.53549F 01	0.481716 62	-0.499618 62	-••	0.	5.45
-0.16842F 07 -0.16599E 02	0.56892E 01 0.60161E 01	9.39484E 02 9.34848E 02	-0.46324E 02 -3.46722F 02	-0. -0.	o. o.	6.00 6.03
-0.14340E 07	0.63356F 01	0.35266E P	-0.470958 02	-0.	٥.	6.06
-0.14093F 02	0.76840E 01	3.336746 62	-0.48490F 02	-0.	0.	6.08
-0.14463E 02 -0.12552F 02	0.96035E 01 0.11162F 02	0.26630E 02 0.15195E 02	-0.4988E 02 -0.30242E 02	-8. -3.	0. 0.	6.20 6.40
-n.97964F nl	0.12479E 02	9.326246 01	-0.452248 02	-0.	0.	6.60
-0.1977CE 01	-0.11645E Q1	-0-27048E 02	0.146056 02	0.	-0.	7.00
0.87964F 01 -0.10658F 02	0.80864E 01 0.10488E 02	-9.60862E 02 -0.27075E 02	-0.21723E 02 0.11314E 02	0. -0.	g. 8.	7,40 8,20
-0.43411 00	0.517556 01	0.272986 02	-0.79705E 01	-0.	ŏ.	9.00
0.365700 01	a.	0. F1592E 01	0.	0.	0.	10.00

Teble XVIII - - - Continued

(PSI#PS SINUSOIDAL GUST)

GROSS WEIGHT: 107, 210 LB CUTOFF FREQUENCY: 10 CPS
ALTITUDE: 24, 600 FT
MACH NUMBER: 0, 85

PERCENT SEMI-SPANE 27 SEGMENT NUMBER 14

DECEMBERAL SERAN STREET		I PC I PORTA	. AKIAL STRESS			
PEAL	1MAG 3M6 RY	RE4L	1HAG 1 MARY			
						PRINCULARLY CPS
0.72144F 01	0.1473 36 02 0.114146 02	0.18 0946 02 0.122316 03	0.83739E 02 0.49682E 02	-0. -0.	-0. -0.	0.10 0.30
0.25767F n2	0.706277 01	0.14959€ 03	0.419476 02	-0.	-0.	0.36
1. 104115 02	0.25904E 01	0.1741RE 03	0.16432E 02	-0.	-0.	0.44
0.33054F 02 0.34907F 02	-0.45744E 01 -0.10980E 02	0.18890€ 03 0.19611€ 03	-0.23464E 32 -0.37620E 02	-0. -0.	0.	0.50 0.60
C. 140378 07	-0.34726F 02	0.193116 03	-0.86421E 02	-0.	0.	070
0.34347 02 0.33472 02	-0.22119E 02 -0.27456E 02	0.19523F 03 0.17541E 03	-0.11130F 03 -0.13439F 03	-0. -0.	0. 0.	0.80 0.90
0.724176 02	-0.407235 02	0.1448RF 03	-0.18040E 03	-0.	ŏ.	1.00
0.30405. US	-0.54477E 07	2.143998 03	-0.220706 33	-0.	0.	1.20
0.257676 02 0.251606 07	-0.6240cE 02 -0.70586F 02	0.12769F 03 0.11732E 0;	-0.2422AE 03 -0.26337E 03	-0. -9.	0.	1.34 1.40
0.257445 02	-0.74355E 02	0.110416 03	-0.27330F 03	-0.	6.	1.45
0.25767F 02 0.24532F 02	-n_2n673E 02 -0.93330F 02	0.10594E 03 0.10033E 03	-0.28834E C3 -0.31819E 03	-n. -0.	o. o.	1.47 1.50
11.209295 92	-0.109498 03	0. P6136t 02	-0.35505E 03	-0.	0.	1.55
1.14438F /2	-0.13038E 03 -0.21595E 03	0.65738F 02 0.34237E 02	-0.40229E 03 -0.56913E 03	-0. -0.	o. -o.	1.60
-3,128445 03	-0.62800E 03	-0.292345 03	-0.20403E 03	-0.	0.	1.63 1.80
-0.3209EE 03	0.138376 03	-0.4879RE 03	0.191346 03	-0.	0.	1.90
-0.206*7F 01 -0.577F0F 02	0.147/2E 03 0.12347F 03	-0.39013E 03 -0.16673E 03	0.16551E 03 0.10009E 03	-0. -0.	0.	2.00 2.10
-7.4874PF 07	0.10401F 03	-0.800416 02	0.44455E 02	6.	r.	2.20
-0.17644F 02	0.97219E 02 0.90713E 02	-0.54851E 02 -0.4891BF 02	0.20044E 02 0.50322E 01	-0. 0.	0. 0.	2.30
-0.1279CF 02	0.066736 02	-0.464776 02	-n.e2035E 01	0.	0.	2.55 2.40
-0.12297F 22	0.450406 02	-0.52445E 02	-0.13401E 02	0.	0.	2.43
-0.12F34E 02	0.42524E 02 0.86141E 02	-0.57050F 02 -0.7554E 02	-0.21999E 02 -0.13222E 02	o. o.	0.	2.44 2.47
-0.24121F 02	0.1027RE 03	-0.10044E 03	0.250226 02	ŏ.	0.	2.30
-0.2473:1 02 -0.11343F 02	0.11386F 0s	-0.17564E 03 -0.77803E 02	0.42221E 02 0.20300E 02	0.	0.	2.54 2.58
0.944956 01	0.11635E 03 0.11632F 03	-0.462426 02	0.14052E C2	o. o.	0. 0.	2.50 2.65
0.203656 02	0.114928 03	-0.343036 02	-0.05730[0]	0.	0.	2.70
0.390876 02 0.671756 02	0.13715E 03 0.15308E 03	-0.36834E 02 -0.47598E 02	-0.41261E 02 -0.54110E 02	o. n.	0. 0.	2,80 3-00
0.142155 03	0.16928F 03	-0.791458 02	-0.63567E 02	0.	9.	3.10
0.25676E 03 0.34236F 03	0.16548E 03	-0.11565E 03 -0.15211E 03	-0.62919E 02 -0.55769E 02	0. 0.	0. 0.	3.20
0.43105F 03	-0.23713E 01	-0.18045E 03	-0.97554E 01	0.	0.	3.26 3.29
0.64636F 91	-0.43264E 03	-0.244476 03	0.11412F 03	0.	-0.	3.35
0.75525E 03	-0.40781E 03	-0.76865E 03 -0.23287E 02	0.12767E 03 0.01145E 02	0. -0.	-0. -0.	3.40
-n.24054F 03	-0.337976 03	-0.180855 02	0.50090E 02	-0.	-0.	3.52 3.56 3.60
-0.24988 03	-0.1/6218 03	-0.27309E 02	0.127866 02	-0.	-0.	3.60
-0.25764F 03 -0.24177F 03	-0.35099F 02	-0,69704E 02 -0.19547E 03	0.33185E 02 0.19089E 03	-0. -0.	-0. -0.	3.70 3.85
-0.14433F 03	0.46440E 02	-n.15465E 03	0.1689RE 03	0.	-0.	4.00
-0.72 (IRE 02 0.161*75 02	0.10313F 03 0.24391F 02	-0.24126F 62 0.10071E 03	0.22373E 03 0.88183E 02	-0. -0.	0. -0.	4.20 4.50
0.49*407 02	0.643746 01	0.15897E 03	0.484975 02	-0.	-0.	4.70
0.355948 02	-0.34341E 0' -0.74064E 01	0.14454E 03 0.11981E 03	0.1350FE 02 0.73399E 01	-0. -0.	-ņ. -0.	4,80
0.21674F 02 0.18841F 02	-0.10495F 02	7.114136 03	-0.01901E 01	-0.	-0.	4.96 3.00
0.109745 02	-0.11584E 02	0.76769F 02	-0.19110E 02	-0.	-0.	5.15
0.52558F 01 0.12255F-00	-0.11670F 02 -0.11136F 02	0.82028E 02 0.66338E 02	-0.2870RE 02 -0.35433E 02	-o. -o.	0. 0.	5.30
-C.3645#F 01	-0.10566E 02	0.52586E 02	-0.3892ME 02	~0.	0.	5.70 5.85
-0.477321 01	-0.98800F 01	0.43714E 02 0.34623F 02	-0.414546 02	-0. -0.	0. 0.	5.85
-0.76938F 01	-0.97567E 01 -0.96323E 01	0.334575 02	-0.42038F 02 -0.42399F 02	-0.	0.	6.ග 6.03
-n. F2935F G1	-0.95072E 01	0,320036 02	-0.42738E 02	-0.	0.	6.06
-0.45862E 01	-0.89390E 01 -0.30079E 01	0.37559E 02	-0.44003E 02 -0.45272E 02	-0. -3.	0. 0.	6.0 6 6.20
-0.118485 02	-0.70756E 01	0.137896 02	-0.45411E 07	-0.	0.	6.40
-0.140128 02	-0.40812F 01 6.17795E 02	0.29406E 01 -0.24563E 02	-0.41042E 02 0.14797E 08	-0. 0.	0. -0.	6.60
-3.210001 02 -3.29564F 02	0.385936 01	-0.55230€ 02	-0.197156 07	ŏ.	Ģ.	7.00 7.40
-0.140345 02	0.17214F 02	-0.24570E 02	0.104496 02	-0.	0.	8.20
0.1.020F 02	0,53895E 01	0.24772E 02 0.74043E 01	-0.72330f 01	-0. 0.	0. 0.	9.00 10.00
					- •	3

Table XVIII - - - Continued

(PSIAPS SINUSOIDAL GUST)

10 CPS

GROSS WEIGHT: 187, 240 LB CUTOFF FREQUENCY: ALTITUDE: 24,000 FT

MACH NUMBER:

PERCENT SEMISPAN: 40,06 SEGMENT NUMBER 8

DECE THE THREE PAIL ANTAL STREET #F &L 18461442V ... IMAGINARY 0.10 0.85546 02 0.71353F 02 0.147316 02 0.147316 02 0.247246 02 0.0117876 03 0.0117876 03 0.112460 03 0.124612 03 0.259126 03 0.259126 03 0.259126 03 0.279126 03 0.308126 03 0.308126 03 0.379216 03 0.18733F 02 0.12404F 03 0.17487E 03 0.17487E 03 0.17487E 03 0.17487E 03 0.1947E 03 0.1947E 03 0.1967F 03 0.18067F 03 0.1948E 03 0.1974F 03 0.474778F 03 0.474778F 03 0.474778F 03 0.474778F 03 0.67521F 02 0.73914E 07 0.67521F 02 0.67521F 02 0.67521F 03 0.10764F 03 0.15232F 02 0.12847E 02 0.12847E 02 0.80364F 01 0.35452F 01 -0.35015F 01 -0.39015F 01 -0.79237F 01 -0.79237F 02 -0.7918 02 -0.79492F 02 -0.79492F 02 -0.79492F 02 -0.79467E 02 -0.98564E 02 -0.73794E 02 -0.73794E 02 -0.73794E 02 -0.73794F 03 -0.12733F 03 -0.12733F 03 -0.12733F 03 -0.12733F 03 -0.12733F 03 -0.12733F 03 -0.12733F 03 -0.12735F 02 0.79564F 02 0.795784F 02 0.795784F 02 0.795784 02 *22#eF -0. -0. -0. -0. -0. 0.71eApt 0.72eApt 0.3e70ep 0.3e70ep 0.3e70ep 0.3e70ep 0.3e70ep 0.3e70ep 0.310ep 0.310ep 0.310ep 0.79eApt 0.79eApt 0.79eApt 0.79eApt 0.79eApt 0.79eApt 0.1974ep 0.24983E 03 0.30134F 03 0.21384E 03 0.21384E 03 0.15909F 03 0.8332E 02 0.79183F 02 0.14944F 03 0.14945E 03 0.24552E 03 0.44446E 03 0.24552E 02 0.44411E 02 0.44711E 02 0.44711E 02 0.45173E 02 0.02173E 02 0.03217E 02 0.35334E 02 0.36271 0.42213E 02 0.19062F 02 0.19062F 02 0.19062F 03 0.19060F 03 0.17674F 03 0.39060F 03 0.39060F 03 0.79049F 03 -0.11595F 03 -0.11595F 03 -0.11595F 03 -0.10427F 03 -0.10427F 03 -0.40427F 03 -0.4055F 02 -0.40578F 02 -0.40578F 02 -0.40578F 02 -0.40578F 02 -0.40578F 02 -0.40578F 02 -0.41978F 02 -0.47586 02 -0.39078F 02 -0.47586 02 -0.39078F 02 0.84559E 02 0.15331E 03 0.13297E 03 0.13297E 03 0.1021IT 03 -0.26122E 02 -0.47247E 103 -0.14089E 03 -0.46495F 02 -0.20335E 02 -0.20335E 02 -0.22293E 02 -0.27797F 02 0.17769E 02 0.27797F 02 0.31535E 02 0.37767F 02 0.31535E 02 0.32038F 02 0.37464F 02 0.344316F 02 0.38432E 02 9.38197E 02 0.38432E 02 9.38197E 02 0.31545F 02 0.31545F 02 0.31545F 02 0.31545F 02 0.31545F 02 0.31545F 02 0.31545F 02 0.31545F 02 0.31545F 02 0.31545F 02 0.31545F 02 0.31545F 02 -0.41881E 03 -0.45628F 02 0.45528F 02 0.46533F 02 0.40112F 02 -0.14594E 01 -0.48718E 02 -0.74697E 02 -0.753741E 02 -0.13803E 03 0.20929F 03 0.13803E 03 0.13910F 03 0.13910F 03 0.13910F 03 0.13910F 03 0.14459E 02 0.42344E 02 0.42344E 02 0.43344E 02 0.43344E 02 0.43344E 02 0.42344F 02 0.4234F .85 6.00 6.05 6.06 6.40 6.40 6.40 7.00 7.40 8.20 9.00

Table XVIII --- Continued

(PSIFPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

107, 240 1.8 CUTOFF FREQUENCY: 24,000 FT 0, 85

16 CPS

PERCENT SEMISPAN. 40.00 SEGMENT NUMBER 107

STATEMENTAL SERVE & 1888		TIL PATETURE PROPE STREET				
RE SL	1#461%487			REGL	IMAGINARY	
						FRE, VENCY
0.336106-01	0.15610F 02	-0.	-0.	-0.1en0ef 02	-0.73110E 02	CP3
U-55113t US	0.13134F 02	-0.	-0. -0.	-0.104936 03	-0.40965 02	0.10
0.273236 02	0.419385 01	-0.	-0.	-0.130036 03	-0.36760F 02	0.30 0.36 0.4
0. 724145 02	0.76163F 01	- n.	0.	-0.15368E C3	-C-14295E 02	0.4
0.363375 02	-0.16025F 01	-0,	0.	-0.14519F 33	0.211256 02	0.56
0. 141 44 32	-0.99102E 01	-n. -o.	0.	-0.1720AE 03 -0.16982E 03	0.51781E 02 0.77914F 02	0.60
0.167155 02	-0.15441F 02 -0.20532F 02	-0.	o.	-0.163086 03	0.100716 03	0.70
0.350576 02	-0.256998 02	0.	-0.	-0.15443E 03	0.122025 03	0.80
0.147166 92	-0. 17847F 02	n.	-0.	-0.144542 03	0.144595 03	1.00
0.324245 02	-0.50427F 02	c.	-0.	-0.12547t 03	0.20175F 03	1.20
0.216466 05	-0.54066E 02	n.	-0.	-0.10971F 03	0.22141 03	1.34
0.10446 12	-0.55743F 02 -0.70205E 02	n. 2.	-0. -0.	-0.10151E 93 -0.93268E 02	0.24061F 03 0.24144F 03	1.40
0.251518 02	-0.751816 02	ń.	-0.	-0.88425F 02	0.26327 03	1.45 1.47
0.201751 02	-0.4702AF 02	ń.	-0.	-0.829248 02	0.290466 03	1.50
0.2*1571 02	-0.107165 03	0.	-0.	-0.491428 02	0.324015 03	1.55
0.10-07# 02	- 1.171 TOE 03	n.	-0.	-0.49534E 02	0. 365445 01	1.60
0.104758 07	-0.70311E 03 -0.66239E 02	n. -n.	-0. -0.	-0.19614E 02 0.27154F 03	0.508174 03	1.65
-0.24545 48	0.11852F 03	-0.	0.	0.427415 03	-0.17922E 03	1.80 1.90
-0.100A1E 01	0.12154F 03	-0.	0.	0.347716 03	-0.14711E 03	2.00
-6,040745 02	0.100015 03	-^.	0.	0.147778 93	-0.81537E 02	2.10
-0.41417E 02	0.00556F 02	-0.	0.	0.74033E 02	-0.30301E 02	2.20
-0.31373F 02	0.95104F 02 0.79195F 02	- n. - n.	o. o.	0.70353E 02 0.43154° 02	-0.32929E 02 -0.14364E 02	7,30
-0.14146 02	0.740315 02	•0.	o.	0.52445F 02	-0.15-18E 01	2.35 2.40
0.114578 02	0.719236 02	-0.	0.	C.54771E 02	0.33235E 01	2.45
0.1144EE 02	C.A7785F 02	-0.	0.	0.574925 02	0.11717# 02	2,44
-u"lessak us	0.49145F 02	ņ.	0.	0.709878 02	3.673995 01	2.47
-0.211046 02	^.41777F #2 ^.49F17E #2	9. 9.	o. o.	0.5885AE 02 0.91949E 02	-0.19647E 02 -0.29826E 02	2.50
-n.221015 02 -n.10276 02	0.447[76 02	o.	0.	0.711098 02	-0.1:349E 02	2.54 2.58
0.420684 0)	0.847256 02	o.	ŏ.	0.491685 02	-0.226938 01	2.58 2.65
0.142115 02	0.751715 02	n.	0.	0.451116 02	0.18603- 02	2.70
0.211005 02	0.707481 07	2.	0.	0.48820E 02	0.496945 02	2.80
0.434028 02	0.66091F 02	n. n.	6. 9.	0.80511E 02 0.11115E 03	0.420605 02 0.49474E 02	3.00
0.547247 02	0.5360#2 02	0.	0.	0.16398E 03	0.64237F 02	3.10
0 514901 97	0.494125 02	ć.	ō.	0.214126 03	0.503000 02	5.20 5.26
0.46435 02	0.481805 02	o.	0.	0.26110E 03	-0.27869E 02	5.29
0.415445 02	C. 74480F 02	n.	0.	0.357856 03	-0.23057E 03	5.35
0.141905 02	0.141676 03	٥.	0.	0.39986E 03	-0.25747E 03	5.40
U'ifitet U)	0.13442F 03 0.12434F 03	n. n.	0.	-0.23:35: 02	-0.184428 03	3.52
0.14*/36 03	6.96193F 02	ñ.	o. o.	-0.34759E 02 -0.34273F 02	-0.13593E 03 -0.754746 02	3.56
0.235445 03	0.294F1E 02	2.	-0.	0.124696 01	-0.661928 02	3.60 3.70
0.17 186 03	- 0. 40503F 03	n.	-0,	0.587136 02	-0.12769F 03	3.85
U. Sialti us	-0.26215F 03	ņ.	-0.	0.638235 02	-0.12694E 03	4.00
50 350112.0-	-0.14019E 03 -0.46960F 02	-0. -0.	-0. -0.	0.4591AE 02 -0.11794E 03	-0.243965 03 -0.40969E 02	4.20
-0.121901 03	0.42841F 02	-0.	-0.	-0.17445 03	-0. 3A203F 02	4.50 4.70
10.111425 31	0.184605 92	-0,	٠٥.	-0.15*00E 93	-0.3646ZE 01	4.80
0.102516-03	-C.11874E 07	-0.	-0.	-0.12590E 03	0.17233F 01	4.96
on series no	0.12424F 01	-0.	-9.	-0.1i#85# 03	0.147206 02	5.00
0.047CH H2	0.80714F 01	- n. - n.	٥.	-0.97954E 02 -0.81136E 02	0.223635 02	5.15
0.145245 02	0.226077 02	-6.	o. o.	-0.44534F 02	9.27655F 02 9.36199E 02	5.30
.0. 441 9E 02	0-2616NE 02	-9.	o.	-0.31475E 02	0.309196 02	550
-0.444481 07	0.240016 02	- 0.	0.	-0.439398 02	0.310276 02	5.70 5.85
-0.477541 07	0.274278 02	2.	0.	-0.372246 02	0.309916 02	6.00
0.412325 03	7.74837F 02 7.30231F 02	-n. -n.	0.	-0.341 AGE 02	0.30940€ 02	6.05
-0.40370F 07	7.102317 07 7.317747 02	-0.	0.	-0.35167F 02 -0.34183E 02	0.30476F 02	6.06
-0.364738 02	0.338896 02	- 0.	0.	-0.301056 02	0. 202226 02	6.08
0.277101 02	0.3531#6 02	-0.	n.	-0.245396 02	0.275516 02	6.40
-0.207285 02	0.352776 02	-0.	0.	-0.20366E 02	0.242998 62	6.60
-0.4A724F N	0.14660 02	ņ.	0.	-0.17809€ 02	0.304945 02	7.00
0.1 5107 07	0.25496F 02 0.15497F 02	n. n.	o.	-0.24254E 02 0.22354F 02	0.22496F 02 0.80355F 01	7.40
0.44444 00	5.13001C 02	- 0	ô.	0.104018 02	7. 229026 01	1.20
0.111445 02	0,	ń.	o.	0.172416 02	ð.	9.00 10.00
						20.00

Table XVIII --- Continued

(PSI/FPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

107, 260 LB CUTOFF FREQUENCY: 24, 000 FT 0. 85

10 CPS

BODY BALANCE STATION: 540 SEGMENT NUMBER 17

INCR IMENTA	L SEEAR STREES					
PFAL	IMAGINARY					
•	•					HECTORY
-n.1#744F-01	-0.56730E 00	0.	0.	0.	0.	0.10
-0.10561F 01	-0.41030F-00	0.	0.	0.	0.	0.30
-0.124785 01 -0.13875E 01	-0.150136-00 0.813596-01	9. 0.	0. -0.	0. 0.	-0.	0.36
-0.14238F 01	0.35005E-00	0.	-0.	0.	-0.	0.90
-0.13790F 01	0.555785 00	0.	-0.	0.	-0.	0.60
-0.1273/F 01 -0.11541F 01	0.69612F 00 0.F9702F 00	0.	-0. -0.	∪. 0.	-n. -0.	0.70
-0.104315 01	0.863076 20	n. 0.	-0.	ŏ:	-0.	0.80
-0.54720F 00	0.979627 00	o.	-0.	0.	-0.	1.00
-0.41048F 00	0.108198 01	3.	-0.	2.	-0.	1.20
-0.74£40F 00 -0.72374F 00	0.11396F 01 0.11962F 01	r. 0.	-0. -0.	o. o.	-0. -0.	1.34
-0.70505F 00	0.123228 01	n.	-0.	ŏ.	-0.	1.40 1.45
-0.6951 EF 00	0.126975 01	0.	-0.	0.	-0.	1.47
-0.6F452F 00 -0.6F##3F 00	0.13594F 01 0.14733E 01	0.	-0.	0.	-0.	1.50
-0.421775 00	0.14709F 01	n. 0.	-0. -0.	o. o.	-0.	1.55
-0.46095F 00	0.22900E 01	0.	-0.	0.	-0.	1.65
0-21:12F-00	0.15409F 01	-0-	-0.	-0.	-0.	1.80
0.15780F 01 0.95722F 00	0.25628F-00 0.20042E-00	-0. -0.	-0. -0.	-0. -0.	-0. -0,	1.90
0.34589F-00	0.357246-00	-0.	-0.	-0.	-0.	2.00 2.10
0.45755F-01	0.55122F 00	-0.	-0.	-0.	-0.	5.50
-0.57279F-01	0.59644F 00	-0.	- <u>o</u> .	-o.	-0.	2.30
-0.70477F-01 -0.9272FF-01	0.68965F 00	-n. 0.	-0. -0.	-0. 0.	-0. -0.	2.35
-0.69794F-01	0.78576E 00	-0.	-0.	-0.	-0.	2.40 2.43
-0.47820E-01	0.83524E 00	-).	-0.	-0.	-0.	2.44
0.503285-01	9.79130F 00	-0.	-0.	-0.	-0.	2.47
0.19333F-00 0.24186F-00	0.55898E 00 0.42460F-00	-n. -0.	-0. -0.	-0. -0.	-0. -0.	2,50 2,54 2,58
0.77424E-01	0.46483E-00	-0.	-0.	-ŏ.	- 0.	2.44
-0.15676F-00	0.53391E 00	0.	-0.	o.	-0.	2.65
-0.25236F-00 -0.37805F-00	0.65558F 00	r. 0.	-0, -0.	o. o.	-0. -0.	2.70
-0.68174F GO	0.89972E 00	0.	-0.	ŏ.	-0.	2.80 3.00
-0.10167F 01	0.99665F 00	0.	-0.	ö.	-0.	3.10
-0.17704E 01	0.127756 01	0.	-0.	o.	-0.	3.20
-0.27196E 01 -0.35425F 01	0.14811F 01 0.34340F 01	٥.	-0. -0.	o. o.	-n. -o.	3.26
-C.56787F 01	0.849246 01	G.	-0.	ŏ.	-0.	3.29
-0.45616F 01	G. IORALE 02	0.	-0.	0.	-0.	3.39 3.40
0.50602F 01 0.74136F 01	0.92636F 01 0.81376F 01	-0. -0.	+0. -0.	-0.	-0.	3.52 3.56 3.60
0.727315 01	0.62744E 01	-0.	-0.	-0. -0.	-0. -0.	3.56
0,54272F 01	0.12151F 01	-0.	-0.	-0.	-0.	3.70
0.144896 02	-0.16700F 02	-0.	0.	-0.	0.	3.85
0.17309F 02 -0.88812F 01	-n.74417E 01 0.54045E 01	-0. 0.	o. -o.	-0. 0.	-0.	4.00
-0.403956-00	0.193676-60	o.	-0.	ŏ.	-0.	4.20 4.50
0.25759F 01	-0.82772E 00	-0.	0.	-0.	0.	4.70
0.1942#E 01 0.10770F 01	-0.13196E 01 -0.13460E 01	-0.	0.	-0.	۹.	4,80
0.107707 01	-0.13141E 01	-n. -o.	o. o.	-0. -0.	o. c.	4.96 5.00
-00-18713F-00	-0.11903F 01	-0.	ŏ.	-0.	õ.	5.15
0.175436-00	-0.10032E 01	-0,	0.	-0.	0.	5.36
-0.40446F-01 -0.15725F-00	-0.81876E 00 -0.69857E 00	-n. 0.	0. 0.	-0.	0.	10
-0.19890F-00	-0.58294F 00	0.	ŏ.	0. 0.	o. o.	5.70 5.85
-0.21508F-00	-0.56423F 00	0.	٥.	ō.	ŏ.	6.00
-0.21537E-00	-0.545#8E 00 -0.5278#F 00	0.	0.	0.	0.	6.03
-0.21497F-00 -0.21390F-00	-0.5278#F 00 -0.45135E-00	n. 0.	o. o.	o. o.	o. o.	6.06 6.08
-0.20107F-00	-n. 14159F-00	0.	ŏ.	ŏ:	ö.	6.20
-0.15221F-00	-0.25352F-00	0.	0.	0.	0.	6.40
-0.65765E-01 0.29069E-00	-0.18955F-00 -0.12628E 01	0. -0.	8. 0.	0.	0.	6.60
0.10407F 01	-0.126286 01 -0.81816F 00	-0.	0. 0.	-0. -0.	0.	7.00 7.40
-0.76913F 01	0.93376F 00	0.	-0.	ŏ.	-0.	8.30
0.889625 00	-0.45075E- 01	-0.	-0.	-0.	-0.	9.00
0.453906-00	0.	-0.	••,	-0.	0.	30,00

Teble XVIII - - - Concluded

(PSIFFPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: 107, 260 LB CUTOFF FREQUENCY:

10 CPS

ALTITUDE: MACH NUMBER:

24,000 FT

BODY BALANCE STATION: 820 SEGMENT NUMBER 1

I HAG I MARY REAL C78 0.10 0.30 0.34 0.50 0.60 0.60 0.60 0.60 0.60 1.00 1.00 1.00 1.00 1.00 1.00 1.00 1.00 1.00 1.00 1.00 2.10 2.10 2.10 2.10 2.10 2.10 2.10 2.10 2.10 3 0.12998E 02
0.76771E 02
0.92098E 02
0.10579E 03
0.10579E 03
0.1117E 03
0.1117E 03
0.11179E 03
0.11179E 03
0.10139E 03
0.40472E 02
0.7649E 02
0.7649E 02
0.62076E 02
0.62076E 02
0.62076E 02
0.62076E 02
0.64798E 02
0.62076E 03
0.6479E 02
0.7649E 02
0.7649E 02
0.7649E 02
0.7649E 02
0.7649E 02
0.7649E 02
0.77449E 02
0.7749E 01
0.11477E 03
0.11467E 03
0.11467E 02
0.1367E 03
0.1367E 03
0.1367E 03
0.1367E 03
0.1367E 03
0.1367E 03
0.1367E 03
0.1367E 03
0.1367E 03
0.1367E 03
0.1367E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.1378E 03
0.37330F 02
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03
0.34743F 03 0.44903E 02
0.34992E 02
0.38992E 02
0.39799E 02
-0.37300E 02
-0.37300E 02
-0.37300E 02
-0.3730E 02
-0.77393E 02
-0.1644E 03
-0.12473E 03
-0.12473E 03
-0.12473E 03
-0.12473E 03
-0.1345E 03
-0.1345E 03
-0.17771E 03
-0.14007E 03
-0.17771E 03
-0.2404E 03
-0.17771E 03
-0.2404E 03
-0.27002E 02
-0.39338E 01
-0.27002E 02
-0.39338E 01
-0.27002E 02
-0.39338E 01
-0.27002E 02
-0.39338E 01
-0.27002E 02
-0.39338E 02
-0.39338E 02
-0.39338E 02
-0.17745E 03
-0.17178E 03
-0.1818E 03
-0.11734E 03
-0.11734E 03
-0.11734E 03
-0.11734E 03
-0.11734E 03
-0.11734E 03
-0.11734E 03
-0.12441E 03
-0.11734E 03
-0.12441E 03
-0.12441E 03
-0.12441E 03
-0.13931E 03
-0.22324F 03
0.21898E 03
-0.22324F 03
0.21898E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.22324F 03
0.39207E 03
-0.39207E 03

Table XIX Stress Frequency Response Functions (Analysis Condition 5)

(PSI/FPS SINUSOIDAL GUST)

10 CPS

GROSS WEIGHT: 297,000 LB CUTOFF FREQUENCY: ALTITUDE: 24,000 FT MACH NUMBER: 0,50

PERCENT SEMI SPAN: 27 SEGMENT NUMBER 10

DCIMINAL MAL	SEEAR STREET	INCHIDENTAL	. AXIAL STREET			
efal	1 MASS NAKY	AFAL	IMEGINAPY			DEVICT
						CPS
7.566285 00	-0.67701E 00	-4.771316 01	0.56303F 01	0.	- 0.	5.10
7.3414F 02	2.91346F 31 -0.17775F 02	0.319276 35	-0.73260F 02 -0.10975F 03	-0. -0.	٥. ٥.	0.30
7.24242F 02	-0.1430. 02	0.26271F 03 0.21174F 03	-0.10475E 03	-0.	0.	0.56 0.44
1.29474F 02	-0.163221 02	1.1F712F 03	-0.14717F 03	- 0 .	0.	0.50
1.17470F 07	-0.14507E 02	0 181906 03	-C.16944F 03	-0.	0.	0.60 0.70
7.15967E 02	-0.21278F 02 -0.24989E 02	7.14454E 33 0.12494E 03	-0,19659E 03 -0,23148E 03	-0. -0.	0. 0.	0.86
1.13110F 02	- 0. 1051 AF 02	7.11527 6 03	-0.20129E 03	-0.	č.	0.90
3.11550F 02	-0.55160F 02	0.95733€ 02	-0.48798F 03	-0.	0.	1.00
1.27358F 01 1.53650F 02	-0.10740F 03 -0.67373F 02	0.15277E 01 -0.53971E 03	-0.97974F 03 -0.48338F 03	o. o.	o. -o.	1.34
1.1-7446 03	0.14663F 02	-1.141635 04	0.570926 03	0.	-0.	1,40
1.14104F 01	0.86052F 02	-7.11797E 04	0.712326 03	-0.	-0.	1.45
7.10705F 64 7.73339 62	0.47309F 02 0.74150F 02	-n,#1032E 03 -1,53774E 03	G.M9746F 03 C.56075F 03	-n. -n.	o. o.	1.47 1.50
3.311928 02	0.41303F 0Z	-1,241736 03	0.439936 03	-0.	0.	1.55
1.174/9E 07	0.418946 02	-0-11122F 03	0.34272F 03	-0.,	0.	1.60
1.7978AE 01	0.345AZE 02	-0.49315E 02	0.20549F 03	-0.	n.	1.65 1.80
1.579606 07 1.102536 02	0.31711f 02 0.29268f 02	0.23134F 42 0.34651F 02	0.15407F 03 0.11825F 03	o. o.	0. 0.	1.90
1.155445 02	0.279146 02	0.43074E 02	0.92340F 02	ŏ.	o.	2.00
1.27744F 01	0.265176 02	0.418i1F 02	0.724406 02	0.	0.	2.10
1.34408F 02	0.94275E 01 0.45630E 01	0.40479£ 02 0.34447£ 02	0.56396F 02 0.51100F 02	0. 0.	0. 0.	2.20 2.30
1.54134F 02	- 0. 43307F 01	0.274186 32	0.421376 02	ů.	-0.	2.55
3. FROPAR 02	-0.19572F 02	0.173876 02	0.60084F 02	0.	-0.	2.40
1. 7613AF 07	-0.245796 02	1.17239E 01	0.65774F 02	0.	-0.	2.45 2.44
1.76535F 02	-0.45843F 02 -0.51147F 02	0.12245F 02 0.21597E 02	0.77513F 02 0.77895F 02	o. o.	-0. -0.	2.47
1.57580F 02	-0.57243F 02	0.379146 02	0.644246 02	ŏ.	-0.	2.50
1.604586 07	-0.49034F U?	0.506176 92	0.427225 02	0.	-0.	2.54 2.58
1.76453F 11?	-0.138675 07 -0.765945 02	7.53041E 02 0.46435E 02	0.42978E 02 0.4148F 02	0. -0.	-0. -0.	2.56
1.78343f C2	-0.193646 02	0.472786 02	0.34505F 02	-0.	-0.	2.70
1.633795 6.7	O. FRERE D)	0.500126 02	0.23272F 02	-0.	0.	2,80
1.43523E 02	0.11793F 02	0.55563E 02	0.17957E 02	-0.	0.	3.00 5.10
1.36429F 02 1.14364F 0	0.176825 02 0.280345 02	7.53675F 92 0.83036F 92	0.39528F 01 -0.28418F 02	-0. -0.	0. 0.	5.20
1.443078 02	0.41799F 02	977956 02	-0-40647F 02	- 0.	0.	3%
1.465136 02	0.65518F 02	(. 110E 02	-0.792206 02	-0.	0.	5.29
1,290386 92	0.467715 02	0.32795E 02 2.55093E 01	-0.55299F 02 -0.22451F 02	-0. -0.	o. o.	5.35 5.40
1.925; IF 02	-0.13010f 02	0.15965 31	-0.192278 02	0.	-0.	3.52
1.87712F 02	-0.3704AF 02	0.512576 01	-0.17045E 05	0.	-0.	3.52 5.56 3.60
1.41373F 02 1.75382F 01	-0.269625 02 -0.120346 02	7.74714F 01 0.89119E 01	-0.16233F 02 -0.16208F 32	-0.	-0. -0.	5.60
1.444835 01	-0.481946 01	0.457796 01	-0.172496 02	-0.	-0.	5.85
1.647775 11	-4.43381F-11	3.774746 01	-0.19498F 02	-0.	0.	₹.00
1.617657 01	0.145046 01	0.158778 01	-0.24586F 02	-0.	0.	4.20
1_42946F 01 1,26847F 01	0.45577F 01 0.45974F 01	-0.66257F 01 -0.24182F 02	-0.20632F 02 -0.26476F 02	-0. 0.	o.	4.50 4.70
1.171876 01	0.43015E 01	-0.439736 02	0.47680f 02	õ.	-0.	4.80
1. #COTSF-01	0.393816 01	-0.776956 02	0.74716E 02	0.	-0.	4.96
1.17928E-00	0.43676F 01 0.46820F 01	-(1.49663£ 32 3.19753£ 02	0.47178F 02 0.21278F 02	0. -0.	- n. - 0.	5.00 5.15
1.71562F OR	0.456396 01	0.243176 02	0.81841F 01	-0.	0.	5.30
1_21844F 07	0.390011 01	0.223431 02	0.12734F 01	-0.	0.	100
1.621325 01	0.25343F 01 -0.364970 01	0.272946 02	-0.42829F 01 -0.23540F 02	-0. -0.	o. o.	5,70
1. 88948F O7	-0.60269E 01	0.30341f 02	-0.30782F 02	-0.	0.	5.85 6.00
1,831895 01	- 0.8497 OF 01	0.242375 02	-0.38291F 02	0.	0.	6.05
1.44789F 01	-0.1010AF 02 -0.2917GF 01	0.22464E 02 0.12595F 02	-0.43221F 02 -0.21241F 02	o. -o.	- 0.	6.06
1.4789AF 01	0.171505 01	-0.1445AE 02	-0.21201F 02	-0.	o. o.	4.0\$ 6.20
1.170A2E 01	0.209886 01	-0.77317E 01	-6.383215 01	-0.	0.	6.40
1,554245-07	0.153311 01	-0.445345 01	-0.26367F 01	0.	0.	6.60
1.14117F 01 1.18612F 01	0.76594F 00 -0.48085F-00	-0.29772F 01 -0.2622 9 F 01	-C.14298F 01	o.	0. -0.	7,00
1-164925 03	- 0.10970F 01	-0.20482F 01	0.83313F 00	o.	- 0.	7.50
1.97928F BO	-0.11066F 01	-0.10053E 91	0.001007 00	٥.	-0.	9.00
1.163066-00	0.	-0.167796-60	0.	-6,	0.	10.00

Table XIX --- Continued

(PSIFPS SINUSOIDAL GUSTI

GROSS WEIGHT: 297, 000 LB CUTOFF FREQUENCY: 10 CPS ALTITUDE: 24,000 FT MACH NUMBER: 0.50

PERCENT SEMI-SPANE 27 SEGMENT NUMBER 14

DESIGNAL MARK STREET			L ATTAL STREET			
	14021 8004	4F4L	144014444			PRELEDICY
						CPS
-0.21433F 01 0.10416F 03	0.26396F-00 -0.25679F-02	-0.49995F 01 0.28954F 01	9.51093F 01	0.	-0.	0.10
0.4476AF 07	-0.371845 02	0.214415 03	-0.56482F 02 -0. 99 592F 02	-0. -0.	0.	0.36
O. TO SOME OZ	-0.420917 02	C-19234E 01	-0-11443F 03	-0.	0.	0.44
0.42353F 02	-0.48514F DZ	0.159816 03	-0.13354F 03	-0.	0.	0.50
0.53443F 62	-0.5529AF 02	0.146928 03	-0.19376F 03	-0.	٥.	0.60
0.68263F 02 0.63655F 02	-0.43688F 02	0.13125E 01 0.11792E 01	-0.17840F 03 -0.21004F 03	-0. -0.	0. 0.	0. 40 0. 8 0
0.197617 07	- C. 90577F 07	0.104636 03	-0.25527F 03	-0.	0.	0.90
0.3399AF 02	-0.15375F 03	0.877635 02	-0.44283F 03	-0.	o.	1.00
0.575745 01	-0.29304F 03	0.13854F 31	-0.79834F 03	0.	0.	1.20
-0.14691F 73 -0.46096E 03	-n.17307F 03	-n. 48777F 03	-0.43465F 03	0.	-0.	1.36 1.40
-0.34533F 01	0-17790F 03 0-2294AF 03	-0.12671F 04 -0.13794F 04	0.51810F 03	0. -0.	-0. -0.	1.45
-0.27505F 03	0.279276 03	-0.73577# 03	0.632936 01	-0.	o.	1.67
-0.1449% n3	0.197508 03	-0.48917F 31	0.53887F 03	-0.	0.	1.50
-0.84974F 02	0,149A3F 03	-0.21935F 03	0.39923F 03	-0.	0.	1.55
-0.40327F 02 -0.17441F 02	0.12743F 03 0.76512F 02	-0.10255F 03 -0.44753F 02	0.32608F 03	-9. -0.	o. o.	1.60 1.65
0.104915 02	0. A0980F 02	0.18244F 02	0.13981F 03	-0.	0.	1.80
0.188028 02	0.506218 02	0.304016 02	0.10731# 03	0.	0.	1.90
0.257245 02	0.43500F 02	7.34357F OZ	0.63605F 02	0.	0.	2.00
0.335518 02	0.37342F N2	0.37942F 02	0.45/386 02	0.	0.	2.10
0.45FR7F 02	0.1569 \F 02 0.42765F 01	0.35733F 02 0.31762F 02	0.91178F 02	G.	0. 0.	2.20 2.30
0.635856 02	-0.435017 00	0.252446 02	0.47313F 02	ŏ.	-0.	2.35
0.77034F 02	-0.151 RAF 02	7.157758 02	0.545255 02	0.	-0.	2.40
0.#3654E 02	-0.234204 02	0.11136F 02	0.59489F 02	0.	-0.	2.43 2.44
0.940882 02 0.777796 02	-0.18897F 02	0-111176 02	0.70341F 02	0.	-0-	2.44
0.716235 02	-0.546616 02	0.19579F 02 0.34410F 02	0.70688F 02 0.59370F 02	o. o.	-0. -0.	2.47 2. 3 0
0.794285 02	-0.994295 02	0.451156 02	0.478446 02	o.	-0.	2.56
0.87134F 02	-0.16013F 03	0.481346 02	0.39002F 02	0.	-0.	2.56 2.58
-0.369345 02	- N. FR54RE 02	0.421378 02	0.37650F 02	-0.	-0.	2.69
-0.96276E 02 -0.70064E 02	-7.24379F 02 0.57482F 00	0.42934F 02 0.45355F 02	0.31316F 02 0.21118F 02	-0. -0.	-0. 0.	2.70 2.80
-0.50714F 02	6.54294F 01	0.504228 02	0.162965 02	-0.	0.	3.00
-0.54192F 02	0.174345 62	0.472946 02	0.35871 01	-0.	0.	3.10
-0.73360F 02	0.5051AF 02	0.75355F 02	-0.25749F 02	-0.	0.	3.26
-0.44906F 9.	0.89906F 62 0.11375F 03	0.88747F 02 0.90856F 02	-0.55036F 02 -0.71709F 02	-0. -0.	0.	3.86
-n.32908F 02	n.12794F 03	0.297536 02	-0.53183F 02	-0.	ŏ.	3.29 1.35
0.153676 02	0.677275 02	0.50921f 01	-0.20374F 0?	-0.	0.	3.25 3.46
6.13261F 03	-n. 69254F 01	0.144838 01	-0.16540F 02	0.	-0.	3.52
0.12549F 03 0.81114F 02	-0.38263E 02 -0.31244E 02	0.46543E 01 3.67401E 01	-0.15510F 02 -0.14731F 02	0. 0.	-0. -0.	3.56
0.262736 07	-0.17925F 02	0.406735 01	-0.14708F 02	-0.	-O.).60 1.70
0.864975 01	-n.1246/F 07	7.77853F 01	-0.15653F 02	-0.	-0.	3.52 3.56 3.60 3.70 3.85
0.323016 01	- 0. 4540 UE 01	0.619548 01	-0.17694F 02	-0.	0.	
-0.23706F-00 -0.58186F-01	-0.1045RF 02 -0.12122F 02	1.12551F 01 -2.60127F 01	-0.22311F 02 -0.25983F 02	-0.	٥.	4.20 4.30 4.70 4.60
-0.153055 02	-0.10692F 02	-0.219456 02	-0.24024F 02	-0. 0.	0. 0.	1.30
-0.264376 02	0.376966 02	-0.399078 02	0.432696 02	ŏ.	-0-	3.66
-0.4545]# 02	0.48479F 02	-n.tosotf 02	0.67803F 02	0.	-0.	4.96
-0.289111 02	0. 12324F 02	-0.45069E 02 0.17925E 02	0.42413F 02	٥٠	-0.	4.96 5.00
0.12220F 02	0.17012F 02	0.220472.02	0.19309F 02 0.74269F 01	-0. -0.	-0.	5.15
0.151036 02	0.438658 01	0.707946 02	0.11554F 01	-0.	0.	5.15 5.30 £.50 5.70
0.159001 02	0.458E5F-00	0.20231# 02	-0-38866F 01	-0.	0.	5.70
0.184198 02	-0.13037/ 02	0.77901F 02	-0.21 362F 02	-0.	0.	5.89
0.22433F 02 0.21033F 02	-0.18089F 02 -0.23339F 02	0.27573F 02 0.25674F 02	-0.27934F 02 -0.34749F 02	ð.	0.	6.00
0.171236 62	-0.269206 02	0.203056 02	-0.39222F 02	0. 0.	0. -0.	6.03
0.103735 02	-0.11940E 02	0.114336 02	-0.19294F 02	~0.	0.	6.06 6.09
-0.41443F 01	-0.19474F 01	-0.13120E 02	-0.57002F 01	-0.	0.	5.20
-0.32929E 01 -0.92626F 00	-0.63878F 00 -0.42853F-00	+0.70141F 31 -0.41389F 01	-0.34775F 01 -0.22108F 31	-0. 0.	0.	6.40
0.353546.00	- 0. 464 00E - 00	-0.26501F 01	-0.13892F 01	0.	0.	6.60
0.496926-00	-n.12562F-00	-0.239726 01	0.3394RF-01	ë.	٠٥.	7.00 7.40
0. 1705 1F- 00	- 0.13n41E-nn	-n.18758F 31	0.75405F 90	0.	-0.	8.20
0.41146/-00	-0.70479F-00	-n.98485F 00	0.726897 00	0.	-0.	9.00
0.69694F 00	0.	-0.152246-00	0.	~0•	0.	10.co

Table XIX - - - Continued

(PSI/FPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER: 297, 000 LB 24, 000 FT 0, 50 CUTOFF FREQUENCY: 10 CPS

PERCENT SEMISPAN: 40.06 SEGMENT NUMBER 8

DETERMINATION OF THE PERSON OF	-	DETROOPS	L AILIAL STREET			
PEAL	1 MAG 1 MARY	OF AL	1MAG1MARY			5.0
						PAGE RESIGN
-0.147105 01	0. 524936-01	-0.901545 01	0.63507F 01	-0.	-0.	0.35
0.75289F 02	-n.14871E 02	P. 10784E 03	-0.73698E 02	e.	-0.	0.30
0.***64F 07	-0.24492E C2 -0.27519E 02	0.25355F 03 0.20492F 03	-0.10686E 87 -0.12303E 03	o.	-0. -0.	0,35 0,44
0.469475 07	-0.32354F 02	0.149745 03	-0.14436F 03	0.	-0.	0.30
0.15507E 02	-0.37747F 02	9.15593F 03	-0.105527 03	0.	-0.	0.60
0.32002F 07 0.29147F 02	-0.43414E 02 -2.51583E 02	0.13857F 03 0.12372F 23	-0.19334E 03 -0.22759F 03	0. 0.	- 0. - 0.	0.70 0.80
0.241475 02	-0.435476 02	0.104516.03	-0.27625E 03	0.	-0.	0.90
0.228556 02	-0.11541E 03	0.894756 02	-0.47702E 03	0.	-0-	1,00
0.23863F 01 -0.13033F 03	-0.22054F 03 -0.12834F 03	-0.4#694F <i>9</i> } -7.43777F <i>0</i> 3	-0.85297F 03 -0.45666F 03	0.	-0. -0.	1.30
-0.349006 03	0.144716 01	-n.13978E 04	0.56884E 03	-0.	0.	1.74 1.40
-1.10972F 01	0.195346 03	-7.11439E 04	0.70268F 03	-0.	0.	1.95
-n.21585E 03 -n.14569E 03	0.18493E 03 0.18397E 03	-0.79479E 03 -3.52204E 03	0.68524E 03 0.54881E 03	-0. -0.	0. 0.	1.47
-0-47565F 02	0.125006 03	-7, 23513E 03	0.42975F 03	-0.	ŏ.	1.56 1.55 1.60
-0.4740RE 07	0.104016-03	-9.11146E 03	0.344216 03	-0.	•.	1.60
-0.14189E 02	0.69042F 02	-7.50824E 02 0.13503E 02	0.20049E 03	-0. 0.	o.	1.65 1.80
0.159796 07	0.498278 02	2.252436 02	0.11684F 03	0.	0.	1.90
0.221076 02	0.447046 02	0.293718 02	0.924638 02	0.	0.	2.00
0.28696F 02	0.40A33F 02	7.27421E 02 0.19547E 02).74897E 02 0.80925E 02	0. 3	o. o.	2.10 2 .20
0.507096 02	0.299476 02	7.509136 01	0.791156 02	ő.	2.	2.30
0.04946 02	0.17819F 02	C-13479E 02	0.76369E 02	0.	•.	2.50
0.75666F 07	7.29372E 01 -0.54733E 01	7.44774E 01 3.15951E 01	0.82839F 02 0.87641F 02	o. o.	-0. -0.	2.40 2.65
0.825875 02	-0.21377F 02	2,148376 01	0.973116 02	0.	-0.	2.44
0.74838F 02	-n.76766F 02	0.813718 61	0.98473E 02	0-	-0.	2.47
0.65045E 02 0.66533E 02	-0.30417E 02 -0.55500E 02	0.180ASE 02 0.21229E 02	0.95793F 02 0.10511E 03	o. o.	-0. -0.	2.50
0.7111AF 07	-n. 921 97E 02	0.19646 03	0-125076 03	o.	-0.	2,90 2,94 2,98
-0.86452F C1	- 0.42873F 02	0.66269F 02	0.944396 02	-0.	- 0.	2.65
-0.38193F 02	0.14516E 01 0.20395E 02	n. 37731E 02 C-#5771E 02	0.65462E G2 0.52252E G2	-o. -o.	-0. 0.	2.70
0.165516 07	0.215426 02	0.10034 £ 03	0.47167E 02	0.	0.	3.60
0.472215 07	0.34797E 01	1.12A72E 03	0.14235F 02	0.	0-	3.10
0.91730F 02 0.13273E 03	-0.58302F 62 -0.12692E 03	0.19547E 03 0.24814E 03	-0.82723E 02 -0.18378E 03	o. o.	-0. -0.	5.20 3.26
0.173266 03	-0.1794 PF 03	0.724776 03	-0.24092E 03	0.	-0.	1.29
0.103426 02	-0.14005E 01 -0.5776#F 02	n. 47543E 02	-0.16023E 03	o. -o.	-0.	3.35 3.40
-0.54359E 02 -0.12658E 03	- 0. F1974F G1	-0.35617E 02 -0.25153E 02	-0.57533F 02 -0.55510E 02	-0.	-0. -0.	5,40 1,40
-0.11277F 03	0.14323F 02	-0.14941F 02	-0.42328E 02	-0.	0.	3.52 3.56 3.60
-0.84540E 92	0.18320F 02 0.16591E 02	-0.23957F 02 -1.22794F 02	-0.36603E 02 -0.22059E 02	-0. -0.	0.	3.60
-0.49987E 07	0.165416 02 0.176346 02	-0.189196 02	-0.13689E 02	-0.	o. o.	3.70 3.85
-0.21406F 02	0.179435 02	-1-1521AE 02	-0.68018F 01	-0.	0.	4.00
-0.13756F 07 -0.46393F 01	0.19700E 02	-0.11499E 02 -0.54252E 01	0.79373F 00 0.52423F 01	-0. -0.	o.	4.20
0.376475 01	0.189767 02	0.297576 01	0.55050F 01	0.	0.	4.50 4.70
0.1109AF 02	-0.45013F 01	0.120726 02	-0.25587E 02	0.	-0.	4.80
0.223965 02	-0.12E61E 02	0.28351F 02 0.15953E 02	-0.37370E 02 -0.24338E 02	0. 0.	-0. -0.	4.96
-0.50277F 01	0.24309F C1	-0.14377E 02	-0.11730F G2	-0.	-0.	5.00 5.19
-0.500326 01	0.536/58 01	-0.15700F 02	-0.47366E 01	-0.	J.	3.30
-0.278925 01 -0.113716 01	0.61499F 01 0.64857F 01	-0.13354E 02 -0.10745E 32	-0.11845F 01 0.98407E-01	-0.	o. o.	5.60
-0.62599F 00	0.661196 01	0.47793F 01	-0.69476E 00	-0.	o.	5.70 5. 8 5
-0.541176 00	0.95252F 01	-0.687317 01	-0.120248 01	-0.	0.	6.00
-0.104306-00 0.84755F 00	0.1046'f 07 0.11031f 02	-0.69194E 01 -0.74163F 01	-0.165717 01 -0.208576 01	-9. -0.	6. 0.	6.03
0.237597 03	0.72579F 01	-0.13570F 01	0-154866 01	-0.	ð.	£.06 6,08
0.442556 03	0.411986 01	-n.\ 17E 01	0.396186 01	0.	0.	6.20
0.6[338F 0] 0.59680F 0]	0.27843E 01 0.65931E 00	-0.7247E 03 -0.52143E 01	0.46349F 01 0.48294F 01	o. o.	o. o.	6.40
0,577536 01	-0.11720f 01	-0.27534E 01	0.42199F 01	0.	- 0.	6.60 7.00
0.520706-01	-0.344341 01	0.140346-00	0.193336 01	0.	- 0.	7.46
0.29862F 01	-0.39543F 01 -0.24656F 01	0.71444E 01 0.22519E 01	-0.57253F 00 -0.23255F 01	0.	-0. -0.	4.20
-0.207217 01	0.	0.131796-00	0.	-0.	0.	9.00 10.00

Table XIX - - - Continued

(PSIFFPS SINUSOIDAL GUST)

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

297,000 LB CUTOFF FREQUENCY: 24,000 FT 0.50

10 CPS

PERCENT SEMISPAN: 40.06 SEGMENT NUMBER 107

Dichery			DUMBERAL STILL STILL			
RFAL	1-4G1MARY			4 FAL	1.421M484	PRESENCE
						CPS
-6.13669F 01	- r. 1045 7F - 01	-0.	-0.	0.68494F 01	-2.542424 01	0.10
0.4774#F 02 0.5491#F 02	-0.161076 02	0.	-9.	-0.2630% 03	0.60407E 0?	0.30 0.36
0.44179 07	-n, 232928 02 -n, 244196 02	n. 0.	-o.	-0.21692E 03 -0.17509E 03	0. 91302E 02 0.10512E 03	0.44
0. 19016F 07	- 0, 30 A 3 OF 02	ŏ.	-n,	-0.1544 X 03	0-12335E 03	0.50
0.339366 02	-0. 351 98F 02	n,	-0.	-0.133236 03	0 1^22 BE 03	0.60
0.304756 02	- 0. 4094RF 02	n,	⊸.	-0.1184 % 03	0.16519E #5	ď. 70
0.274536 27	-C. 48627F 02	o.	-0.	-0.10571E 03	G.19446E 03	0.80
0.25:40F 02 0.27134F 02	-C.10845F 03	o.	-0. -0.	-0.92801E 02 -0,26450E 02	0.4075 NE 03	0.90 1.00
0. 11462F 01	-C. 20801F 03	o.	-0.	0.584946 01	0. 725 806 03	1.20
-0.12099F 03	-0.12243F 03	-0.	-o.	0.441386 03	2. 7901 0F 03	1.56
-0.4341PF 03	0.13474F @3	-0.	0.	0.11900E 04	-0.48404E 03	1.40
-6.29057F 03	C- 17293E 03	- c. -c.	0.	0.981616 03	-0.406375 53	1.45
-0.20303F 03	0.17781F 03 0.14395F 03	-e.	0.	0.67243E 03 0.44606F 03	-0.54549E 03 -0.44892E 03	1.47 1.50
-0.43540F 02	0.116955 03	.0.	0.	0.200956 03	-0.347198 03	1.55
-0.303578 07	0. 97285F 02	-0.	0.	0.95234€ 02	-C.29411E 03	1.40
-0.13170F 02	0.44455 02	-0.	0.	0.43410F 02	-0-17,40E 03	1.65
9.PA775F 01	0. 534 747 02	n.	0.	-0.11623€ 02	-0.12704E 03	1.80
0.15609F 07	0.46227E 02	n. c.	0. 0.	-0.21602E 02 -0.24711E 02	-0-99828E 02 3-79004E 02	1.99
0.778195 02	r. 37241f 02	n.	ŏ.	-0.23431E 02	-0.43994E 02	2.10
0.343416 02	0. 30197F 02	0.	0.	-0.16701E 02	-0.490597 02	2.20
0.489716 02	0. 267 has 02	0.	0.	-0.434996 01	-0.475986 07	2.30
0.57718F 02	0, 149975 02	0.	0.	-0.11517E 62	-0.44251E 02	2.35
0.71740F 07	0.11992E 01 -0.65290E 01	o. c.	-o. -o.	-0.55362E 01 -0.14492E 0i	-0.20750E 02 -0.74883E 02	2.40 2.43
0.781435 02	-0. 21042F 02	o.	-ŏ.	-0.126790	-0. #31 438 02	2.44
0.712965 02	-0.25715F 02	o.	-6 .	-0.692498 01	- 5. 84134E 02	2.47
0.43013f 02	- 0. 31136E 02	0.	-0.	-0.15452E 02	-0.818492 02	2.90
0.655796 07	-0.565986 02	o.	-0.	-0.1813# 32	-0.898096 02	2.54 2.58
0.70274F 02 -0.97309F 01	-0.93742E 02	o. -o.	-0. -0.	-0.1678¥€ 02 -0.56623€ 02	-0.10584E 03	2.65
-0.19305F 02	-0.11581F 01	-0.	-0.	- 0, 74 940€ 02	-0.559236 02	2.70
-0.27248F 02	0.16751F 07	-0.	o.	-0.74140E 02	-0.446466 02	2.80
0.17423f 02	N-17724€ 02	n,	0.	-0.857376 02	-0.40301F 02	3.00
0.352516 02	n. 19343F 01	n.	0.	-0.1099# 03	-0.12163E 02	3.10
0.78724F 02 0.11498F 03	-0.51974F 02 -0.11728E 03	o.	-0. -0.	-0.16701E 03 -0.21202E 03	0.20581E 02 0.15703E 03	3.20 3.26
0.107306 03	-0.160016 03	0.	-0.	-0.1915 # 03	0. 205 85E 03	5.29
0.949845 01	-0.12751f 03	n.	-0.	-0.40619€ 02	0.136918 03	3.35
-0.48799F 07	-0. 531 79F 02	-0.	-0.	0.31286 02	0.491587 62	3.40
-0.11667F 03	0.40481F 01	-0. -0.	-0. 0.	0.22352€ 02 0.15183€ 02	0 47430E UZ 0.44711E B2	3.52 3.56
-0.80AR6F 02	0. 203426 02	-0.	v.	0.179046 02	0. 31 2 74E 02	3.60
-G.43023F 02	0-172446 02	-0.	0.	0.194766 02	0. 1 8848E 02	3.70
-M.24571F 07	0.163796 02	-0.	0.	0.161826 02	0.11697E 07	3.65
-0.18701F 02	0.14149F 02	-0. -0.	o.	0.13020€ 02 0.98169€ 01	0.58116E 01 -0.67636E 02	4.00
-0.44090F 01	0.155216 02	-0.	0.	0.4435% 01	-9.44792E 23	4.20
0.41803F 00	0.151406 02	0.	0.	-0.25436F 01	-0.470±7F 01	4.70
0.44778F 01	0.36237E 01	0.	⊸.	-0.10315E 02	0. 21 862E 02	1.70
0.10977F 02	0. 160426-00	0.	-0.	-0.239A7E 02	0.31930F 05	4.96
0.72540F 01 -0.12913F 01	0.313556 01	°. ~n.	-0. -0.	-0.13628E G2 0.12280E G2	0.20795E 02 0.10022F 02	5.00
-0.416876 00	0.474826 01	-0.	0.	0.13414	0.404715 01	5.15 5.30
0.111325 01	0.638196 01	-0.	0.	0.11410E 02	0.101215 01	5 17
0.248176 01	0. 595076 01	·O.	0.	0.918196 01	-0. #- ?536 -01	5.70
0.313526 01	0.59110# 01	-n.	0.	9.25002F 01	0.55363F 00	5.85
0.3A23AF 01 0.3R1AGF 01	0.60342F 01 0.41581E 01	-0. 0.	0.	0.58288€ 01	0.15046E 01	4.00
0.414715 01	O. ALARRE OL	-3.	0.	0.633676 01	0.178216 01	6.03 6.06
0.440716 01	0.441 87E 01	-0.	0.	0.2140% 01	~ 9. 33232F 0;	6.08
0.584866 01	0. 302 RZE 01	0.	0.	0.852016 01	-0-33051F 01	6,20
0.55752F 01 0.588635 01	0.190666 01	n. n.	o. o.	0.61 () 4E 01 0.44550E 01	-0.34549E CI	6.40
0.540815 01	- 0. 1441 RF OI	0.	⊸.	0.192545 21	0.160366 01	6.60
0.488A2F 01	-0. 367 87F 01	o.	-ŏ.	0.170346-00	G- 10919E 01	7.00 7.40
0.254906 01	-0.37856E 01	0.	-o.	-0.1K. '~* .	0.444196-00	4.20
-0.580056-01	-0-51843£ 93	o. -0.	-o. o.	-2,945 M	0.196706 01	9.00
-0.20921F 01	0.	-0.	J.	-0.154-46-03	0.	10.00

Table XIX --- Continued

(PSI/FPS SINUSOIDAL GUST)

297,000 LB CUTGFF FREQUENCY: J CPS 24,000 FT 0.50

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

BODY BALANCE STATION: 545 SEGMENT NUMBER 17

THEMSETA	L SSEAN STREET					
#FAL	14421 VARY			~		TEXTE
-0.4]362F-00	0.21P**F 01	٥.	~0.	0.	-0.	C78
-0.17055F 02	0.47525F 01	0.	~.	0.	- 0.	0.30
-0.12353F 02	0.5104+6-01	0.	-ŏ:	ö.	-0.	0.36
-0.101 PAE 02	0.401926 01	n.	-0.	0.	- 0.	0.44
-9.91232F 01	0.45184F 01	0.	-o.	0.	-0.	0.50
-0.84287F 01	0.52/3.6 01	٦.	-0.	0.	- 0.	0.60
-0.82977F 01 -0.84312F 01	0.546495 01	e.	-o.	0.	- 0.	0.70
-0.PARZZE 01	0. A6783F (0) 0. A47~3F (0)	ი. ი.	-•. -•.	o. o.	-0.	0. 8 0 0. 9 0
-0.89332F n	0.15061E 02	0.	~·.	ð.	-0.	1.00
-0.61770E 01	0.27250F 02	ň.	-ŏ:	ŏ.	-0.	1.20
0.39487F 01	0.195914 02	-0.	-o.	-0.	-0.	1.34 1.40
0.265P#E 02	-0.68790F 01	-9,	0.	-0.	0.	1,40
0.22325F 02	-0.10430E 02	-0.	0.	-0.	0.	1.45
0.13154F 02	-0.10640F 07	-ė.	0.	-0.	0.	1.47
0. A] 509F 01	-0.70439F 01	-5.	٥.	-0.	. 0.	1.50
-0.17455F C1 -0.33592F C1	-0.41042F 00	0. 0.	o. -o.	o. o.	° 0.	1.55 1.60
-G.72321F 01	0.55539F 01	0.	⊸.	2.	-0.	1.65
-0.9326AF 01	P. 4 2075F 01	0.	-o.	ő.	-0.	1.80
-0.961 11F D1	0.12024F 02	n.	-ō.	ŏ.	-0.	1.90
-0.949A7F 01	0-15111F 02	n.	-ō.	o.	-0	2.60
-0.913096 01	0.1#391# 02	0.	-o.	o.	-0.	2.10
-0.87259F 01	0.231426 02	٥,	-e.	0.	-0.	2.20
-0.Alli3F 01	0-253156 02	0.	⊸.	0.	-0.	2.50
-0.49002F 01	0.27670F 02	0.	-o.	o.	-0.	2.35
-0.62726F 01 -0.57700F 01	0.29245F 02 0.29939F 02	o. o.	-o. -o.	٥.	-0.	2.40
-0.57700F 01	0.31197F 02	0.	~. ~o.	0. 0.	-0. -0.	2.43 2.44
-0.51902F 01	0.329355 02	0.	-ŏ.	o.	-0.	2.47
-0.58220F 01	0.3893FF 02	0.	-ŏ.	o.	-0.	2.36
-0.75716F 01	0.4840AE 02	0.	-ō.	9.	.0.	2,56
-0.72"68F 01	0.4273-F 62	0.	-o.	0.	-0.	2.58
0.191336 07	0.52247F 02	-n.	-0.	-0.	-0.	2.65
0.31120F 02	0.459458 02	-0.	~0.	-0.	· ə.	2.70
0.36949F 02	0.57187F 02 0.57737F 02	-0.	-o.	-0.	-0.	2.80
0.6842]F 02 0.10874F 03	0.18364F 02	-0. -0.	-0. -0.	-0. -0.	-0. -0.	3.00
0.196315 63	-0.975356 02	-0.	0.	-0.	-0.	3.10 3.20
0,24641F 03 0,24127F 03	-0.22197F 03	-0.	e.	-0.	ŏ.	3. %
		-n.	9.	- 0.	ō.	3.29
G.2729AF 02	-0.200645 03	-0.	G.	-0.	0.	3.35
-0.78747E 02	-0.48254F 02 -0.45143F 02	0.	0.	o.	0.	3.40
-0.47986F 02	-0.45449F 02	n. n.	0. 0.	0.	0.	3.52 3.56 3.60
-0.63341F 07	- G. 27072F 02	n.	ŏ.	o. o.	0. 0.	3.56
-0.55173F 02	-0-101556 02	0.	ŏ.	0.	0.	3.60
-0.43935F 02	-0.4P793F-00	n,	-ō.	ő.	- 0.	3.70 3.85
-0.34768F 02	0.65665 61	0.	-o.	ō.	- 0.	4.00
-0.25570F 02	0.11495F 02	n.	~O.	0.	-0.	4,20
-0.15601F 02	0.126516 02	ņ.	-o.	0.	-0.	4.50
-0.11071F 02	0.131 FOF 02 0.20225F 02	o.	-•.	0.	-0.	4.70 4.60
-0.10124F 07	0.228145 02	0.	-0. -0.	o. o.	-0. -0.	4.80
O.5488]F O	0.199995 02	n.	-ŏ:	ŏ.	-0.	4.96
0.50463F 01	0-1669PF 02	-0.	-ŏ.	- 0.	-0.	3.00 3.15
0.97195F 01	0.13537F 02	-0,	-0.	- ŏ.	-0.	3.30
0.15819F 02	0.87449F 01	-0.	⊸.	-0.	-0.	6.0
0.25065F 02	0.88665E-01	-0.	-0.	-0.	-0.	9,70
0.367956 02	-0.39841F 02	-0.	0.	-0.	0.	5.85
0.324795 07	-0.55355F 02 -0.71384F 62	-0.	٥.	-0.	٥.	6.00
0.48409F 02 0.36502F 02	-0.81800F 02	-0. -0.	o. o.	-0. -0.	o. o.	6.03 6.06
4.15736F 02	-0.33591F 02	-0.	0.	-0.	0.	6.06
-0.39733F 07	-0.16024F 01	0.	-o.	0.	-0.	6.08
-0.21146E 02	0.26640F 01		-o.	ŏ.	-0.	6.20 6.40
-0.11265F 02	0.27549F 01	٥.	-0.	0.	- 0.	6.60
-0.38270F 01	0.159696 01	-0.	-0 ,	-0.	. 0.	7.00
-0.145A9E 01	0.22640F-00	÷0.	-0.	-0.	-0.	7.40
-D.A0630F 00	0.317796-01	-0.	-0.	*1**	-0.	8.20
-0.61647F 00	-0.209416-01	-0.	-0.	-0.	- 0.	9.00
-0.42690F-00	n .	-0.	0.	-0.	0.	10.00

Teble XIX --- Concluded

(FSIFFS SINUSOIDAL GUST)

297, 000 LB CUTOFF FREQUENCY: 10 CPS 24,000 77 0,50

GROSS WEIGHT: ALTITUDE: MACH NUMBER:

BODY BALANCE STATION: 820 SEGMENT NUMBER 1

		TICHNOTAL	ATTAL STREET			
		RFAL	IMACIMARY			navact
						CPS
٠٠.	-n.	-n.64532F 53	-0.37354F #1	-0.	-0.	0.15
٥.	-0.	0.111126 03	-0.27342F 02	0.	-0.	0,30 0,36
n.	-n. -n.	0.89432F 02 0.72751F 02	-0.36570F 02	o. o.	-0. -0.	0.44
n.	٠٥.	0.44755F 02	-0.45995F 02	ŏ.	- 0.	0.50
e.	-0.	0.50275F 02	-0.52430F 02	0.	٠٥.	0.60
0.	-0.	0.54931F 02	-9.62109E 02	٥.	-0.	0.70 0. 8 0
n. n.	-n. -n.	0.52477F 02 0.50185F 02	-0.74731F 02 -0.93173E 02	o. o.	-0. -0.	0.90
n.	-0.	0.45633F 02	-0.16979E 03	ŏ.	- U.	1.00
n.	- 0.	0-184725 02	-0.30810F 03	ð,	-0.	1.20
-0.	٠٠.	-0.16453F 03	-0.17367F 63	-0.	-0.	1.34 1.40
-n. -n.	n. n.	-0.46555F 03 -1.38233F 03	0.18689F 03 0.23427F 03	-0. -6.	0. 0.	1.45
- n .	" .	-0.25472F 03	0.22788F 03	-0.	ŏ.	1.47
-n.	n.	-0.15051F 93	0.17874F 03	-0.	٥٠.	1.90
-n.	r.	-0.5944PF 0Z	0.13400F 03	-0.	0.	1.55 3.60
n. n.	a. c.	-n.139;7F nz -n.139;7F nz	0.10171F 03 0.41327F 07	-0. 6.	o. o.	1.65
0.	n.	7.31133F 02	0.16867F 02	o.	0.	1.80
с.	-0.	0.341336 02	-0.29970F O:	0.	-0.	1.90
٥.	- O.	9433107F 02	-0.23064F 02	0.	-0.	2,00
٥.	- n. - o.	0.288356 02	-0.35515F 02 -0.43678F 02	0. 0.	-0. -0.	7.10 2.20
Λ. Λ.	-0.	3.53397F 01	-0.48643F 02	n.	-0.	2.30
-0.	· n.	-0.79345F-01	-0.49504F 62	-0.	-0.	2.35
-n.	- n.	-1.14135E 02	-0-41474F 02	-0.	-0.	2.40
-n. -n.	-n. -n.	-0.21557F 02 -0.22415F 02	-C.35922F 02 -0.24637F 02	-0. -0.	-0. -9.	2.43 2.44
-n.	-0.	-5.14254F 02	-0.26219F 02	-0.	-0.	2.47
n.	· r.	7.594136 00	-0.42131F 02	õ.	-0.	2,30
٥.	• n.	0.748136 31	-0.57721F 02	0.	- <u>(</u> .	2.54 2.58 2.65
n.	-r. -r.	0.65911F 91 -0.17011F 02	-0.73309F 07 -0.80101F 07	o. -o.	-0. -0.	2,56
-n. -n.	-r. -n.	-0.17347F 02	-0.97217F CZ	-0.	-0.	2.70
-6	-n.	-0.41499F 02	-0.13161F 03	-0.	-0.	2.80
-0.	- n.	-0.13450F 03	-0.13753F 03	-0.	-0.	3.00
-n. -n.	-0. n,	-n.23933F 03 -n.45955F 03	-0.55756F 02 0.22871E 03	-0. -0.	-0.	3.10 3.20
-0.	ň.	-0./305F 03	0.53228F 03	-0.	0. 0.	3.26
-0.	٠.	-0.55655 03	0.70544F 03	-0.	0.	3.29
-n.	Ç.	-0.402936 02	0.45755F 03	-0.	9.	3.35
n.	n.	0.20977F 03 0.15711F 03	0.14242F 03 0.14095F 03	o. o.	o. o.	3.45 3.52
n.	n.	0.13352F 03	C.13074F 03	ŏ.	ŏ.	3.56
n.	٥.	0-141275 03	0.75164E 02	0.	0.	3.60
e.	n.	0.14249F 13 0.11437F 03	0.23699F 02	0.	0.	3.70
ñ.	-0. -0.	0.943735 92	-0.51825F 31 -0.26377F 02	o. o.	-0. -0.	3.85 4.00
n.	- n.	0.485555 02	-0.42544F 02	n.	-0.	4,20
n.	- 0.	0.35313F 02	-0.47946F 02	0.	-0-	4,50
n.	-0,	0.13154F 02 -0.22433F 31	-0.47551F 02	0.	~ O•	1.70 1.80
-n. -n.	- n. - n.	-0.25437F 02	-0.14743F 02 -0.21199F 01	-C. -O.	-0. -n.	4.96
-a.	٠.	-7.15455F 02	-0.133JAF 02	-0.	-0.	3.00
ο.	- O.	0.67912F 01	-0.22428F n2	٥.	-0.	5.15
6.	-0.	7.1744F-00	10.24146F 02 -0.19803F 02	0.	- 0-	5.30
- n.	-n. -n.	-0.11474F 02 -0.25554F 02	-0.1980 F 02	-0. -0.	-0, -0.	5.50
-n.	o,	-0.49372E 02	0.35682F 02	-0.	ŏ.	5.85
-A.	0.	-0.545236 02	0.50360F 02	-0,	c.	6,00
-0.	٥.	-0.54344F 02	0.675725 02	-0.	0.	6.03
-n. -n.	۰. •	-7.41729E 32 -0.19479F 02	0.74/10F 02 0.27564F 02	-0. -0.	o. o.	5.06 5.09
n.	-0,	0,383116 02	.0.50559F 01	0.	- 0,	6.20
ο.	-0.	0.171746 02	-0.81817F 01	0.	-0.	6.40
Λ.	-0-	0.673496 01	-0.403F 0]	0.	-0.	6.60
-n.	- 0. - 0.	-0.29734F 01 -0.41476F 01	-0.*.171F 01 -0.*:36F 00	-0. -0.	- 0. - 0.	7.00
-0.	-0.	-0.426515 01	-0.76482E-02	-0.	-0.	7.40 8.30
-0.	۰.	-0.41453E 01	0.13460F 01	-0.	0.	9.00
-0,	۰.	-0.3#230€ 01	0.	-0.	0.	10,00

APPENDIX VI STRESS RESPONSE PARAMETERS AND ZERO-CROSSING RATES

Table XX. Stress Response Parameters and Zero-Crossing Rates (Analysis Condition 1)

GROSS WEIGHT: 297,000 LB MACH NUMBER: 0.85
ALTITUDE: 24,000 FT

Location		Axial stress		Shear stress		
Body station	Percent semispan	Segment number	A (psi)	N ₀ (Zero crossings per second)	A (psi)	N _o (Zerc crossings per second)

SCALE OF TURBULENCE: 1,000 FT CUTOFF FREQUENCY 10 CPS

	27	10	421	1.02	62	1.81
	27	14	382	1.02	152	1.25
	40.06	8	397	1.96	111	1.32
	40.96	107	359	1.06	106	1.30
540		S-17	0	0	31.17	2.64
820		S-1	159	1.38	0	0

SCALE OF TURBULENCE: 3,000 FT CUTOFF FREQUENCY: 10 CPS

	27	10	339	0.836	47.5	1.64
	27	14	308	0.886	122	1.09
	40.06	8	318	0.921	87.6	1.16
	40.06	107	272	0.921	84.1	1.15
540		S-17	0	0	24.09	2.37
820		S-1	128	1.19	0	0

SCALE OF TURBULENCE: 5,000 FT CUTOFF FREQUENCY: 10 CPS

	27	10	293	0.863	40.7	1.62
	27	14	266	0.863	195	1.06
	40.06	8	274	0.903	75.4	1.14
	40.06	107	234	0.903	72.4	1.12
540		S-17	0	0	20.67	2.33
820		S-1	111	1.16	0	. 0

Table XX --- Concluded

GROSS WEIGHT: 297,000 LB MACH NUMBER: 0.85

ALTITUDE:

24,000 FT

Location	Seemen'	Axial stress		Shear stress		
Body station	Percent semispan	Segmen(number	A (psi)	N _o (Zero crossings per second)	A (psi)	N _o (Zero crossings per second)

SCALE OF TURBULENCE: 1,000 CUTOFF FREQUENCY: 15 CPS

	27	10	421	1.02	62	1.82
1						
	27	14	382	1.02	152	1.25
~	40.06	8	397	1.06	111	1.33
	40.06	107	339	1.06	106	1.31
540		S-17	0	0	31.17	2.64
820		S-1	159	1.38	Ü	0

SCALE OF TURBULENCE: 1,000 FT CUTOFF FREQUENCY: 20 CPS

	27	10	421	1.04	52	1.82
	27	14	382	1.04	152	1.22
	40.06	8	397	1.04	111	1.30
	40.06	107	339	1.04	106	1.30
540		S-17	0	0	31.17	2.61
820		S-1	159	1.39	0	0

Table XXI. Stress Response Parameters and Zero-Crossing Rates (Analysis Condition 2)

GRUSS WEIGHT: 268,000 LB MACH NUMBER: 0.85 ALTITUDE: 24,000 FT

Loc	ation		Axial stress		Shear stress	
Body station	Percent semispan	Segment number	A (psi)	N _O (Zero crossings per second)	A (psi)	N ₀ (Zero crossings per second)

SCALE OF TURBULENCE: 1,000 FT CUTOFF FREQUENCY: 10 CPS

	27	10	377	1.03	71	2.21
	27	14	342	1.03	152	1.50
	40.06	8	342	1.03	111	1.79
	40.06	107	293	1.03	107	1.75
540		S-17	0	0	43.56	3
829		S-1	179	2.02	0	0

SCALE OF TURBULENCE: 3,000 FT CUTOFF FREQUENCY: 10 CPS

	27	10	312	0.868	52.9	2.06
	27	14	283	0.868	121	1.30
	40.06	8	283	0.863	86.6	1.59
	40.06	107	242	0.863	83.6	1.55
540		S-17	0	0	32.44	2.80
820		S-1	143	1.76	0	0

SCALE OF TURBULENCE: 5,000 FT CUTOFF FREQUENCY: 10 CPS

	27	10	270	0.846	45.1	2.03
	27	14	245	0.846	105	1.28
	40.06	8	245	0.841	74.3	1.56
	40.06	107	210	0.841	71.9	1.52
540		S-17	0	0	27.69	2.76
820		S-1	123	1.73	Û	0

Table XXII. Stress Response Parameters and Zero-Crossing Rates (Analysis Condition 3)

GROSS WEIGHT: 190,590 LB
MACH NUMBER: 0.85
ALTITUDE: 24,000 FT

Location		Co. Sec. of	Axial stress		Shear stress	
Body station	Percent semispan	Seyment number	A (psi)	N _O (Zero crossings per second)	A (psi)	N ₀ (Zero crossings per second)
		SCALE OF	TURBULENCE	: 1,000 FT		
		CUTOFF F	REQUENCY:	10 CPS		
	27	10	304	1.15	58.1	2.34
	27	14	276	1.15	121	1.62
	40.06	8	274	1.22	95.2	2.09
	40.06	107	234	1.22	91.6	2.08
540		S-17	0	0	51.4	2.55
820		S-1	177	2.07	0	0
	27	T			415	2 28
			FTURBULENCE FREQUENCY:	E: 3,600 FT 10 CPS		
	27	10	239	1.02	41.5	2.28
	27	14	217	1.02	92.1	1.48
	40.06	8	215	1.09	7ü.3	1.97
	40.06	107	183	1.09	68	1.95
540		S-17	0	0	39.2	2.32
820		S-1	138	1.85	0	0
						•
			FTURBULENC Frequency:	E: 5,000 FT 10 CPS		
	27	10	206	1	35.1	2.27
	27	14	187	1	78.8	1.45
	40.06	8	184	1.07	59.9	1.95
	40.06	107	157	1.07	58	1.92
540		S-17	0	υ	33.6	2.28
820		S-1	118	1.82	0	0

Table XXIII Stress Response Parameters and Zero-Crossing Rates (Analysis Condition 4)

GROSS WEIGHT: 107,260 LB MACH NUMBER: 0.85
ALTITUDE: 24,000 FT

Location		Connel	Axial stress		Shear stress	
Body station	Percent semispan	Segment number	A (psi)	K _o (Zero crossings per second)	A (psi)	N _o (Zero crossings per second)

SCALE OF TURBULENCE: 1,000 FT CUTOFF FREQUENCY: 10 CPS

	27	10	177	1.39	58.1	3.14
	27	14	161	1.39	77.9	2.89
	40.06	8	169	1.50	63.2	2.93
	40.06	107	145	1.50	60.5	2.91
540		S-17	0	0	31.25	3.46
820		S-1	116	2.90	0	0

SCALE OF TURBULENCE: 3,000 FT CUTOFF FREQUENCY: 10 CPS

	27	10	135	1.27	40.5	3.12
	27	14	122	1.27	54.8	2.85
	40.06	8	128	1.37	44.8	2.86
	40.06	107	110	1.37	43.1	2.84
540		S-17	0	0	22.39	3.34
820		S-1	85.8	2.73	0	0

SCALE OF TURBULENCE: 5,000 FT CUTOFF FREQUENCY: 10 CPS

	27	10	115	1.25	34.2	3.12
	27	14	105	1.25	46.3	2.65
	40.96	8	110	1.36	38	2.85
	40.06	107	93.8	1.36	36.5	2.82
540		S-17	0	0	18.99	3.33
820		S-1	73.1	2.70	0	0

Table XXIV. Stress Response Parameters and Zero-Crossing Rates (Analysis Condition 5)

GROSS WEIGHT: 297,000 LB MACH NUMBER: 0.50 ALTITUDE: 24,000 FT

Loca	ation	C	Axial	stress	Shea	r stress
Body station	Percent semispan	Segment number	N _o (Zero crossings per second)			
			TURBULENC FREQUENCY:	E: 1,000 FT 10 CPS		
	27	10	226	1.08	29.5	1.53
	27	14	205	1.08	76.7	1.21
	40.06	8	221	1.10	56.6	1.34
	40.06	107	189	1.10	53.3	1.33
540		S-17	0	0	21.1	3.08
820		S-1	88.1	1.87	0	ŋ
			TURBULENC Frequency:	E: 3,000 FT 10 CPS		
	27	10	158	1.08	20.6	1.54
	27	14	144	1.08	53.6	1.18
	40.06	8	154	1.08	39.5	1.34
	40.06	107	132	1.08	37.2	1.34
540		S-17	0	0	14.68	3.08
820		S-!	61.5	1.85	0	0
			F TURBULENC FREQUENCY:	E: 5,000 FT 10 CPS	· · · · · · · · · · · · · · · · · · ·	
	27	10	133	1.07	17.3	1.52

			FREQUENCY:	10 CPS		
	27	10	133	1.07	17.3	1.52
	27	14	121	1.07	45.3	1.20
	40.05	8	130	1.10	33.3	1.34
	40.06	107	111	1.10	31.4	1.31
540		S-17	0	0	12.4	3.06
820		S-1	59.9	1.86	0	0

Table XXV. Incremental Limit Allowable Stresses

							Analysis	Analysis condition				
-	1					2				4		
٢	Location	,	Gross wt:	Gross wt: 297,000 lb Gross wt: 268,000 lb Gross wt: 190,599 lm Gross wt: 107,000 lb Gross wt: 297,000 lb	Gross wt:	268,000 lb	Gross wt:	190,590 IN	Gross wt:	107,000 16	Gross wt:	297,000 lb
		Segment	Mach number: 0.85	per: 0.85	Mach number: 0.85	er: 0.85	Mach number: 0.85	er: 0.85	Mach number: 0.85	er: 0.85	Mach number: 0.50	er: 0.50
4	,	пишрет	Axial	Shear	Axial	Shear	Axial	Shear	Axial	Shear	Axial	Shear
body	reicent		stress	stress	stress	stress	stress	stress	stress	stress	stress	stress
Station	semispan		(psi)	(isd)	(bsi)	(psi)	(bsi)	(psi)	(psi)	(psi)	(ps1)	(psi)
				SCALI	S OF TURE	SCALE OF TURBULENCE	1,000 FEET	ET				
1	22	2	24,700	3,800	24,400	4,500	29,400	2,600	33,500	10,300	23,500	3.000
1 :	27	14	24,100	9,500	23,600	10,500	27,230	12,500	31,000	15, 100	23,500	000'6
1 1	40.06	œ	24, 100	9,600	23,600	2,600	27,700	10,000	32,400	11,100	22,300	2,700
 	40. 0%	197	18,000	2,800	17,500	6,500	20,400	7,900	23,200	9,900	18,200	2,300
240		S-17	!	4,190		4, 190		4,190		5,635		190
820		S.1	22, 100		21,300		23,400		27,700	-	25,700	:
				SCAL	E OF TURE	SCALE OF TURBULENCE:	3,000 FEET	ET				
!	27	22	24,800	3,500	24,560	4, 100	29,600	4,900	34,300	000'6	23,500	3,000
!	27	14	24,000	9,500	23,400	10,500	27,400	12,400	31,100	15,100	23,500	2,000
1	40.06	œ	24,000	6,700	23,600	7,300	27,800	009'6	32,300	11,300	22,300	5,200
[[:	40.06	101	18,300	2,700	17,800	6,100	20,900	7,500	24,200	9,300	18,200	2,300
3		5.17		₹ ,190		4.190		4,190	-	5,635		190
820		ાંડ	22,100		21,900		23,400		27,700		25,700	
				SCAL	E OF TUR	SCALE OF TURBULENCE:	5,000 FEET	ET				
	12	92	24,700	3,800	24,500	4,200	29,500	5,100	34,100	9,100	23,500	3, 100
1	12	14	24,100	9,500	23,800	10,100	27,900	11,600	31,500	14,400	23.600	8,600
1 1	40.06	æ	24,300	2,300	23,700	7,300	28,200	9,100	32,600	10,700	22,300	5,800
1 1	40.06	107	18,300	5,5.0	17,900	9,000	21,100	7,300	24,300	9,000 6	18,200	5,100
25		۲I.۶		4,190		4,190	! !	4,190	1	5,635	1 1	61.
ğ	1 1	ي ت	22,100	1	21,900	!	23,400	!	27,700		25,700	

APPENDIX VII INCRI MENTAL LIMIT ALLOWABLE STRESSES

APPENDIX VIII CORRELATION COEFFICIENTS BETWEEN AXIAL AND SHEAR STRESSES

Table XXVI. Correlation Coefficients Between Axial and Shear Strosses

Loca	ation		W _c = 10 cps		W _c = 15 cps	₩ _C = 20 cps			
Percent wing semispan	Segment number	L = 1,000 ft	L = 3,000 ft	L = 5,000 ft	L. = 1,000 ft	L = 1,000 ft			
		ANA	LYSIS CONDITIO	DN 1					
27	10	0.835	0.866	0.871	-0.835	-0.835			
27	14	0.961	0.97 1	0.972	0.961	0.961			
40.06	8	0.962	0.971	0.972	0.962	0.962			
40.06	107	-0.962	-0.971	-0.972	-0.962	-0.952			
		ANA	ALYSIS CONDITIO	ON 2	- · · · · · · · · · · · · · · · · · · ·				
27	10	0.746	0.780	0.786					
27	14	0.941	0.956	0.958					
40.06	8	0.887	0.910	9.913					
40.06	107	-0.894	-0.916	-0 019	<u> </u>				
		AN /	ALYSIS CONDITION	ON 3					
27	10	0.673	0.687	0.689					
27	14	0.922	0.935	0.937					
40.06	8	0.831	0.851	0.854					
40.0€	107	-0.837	-0.859	-0.862					
ANALYSIS CONDITION 4									
27	10	-0.0127	-0.0499	-0.0558					
27	14	0.475	0.493	0.495					
40.06	8	0.575	0.598	0.602					
40.06	107	-0.585	-0.611	-0.615					
		ANA	ALYSIS CONDITIO	ON 5					
27	10	0.904	0.905	0.905					
27	14	0.975	0.975	0.975					
40.06	8	0.972	0.972	0.972	-				
40.06	107	-0.372	-0.972	-0.972					

APPENDIX IX STRESS INFLUENCE COEFFICIENTS

(a) 27 PERCENT WING SEMISPAN, SEGMENT NUMBER 10

$$\begin{cases} \frac{\text{Skin}}{\text{Stress}} \\ \frac{\text{Shear}}{\text{Stress}} \end{cases} = \begin{bmatrix} 0.00197 & \frac{\text{PSI}}{\text{In-Lb}} & 0 & 0 \\ 77 \times 10^{-6} & \frac{\text{PSI}}{\text{In-Lb}} & 0.0228 & \frac{\text{PSI}}{\text{Lb}} & 865 \times 10^{-6} & \frac{\text{PSI}}{\text{In-Lb}} \end{bmatrix} \begin{cases} \frac{\text{Bending Moment}}{\text{Moment}} \\ \frac{\text{Shear}}{\text{Torsion}} \end{cases}$$

(b) 27 PERCENT WING SEMISPAN, SEGMENT NUMBER 14

$$\begin{cases} \frac{\text{Skin}}{\text{Stress}} \\ \frac{\text{Shear}}{\text{Stress}} \end{cases} = \begin{bmatrix} 0.000971 & \frac{\text{PSI}}{\text{In-Lb}} & 0 & 0 \\ 188 \times 10^{-6} & \frac{\text{PSI}}{\text{In-Lb}} & 0.0655 & \frac{\text{PSI}}{\text{Lb}} & 865 \times 10^{-6} & \frac{\text{PSI}}{\text{In-Lb}} \end{bmatrix} \begin{cases} \frac{\text{Bending Moment}}{\text{Moment}} \\ \frac{\text{Shear}}{\text{Torsion}} \end{cases}$$

(c) 40.06 PERCENT SEMISPAN, SEGMENT NUMBER 8

$$\begin{cases}
Skin \\
Stress
\end{cases} = \begin{bmatrix}
0.00158 & \frac{PSI}{In-Lb} & 0 & 0 \\
123 \times 10^{-6} & \frac{PSI}{In-Lb} & 0.0359 & \frac{PSI}{Lb} & 1370 \times 10^{-6} & \frac{PSI}{In-Lb}
\end{bmatrix} \begin{cases}
Bending \\
Moment \\
Shear
\end{cases}$$
Torsion

(d) 40.06 PERCENT SEMISPAN, SEGMENT NUMBER 107

$$\begin{cases}
Segment \\
Stress
\end{cases} = \begin{bmatrix}
-0.00135 & \frac{PSI}{In-Lb} & 0 & 0 \\
43.4 \times 10^{-6} & \frac{PSI}{In-Lb} & 0.0525 & \frac{PSI}{Lb} & 1286 \times 10^{-6} & \frac{PSI}{In-Lb}
\end{bmatrix}
\begin{cases}
Bending \\
Moment \\
Shear
\end{cases}$$
Torsion

APPENDIX IX --- CONCLUDED

(e) BODY BALANCE STATION 540, STRINGER S-7

$$\left\{ \begin{array}{l} \text{Axial Stress} \\ \text{Shear Stress} \end{array} \right\} = \begin{bmatrix} 0 & 0 \\ 0 & 0.0516 \frac{\text{PSI}}{\text{Lb}} \end{bmatrix} \quad \left\{ \begin{array}{l} \text{Bending } \\ \text{Moment} \\ \text{Shear} \end{array} \right\}$$

(f) BODY BALANCE STATION 820, STRINGER S-1

$$\begin{cases}
Axial Stress \\
Shear Stress
\end{cases} = \begin{bmatrix}
0.000302 \frac{PSI}{In-Lb} & 0 \\
0 & 0
\end{bmatrix}
\begin{cases}
Bending \\
Moment \\
Shear
\end{cases}$$

Sign Convention

- + Segment Stress Tension
- + Bending Moment Tension in lower skin
- + Shear Outboard wing sheared up relative to inboard wing
- + Torsion Leading edge up

REFERENCES

- 1. D. Espey and A. Laier, External Loads Criteria, D-16809, Vol. I, The Boeing Company, April 1956
- 2. Military Specification, Airplane Strength and Rigidity, MIL-A-8860 (ASG), U. S. Air Force, May 1960
- 3. R. E. Baum and J. R. Newell, Jr., Mass Properties for Power Spectral Gust Analysis, D6-18249, The Boeing Company, September 1965
- 4. D. O. Neilson and E. Plunkett, <u>Ground Vibration Test of the KC-135</u> Airplane, T6-1044, The Boeing Company, December 1957
- 5. D. R. Magee and W. W. Bingham, <u>Ground Vibration Test of the 707-321</u>
 Airplane, T6-1588, The Boeing Company, June 1959
- 6. J. T. Rogers, Review of 707 Design Aeroelastic Analysis Based on Comparison with Measured Flight Loads on the KC-135 Airplane, D6-1042, The Boeing Company, September 1958
- 7. L. D. Richmond, <u>A Rational Method of Obtaining Three-Dimensional Unsteady Aerodynamic Derivatives of Intersecting Airfoils in Subsonic Flow</u>, D6-7401, The Boeing Company, April 1962
- 8. J. C. Houbolt, R. Steiner, and K. G. Pratt, <u>Dynamic Response of Air-planes to Atmospheric Turbulence Including Flight Data on Input and Response</u>, TRR-199, National Aeronautics and Space Administration, 1965
- 9. R. Coffman, G. Gillette, G. Houlihan, <u>Fuselage Stress Analysis</u>, D-16811, Vol. II, The Boeing Company, May 1957
- 10. A. R. Wright, Wing Stress Analysis, D6-16810, Vol. I, The Boeing Company, January 1957
- 11. J. R. Fuller et al., Contribution to the Development of a Power Spectral Gust Design Procedure for Civil Aircraft, FAA-ADS-54, Contract FA-WA-4768, January 1966
- T. A. Montgomery, External and Internal Loads for C-135 Airplane, D6-7267, Vol. II, The Boeing Company, April 1981
 - 13. A. W. Byrski, Design Loads Criteria for Growth 707-100/200 Series Airplanes, D6-3212, Vol. Π, The Boeing Company, July 1959

	ennotation wast be entered when the overall report is classified;
1 ORIGINATING ACTIVITY (Corporate author) The Boeing Company	24 REPORT SECURITY CLASSIFICATION UNCLASSIFIED
Commercial Airplane Division	2h GROUP
Renton, Washington	none
3 REPORT TITLE	
Volume 1.	
KC-135 Power Spectral Vertical Gust Lo	ad Analysis
Detailed Analysis and Results DESCRIPTIVE NOTES (Type of report and inclusive dates)	
Final Report July 1965 to March 1966	
5 AUTHOR(S) (Lass name, first name, initial)	
Latz, Robert N.	
6 REPORT DATE July 1966	74 TOTAL NO. OF PAGES 75. NO. OF REFS.
	121 13
Ma. CONTRACT OR GRANT NO.	SA ORIGINATOR'S REPORT NUMBER (5)
AF 33 (615) - 2454	AFFDL-TR-66-57, Vol. I
System No. 5 (611367 62405334)	
Project No. 1367	95. OTHER REPORT NO. (\$) (Any other numbers that may be assigned this report)
	D6-18252
10. AVAILABILITY/LIMITATION NOTICES	
	t controls and each transmittal to foreign
	made only with proper approval of the Air Force
Flight Dynamics Laboratory (FDTR). W	
11. SUPPLEMENTARY NOTES	12. SPONSORING MILITARY ACTIVITY Air Force Flight Dynamics Laboratory, Research
	and Technology Division, Air Force Systems
	Command, Wright-Patterson AFB, Caio
13 ABCTRACT	

DOCUMENT CONTROL DATE BED

This report presents the results of an anilysis to obtain the stress response parameters (level of stress per level of turbulence) and zero-crossing rates at two wing stations and two body stations of the KC-135 airplane where the margins of safety for gusts are minimum. Five combinations of gross weight, speed, and altitude were selected. The results of the computer analysis pre sent the effects of changes in scale of turbulence and upper cutoff frequency on the response parameters and zero-crossing rates. Results indicate a large reduction in stress response parameter and small reduction in zerocrossing rate with increasing scale of turbulence. Variatons of upper cutoff frequence above the highest modal frequency used in the analysis indicate negligible change in eigher stress response parameter or zero-crossing rates. The ratios of incremental limit allowable stress to stress response parameter obtained over a wide range of gross weight, speed, and scale of turbulence result in a minimum value of 53. This document (volume I) presents the analyses and specific results described above. Volume II presents response parameters, zero-crossing rates, frequency response functions, and power spectra of bending moment, shear, and torsion.

Unclassified

14	KEY WORDS	LINK A		LINK 8		LINK C	
	RET WORDS	ROLE	WT	ROLE	WT	ROLE	WT
	Gust Loads Power Spectra Stresses Response Parameters Zero Crossings Analysis						

INSTRUCTIONS

- I. ORIGINATING ACTIVITY: Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (corporate author) issuing the report.
- 28. REPORT SECURITY CLASSIFICATION: Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.
- 2b. GROUP: Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.
- 3. REPORT TITLE: Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.
- 4. DESCRIPTIVE NOTES: If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.
- 5. AUTHOR(S): Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.
- REPORT DATE: Enter the date of the report as day, month, year; or month, year. If more than one date appears on the report, use date of publication.
- 7a. TOTAL NUMBER OF PAGES: The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.
- 7b. NUMBER OF REFERENCES: Enter the total number of references cited in the report.
- 8s. CONTRACT OR GRANT NUMBER: If appropriate, enter the applicable number of the contract or grant under which the report was written.
- 8b, 8c, & 8d. PROJECT NUMBER: Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.
- 9a. ORIGINATOR'S REPORT NUMBER(S): Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.
- 9b. OTHER REPORT NUMBER(S): If the report has been assigned any other report numbers (either by the originator or by the sponsor), also enter this number(s).
- 10. AVAILABILITY/LIMITATION NOTICES: Enter any lim-

itations on further dissemination of the report, other than those imposed by security classification, using standard statements such as

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through
- (4) "U. S. military agencies may obtain copies of this report directly from DPIC. Other qualified users shall request through
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

- 11. SUPPLEMENTARY NOTES: Use for additional explanatory notes.
- 12. SPONSORING MILITARY ACTIVITY: Enter the name of the departmental project office or laboratory sponsoring (paying for) the research and development. Include address.
- 13. ABSTRACT: Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested tength is from 150 to 225 words.

14. KEY WORDS: Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, rules, and weights is optional.

SUPPLEMENTARY

INFORMATION

KC-135
POWER SPECTRAL
VERTICAL GUST LOAD
ANALYSIS

DETAILED ANALYSIS AND RESULTS VOLUME I

ROBERT N. LATZ
THE BOEING COMPANY

"This document is subject to special export controls and each transmittal to foreign governments or foreign nationals may be made only with proper approval of the Air Force Flight Dynamics Laboratory (FDTR), Wright-Patterson AFB, Ohio 45433."